

Report 11669
July 2000

AEROJET

**Integrated Advanced Microwave Sounding Unit-A
(AMSU-A)**

Performance Verification Report

**Final Comprehensive Performance Test Report,
P/N 1331720-3-TST, S/N 109/A1**

**Contract No. NAS 5-32314
CDRL 208**

Submitted to:

**National Aeronautics and Space Administration
Goddard Space Flight Center
Greenbelt, Maryland 20771**

Submitted by:

**Aerojet
1100 West Hollyvale Street
Azusa, California 91702**

Aerojet

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Electronic Systems Plant

P.O. Box 296

Azusa, California 91702-0296

CAGE/Facility Ident: 70143



AE-26156/3C

6 April 1999

Superseding
AE-26156/3B
10 March 1999

PROCESS SPECIFICATION

**METSAT/KLM/AMSU-A1, SYSTEM COMPREHENSIVE
AND LIMITED PERFORMANCE TESTS
TEST PROCEDURE**

Contract No.: NAS5-32314

Prepared for:

**NASA/Goddard Space Flight Center
Greenbelt Road
Greenbelt, MD 20771**

TABLE OF CONTENTS

Paragraph		Page
1.	SCOPE	1
1.1	Scope	1
1.2	Test procedure sequence	1
2.	APPLICABLE DOCUMENTS	3
2.1	Government documents	3
2.2	Non-Government documents	3
2.2.1	Aerojet documents	3
3.	REQUIREMENTS	5
3.1	General test requirements	5
3.1.1	Equipment and test facilities	5
3.1.2	Required procedures and operations	6
3.1.2.1	Limited performance test (LPT)	6
3.1.2.2	Comprehensive performance test (CPT)	6
3.1.3	Inspection instructions	6
3.1.4	Test conditions	6
3.1.4.1	Standard ambient conditions	6
3.1.4.2	Test tolerances	7
3.1.4.3	Read-out accuracy	7
3.1.5	Electrostatic Sensitive Device (ESD) handling	8
3.2	Detailed Procedures	8
3.2.1	Responsibility for inspection	8
3.2.2	Monitoring procedures for equipment	8
3.2.3	Test preparation	8
3.2.3.1	STE connection	8
3.2.3.2	Signal sources	8
3.2.3.3	Signal outputs	8
3.2.3.4	Test software	8
3.2.3.5	Initial turn-on	8
3.2.3.6	Turn-off methods	10
3.2.4	Detailed performance tests	10
3.2.4.1	Grounding test	10
3.2.4.2	Power system, transient susceptibility, power quality, and instrument feedback tests	11
3.2.4.2.1	+28 V main load bus test	12
3.2.4.2.1.1	+28 V MLB during turn on transient	12
3.2.4.2.1.2	+28 V MLB operating power	17
3.2.4.2.1.3	Instrument feedback test	18
3.2.4.2.1.4	Transient susceptibility and power quality tests	18
3.2.4.2.1.4.1	Equipment setup	18
3.2.4.2.1.4.2	Low frequency load induced transients	18
3.2.4.2.1.3.3	High frequency load induced transients	21
3.2.4.2.2	+28 V pulse load bus test	21
3.2.4.2.2.1	PLB during the first two seconds	21
3.2.4.2.2.2	PLB measured from 2 to 4 seconds	24
3.2.4.2.2.3	PLB measured from 4 to 6 seconds	24
3.2.4.2.2.4	PLB measured from 6 to 8 seconds	26
3.2.4.2.2.5	Eight second integrated current measurement	26
3.2.4.2.2.6	PLB turn-on transient	26
3.2.4.2.2.7	PLB current in warm cal, cold cal and Nadir mode	30
3.2.4.2.2.8	Instrument feedback test (PLB)	30

3.2.4.2.2.9	Transient susceptibility and power quality tests	30
3.2.4.2.2.9.1	Equipment setup	30
3.2.4.2.2.9.2	Low frequency induced transients	30
3.2.4.2.2.9.3	High frequency load induced transients	32
3.2.4.2.3	Analog telemetry bus	33
3.2.4.2.3.1	Operating power measurements	33
3.2.4.2.3.2	Instrument feedback test (ATB)	33
3.2.4.2.3.3	Transient susceptibility and power quality tests (ATB)	33
3.2.4.2.3.3.1	Equipment setup	33
3.2.4.2.3.3.2	Low frequency induced transients	33
3.2.4.2.3.3.3	High frequency load induced transients	36
3.2.4.2.4	+10 volt interface bus test	37
3.2.4.2.4.1	Operating power measurements	37
3.2.4.2.4.2	Instrument feedback test	37
3.2.4.2.5	Power input test for LPT	37
3.2.4.3	Clock, commands, and data system test	37
3.2.4.3.1	Test sequence	37
3.2.4.3.2	Clock signals test	40
3.2.4.3.2.1	1.248 MHz synchronization clock	40
3.2.4.3.2.2	C1 shift pulse verification	42
3.2.4.3.2.3	A1 select pulse verification	42
3.2.4.3.2.4	8-seconds frame sync pulse verification	42
3.2.4.3.2.5	Synchronization signal relationship	42
3.2.4.3.3	Commands and digital-B telemetry test	44
3.2.4.3.3.1	Module totally off	44
3.2.4.3.3.2	Survival heater power ON/OFF command	44
3.2.4.3.3.3	Module power connect command	44
3.2.4.3.3.4	Phase lock loop (PLL) PLLO No. 1 / PLLO No. 2	45
3.2.4.3.3.5	Scanner commands verification	45
3.2.4.3.3.6	Scanner position commands (A1-1 and A1-2) verification	45
3.2.4.3.4	Digital-A data output verification	46
3.2.4.3.4.1	Full scan mode	47
3.2.4.3.4.2	Warm cal mode	48
3.2.4.3.4.3	Cold cal mode	49
3.2.4.3.4.4	Nadir cal mode	50
3.2.4.3.5	Analog telemetry test	50
3.2.4.3.5.1	Analog TLM signals measurements connector J6	51
3.2.4.3.5.2	Analog TLM signal measurements using the STE	51
3.2.4.3.6	Test point verification	51
3.2.4.3.6.1	Integration/hold and dump clock signals	53
3.2.4.3.6.2	Integration time (analog outputs)	53
3.2.4.3.6.3	PLLO No. 1 verification	56
3.2.4.3.6.4	PLLO No. 2 verification	58
3.2.4.3.7	GSE mode verification	58
3.2.4.3.7.1	Equipment preparation and instrument turn-on procedure	59
3.2.4.3.7.2	GSE Mode-1	61
3.2.4.3.7.3	GSE Mode-2	61
3.2.4.3.7.4	GSE Mode-3	62
3.2.4.3.7.5	GSE Mode-4	62
3.2.4.3.7.6	GSE Mode-5	62
3.2.4.3.7.7	GSE Mode-7	63
3.2.4.4	Radiometer functional test	64
3.2.4.4.1	PLLO frequency measurements	64
3.2.4.4.2	Relative radiometer NEAT measurements	66
3.2.4.4.2.1	Equipment preparation and setup configuration	67

6 Apr 99

3.2.4.4.2.2	Relative NEAT data collection	68
3.2.4.5	Channel identification test	68
4.	QUALITY ASSURANCE PROVISIONS	72
4.1	Responsibility for inspection	72
4.1.1	Test facilities	72
4.1.2	Electrostatic Device (ESD) handling	72
4.2	Monitoring procedures	72
4.2.1	Test equipment	72
4.2.2	Software	72
4.3	Monitoring procedures for materials	72
4.4	Certification	72
4.5	Test methods	72
4.5.1	Accept-reject criteria	72
4.5.2	General	73
4.5.2.1	Test data	73
5.	PREPARATION FOR DELIVERY	74
6.	NOTES	74
6.1	Acronyms and abbreviations	74
6.2	Changes	75

FIGURES

Figure		Page
1.	Test Procedure Sequence.....	1
2.	Signal Output at J7	9
3.	Grounding Test Setup.....	11
4.	+28 V Main Load Bus Verification Setup.....	13
5.	+28 V Main Bus Load Peak Power for KLM (S/N 102, 103 and 104)	15
6.	+28 V Main Bus Load Peak Power for METSAT (S/N 105 and up)	16
7.	+28 V MLB Transient Susceptibility and Power Quality Tests Setup.....	19
8.	Load Induced Transient (Main Bus).....	20
9.	+28 V Pulse Load Verification Setup.....	22
10.	Typical Load Current Waveforms from the +28 V Pulse Load Bus.....	25
11.	+28V Pulse Load Bus Turn-on Transient.....	29
12.	+28 V PLB Transient Susceptibility and Power Quality Tests Setup.....	31
13.	Load Induced Transient (Pulse Load).....	32
14.	+28V Analog Telemetry Bus Test Setup.....	34
15.	+28 Vdc Analog Telemetry Bus Ripple Current and Transient Susceptibility Test Setup	35
16.	Load Induced Transient (Main Bus).....	36
17.	+10V Interface Bus Operating Power Measurements Test Setup	38
18.	+28 V Main Load Bus Test Setup (For LPT Only)	39
19.	Clock Pulses Timing and Synchronization	41
20.	Clock Signals Test Setup.....	41
21.	Synchronization Signal Relationships Test Setup.....	43
22.	Analog Telemetry Signal Verification Test Setup	52
23.	Integration/Hold and Dump Signals Verification Test Setup	54
24.	Integration Time (Analog Output) Verification Setup.....	55
25.	PLL0 No. 1/No. 2 Test Setup	57
26.	GSE Modes Verification Test.....	60
27.	Configuration for RF Measurements	65
28.	Sample Plot.....	66
29.	NEAT Setup Configuration.....	68
30.	Digital-A Data Screen.....	70
31.	Radiometric Data Screen	71
32.	Channel Identification Setup	71

TABLES

Table		Page
I.	Equipment List	5
II.	AMSU-A1 Performance Tests.....	7
III.	Location and Frequency of Channel 3 through 15 Analog Outputs.....	56

TEST DATA SHEETS

TDS

	Page
1 Grounding System Test.....	A-2
2 +28 MLB During Turn-on Transient.....	A-11
3 +28 MLB Operating Power.....	A-12
4 +28 Pulse Load Bus.....	A-13
5 +28 V Analog Telemetry Bus.....	A-15
6 +10 V Interface Bus Voltage.....	A-16
7 Power Input Test for LPT.....	A-17
8 1.248 MHz Clock Signal Verification.....	A-18
9 "C1" Shift Pulse Verification.....	A-19
10 "A1" Select Pulse Verification.....	A-20
11 "8 Seconds" Frame Sync Pulse.....	A-21
12 Synchronization Signals Relationship.....	A-22
13 Synchronization Signals Relationship.....	A-24
14 Commands and Digital-B Telemetry Verification.....	A-25
15 Scanner Commands Verification.....	A-26
16 Scanner Commands Verification.....	A-27
17 Scanner Commands Verification.....	A-28
18 Scanner Positions Commands.....	A-29
19 Digital-A Data Output Full Scan Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification.....	A-30
20 Reflector Positions Section [IV].....	A-31
21 Digital-A Data Output Radiometer Data Section [V].....	A-32
22 Full Scan Mode Temperature Sensors Section [VI].....	A-33
23 Digital-A Data Output Warm Cal Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification.....	A-35
24 Reflector Position Warm Cal Mode Section [IV] and Reflector Position Nadir Mode Section [IV].....	A-36
25 Digital-A Data Output Warm Cal Mode Radiometer Data Section [V].....	A-37
26 Warm Cal Mode Temperature Sensors Section [VI].....	A-38
27 Digital-A Data Output Cold Cal Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification.....	A-40
28 Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV].....	A-41
29 Digital-A Data Output Cold Cal Mode Radiometer Data Section [V].....	A-43
30 Cold Cal Mode Temperature Sensors Section [VI].....	A-44
31 Digital-A Data Output Nadir Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification.....	A-46
32 Digital-A Data Output Nadir Mode Radiometer Data Section [V].....	A-47
33 Nadir Mode Temperature Sensors Section [VI].....	A-48
34 Analog Telemetry Verification by Way of Connector J6.....	A-50
35 Analog Telemetry Signals by Way of the STE.....	A-51
36 Integrate/Hold and Dump Signal Verification.....	A-53
37 Integration Time (Analog Output) Verification.....	A-54
38 Integration Time (Analog Output) Verification.....	A-55
39 Integration Time (Analog Output) Verification.....	A-56
40 Integration Time (Analog Output) Verification.....	A-57
41 Integration Time (Analog Output) Verification.....	A-58
42 Integration Time (Analog Output) Verification.....	A-59
43 Integration Time (Analog Output) Verification.....	A-60
44 PLL0 No. 1 Verification and PLL0 No. 2 Verification.....	A-61
45 Digital-A/GSE Mode-1 Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification.....	A-62
46 Reflector Position.....	A-63
47 Digital-A/GSE Mode-1 Radiometer Data Section [V].....	A-65

TEST DATA SHEETS (CONT)

TDS

Page

48	Digital-A/GSE Mode-1 Temperature Sensors Section [VI]	A-66
49	Receiver Input Signals.....	A-68
50	Radiometer "Relative" NEDT Verification.....	A-69
51	Transient Susceptibility Test.....	A-71
52	Channel Identification Test	A-73

1. SCOPE

1.1 Scope. This specification establishes the requirements for the Comprehensive Performance Test (CPT) and Limited Performance Test (LPT) of the Advanced Microwave Sounding Unit-A1 (AMSU-A1), referred to herein as the unit. The unit is defined on Drawing 1331720.

1.2 Test procedure sequence. The sequence in which the several phases of this test procedure shall take place is shown in Figure 1, but the sequence can be in any order.

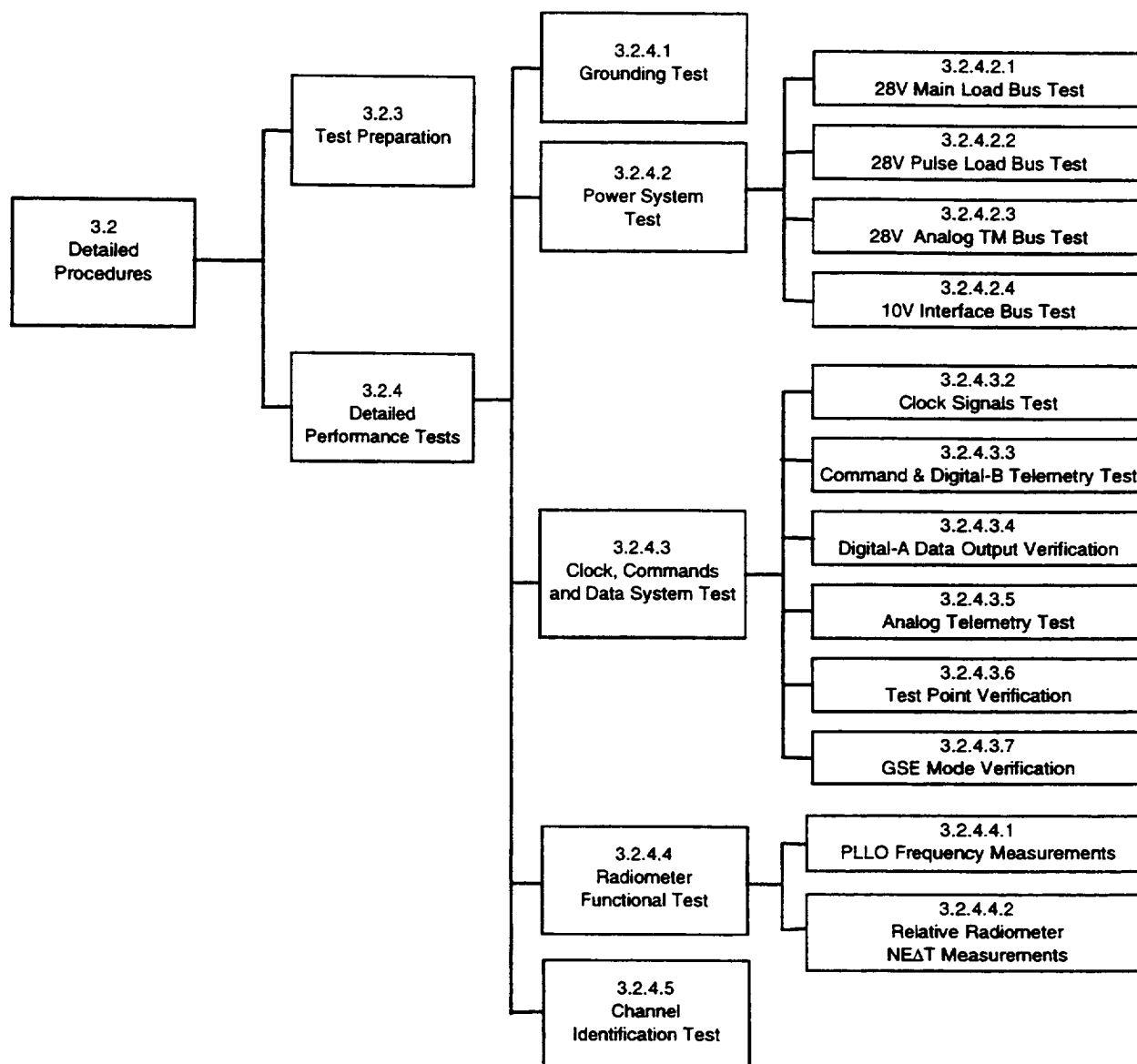


Figure 1. Test Procedure Sequence

AE-26156/3C
6Apr 99

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2. APPLICABLE DOCUMENTS

2.1 Government documents. The following documents form a part of this specification to the extent specified. Unless otherwise specified, the issue shown shall apply.

STANDARDS

Military

MIL-STD-45662	Calibration Systems Requirements
---------------	----------------------------------

OTHER DOCUMENTS

S-480-79	Performance Assurance Requirements for the EOS/METSAT Integrated Programs Advanced Microwave Sounding Unit-A (AMSU-A) (PAR)
S-480-80	Performance and Operation Specification for the EOS/METSAT Integrated Programs Advanced Microwave Sounding Unit-A (AMSU-A) (POS)
IS-2617547	AMSU-A1 Unique Instrument Interface Specification (UIIS)
IS-3267415	ATN-KLM General Instrument Interface Specification (GIIS)

(Copies of government documents should be obtained as indicated in the Department of Defense Index of Specification and Standards.)

2.2 Non-Government documents. The following documents form a part of this specification to the extent specified herein. Unless otherwise specified, the issue in effect on the date of testing shall apply.

2.2.1 Aerojet documents

SPECIFICATION

AE-26002/1	Test Procedure, Subsystem, Antenna Drive for AMSU-A1
AE-26151/5	Test Procedure, EMI/EMR & EMC for the METSAT/METOP Advanced Microwave Sounding Unit-A (AMSU-A)
AE-26157	Special Test Equipment (STE), Operation and Maintenance Manual
AE-26357	Transportation Handling Procedure for the AMSU-A System Integrated Program

STANDARD

STD-2454	Requirements for Electrostatic Discharge Control
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AE-26156/3C
6Apr 99

REPORT

10353	Contamination Control Plan for the Advanced Microwave Sounding Unit-A (AMSU-A)
-------	--

DRAWINGS

1331720	Advanced Microwave Sounding Unit A1 (AMSU-A1)
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1335695	Special Test Equipment
---------	------------------------

1356655	Console Assembly, METSAT and EOS STE
---------	--------------------------------------

(Copies of Aerojet documents may be obtained from Gencorp Aerojet, Azusa Operations, CAGE 70143, P.O. Box 296, Azusa, California, 91702-0296).

3. REQUIREMENTS

3.1 General test requirements

3.1.1 Equipment and test facilities. The tests described herein shall be performed at Aerojet under laboratory conditions in an EMI shielded chamber for the first and final CPT. Other tests need not be accomplished in the EMI shielded chamber. The test equipment listed in Table I shall be used when performing the tests. If the specified equipment is not available, the equipment substituted shall provide a measurement accuracy equal to or greater than that of the specified equipment. The AMSU-A Special Test Equipment (STE) shall be used for activation and control of the unit and monitoring of its performance.

Table I. Equipment List

Item	Quantity	Item Description	Mfg.	Model
1	1	Dynamic signal analyzer	Hewlett-Packard	3562A
2	1	Signal Generator	Hewlett-Packard	3314A
3	1	Oscilloscope	Tektronix	2225A
4	1	9-pin breakout box	Aerojet	2536-3743/SK1358702-1
5	1	15-pin breakout box	Aerojet	2536-3744/SK1358703-1
6	2	25-pin breakout box	Aerojet	2336-3746/SK1358704-1
7	1	37-pin breakout box	Aerojet	2536-3745/SK1358705-1
8	1	Relay Board	Aerojet	—
9	1	Double Shielded Connector	—	—
10	1	Lab. General Purpose Power Supply	Hewlett-Packard	6114
11	1	Oscilloscope	Tektronix	466A
12	1	Power Supply	Power Designs	3650-S
13	1	WR19 Harmonic Mixer (40-60 GHz)	Hewlett-Packard	HP11970V
14	1	Power Meter	Anritsu	ML83A
15	1	WR19 Feed Horn	TRG	V861
16	1	LN2 Container	Cole	N03726-20
17	1	Spectrum Analyzer	Hewlett-Packard	8566B
18	1	STE Computer	Aerojet	1336695
19	1	STE Interface Cable J1	Aerojet	1335758-1
20	1	STE Interface Cable J2	Aerojet	1335752-1
21	1	STE Interface Cable J3	Aerojet	1335756-1
22	1	STE Interface Cable J4	Aerojet	1335755-1
23	1	STE Interface Cable J5	Aerojet	1335753-1
24	1	STE Interface Cable J6	Aerojet	1335754-1
25	1	STE Interface Cable J7	Aerojet	1335757-1
26	1	Oscilloscope Camera	Tektronix	—
27	1	Current Probe	Tektronix	AM503
28	1	Plotter	Hewlett-Packard	7475A
29	1	Frequency Counter	Hewlett-Packard	5316A
30	1	Multimeter (Digital volt-ohm meter)	Fluke	77

6Apr 99

Table I. Equipment List (Continued)

Item	Quantity	Item Description	Mfg.	Model
31	1	Cold Target Stand A1-1	Aerojet	T-1291001-3
32	1	Cold Target Stand A1-2	Aerojet	T-1291001-2
33	2	Cold Target Support	Aerojet	T-1291000-1
34	1	Sweeper	Hewlett-Packard	83623A
35	1	Multiplier	Hewlett-Packard	83557A/83558A
36	1	Coupler/Detector	Hewlett-Packard	83557-60001
37	1	Spectrum Analyzer	Hewlett-Packard	8563E

3.1.2 Required procedures and operations. The unit shall be subjected to the examinations and tests specified in 3.2.4 and Table II.

3.1.2.1 Limited performance test (LPT). The Limited Performance Test shall consist of the test procedures specified in the LPT column of Table II.

3.1.2.2 Comprehensive performance test (CPT). Three versions of the Comprehensive Performance Test are identified in Table II. These are applicable for different test stages. The test procedures to be performed for each version are specified in the 1st CPT, Sub CPT, and Final CPT columns of Table II. See 3.1.1 for required location of the first and the final CPT.

3.1.3 Inspection instructions. The following shall apply to all inspections performed under this specification.

- a. **Personnel familiarization:** All personnel directly concerned with the conduct of the inspection shall become familiar with the entire content of this document before beginning the tests. Each step, including all notes, warnings, and cautions, shall be understood thoroughly before starting.
- b. **Referenced documents:** Performance of the tests specified herein may require reference to the documents listed in Section 2. It is recommended that the applicable issues of these documents be available at the time and place of testing.

3.1.4 Test conditions. The following paragraphs shall apply to all testing described in this document.

3.1.4.1 Standard ambient conditions. Unless otherwise specified in a detailed method paragraph, all handling shall be performed under the following laboratory ambient conditions.

- a. Handling in accordance with AE-26357
- b. Contamination control in accordance with Report 10353
- c. Temperature: $+23 \pm 10^{\circ}\text{C}$
- d. Pressure: 610 to 810 torr
- e. Humidity: $50 \pm 20\%$ (no condensation)
- f. The instrument shall be placed in its protective bag (1338427) when not in use.

Table II. AMSU-A1 Performance Tests

Paragraph	Test Description	1st CPT	LPT	Sub CPT	Final CPT
3.2.4.1	Grounding	X	X	X	X
3.2.4.2.1.1	+28 Main Load Bus (MLB) Turn-On Transient	X			X
3.2.4.2.1.2	+28 MLB Operating Power	X	Note 2	Note 3	X
3.2.4.2.1.3	Instrument Feedback Test	Note 8			
3.2.4.2.1.4	Transient Susceptibility Test	X			
3.2.4.2.2	+28 Pulse Load Bus (PLB) Peak Current	X		Note 4	X
3.2.4.2.2.8	Instrument Feedback Test (PLB)	Note 8			
3.2.4.2.2.9	Transient Susceptibility Test	X			
3.2.4.2.3	+28 Analog Telemetry Bus (ATB)	X		X	X
3.2.4.2.3.2	Instrument Feedback Test (ATB)	Note 8			
3.2.4.2.3.3	Transient Susceptibility Test	X			
3.2.4.2.4	+10 V Interface Bus	X		X	X
3.2.4.2.4.2	Instrument Feedback Test	Note 8			
3.2.4.2.5	Power Input Test for LPT		X		
3.2.4.3.2	Clock Signals	X			X
3.2.4.3.3	Commands and Digital-B Telemetry	X	X	X	X
3.2.4.3.4	Digital-A Data Output	X	Note 5	Note 5	X
3.2.4.3.5	Analog Telemetry	X	Note 6	Note 6	X
3.2.4.3.6	Test Points	X		X	X
3.2.4.3.7	GSE Mode	X Note 7			
3.2.4.4	Radiometer Functional	Title			
3.2.4.4.1	PLLO Frequency Measurement	X			X
3.2.4.4.2.2	Relative NE Δ T	X	X	X	X
3.2.4.5	Channel Identification Test	X			
Notes: 1. Test Data Sheets for CPT/LPT located in Appendix A. 2. 3.2.4.2.5 (Power input test for LPT). 3. At 28 V only. 4. 3.2.4.2.2 except 3.2.4.2.2.6. 5. Only full scan. 6. STE only. 7. GSE mode test/verification is not required and is for engineering use only. 8. Instrument feedback test will be performed in the EMI/RFI chamber using EMI/RFI test procedure AE-26151/5.					

3.1.4.2 Test tolerances. The tolerances allowed on test conditions are intended only to provide for accuracy of such items as instrumentation and controls. Conditions shall be as close as possible to the nominal or center values specified, and in no instance shall they exceed the tolerances specified. Unless otherwise specified, the tolerances shall be within $\pm 10\%$.

3.1.4.3 Read-out accuracy. Parameters are specified either as limits or as nominal values with plus-or-minus tolerances. These limits and tolerances shall be regarded as absolute, and the inaccuracies of measuring equipment shall not be interpreted as part of measured values in such a way that out-of-limit measurements may appear in-limit.

3.1.5 Electrostatic Sensitive Device (ESD) handling. All electronic hardware shall be handled in accordance with Aerojet Standard STD-2454.

3.2 Detailed Procedures

3.2.1 Responsibility for inspection. All tests specified herein shall be performed under the cognizance of Aerojet Quality Assurance.

3.2.2 Monitoring procedures for equipment. Test equipment calibration schedules and procedures shall comply with the requirements of MIL-STD-45662. Before performing examinations and tests in accordance with this procedure, all test equipment to be used shall be verified as being within their current calibration period. Calibration or alignment, necessary for operation of the equipment within the requirements of this document, shall be performed when required.

3.2.3 Test preparation

3.2.3.1 STE connection. The power sources, signal sources, and loads are provided to the unit under test by the AMSU-A Special Test Equipment (STE) (Drawing 1335695 or 1356655), in accordance with paragraph 5.2 of S-480-80. The STE is automated test equipment controlled by a MicroVax computer. The unit shall be connected to the STE in accordance with AE-26157 and the detailed test procedures in 3.2.4.

3.2.3.2 Signal sources. Signal sources required during the performance test but not provided by the STE are as follows:

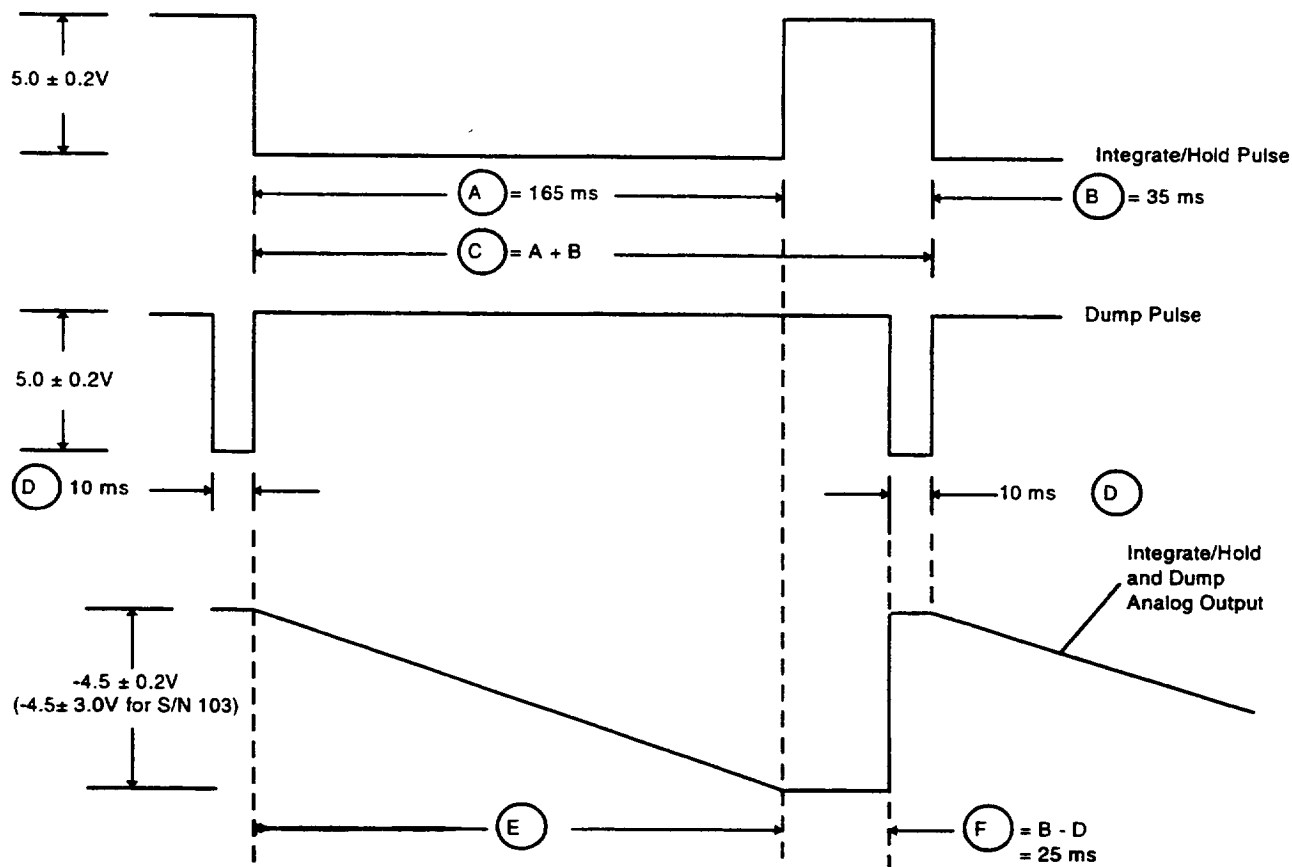
- a. Cold background at LN₂ temperature at room ambient.
- b. +28 ± 1 Vdc, 3 Amps.

3.2.3.3 Signal outputs. Signal outputs, except for the test signals at J7, shall be monitored by the STE. The signal outputs at J7 are shown in Figure 2.

3.2.3.4 Test software. AMSU-A1 bonded software shall be used to operate the STE. During initialization of the STE, as specified in AE-26157, the A1 software shall be selected. The bonded software is being selected by the STE computer automatically during initialization of the STE.

3.2.3.5 Initial turn-on. When called for in the individual test procedures, turn on the unit as follows:

1. Turn on the STE and initialize the STE as specified in AE-26157.
2. Connect breakout box to J1 on the STE +28 V power supply cable J1.
3. Connect DVM to J1-1 (+) and J1-3 (RTN).
4. Verify that the STE power supply POWER switch on the STE +28 V power supply is ON and the power supply is adjusted to +28 ±0.5 Vdc.
5. Verify that the PWR and SW/TM switches on the STE power distribution unit are ON.
6. Enter the serial number (decimal equivalent of the identification number provided in the UIIS) for the unit under test using AE-26157, if necessary. Verify that the Main Menu (AMSU-A1 WHAT TYPE OF TEST?) is displayed on the STE CRT terminal display.
7. On the Main Menu, press the [2] MONITOR ONLY touch area (or type the number). The Monitor Only Menu will be displayed, with Block Monitor Data Select options shown in the middle (window) area of the screen.



NOTE: Timing Tolerances are $\pm 10\%$.

Figure 2. Signal Output at J7

8. On the Monitor Only Menu, press [14] COMMANDS. The Commands Menu will be displayed in the window area.
9. On the Commands Menu, press [9] MODULE POWER = CONNECT. Wait at least 18 seconds for command execution. This applies power to the unit.
10. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

11. Wait at least 18 seconds and observe the commands are acknowledged by STE.

12. Verify that the STE power supply is adjusted to $+28 \pm 0.5$ Vdc (see steps 2 through 4) .
13. Verify that all breakout box switches are in the closed position.
14. According to the individual test procedures, execute commands as necessary to obtain the required commands configuration. Several commands can be executed at the same time.

3.2.3.6 Turn-off methods. The unit can be turned off immediately by pressing [9] MODULE POWER = DISCONNECT on the Commands Menu. For a phased shutdown, press [11] MODULE TOTALLY OFF = OFF on the Commands Menu or press POWER [4] OFF on any display. When connecting breakout boxes to the unit or STE connectors, verify that the unit power is off and the STE +28 V power supply is manually turned off.

NOTE

If power of the unit is turned off by command [9] MODULE POWER = DISCONNECT or the STE program is interrupted, then perform a phased shutdown after turn-on before starting next step.

3.2.4 Detailed performance tests. The comprehensive performance tests for the AMSU-A1 system are to be carried out on the fully assembled and operational unit. The tests to be performed are as follows:

- a. Grounding/Isolation system test.
- b. Power system test.
- c. Clock commands and data system test.
- d. Radiometer functional test.
- e. Transient susceptibility test.
- f. Instrument feedback test.

3.2.4.1 Grounding test. This test provides the verification of the unit grounding requirements of GHS IS-3267415 Paragraph 3.1.1 and UIIS IS-2617547 paragraph 3.1.1.

1. Connect breakout boxes to each of the spacecraft interface connectors J1 through J7 as shown in Figure 3. Verify that all connectors are protected with connector savers.
2. Measure and record continuity or isolation between the points shown on Test Data Sheet (TDS) 1.

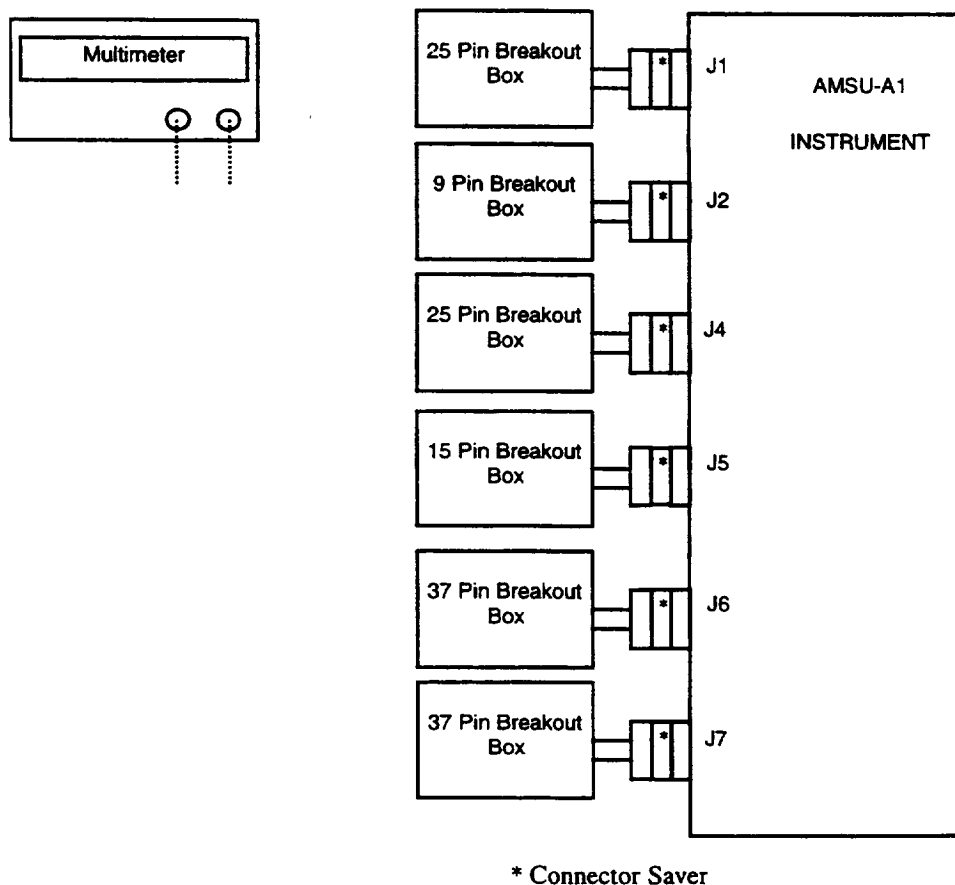


Figure 3. Grounding Test Setup

3.2.4.2 Power system, transient susceptibility, power quality, and instrument feedback tests. The purpose of these tests is to verify power system compliance in regard to:

- Turn-On transients
- Operating power
- Transient susceptibility
- Current ripple

The following DC voltage lines will be tested for the above parameters:

- +28 V Main Load Bus (parameters a, b, c, d)
- +28 V Pulse Load Bus (parameters a, b, c, d)
- +28 V Analog Telemetry Bus (parameters b, c, d)
- +10 V Interface Bus (parameters b, d)

3.2.4.2.1 +28 V main load bus test

3.2.4.2.1.1 +28 V MLB during turn on transient. The +28 V MLB (at 28.56 Vdc) during turn on, shall be verified as follows:

1. Configure the unit and test equipment as shown in Figure 4. Obtain DSA trigger from J4-14. Verify that switches 1, 2, 14 and 15 of the breakout box are in the OPEN position. Disconnect +28 Vdc external power supply output at J1 and adjust the power supply to read 28.56 ± 0.05 Vdc on voltmeter. Re-connect the power supply output (J1) as shown in Figure 4.
2. Configure the Dynamic Signal Analyzer (DSA) as follows:

Select MEAS MODE	Select INPUT COUPLE
Select <i>Time Capture</i>	Select <i>CH1 DC</i>
Select <i>Capture Select</i>	Select <i>CH1 Ground</i>
Select <i>Capture Length</i> ; Enter 300.0; Select <i>msec</i>	Select INPUT TRIG
Select FREQ	Select <i>Trig Level</i> ; Enter 100; Select <i>mV</i>
Select <i>E SMPL Off</i>	Select <i>Arm AU</i>
Select <i>Freq Span</i> ; Enter 25; Select <i>kHz</i>	Select <i>Ext</i> ; Select <i>(-) Slope</i>
Select SELECT MEAS	Select TRIG DELAY
Select <i>Power Spec</i>	Enter 0; Select <i>μSec</i>
Select <i>CH1 Active</i>	Select COORD
Select WINDOW	Select <i>Real</i>
Select <i>Hann</i>	Select VIEW INPUT
Select SOURCE	Select <i>Time Buff</i>
Select <i>Source Off</i>	Select SCALE
Select AVG	Select <i>X Fixd Scale</i> ; Enter 0.0, 300; Select <i>msec</i>
Select <i>Avg Off</i>	Select <i>Y Fixd Scale</i> ; Enter 0,80; Select <i>mV</i>
Select <i>Tim Av Off</i>	Select UNITS
Select RANGE	Select <i>Hz (sec)</i>
Select <i>Chan 1 Range</i> ; Enter 1; Select <i>V</i>	

NOTE

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- a) Select 1.0 A/10mV per div. on the current amplifier.
 - b) Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
 - c) Adjust the "y" axis voltage range to ± 4 mV.
 - d) Place the DSA in "Free Run" Trigger and depress "Start Capture" on the DSA.
 - e) With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
 - f) Position the current probe to its original location in accordance with Figure 4, and return the DSA to "Ext" trigger.
3. Turn the unit ON by selecting [9] MODULE POWER; set up the operating modes as defined in paragraph 3.2.3.5 (reference the command screen parameters below). If necessary, adjust the external power supply for 28 Vdc.



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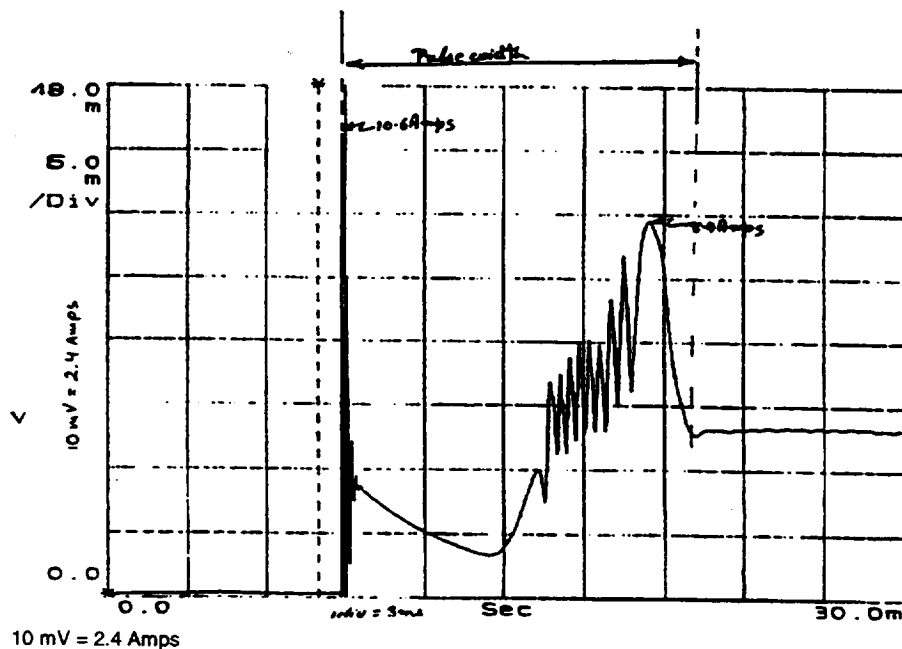
COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

4. Turn the unit OFF by executing command [9] MODULE POWER. Confirm the command has been executed on the STE display.
5. Start the DSA signal capture by depressing "Start Capture"; wait for the DSA message "waiting for trigger" before proceeding.
6. On the STE computer, select [9] MODULE POWER and obtain a record of the +28 MLB Turn-On current waveform. On the STE computer, select [9] MODULE POWER to turn the instrument's power OFF. Adjust the display time base and voltage sensitivity to allow for adequate current and pulse duration measurements (refer to Figure 5 or Figure 6 for an example of per division values). Plot the obtained waveform and attach a hard copy of the scan to TDS 2.
7. Measure the Turn-On time to reach steady state current; record this value on TDS 2.
8. Compute the peak current as follows:
 Measure the maximum Y value by the current/div as selected on the current amplifier. As an example, if the current amplifier is set up to display 1.0 A/10 mV per division, and the maximum Y value = 46.8 mV:

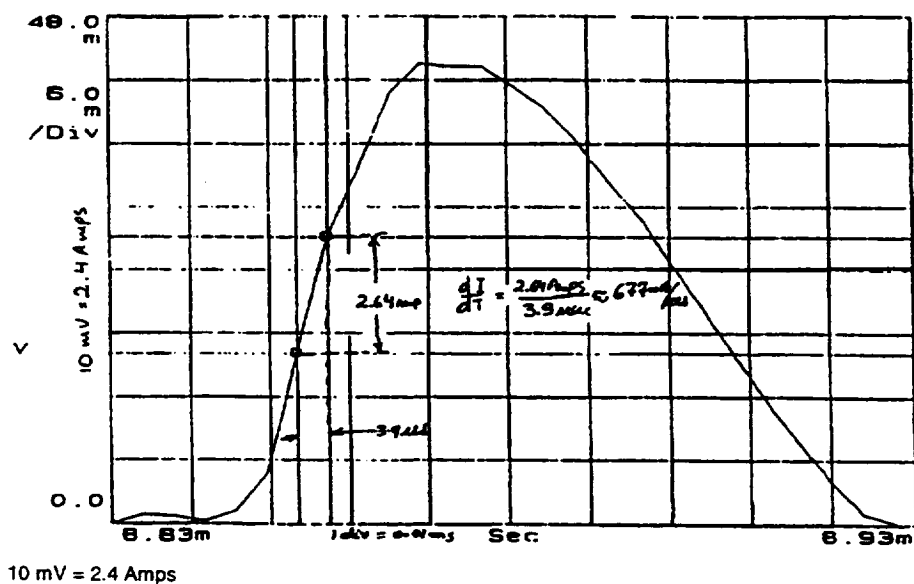
$$46.8 \text{ mV} \times (1.0 \text{ A}/10 \text{ mV}) = 4.68 \text{ amps}$$
 Record this value on TDS 2.
9. The 1st derivative of the current waveform must be calculated. Compute the dI/dT as follows:
 The most probable location of the greatest current demand is during the first positive transition after voltage application. If this is the case, expand that segment of the display and measure the greatest voltage transition in the smallest time transition. The change in voltage times the current/div as selected on the current amplifier produces the change in current. Next divide this change in current by the change in time (in microseconds). This value is dI/dT. Example:

$$\begin{aligned} \text{Change in voltage} & \dots\dots\dots 35.29 \text{ mV} \\ \text{Change in time (microseconds)} & \dots\dots\dots 31.25 \mu\text{s} \\ \text{Current/div on current amplifier} & \dots\dots\dots 1000 \text{ mA}/10 \text{ mV} \end{aligned}$$

$$35.29 \text{ mV} \times (1000 \text{ mA}/10 \text{ mV})/31.25 \mu\text{s} = 112.9 \text{ mA}/\mu\text{s}$$
10. Record the computed value on TDS 2.
11. With the multimeter, adjust the external power supply to $27.44 \pm 0.05 \text{ Vdc}$ as measured between J1-1 (high) and J1-3 (low).
12. Repeat steps 3 through 10.
13. With the multimeter, adjust the external power supply to $28.00 \pm 0.05 \text{ Vdc}$ as measured between J1-1 (high) and J1-3 (low).
14. Repeat steps 3 through 10.

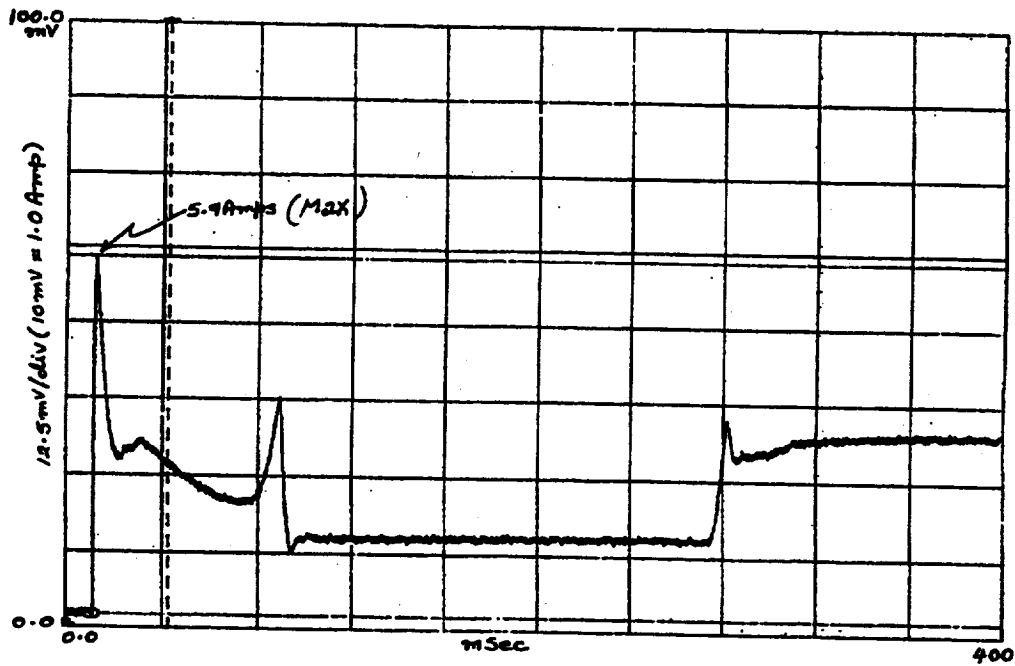


AMSU-A1 (S/N 102) Main Load Bus Worst Case Transient

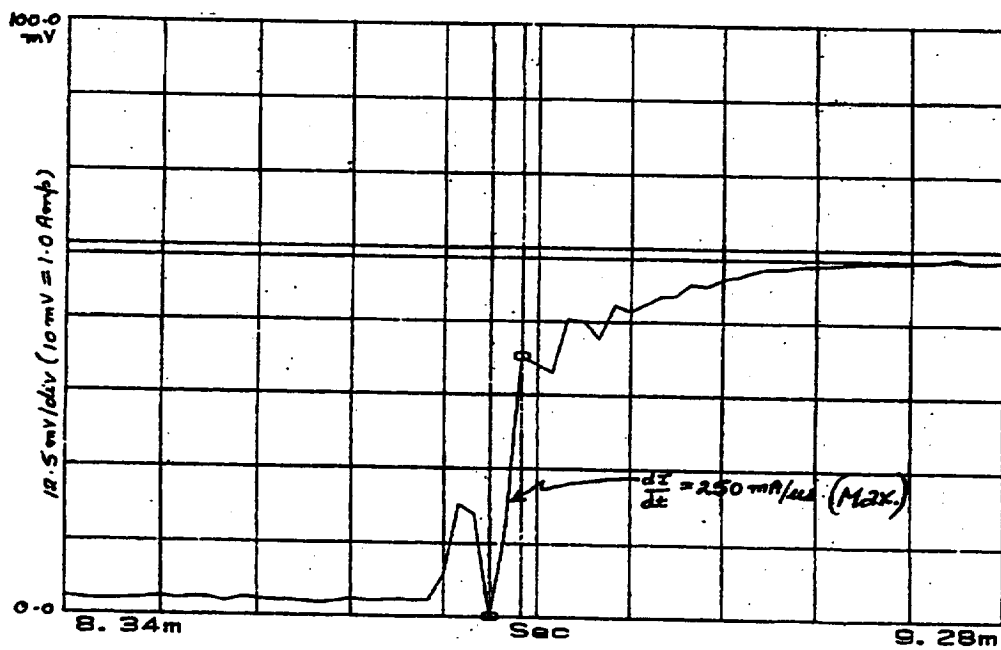


AMSU-A1 (S/N 102) Main Load Bus $\frac{dI}{dT}$ at Worst Case Transient

Figure 5. +28 V Main Bus Load Peak Power for KLM (S/N 102, 103 and 104)



AMSU-A1 Main Load Bus Worst Case Turn-on Transient



AMSU-A1 Main Load Bus $\frac{dI}{dt}$ at Worst Case Turn-on Transient

Figure 6. +28 V Main Bus Load Peak Power for METSAT (S/N 105 and up)

3.2.4.2.1.2 +28 V MLB operating power. Measure the steady state current, voltage, and power as follows:

1. Configure the unit and test equipment as shown in Figure 4. Verify that switches 1, 2, 14 and 15 of the breakout box are in the OPEN position.
2. Turn off power supplies. Insert current meter in positive lead of external power supply, turn power supplies on. Place the unit in operating condition as described in 3.2.4.2.1.1, step 3. While monitoring voltmeter No. 1, adjust the external power supply to 27.0 ± 0.1 volts (see Figure 4). Record the voltage displayed on voltmeter No. 1 on TDS 3 (MLB voltage at 27 Vdc).
3. Record the operating current on TDS 3.
4. Compute the operating power (in watts) as explained on TDS 3.
5. Execute command [18] PLL POWER to change from PLLO#1 to PLLO#2. Allow the instrument to stabilize for a minimum of two minutes.
6. Record the operating current on TDS 3.
7. Compute the operating power (in watts) as explained on TDS 3.
8. Execute command [18] PLL POWER to change from PLLO#2 to PLLO#1. Allow the instrument to stabilize for a minimum of two minutes.
9. Adjust the external power supply to 28.0 ± 0.1 Vdc and record voltage on TDS 3.
10. Record the operating current on TDS 3.
11. Compute the operating power (in watts) as explained on TDS 3.
12. Execute command [18] PLL POWER to change from PLLO#1 to PLLO#2. Allow the instrument to stabilize for a minimum of two minutes.
13. Record the operating current on TDS 3.
14. Compute the operating power (in watts) as explained on TDS 3.
15. Execute command [18] PLL POWER to change from PLLO#2 to PLLO#1. Allow the instrument to stabilize for a minimum of two minutes.
16. Adjust the external power supply to 29.0 ± 0.1 Vdc and record voltage on TDS 3.
17. Record the operating current on TDS 3.
18. Compute the operating power (in watts) as explained on TDS 3.
19. Execute command [18] PLL POWER to change from PLLO#1 to PLLO#2. Allow the instrument to stabilize for a minimum of two minutes.
20. Record the operating current on TDS 3.
21. Compute the operating power (in watts) as explained on TDS 3.
22. Execute command [18] PLL POWER to change from PLLO#2 to PLLO#1. Allow the instrument to stabilize for a minimum of two minutes.

23. Adjust the external power supply to 28.0 ± 0.5 Vdc.
24. Turn the unit off by executing [9] MODULE POWER = DISCONNECT.

3.2.4.2.1.3 Instrument feedback test. Instrument feedback test will be performed in the EMI/RFI chamber using EMI/RFI test procedure AE-26151/5.

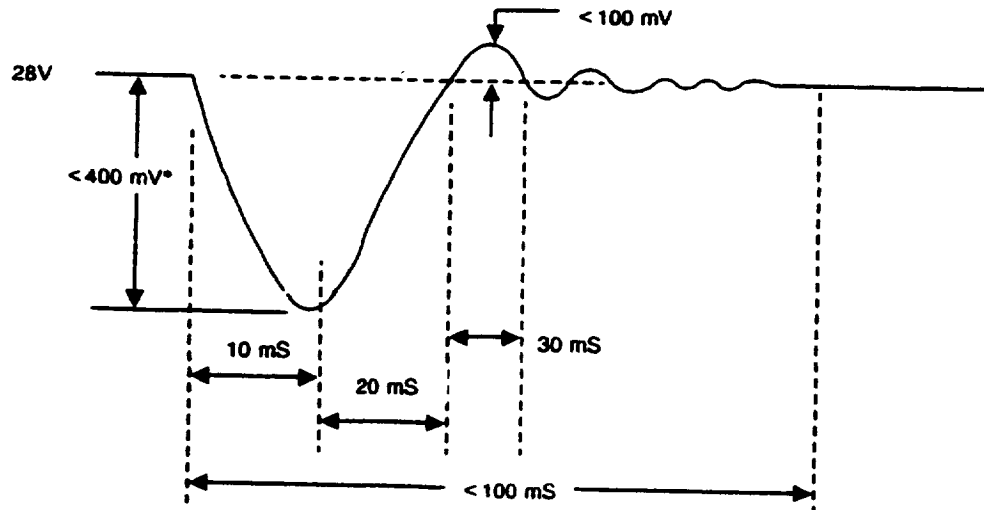
3.2.4.2.1.4 Transient susceptibility and power quality tests. The power tests that follow will demonstrate the AMSU-A1 instrument will operate within specified parameters when the transients (low and high frequency) are applied directly to the power lines.

3.2.4.2.1.4.1 Equipment setup. Set up the test equipment and connect to the instrument as shown in Figure 7.

3.2.4.2.1.4.2 Low frequency load induced transients. The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the power line at the amplitude and duration specified in Figure 8. Perform the Low Frequency Load Induced Transients test as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. Place the signal generator in ARB 0 mode. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator to attain the signal characteristics as shown in Figure 8.
3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three scans. Remove the signal generator output from the power supply.
7. Acquire one Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
8. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

19



* Typical transients occurring a number of times per orbit are on the order of 200 mV zero-to-peak for a 1.5A load change.

Figure 8. Load Induced Transient (Main Bus)

3.2.4.2.1.4.3 High frequency load induced transients. The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the power line. The interfering frequencies are simulated by using the triangular wave output of the signal generator. There are three signals to be sequentially injected; the frequencies and amplitudes as produced by the signal generator and measured by the oscilloscope are:

<u>Frequency (Hz)</u>	<u>Amplitude</u>
1.43	200 mVpp
2.86	1.00 Vpp
6.67	1.50 Vpp

Tolerance on the above values is $\pm 10\%$.

Perform High Frequency Load Induced Transients as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator output as follows:

amplitude 200 mVpp
offset 0.000 V
frequency 1.430 Hz

3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three scans. Remove the signal generator output from the power supply.
7. Acquire one Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
8. Repeat steps 2-4 and 6-7 for 2.86 Hz and 1.0 Vpp.
9. Repeat steps 2-4 and 6-7 for 6.67 Hz and 1.5 Vpp.
10. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

3.2.4.2.2 +28 V pulse load bus test

3.2.4.2.2.1 PLB during the first two seconds. The PLB operation, during the first two seconds, shall be verified as follows:

1. Configure the unit and test equipment as indicated in Figure 9. Obtain DSA trigger from J2-7. Verify that switches 5, 6, 18 and 19 of the breakout box are in the OPEN position.
2. Disconnect +28 Vdc external power supply output and adjust the power supply to read 28.00 ± 0.05 Vdc by using DVM. Re-connect power supply output as shown in Figure 9.

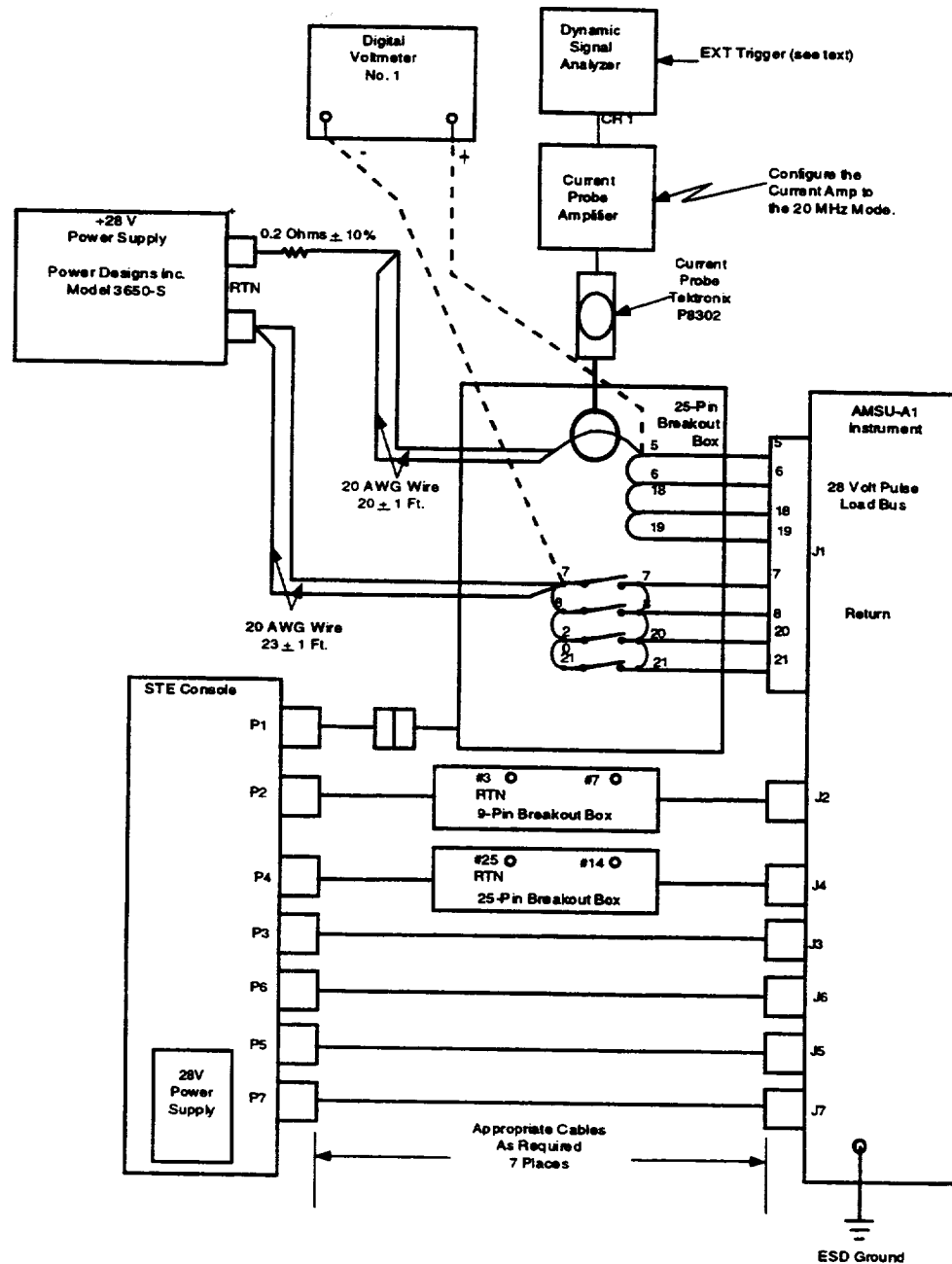


Figure 9. +28 V Pulse Load Verification Setup

3. Configure the dynamic signal analyzer as follows:

Select **MEAS MODE**

Select *Time Capture*
Select *Capture Select*
Select *Capture Length*; Enter *1*; Select *Record*

Select **FREQ**

Select *Freq Span*; Enter *100*; Select *Hz*
Select *E SMPL Off*
Select *Time Length*; Enter *8.0*; Select *Sec*

Select **SELECT MEAS**

Select *Power Spec*
Select *CH1 Active*

Select **WINDOW**

Select *Hann*

Select **SOURCE**

Select *Source Off*

Select **AVG**

Select *Avg Off*
Select *Tim Av Off*

Select **RANGE**

Select *Aut 1 Rng up*

Select **INPUT COUPLE**

Select *CH1 DC*
Select *CH1 Ground*

Select **INPUT TRIG**

Select *Trig Level*; Enter *1.5*; Select *V*
Select *Arm AU*
Select *Ext*
Select *Slope -*

Select **TRIG DELAY**

Enter *0.0*; Select *Sec*

Select **COORD**

Select *Real*

Select **VIEW INPUT**

Select *Time Buff*

Select **SCALE**

Select *X Fixd Scale*; Enter *0.0*, *8.0*; Select *Sec*
Select *Y Fixd Scale*; Enter *-10.0*, *70.0*; Select *mV*

Select **UNITS**

Select *Hz (sec)*

NOTE

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- Select 200 mA/10mV per div. on the current amplifier.
- Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
- Adjust the "y" axis voltage range to ± 4 mV.
- Place the DSA in "Free Run" Trigger and depress "Start Capture" on the DSA.
- With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
- Position the current probe to its original location in accordance with Figure 9, and return the DSA to "Ext" trigger.

The instrument is now ready to capture and plot 8 seconds of data.

6Apr 99

4. Adjust external power supply for +28 Vdc. Turn the unit ON by selecting [9] MODULE POWER, set up the operating modes as defined in paragraph 3.2.3.5 (reference the command screen parameters below). If necessary, re-adjust the external power supply for 28 Vdc.

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

5. Start the DSA signal capture by depressing "Start Capture".
6. Obtain the first 2 second PLB current waveform by selecting 0 to 2 seconds time span. Refer to Figure 10 for a typical waveform. Turn OFF the "X" cursor if it is ON. Turn the "X" cursor ON. The cursor will appear at the highest peak. Ensure this value is less than or equal to 1.3 amps. Record value on TDS 4.
7. Compute the peak current as follows:
 Multiply the maximum Y value by the current/div as selected on the current amplifier. As an example, if the current amplifier is set up to display 200 mA/10 mV per division, and the maximum Y value = 276 mV:

$$60 \text{ mV} \times (200 \text{ mA}/10 \text{ mV}) = 1200 \text{ mA} = 1.20 \text{ amps}$$

3.2.4.2.2.2 PLB measured from 2 to 4 seconds. The PLB operation, from 2 to 4 seconds, shall be verified as follows:

1. Reset the dynamic analyzer in accordance with 3.2.4.2.2.1(2).
2. Change the PRE-TRIGGER DELAY setting of the dynamic signal analyzer to 1.9 seconds.
3. Obtain a hard copy of the signal displayed on the dynamic signal analyzer (refer to Figure 10 for typical waveform).
4. From the hard copy obtained in step 3, calculate the peak current. Record the peak current and bus current values during the integrate/hold, dump (I/H, D) time period (refer to Figure 10) on TDS 4.

3.2.4.2.2.3 PLB measured from 4 to 6 seconds. The PLB operation, from 4 to 6 seconds, shall be verified as follows:

1. Reset the dynamic analyzer in accordance with 3.2.4.2.2.1(2).
2. Change the PRE-TRIGGER DELAY setting of the dynamic signal analyzer to 3.9 seconds.
3. Obtain a hard copy of the signal displayed on the dynamic signal analyzer (refer to Figure 10 for typical waveform).
4. From the hard copy obtained in step 3, calculate the peak current. Record the peak current and bus current values during the integrate/hold, dump (I/H, D) time period (refer to Figure 10) on TDS 4.

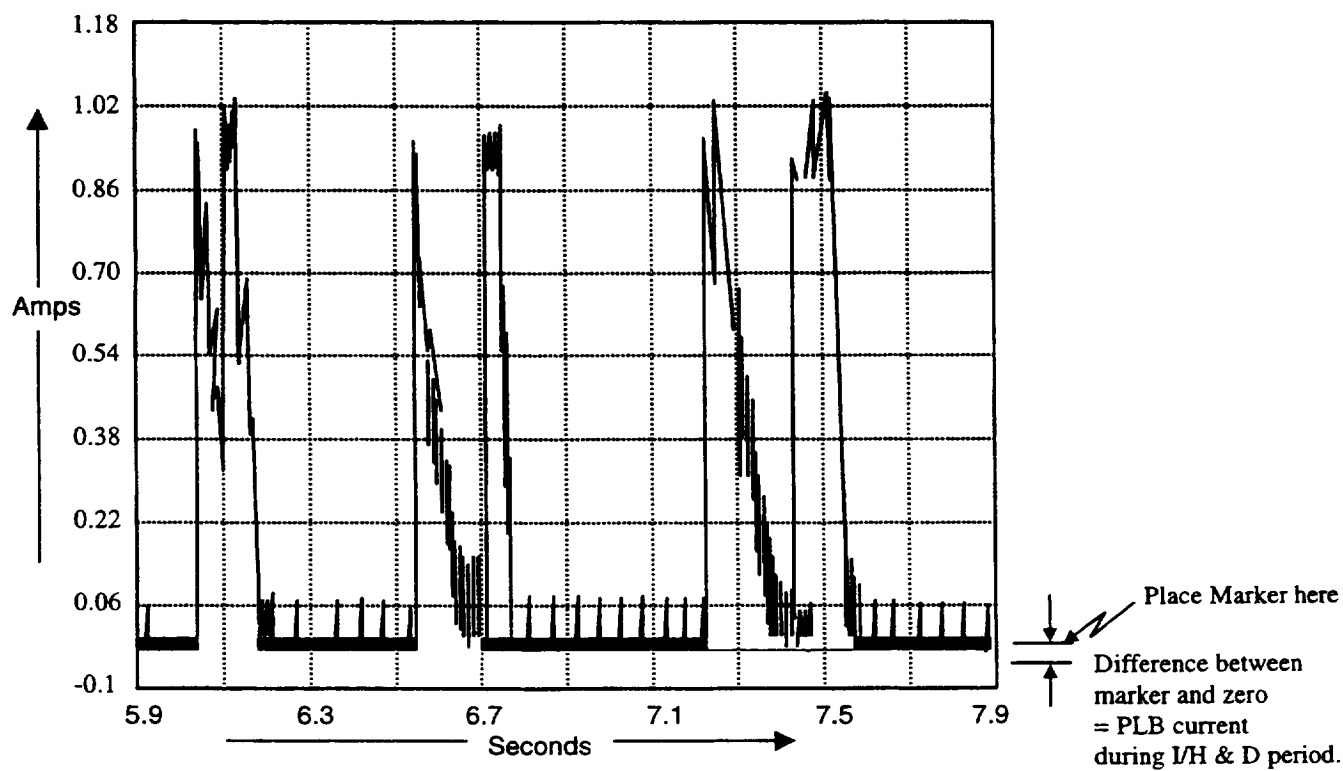
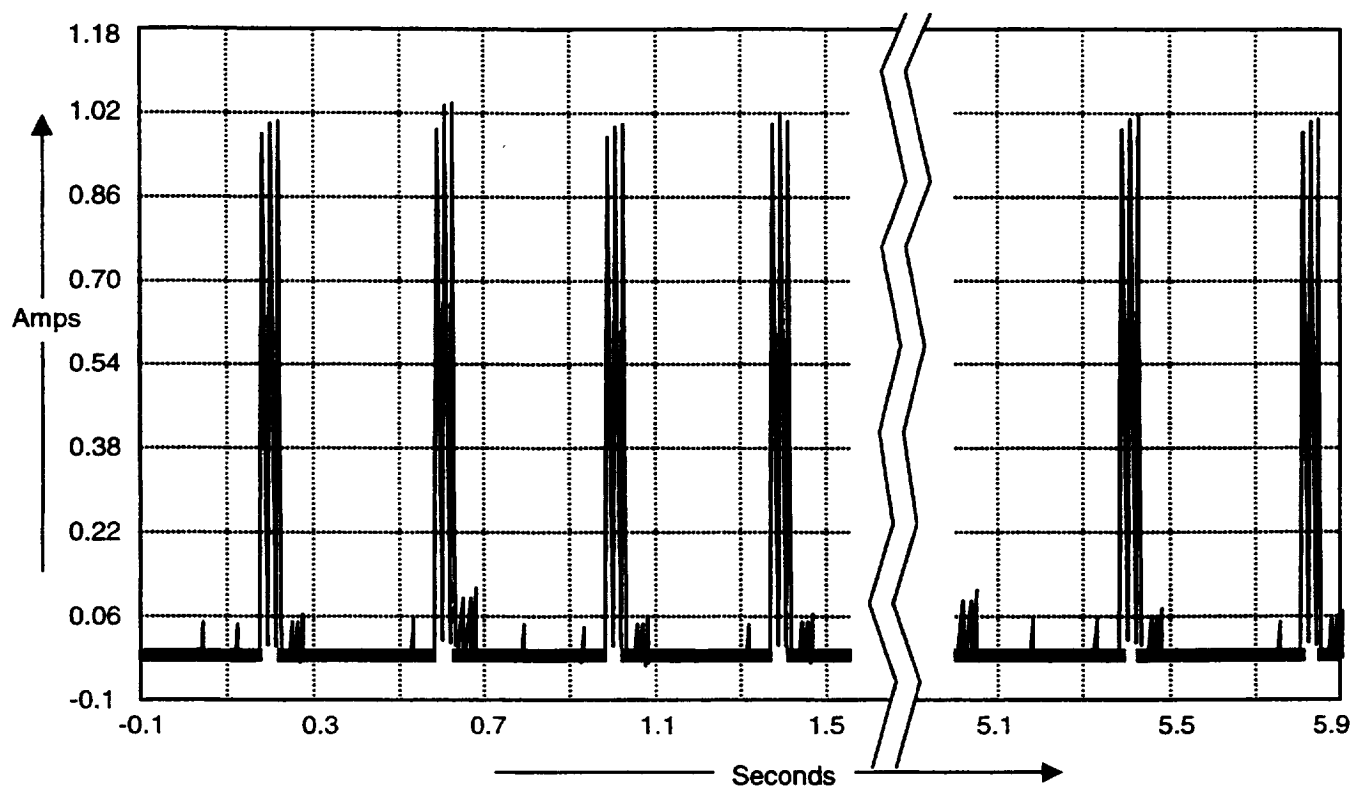


Figure 10. Typical Load Current Waveforms from the +28 V Pulse Load Bus

6Apr 99

3.2.4.2.2.4 PLB measured from 6 to 8 seconds. The PLB shall be measured as follows:

1. Reset the dynamic analyzer in accordance with 3.2.4.2.2.1(2).
2. Change the PRE-TRIGGER DELAY setting of the dynamic signal analyzer to 5.9 seconds.
3. Obtain a hard copy of the signal displayed on the dynamic signal analyzer.
4. From the hard copy obtained in step 3, calculate the peak current. Record the peak current and bus current values during the integrate/hold, dump (I/H, D) time period (refer to Figure 10) on TDS 4.

3.2.4.2.2.5 Eight second integrated current measurement. To observe the PLB integrated (8 sec.) current waveform on the dynamic signal analyzer, configure the dynamic signal analyzer as follows:

Select **SCALE**

Select X Fxd Scale; Enter 0.0, 8; Select Sec

Select Y Fxd Scale; Enter -10, 70; Select mV

Select **VIEW INPUT**

Select Time Record: Note – the display heading changes to read “Cap Tim Rec”

Select **MATH**

Select Next

Select **Intgrt:**

Note – the display changes to present an integrated value of the current waveform.

Select **X (cursor)**

Move the X marker to the maximum right of the display. The Y value is indicative of the integrated current value over the entire 8 second period (in amp-sec).

Multiply the maximum Y value by the current/div as selected on the current amplifier, then divide by 8 seconds to acquire the average current value. As an example: if the current amplifier is set up to display 200 mA/10 mV per division, and the maximum Y value = 32.4 mV-sec:

$$[32.4 \text{ mV-sec} \times (200 \text{ mA}/10 \text{ mV})]/8 \text{ sec} = 81 \text{ mA}$$

Enter the calculated integrated value on TDS 4.

3.2.4.2.2.6 PLB turn-on transient

1. Configure the unit and test equipment as shown in Figure 9. Obtain DSA trigger from J4-14. Verify that switches 5, 6, 18 and 19 of the breakout box are in the OPEN position.
2. Configure the Dynamic Signal Analyzer (DSA) as follows:

Select MEAS MODE	Select INPUT COUPLE
Select <i>Time Capture</i>	Select <i>CH1 DC</i>
Select <i>Capture Select</i>	Select <i>CH1 Ground</i>
Select <i>Capture Length</i> ; Enter 500.0; Select <i>msec</i>	Select INPUT TRIG
Select FREQ	Select <i>Trig Level</i> ; Enter 1; Select <i>V</i>
Select <i>Freq Span</i> ; Enter 20; Select kHz	Select <i>Arm AU</i>
Select <i>E SMPL Off</i>	Select <i>Extrenal</i>
Select <i>Time Length</i> ; Enter 32.0;	Select <i>Ext</i> ; Select <i>Slope(-)</i>
Select <i>msec</i>	Select TRIG DELAY
Select SELECT MEAS	Enter 0; Select <i>μSec</i>
Select <i>Power Spec</i>	Select COORD
Select <i>CH1 Active</i>	Select <i>Real</i>
Select WINDOW	Select VIEW INPUT
Select <i>Hann</i>	Select <i>Time Buff</i>
Select SOURCE	Select SCALE
Select <i>Source Off</i>	Select <i>X Fixd Scale</i> ; Enter 0.0, 25
Select AVG	Select <i>msec</i>
Select <i>Avg Off</i>	Select <i>Y Fixd Scale</i> ; Enter -10, 470
Select <i>Tim Av Off</i>	Select <i>mV</i>
Select RANGE	Select UNITS
Select <i>Chan 1 Range</i> ; Enter 1; Select <i>V</i>	Select <i>Hz (sec)</i>

NOTE

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- a) Select 200 mA/10mV per div. on the current amplifier.
- b) Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
- c) Adjust the "y" axis voltage range to ± 4 mV.
- d) Place the DSA in "Free Run" Trigger and depress "Start Capture" on the DSA.
- e) With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
- f) Position the current probe to its original location in accordance with Figure 9, and return the DSA to "Ext" trigger.

3. Adjust external power supply for +28 Vdc. Turn the unit ON by selecting [9] MODULE POWER; set up the operating modes as defined in paragraph 3.2.3.5 (reference the command screen parameters below). If necessary, re-adjust the external power supply for 28 Vdc.

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

4. Turn the unit OFF by executing command [9] MODULE POWER. Confirm the command has been executed on the STE display.
5. Start the DSA signal capture by depressing "Start Capture"; wait for the DSA message "waiting for trigger" before proceeding.
6. On the STE computer, select [9] MODULE POWER and obtain a record of the +28 PLB Turn on current waveform. On the STE computer, select [9] MODULE POWER to turn the instrument's power OFF. Adjust the display time base and voltage sensitivity to allow for adequate current and pulse duration measurements. Plot the obtained waveform and attach a hard copy of the scan to TDS 4. Refer to Figure 11 for an example of the expected waveform.
7. Measure the Turn-On pulse width; record this value on TDS 4.
8. Compute the peak current as follows:

Measure the maximum Y value by the current/div as selected on the current amplifier. As an example, if the current amplifier is set up to display 200 mA/10 mV per division, and the maximum Y value = 276 mV:

$$276 \text{ mV} \times (200 \text{ mA}/10 \text{ mV}) = 5520 \text{ mA} = 5.52 \text{ amps}$$

Record this value on TDS 4.

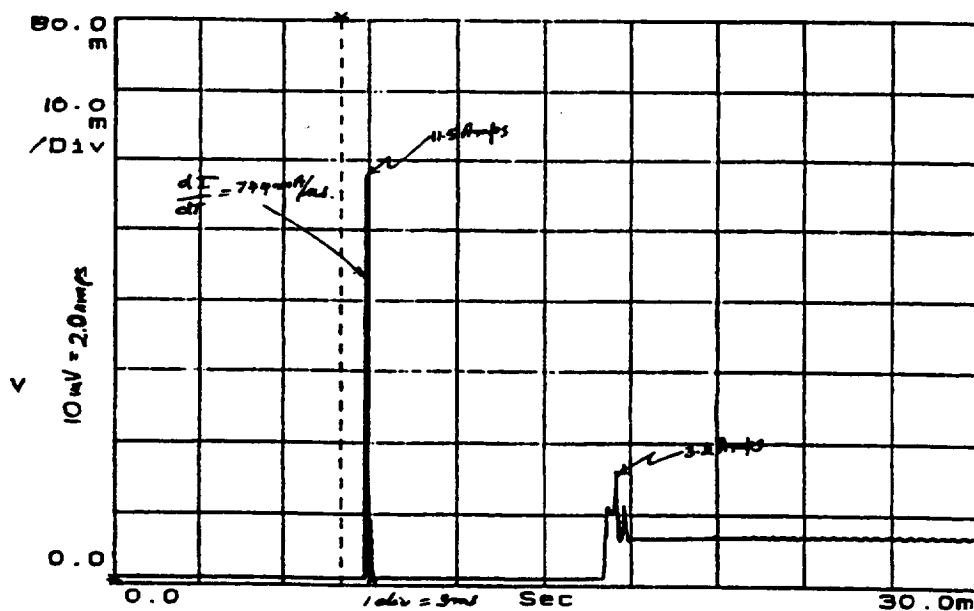
9. The 1st derivative of the current waveform must be calculated. Compute the dI/dT as follows:

The most probable location of the greatest current demand is during the first positive transition after voltage application. If this is the case, expand the segment of the display and measure the greatest voltage transition in the smallest time transition. The change in voltage times the current/div as selected on the current amplifier produces the change in current. Next divide this change in current by the change in time (in microseconds). This value is dI/dT. Example:

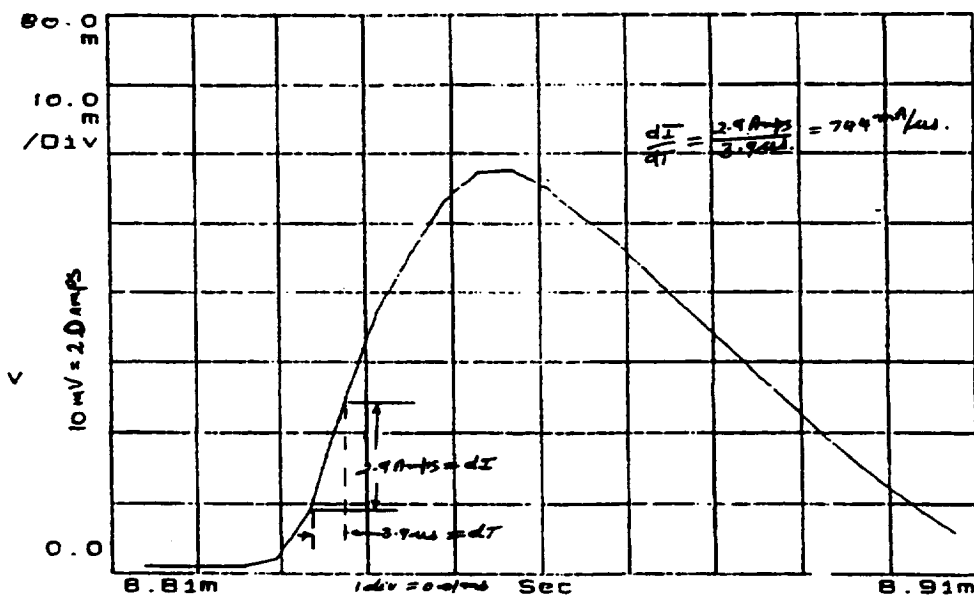
Change in voltage 144 mV
Change in time (microseconds) 19.5 μ s
Current/div on current amplifier 200 mA/10 mV

$$144 \text{ mV} \times (200 \text{ mA}/10 \text{ mV})/19.5 \mu\text{s} = 147.7 \text{ mA}/\mu\text{s}$$

10. Record the computed value on TDS 4.



AMSU-A1 PLB Worst Case Transient



AMSU-A1 PLB $\frac{dI}{dt}$ Worst Case Transient

Figure 11. +28V Pulse Load Bus Turn-on Transient

3.2.4.2.2.7 PLB current in warm cal, cold cal and Nadir mode

1. Place instrument in Warm Cal mode.
2. Measure and record PLB steady state current on TDS 4 with a multimeter in the Current mode.
3. Repeat step 2 after placing instrument in Cold Cal mode.
4. Repeat step 2 after placing instrument in Nadir mode.
5. Repeat step 2 after placing instrument in Warm Calm mode and commanding both motors off.
6. After stabilizing for a minimum of 20 scans, acquire one Full Scan mode printout, and attach it to TDS 4.

3.2.4.2.2.8 Instrument feedback test (PLB). Instrument feedback test will be performed in the EMI/RFI chamber using EMI/RFI test procedure AE-26151/5.

3.2.4.2.2.9 Transient susceptibility and power quality tests. The tests that follow will demonstrate the AMSU-A1 instrument will operate within specified parameters when the transients (low and high frequency) are applied directly to the power lines.

3.2.4.2.2.9.1 Equipment setup. Set up the test equipment and connect to the instrument as shown in Figure 12.

3.2.4.2.2.9.2 Low frequency load induced transients. The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the Pulse Load Bus power line at the amplitude and duration specified in Figure 13. Perform the Low Frequency Load Induced Transients as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. Place the signal generator in ARB 1 mode. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator to attain the signal characteristics as shown in Figure 13.
3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three (3) scans. Remove the signal generator output from the power supply.
7. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
8. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

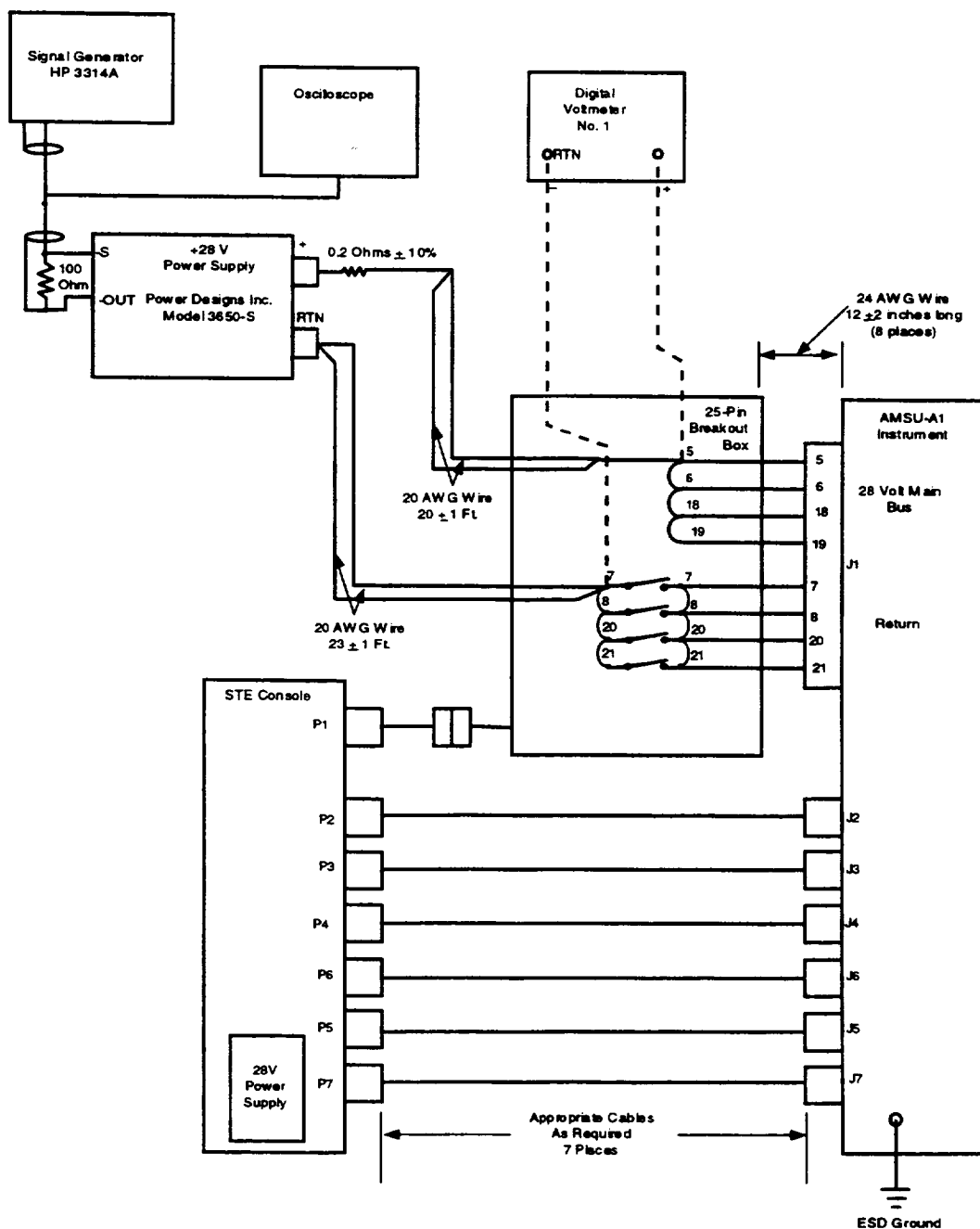


Figure 12. +28V PLB Transient Susceptibility and Power Quality Tests Setup

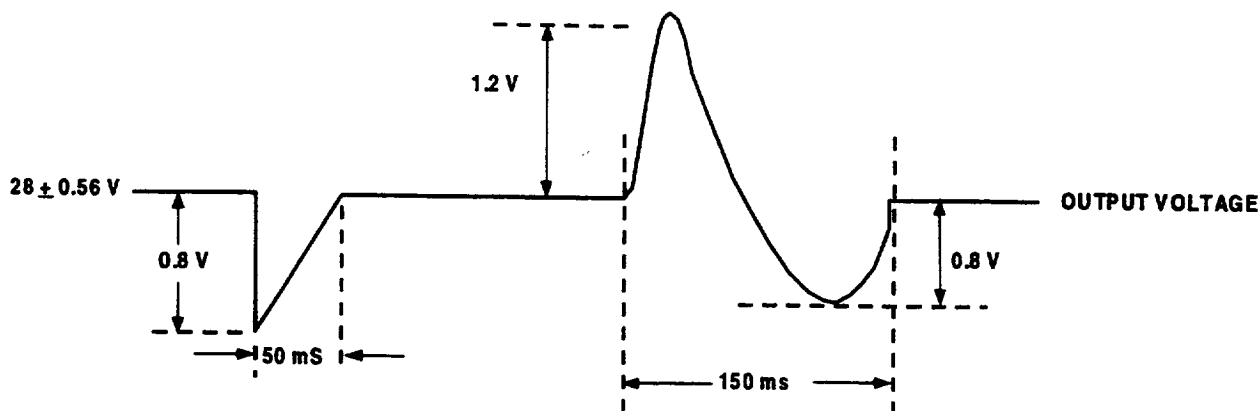


Figure 13. Load Induced Transient (Pulse Load)

3.2.4.2.2.9.3 High frequency load induced transients. The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the power line. The interfering frequencies are simulated by using the triangular wave output of the signal generator. There are three signals to be sequentially injected; the frequencies and amplitudes as produced by the signal generator and measured by the oscilloscope are:

Frequency (Hz)	Amplitude
1.43	200 mVpp
2.86	1.00 Vpp
6.67	1.50 Vpp

Tolerance on the above values is $\pm 10\%$.

Perform the High Frequency Load Induced Transients as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator output as follows:

amplitude	200 mVpp
offset	0.000 V
frequency	1.430 Hz
3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three (3) scans. Remove the signal generator output from the power supply.
7. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
8. Repeat steps 2-4 and 6-7 for 2.86 Hz and 1.0 Vpp.
9. Repeat steps 2-4 and 6-7 for 6.67 Hz and 1.5 Vpp.

10. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

3.2.4.2.3 *Analog telemetry bus*

3.2.4.2.3.1 *Operating power measurements.* The purpose of this test is to calculate the operating power of the Analog Telemetry Bus from measurements taken of the bus voltage and current.

1. Configure the instrument as shown in Figure 14.
2. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
3. Measure the bus current and record on TDS 5.
4. From the measurements recorded on TDS 5, calculate the operating power for the telemetry bus and record on TDS 5.

3.2.4.2.3.2 *Instrument feedback test (ATB).* Instrument feedback test will be performed in the EMI/RFI chamber using EMI/RFI test procedure AE-26151/5.

3.2.4.2.3.3 *Transient susceptibility and power quality tests (ATB).* The tests that follow will demonstrate the AMSU-A1 instrument will operate within specified parameters when the transients (low and high frequency) are applied directly to the power lines.

3.2.4.2.3.3.1 *Equipment setup.* Set up the test equipment and connect to the instrument as shown in Figure 15 (exceptions: remove the current probe and amplifier; connect the oscilloscope to monitor output of the signal generator).

3.2.4.2.3.3.2 *Low frequency load induced transients.* The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the power line at the amplitude and duration specified in Figure 16. Perform the Low Frequency Load Induced Transients as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. Place the signal generator in ARB 0 mode. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator to attain the signal characteristics as shown in Figure 16.
3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three (3) scans. Remove the signal generator output from the power supply.
7. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22. Attach printouts to TDS 51.
8. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

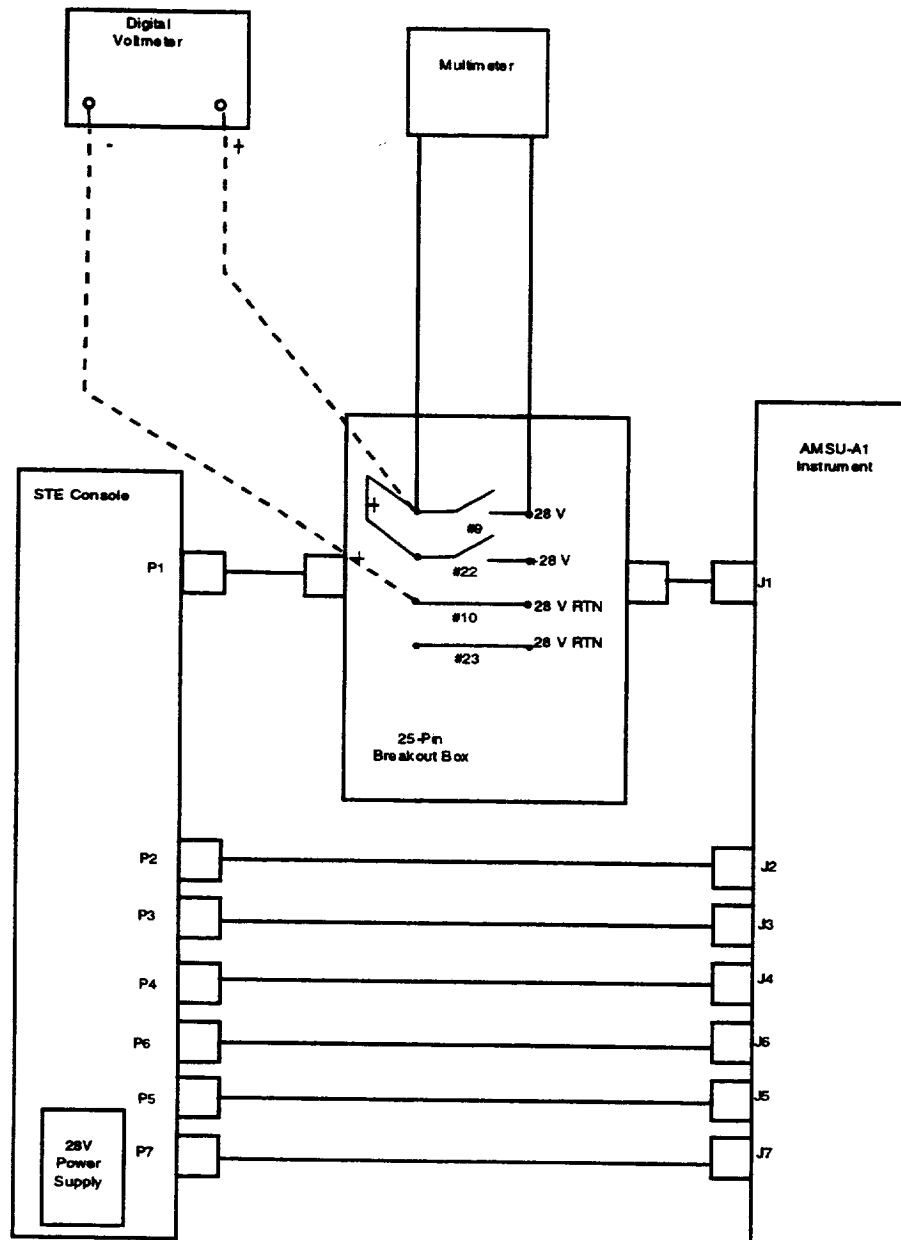


Figure 14. +28V Analog Telemetry Bus Test Setup

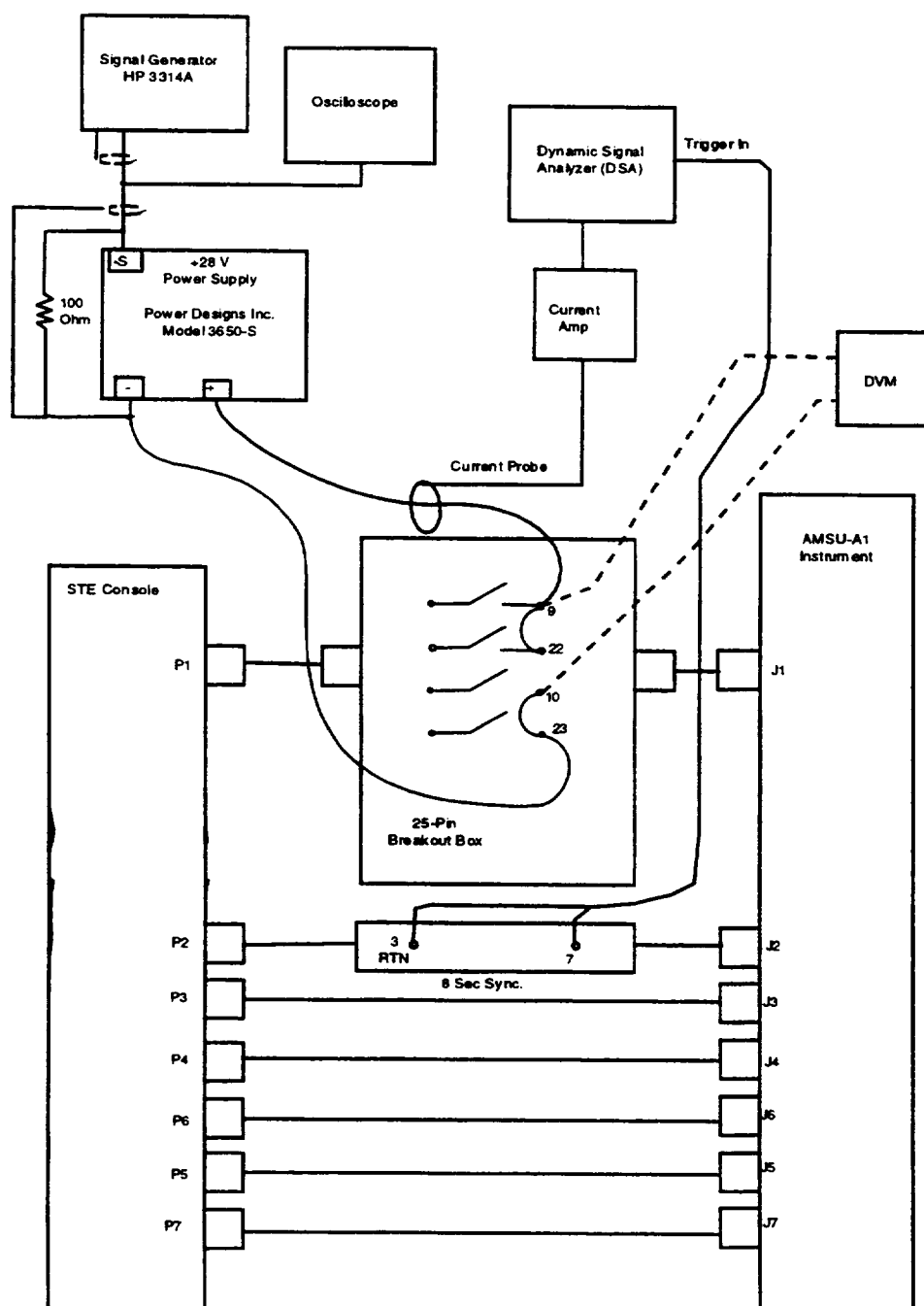
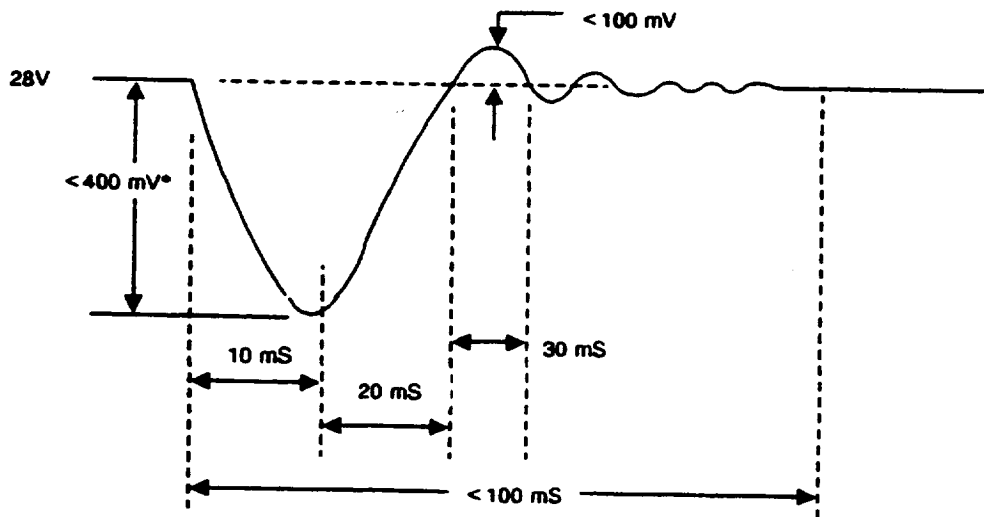


Figure 15. +28 Vdc Analog Telemetry Bus Ripple Current and Transient Susceptibility Test Setup



* Typical transients occurring a number of times per orbit are on the order of 200 mV zero-to-peak for a 1.5A load change.

Figure 16. Load Induced Transient (Main Bus)

3.2.4.2.3.3.3 High frequency load induced transients. The AMSU instrument shall be capable of normal operation before and after positive and negative transients are injected into the power line. The interfering frequencies are simulated by using the triangular wave output of the signal generator. There are three signals to be sequentially injected; the frequencies and amplitudes as produced by the signal generator and measured by the oscilloscope are:

<u>Frequency (Hz)</u>	<u>Amplitude</u>
1.43	200 mVpp
2.86	1.00 Vpp
6.67	1.50 Vpp

Tolerance on above values is $\pm 10\%$.

Perform the High Frequency Load Induced Transients as follows:

1. With the exception of the external power supply, turn ON all the test equipment.
2. With the external power supply OFF, while monitoring the oscilloscope, adjust the amplitude and frequency output of the signal generator output as follows:

..... amplitude	200 mVpp
offset	0.000 V
frequency	1.430 Hz

3. Remove the signal generator output connection from the power supply. While monitoring the external power supply dc voltage with the meter, turn the external power supply ON.
4. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
5. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
6. Connect the signal generator to the external power supply. Wait for the instrument to complete three (3) scans. Remove the signal generator output from the power supply.

7. Acquire one (1) Full Scan Mode printout; verify the printout meets the requirements of TDS 19 thru 22.
8. Repeat steps 2-4 and 6-7 for 2.86 Hz and 1.0 Vpp.
9. Repeat steps 2-4 and 6-7 for 6.67 Hz and 1.5 Vpp.
10. Record any deviations in the functional performance of the AMSU instrument on TDS 51.

3.2.4.2.4 +10 volt interface bus test

3.2.4.2.4.1 Operating power measurements. The purpose of this test is to calculate the operating power of the +10 Vdc Interface Bus from measurements taken of the bus voltage and current.

1. Configure the instrument as shown in Figure 17.
2. Turn the instrument ON and place the instrument in the modes congruent with paragraph 3.2.3.5.
3. Measure the bus current and record on TDS 6.
4. From the measurements recorded on TDS 6, calculate the operating power for the telemetry bus and record on TDS 6.

3.2.4.2.4.2 Instrument feedback test. Instrument feedback test will be performed in the EMI/RFI chamber using EMI/RFI test procedure AE-26151/5.

3.2.4.2.5 Power input test for LPT. For LPT, test the power input as follows:

1. Configure the unit and test equipment as indicated in Figure 18.
2. Turn the unit ON as described in 3.2.3.5. Set the STE power supply voltage at 28.00 ± 0.05 Vdc using 25-pin breakout box and DVM #1.

NOTE

Do not proceed without successful completion of step 2.

3. Record the voltage from DVM #1 and current in Amps from STE current meter on TDS 7.

3.2.4.3 Clock, commands, and data system test. This procedure verifies the clock signal, the commands, and the data requirements specified in S-480-80, GHS IS-3267415, and UHS IS-2617547.

3.2.4.3.1 Test sequence. The test sequence shall be as follows:

- a. Clock signals verification
- b. Commands and Digital-B telemetry verification
- c. Data output verification
 - (1) Digital-A
 - (2) Analog telemetry
 - (3) Test points
- d. GSE modes.

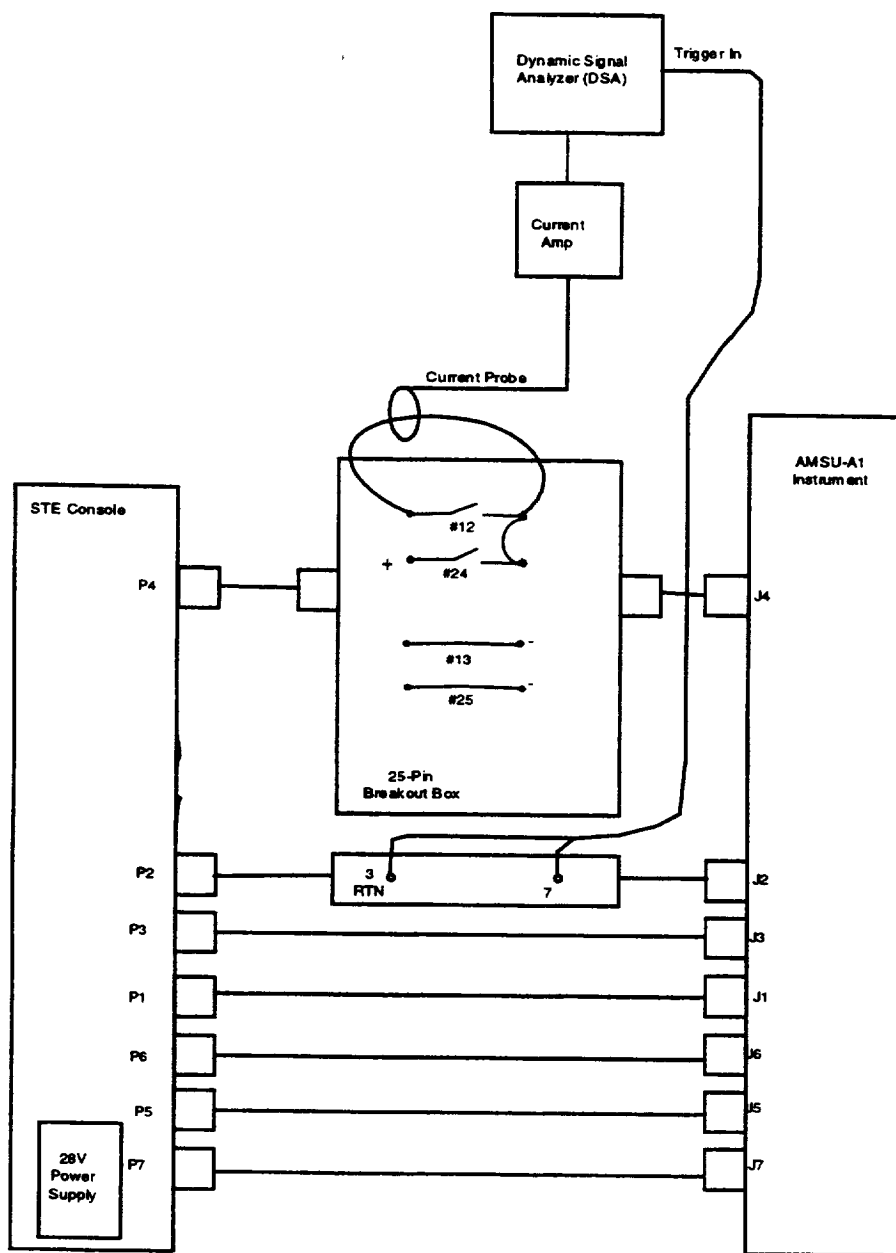


Figure 17. +10V Interface Bus Operating Power and Ripple Current Measurements Test Setup

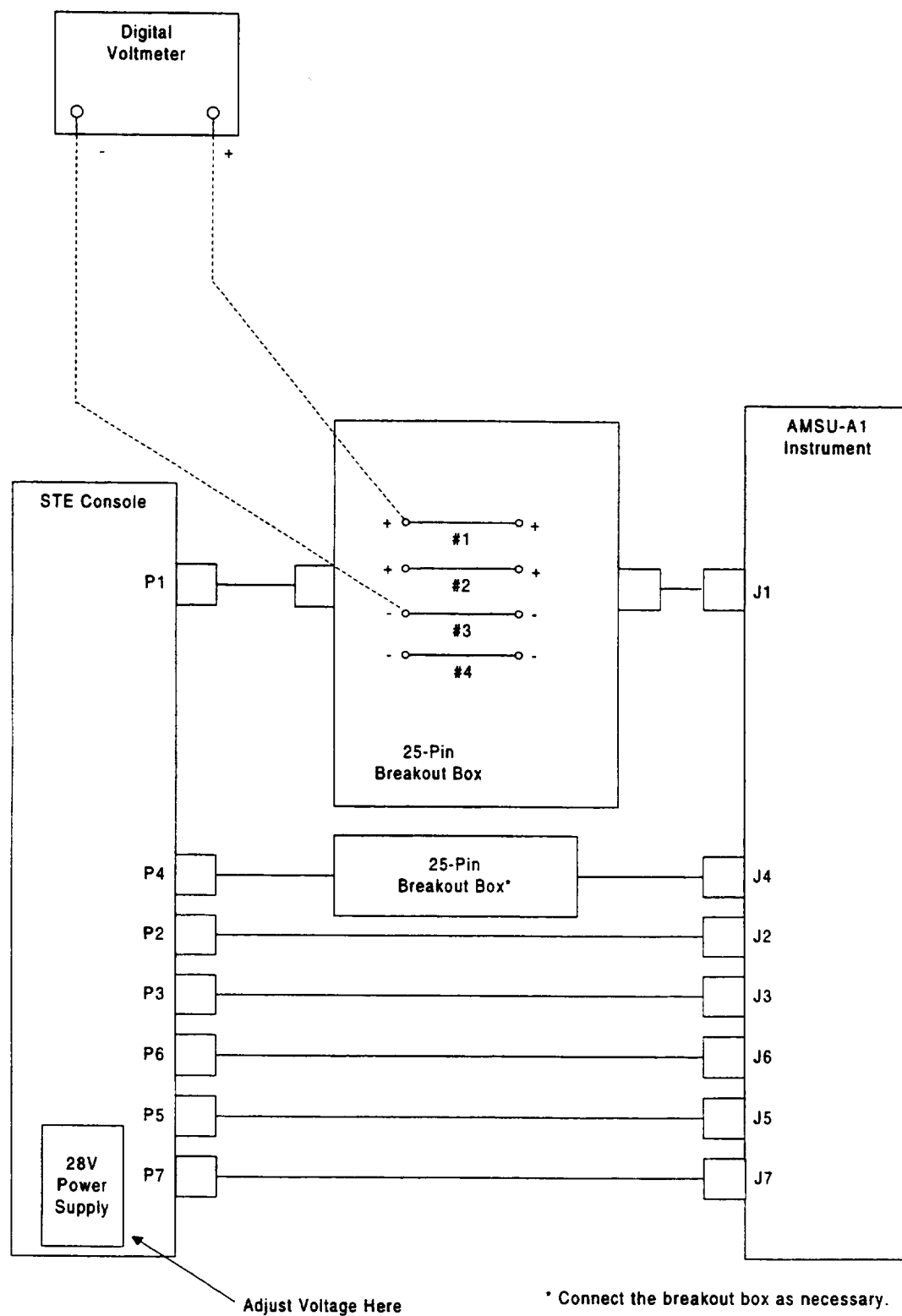


Figure 18. +28 V Main Load Bus Test Setup (For LPT Only)

6Apr 99

3.2.4.3.2 Clock signals test. The following items shall be tested to verify the clock signals. Refer to Figure 19 for graphical representation of these pulses.

- a. 1.248 MHz clock
- b. 8 seconds frame pulse
- c. A1 select pulse
- d. C1 shift pulse

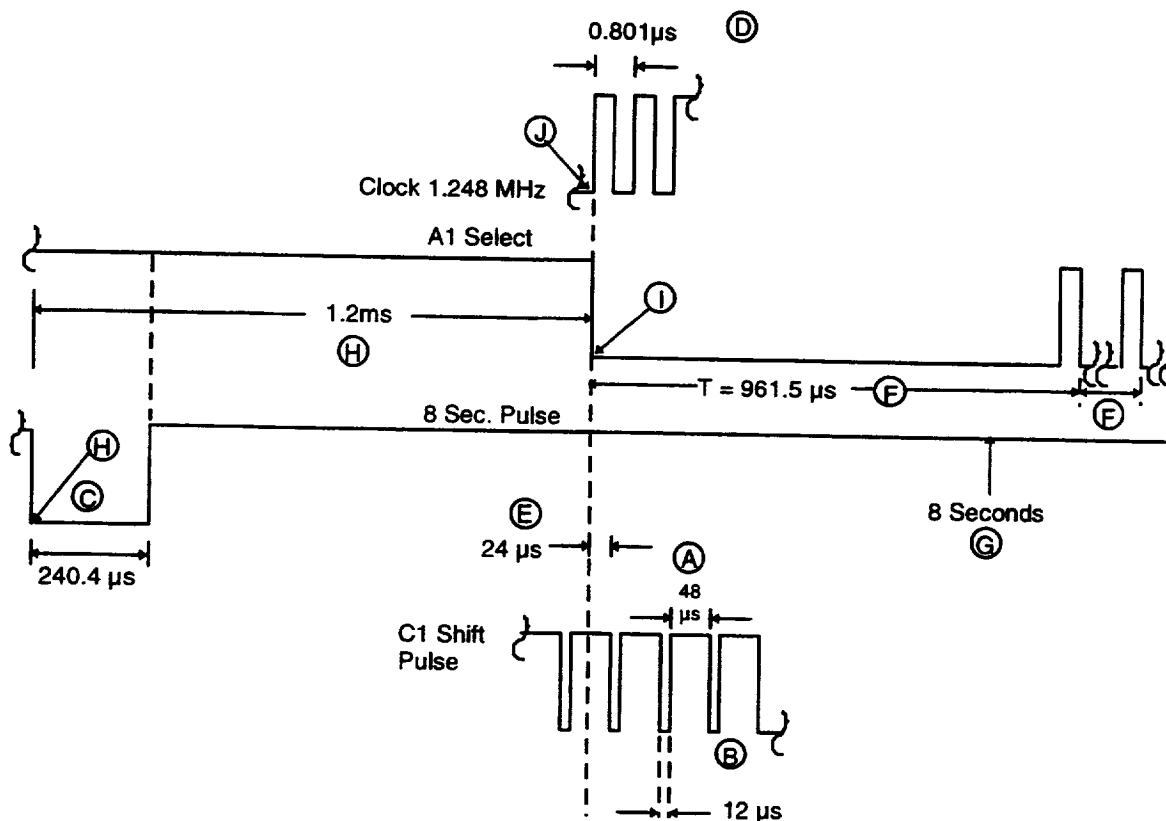


Figure 19. Clock Pulses Timing and Synchronization

3.2.4.3.2.1 1.248 MHz synchronization clock. Perform the following procedures:

1. Configure the unit and the test equipment as indicated in Figure 20.
2. Connect CHANNEL-1 of the oscilloscope to the 1.248 MHz clock signal as shown in Figure 20.
3. Turn the unit ON as described in 3.2.3.5.

NOTE

Do not proceed without successful completion of step 3.

4. Using the oscilloscope, measure the 1.248 MHz clock signal. Record the data and attach the photograph or plot on TDS 8.

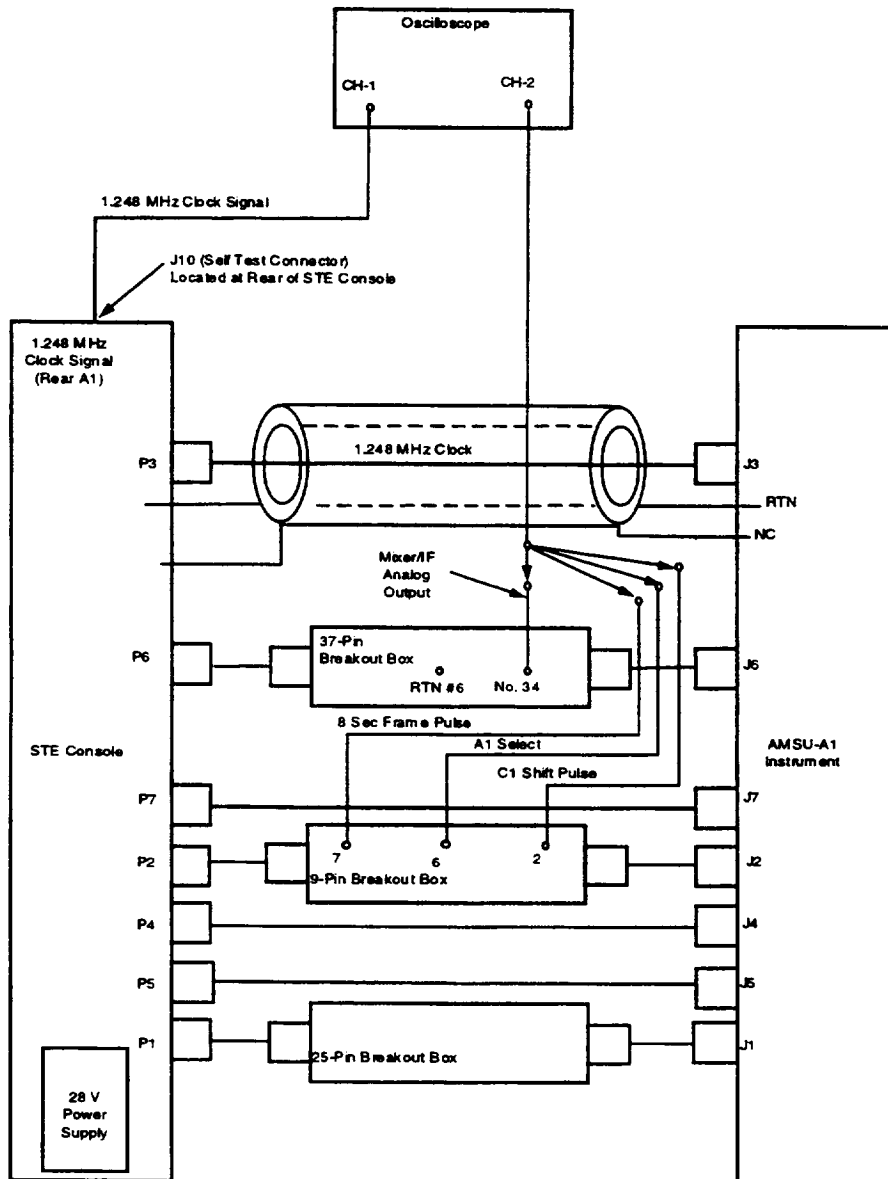


Figure 20. Clock Signals Test Setup

3.2.4.3.2.2 C1 shift pulse verification. Connect CHANNEL-2 of the oscilloscope to Pin 2 of the 9-pin breakout box (P2-J2). Photograph or plot the oscilloscope display and record the information indicated on TDS 9.

3.2.4.3.2.3 A1 select pulse verification. Connect CHANNEL-2 of the oscilloscope to Pin 6 of the 9-pin breakout box (P2-J2). Photograph or plot the oscilloscope display and record the information indicated on TDS 10.

3.2.4.3.2.4 8-seconds frame sync pulse verification

1. Connect CHANNEL-2 of the oscilloscope to Pin 7 of the 9-pin breakout box (P2-J2). Photograph or plot the oscilloscope display and record the information indicated on TDS 11. (Record of "C" timing only, is required.)
2. Turn the unit OFF by executing the softkey command [11] MODULE TOTALLY OFF to OFF. Leave both breakout boxes in place.

3.2.4.3.2.5 Synchronization signal relationship. The following synchronization signal relationship shall be verified.

- a. A1 select pulse and the 8-second frame sync pulse
 1. With the unit off, configure the unit and the test equipment as indicated in Figure 21.
 2. Connect CHANNEL-1 of the oscilloscope to the breakout box, Pin 6 (A1).
 3. Adjust the amplitude and the trigger level of the oscilloscope for best picture.
 4. Photograph or plot the oscilloscope display and attach the photograph or plot in the space provided on TDS 12.
 5. From the photograph or plot, verify the synchronization as described in TDS 12. Record pass or fail.
- b. A1 select pulse and C1 shift pulse
 1. Connect CHANNEL-2 of the oscilloscope to the breakout box Pin 2 (C1 shift pulse).
 2. Adjust the amplitude and the trigger level of the oscilloscope for best picture.
 3. Photograph or plot the oscilloscope display and attach the photograph or plot in the space provided on TDS 12, sheet 2.
 4. From the photograph or plot, verify the synchronization as described in TDS 12, sheet 2. Record pass or fail.
- c. A1 select pulse and 1.248 MHz clock.
 1. Connect CHANNEL-2 of the oscilloscope to the clock connector located at the rear of the STE.
 2. Adjust the amplitude and the trigger level of the oscilloscope for best picture.
 3. Photograph or plot the oscilloscope display and attach the photograph or plot in the space provided on TDS 13.
 4. From the photograph or plot, verify the synchronization as described in TDS 13. Record pass or fail.

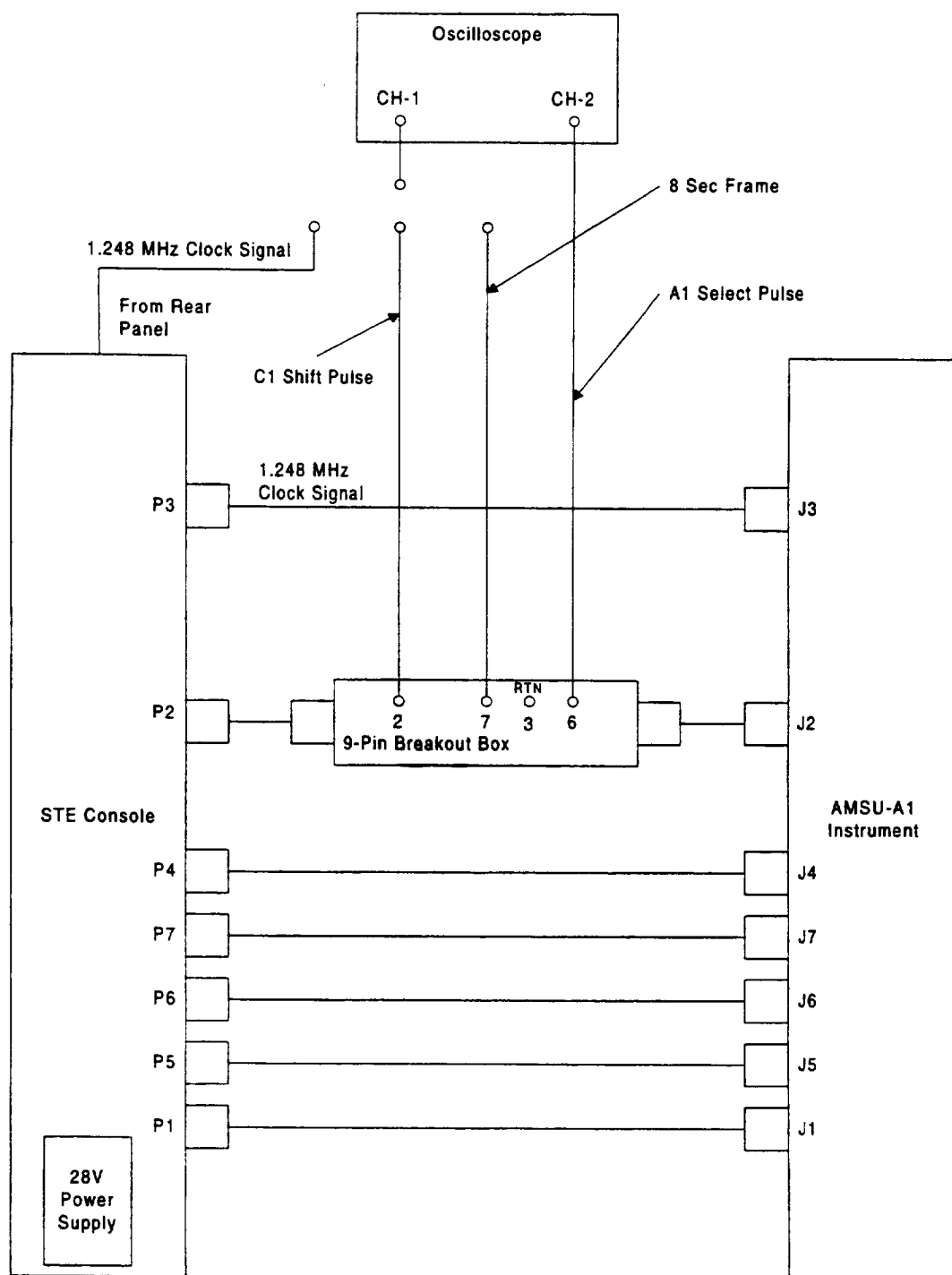


Figure 21. Synchronization Signal Relationships Test Setup

3.2.4.3.3 Commands and digital-B telemetry test. Commands and digital-B telemetry shall be verified in accordance with the following paragraphs.

3.2.4.3.3.1 Module totally off. Commands and digital-B telemetry, with the module totally off, shall be tested as follows:

1. Turn the unit on as follows:
 - a. Press [12] POWER ON (from 1st screen).
 - b. Press [2] MONITOR ONLY (from 1st screen)
 - c. Press [14] COMMANDS (from 2nd screen)

Verify the screen displays the default parameters below.

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	YES	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

2. From the Commands Menu, execute command [11] MODULE TOTALLY OFF to OFF mode.
3. Wait at least 18 seconds, then verify that the following events are in effect:
 - a. [11] MODULE TOTALLY OFF = OFF
 - b. [12] SCANNER A1-1 POWER = OFF.
 - c. [13] SCANNER A1-2 POWER = OFF.
 - d. [10] SURVIVAL HEATER POWER = OFF

Antenna reflectors for A1-1 and A1-2 pointing toward the warm load.

4. Record the above observations on TDS 14.

3.2.4.3.3.2 Survival heater power ON/OFF command. The survival heater power ON/OFF command shall be tested as follows:

1. Execute command [10] SURVIVAL HEATER POWER to ON mode. Wait at least 18 seconds. Verify that the command is in effect. Record observation on TDS 14.
2. Execute command [10] SURVIVAL HEATER to OFF mode. Wait at least 18 seconds. Verify that the command is in effect. Record observation on TDS 14.

3.2.4.3.3.3 Module power connect command. The module power connect command shall be tested as follows:

1. Execute command [9] MODULE POWER to CONNECT mode. Wait at least 18 seconds. Verify that the command is in effect. Record observation on TDS 14.
2. Verify that the current at the STE power supply is 0.5 to 4.3 Amperes. Record this information on TDS 14.

3.2.4.3.3.4 Phase lock loop (PLL) PLLO No. 1 / PLLO No. 2. The PLL PLLO No. 1/PLLO No. 2 command shall be tested as follows:

1. Execute [18] PLL POWER = PLLO#2
Wait at least 18 seconds. Verify that the command is in effect. Record observation on TDS 14.
2. Execute [18] PLL POWER = PLLO#1
Wait at least 18 seconds. Verify that the command is in effect. Record observation on TDS 14.

3.2.4.3.3.5 Scanner commands verification. The scanner commands shall be tested as follows:

1. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

Wait at least 18 seconds. Verify that the commands are in effect. Record observations on TDS 15.

2. Execute. [12] SCANNER A1-1 POWER = OFF
[13] SCANNER A1-2 POWER = OFF

Wait at least 18 seconds. Verify that the commands are in effect. Record observations on TDS 16.

3. Execute. [12] SCANNER A1-1 POWER = ON
[13] SCANNER A1-2 POWER = ON

Wait at least 18 seconds. Verify that the commands are in effect. Record observations on TDS 17.

3.2.4.3.3.6 Scanner position commands (A1-1 and A1-2) verification. Verify scanner position command operation as follows:

NOTE

Verification of the scan position is applicable to both antenna reflectors located at the high and low bays of the instrument (A1-1 and A1-2).

1. Execute: [14] ANTENNA WARM CAL POS = YES
[17] ANTENNA FULL SCAN MODE = NO

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

2. Execute: [15] ANTENNA IN COLD CAL POS = YES
[14] ANTENNA WARM CAL POS = NO

Execute: [19] COLD CAL POS MSB = zero
[20] COLD CAL POS LSB = one

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

3. Execute: [19] COLD CAL POSITION MSB = ONE
 [20] COLD CAL POSITION LSB = ZERO

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

4. Execute: [19] COLD CAL POSITION MSB= ONE
 [20] COLD CAL POSITION LSB= ONE

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

5. Execute: [19] COLD CAL POSITION MSB= ZERO
 [20] COLD CAL POSITION LSB= ZERO

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

6. Execute: [16] ANTENNA IN NADIR POSITION = YES
 [15] ANTENNA IN COLD CAL POS = NO

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

7. Execute: [14] ANTENNA WARM CAL POS = YES

Wait at least 18 seconds. Verify that the commands are in effect. Record observation on TDS 18.

3.2.4.3.4 Digital-A data output verification. The following items shall be tested to verify the digital-A data output:

- a. Full scan (3.2.4.3.4.1)
- b. Warm load (3.2.4.3.4.2)
- c. Cold cal (3.2.4.3.4.3)
- d. Nadir (3.2.4.3.4.4).

For each of the above scan modes, the following parameters will be subject to pass/fail criterion:

- [I] Sync. sequence
- [II] Unit I.D. and serial number
- [III] Digital-B serial data verification
- [IV] Reflector positions

[V] Radiometric data (scene data)

Radiometric data shall be obtained from two channels only, Channels 9 and 3. Channel 9 is physically located at the high bay of the sensor (A1-1 location) and Channel 3 is located at the lower bay of the sensor (A1-2 location).

[VI] Temperature sensors.

For the cold cal mode, reflector position [IV], verify the following:

- (a) Cold cal position with MSB=1 and LSB=0
- (b) Cold cal position with MSB=0 and LSB=1
- (c) Cold cal position with MSB=1 and LSB=1.

NOTE

The calibration data for the selected AMSU-A1 sensor serial number is required prior to the start of this test. Refer to 3.2.4.3.4.1.

3.2.4.3.4.1 Full scan mode. The digital-A data output in full-scan mode shall be tested as follows:

1. Turn the unit on. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

2. Obtain a full printout (9 pages) of all the parameters ([I] through [VI]) described above, by touching the PRINT [3] FULL touch area. The computer will start printing all 9 pages of data.
3. Label 1st page of 9 pages with the unit serial number and the paragraph number corresponding to this test.

(I), (II), and (III) Sync, Unit ID, and Digital-B Data

4. Using Page 1 of the printout, verify that elements 0001 through 0008 are within the required values specified in TDS 19. Record pass or fail.

[IV] Reflector position

NOTE

To verify the following steps, the operator may print out the individual parameters by using AE-26157 and attach the data to each TDS.

5. Using the individual printout, verify that there is no "E" ERROR Flag (for S/N 102 through 104) on the computer printout. Record pass or fail on TDS 20. For S/N 105 and up, verify that position values are within ± 10 counts from requirement provided in TDS 6, AE-26002/1.

[V] Radiometric data

6. Using the individual printout, verify that the data are within the values specified on TDS 21. Record pass or fail.

[VI] Temperature sensors

7. Using the individual printout, verify that elements 1090 through 1180 are within the values specified on TDS 22 (sheets 1 and 2). Record pass or fail.

3.2.4.3.4.2 Warm cal mode. The digital-A data output, in warm-cal mode shall be tested as follows:

1. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	YES	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

[II], [III], and [III] Sync, Unit ID, and Digital-B Data

2. Using Page 1 of the printout, verify that elements 0001 through 0008 are within the required values specified in TDS 23. Record pass or fail.

NOTE

To verify the following steps, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS.

[IV] Reflector position

3. Using the individual printout, verify that there is no "E" ERROR Flag (for S/N 102 through 104) on the computer printout. Record pass or fail on TDS 24. For S/N 105 and up, verify that position values are within ± 10 counts from requirement provided in TDS 6, AE-26002/1.

[V] Radiometric data

4. Using the individual printout, verify that the data are within the values specified on TDS 25. Record pass or fail.

[VI] Temperature sensors

5. Using the individual printout, verify that elements 1090 through 1180 are within the values specified on TDS 26 (sheets 1 and 2). Record pass or fail.

3.2.4.3.4.3 Cold cal mode. The digital-A data output, in cold-cal mode, shall be tested as follows:

1. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	YES [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

[II], [III] and [III] Sync, Unit ID, and Digital "B" data

2. Using Page 1 of the printout, verify that elements 0001 through 0008 are within the required values specified in TDS 27. Record pass or fail.

NOTE

To verify the following steps, the operator may print out the individual parameters by using AE-26157 and attach the data to each TDS.

[IV] Reflector position

3. Using the individual printout, verify that there is no "E" ERROR Flag (for S/N 102 through 104) on the computer printout for steps 4a, 4b, 4c, and 4d. For S/N 105 and up, verify that position values are within ± 10 counts from requirement provided in TDS 6, AE-26002/1.
4. To test the cold cal reflector position, perform the following substeps:
 - a. Using AE-26157; select reflector position screen, execute PRINT [2] SCREEN ONLY, and attach the data to TDS 28. Verify that there is no "E" ERROR Flag (for S/N 102 through 104) on the computer printout. Record pass or fail on TDS 28. For S/N 105 and up, verify that position values are within ± 10 counts from requirement provided in TDS 6, AE-26002/1.
 - b. Execute commands [19] COLD CAL POSITION MSB to 0 and [20] COLD CAL POSITION LSB to 1. Repeat substep a. then proceed to substep c.
 - c. Execute commands [19] COLD CAL POSITION MSB to 1 and [20] COLD CAL POSITION LSB to 0. Repeat substep a., then proceed to substep d.
 - d. Execute commands [19] COLD CAL POSITION MSB to 1 and [20] COLD CAL POSITION LSB to 1. Repeat substep a., then proceed to substep e.
 - e. Execute commands [19] COLD CAL POSITION MSB to 0 and [20] COLD CAL POSITION LSB to 0.

[V] Radiometric data

5. Using the individual printout, verify that the data are within the values specified on TDS 29. Record pass or fail.

6Apr 99

[VI] Temperature sensors

6. Using the individual printout, verify that elements 1090 through 1180 are within the values specified on TDS 30 (sheets 1 and 2). Record pass or fail.

3.2.4.3.4.4 Nadir cal mode. The digital-A data output, in nadir-cal mode, shall be tested as follows:

1. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	YES [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

[II], [III] and [III] Sync, Unit ID, and Digital "B" data

2. Using the individual printout, verify that elements 0001 through 0008 are within the required values specified in TDS 31. Record pass or fail.

NOTE

To verify the following steps, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS.

[IV] Reflector position

3. Using the individual printout, verify that there is no "E" ERROR Flag (for S/N 102 through 104) on the computer printout. Record pass or fail on TDS 24. For S/N 105 and up, verify that position values are within ± 10 counts from requirement provided in TDS 6, AE-26002/1

[V] Radiometric data

4. Using the individual printout, verify that the data are within the values specified on TDS 32. Record pass or fail.

[VI] Temperature sensors

5. Using the individual printout, verify that the elements 1090 through 1180 are within the values specified on TDS 33 (sheets 1 and 2). Record pass or fail.

3.2.4.3.5 Analog telemetry test. The purpose of this test is to verify that the 26 analog telemetry signals are within requirements. The purpose of the analog telemetry signals is to provide information about the functionality of the subsystems during normal operation of the unit. The analog telemetry signals shall be verified in two ways: (1) by measuring the analog telemetry signals directly at the interfacing connector and (2) by use of the STE.

3.2.4.3.5.1 Analog TLM signals measurements connector J6. Measure analog TLM signals at connector J6 as follows:

1. Configure the unit and the STE as indicated in Figure 22. Verify that unit power is off prior to the installation of the breakout boxes. To turn the unit off, select the Commands Menu and execute command [9] MODULE POWER = DISCONNECT and POWER [4] OFF. Manually turn off the STE 28 V power supply located inside the STE console.
2. Turn the unit on as follows:
 - (a) Turn on the STE 28 V power supply.
 - (b) On the Commands Menu, execute: POWER [4] ON and [9] MODULE POWER = CONNECT. Verify the display is as follows.

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	YES [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON			

3. Using the "28 V Analog Telemetry Bus Return" (J1-10) as a reference ground, measure and record the six temperature sensor voltages in the order specified on TDS 34.
4. Using the "Signal Ground" (J2-03) as a reference ground, measure and record the remaining analog telemetry voltage levels in the order specified on TDS 34.
5. Leave the unit on in preparation for the next test.

3.2.4.3.5.2 Analog TLM signal measurements using the STE. Analog TLM signal measurements using the STE shall be taken as follows:

1. Using the individual printout, verify that the data matches the values specified on TDS 35. Record pass or fail.
2. Attach computer individual printout to TDS 35.

3.2.4.3.6 Test point verification. The purpose of this test is to verify the performance of the integrator and its associated clock pulses. Figure 2 shows the integration waveform and the clock signals. Test point verification consists of the following parameters:

- a. Integration/Hold and Dump Clock Signals. (3.2.4.3.6.1) (Time and amplitude)
- b. Integration Time (Analog Output). (3.2.4.3.6.2) (Time and amplitude for all 13 channels.)

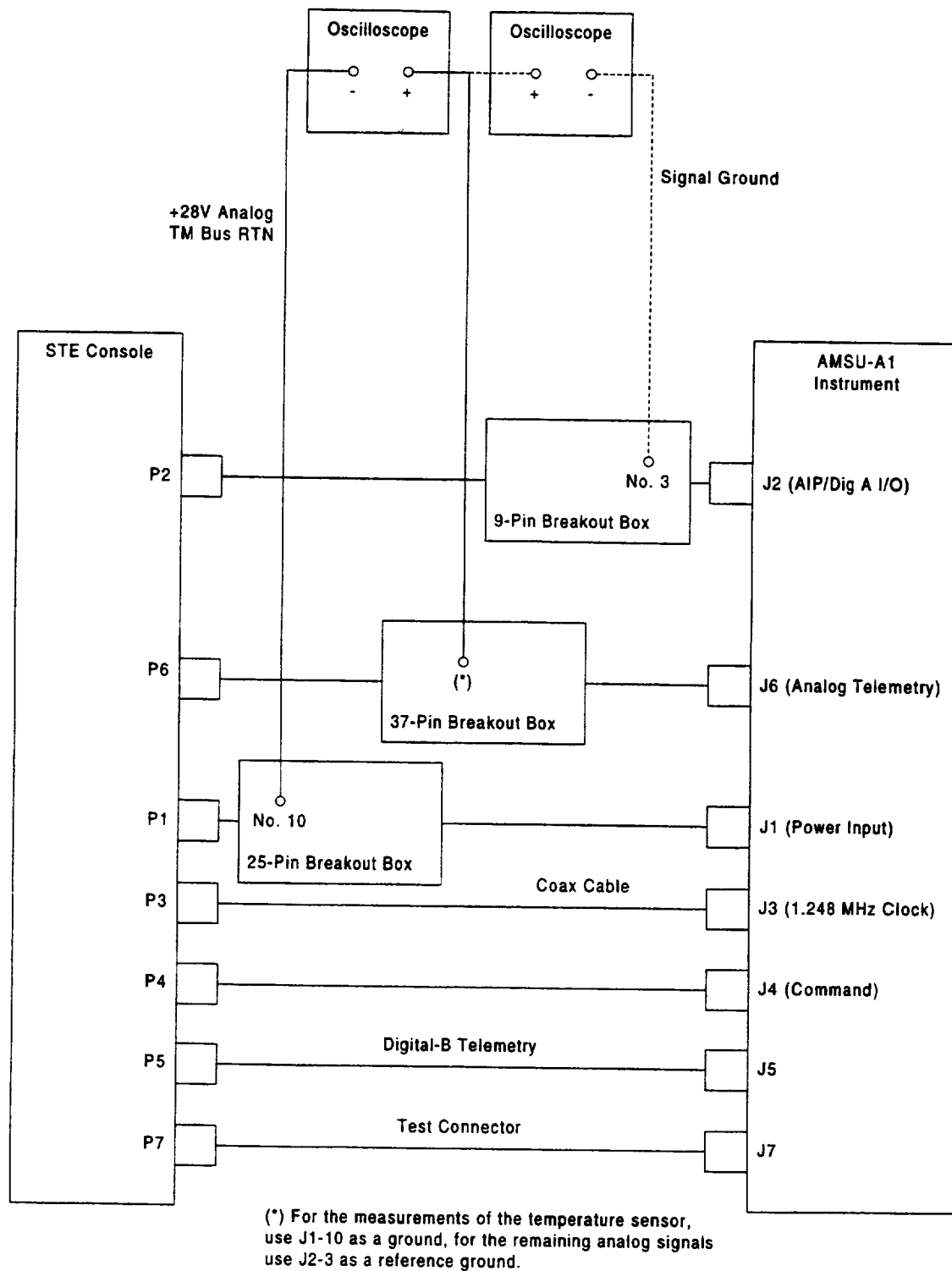


Figure 22. Analog Telemetry Signal Verification Test Setup

3.2.4.3.6.1 Integration/hold and dump clock signals. The integration/hold and dump clock signals shall be tested as follows:

1. Referring to Figure 23, configure the oscilloscope as follows:
 - (a) Channel-2 to J7-06 dump clock signal.
 - (b) Channel-1 to J7-24 integration/hold clock signal.
 - (c) Channel-1 (shielded cable) to J7-05 (I/H and Dump RTN).
 - (d) Internal trigger mode to channel-1.
 - (e) Amplitude and Time optimized for best resolution.
2. Photograph or plot the oscilloscope display and attach the photograph or plot to TDS 36.
3. From the photograph or plot, measure time and amplitude for the integrate/hold and dump clock signals. Verify that the data obtained are within the requirements specified on TDS 36 and Figure 2.
4. Leave the equipment in place and the unit turned on in preparation for the next test.

3.2.4.3.6.2 Integration time (analog outputs). The analog outputs integration time shall be tested as follows:

1. Reconfigure the test equipment as indicated in Figure 24.
2. Connect the oscilloscope, channel-2 positive line to J7-XX of the 37-pin breakout box. Where: XX indicates the pinout distribution for all the 13 channels as shown in Table III.
3. Start with the first channel of the above list. Adjust the oscilloscope for best amplitude and time resolution. The displayed signals should look like Figure 2.
4. Photograph or plot the display and attach it to the corresponding TDS (TDSs 37 through 43).
5. From the photograph or plot, measure the integration time and the amplitude. Verify that the data obtained is within the requirements specified in TDSs 37 through 43.
6. Repeat steps 2 through 5 to measure the integration time (analog output) for the remaining channels.
7. Leave the unit turned on and the test equipment in place in preparation for the next test.

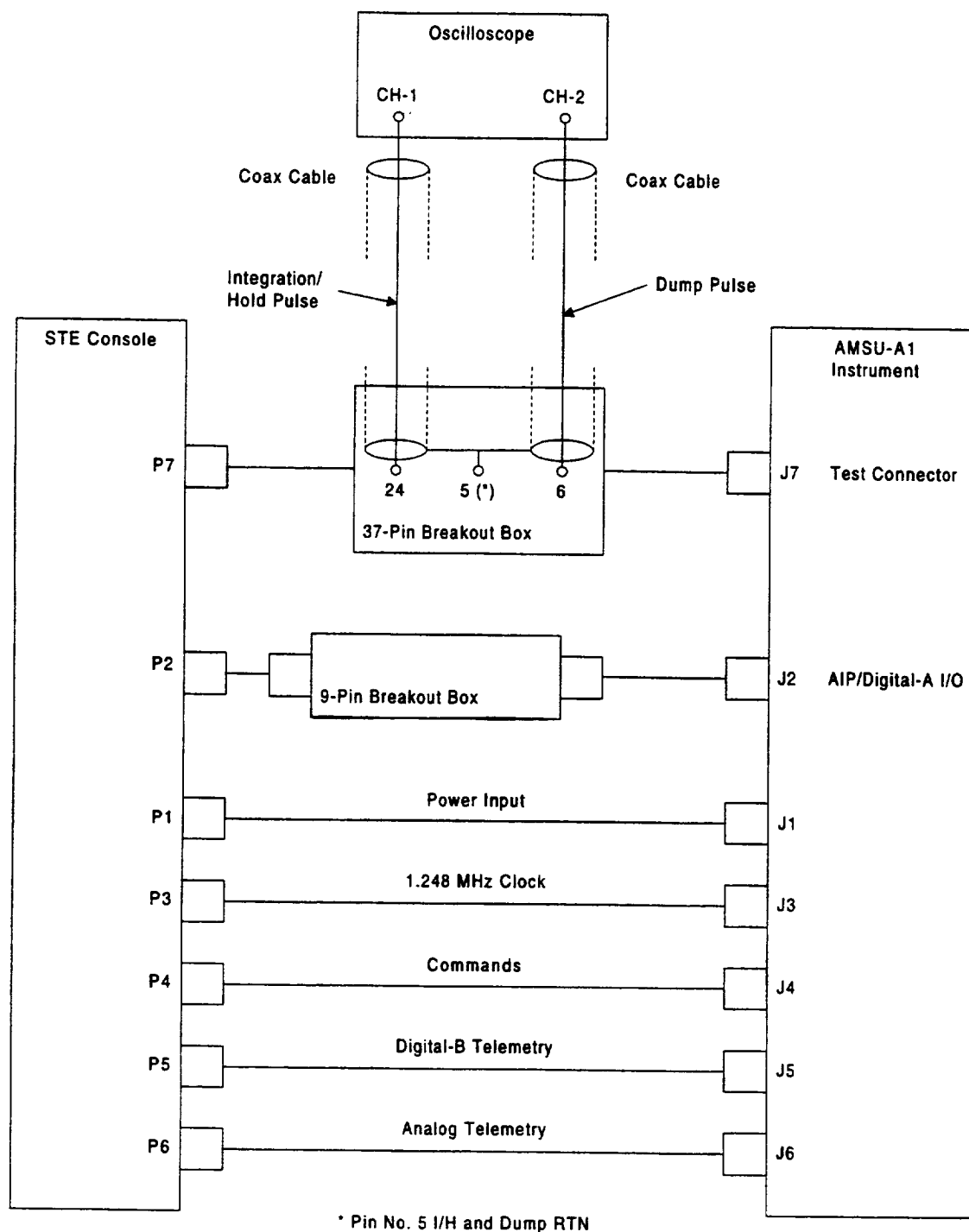


Figure 23. Integration/Hold and Dump Signals Verification Test Setup

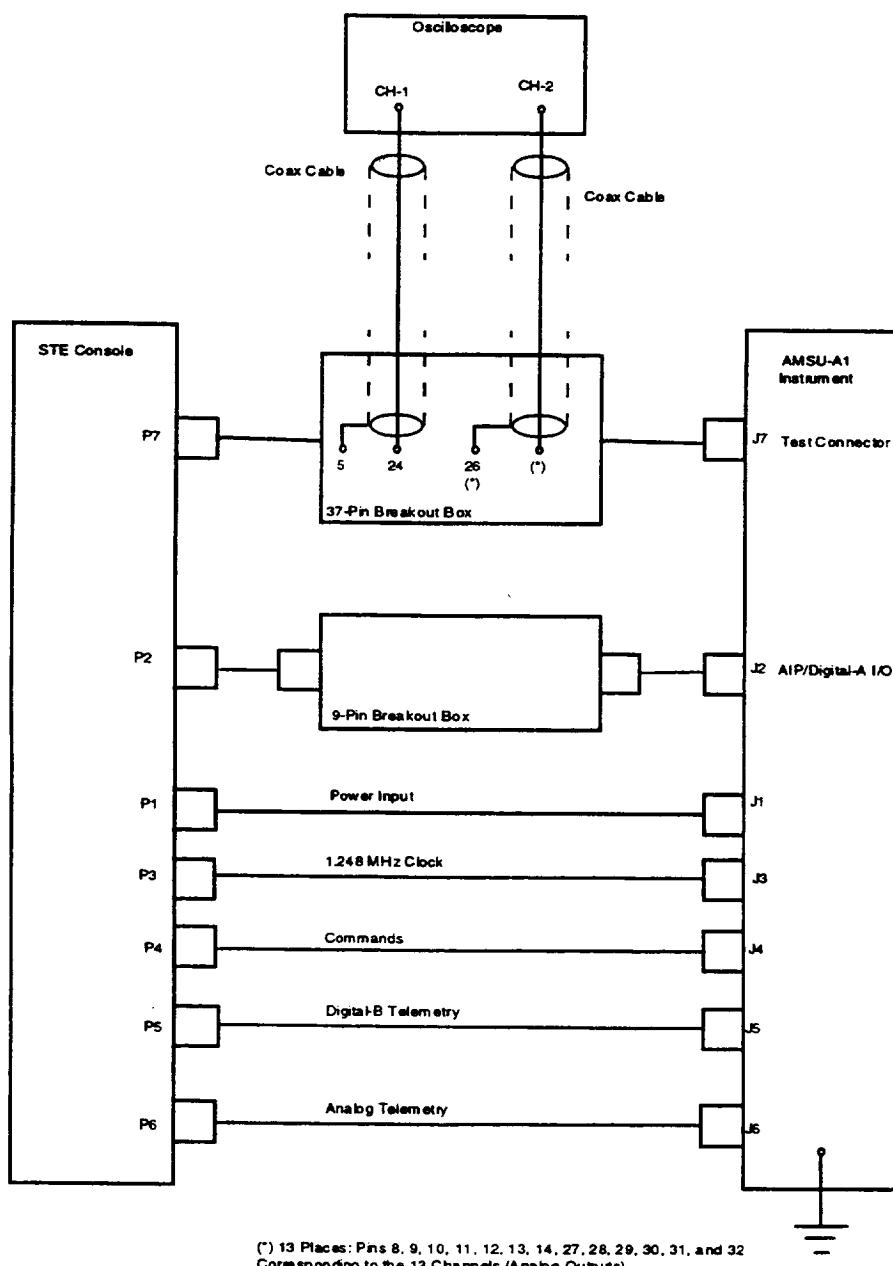
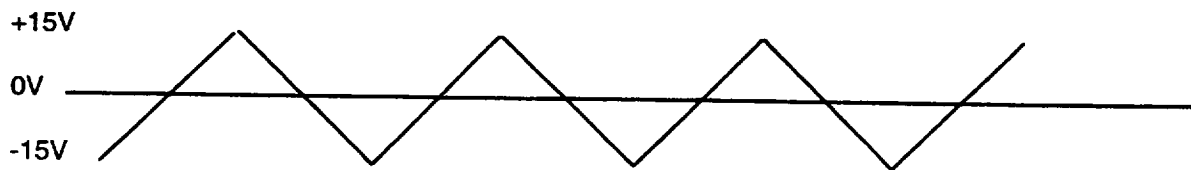


Figure 24. Integration Time (Analog Output) Verification Setup

3.2.4.3.6.3 PLLO No. 1 verification. The PLLO No. 1 shall be verified as follows:

1. Reconfigure the oscilloscope as indicated in Figure 25. Connect the oscilloscope channel-1 to J7-22 (PLLO No. 1).
2. From the Commands Menu of the STE, verify that the PLLO is selected in PLLO No. 1 as follows:

PLL POWER = PLLO#1 [18]
3. For S/N 101 - S/N 104, adjust the oscilloscope for best amplitude and time base. If the PLLO is locked properly, the oscilloscope will display a dc-voltage level of -15 to +15 V. Record the voltage level on TDS 44. Record PASS. (Any dc level recorded is considered PASS). If the PLLO is not locked properly, the scope will display a waveform similar to this:



Record FAIL on TDS 44. Discontinue the test until the deficiency is corrected.

4. For S/N 105 and above, if the PLLO is locked properly, the oscilloscope will display a dc-voltage = 4.0 ± 1 V. If the PLLO is not locked, the oscilloscope will display a dc-voltage of $+0.61 \pm 0.30$ V. If PLO is OFF, the oscilloscope will display a dc-voltage of 0.0 ± 0.2 V. If the PLLO is trying to acquire lock, the oscilloscope will display a various dc level. Record the voltage level on TDS 44.

Table III. Location and Frequency of Channel 3 through 15 Analog Outputs

Breakout Box Pin Location	Channel Distribution	Frequency
J7-08	Channel-03 Analog Output	50.3 GHz
J7-09	Channel-04 Analog Output	52.80 GHz
J7-10	Channel-05 Analog Output	53.596 GHz
J7-11	Channel-06 Analog Output	54.400 GHz
J7-12	Channel-07 Analog Output	54.940 GHz
J7-13	Channel-08 Analog Output	55.500 GHz
J7-14	Channel-09 Analog Output	57.290 GHz PLLO
J7-27	Channel-10 Analog Output	57.290 GHz PLLO
J7-28	Channel-11 Analog Output	57.290 GHz PLLO
J7-29	Channel-12 Analog Output	57.290 GHz PLLO
J7-30	Channel-13 Analog Output	57.290 GHz PLLO
J7-31	Channel-14 Analog Output	57.290 GHz PLLO
J7-32	Channel-15 Analog Output	89.000 GHz

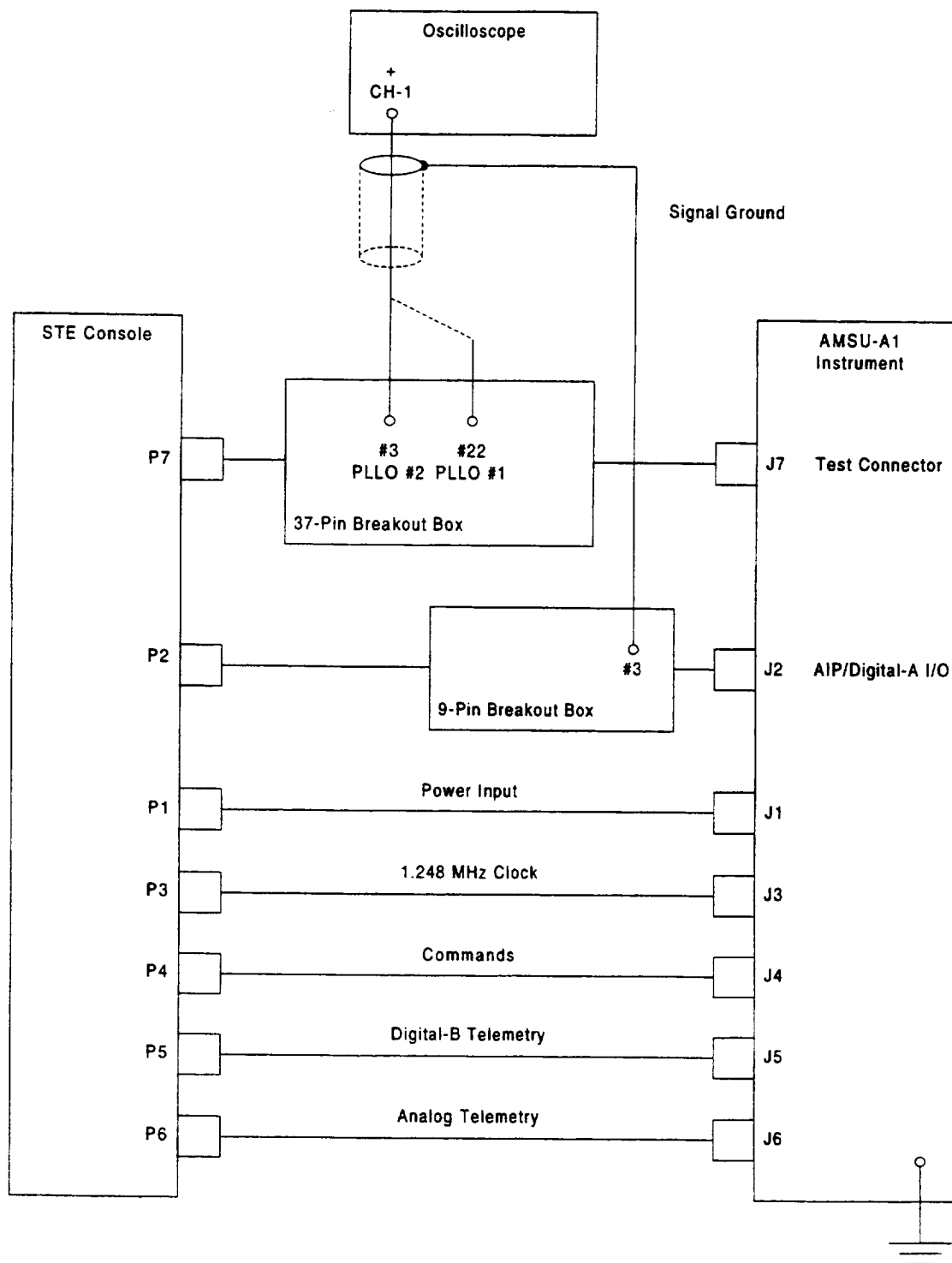
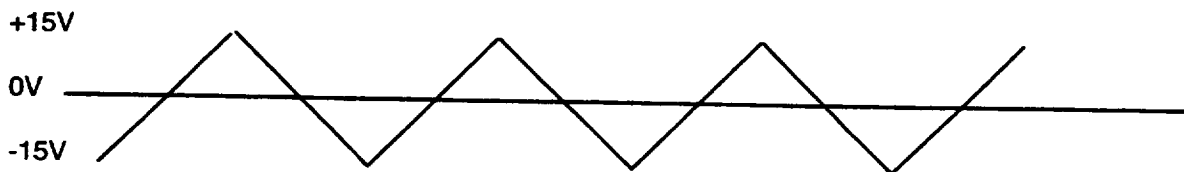


Figure 25. PLLO No. 1/No. 2 Test Setup

3.2.4.3.6.4 PLLO No. 2 verification. The PLLO No. 2 shall be verified as follows:

1. Reconfigure the oscilloscope as indicated in Figure 25. Connect the oscilloscope channel-1 to J7-03 (PLLO No. 2).
2. Select the PLLO No. 2 unit by executing the following command:

[18] PLL POWER = PLLO#2
3. For S/N 101 - S/N 104, adjust the oscilloscope for best amplitude and time base. If the PLLO is locked properly, the oscilloscope will display a dc-voltage level of -15 to +15 V. Record the voltage level on TDS 44. Record pass. (Any dc level recorded is considered PASS). If the PLLO is not locked properly, the scope will display a waveform similar to this:



Record FAIL on TDS 44. Discontinue the test until the deficiency is corrected.

4. For S/N 105 and above, if the PLLO is locked properly, the oscilloscope will display a dc-voltage = 4.0 ± 1 V. If the PLLO is not locked, the oscilloscope will display a dc-voltage of $+0.61 \pm 0.30$ V. If PLO is OFF, the oscilloscope will display a dc-voltage of 0.0 ± 0.2 V. If the PLLO is trying to acquire lock, the oscilloscope will display a various dc level. Record the voltage level on TDS 44.
5. Return to PLLO No. 1 by executing: PLL POWER = PLLO#1 [18]
6. Leave the unit turned on in preparation for the next test.

3.2.4.3.7 GSE mode verification. The purpose of this test is to verify the data obtained from the Ground Support Equipment (GSE), the following modes shall be evaluated. These modes are used for engineering evaluation only.

- GSE-1 (Position: 10, 10, 10)
- GSE-2 (Position: 1)
- GSE-3 (Position: current)
- GSE-4 (Position: 30)
- GSE-5 (Position: 6)
- GSE-7 (Position: required)

For GSE mode-1, the following parameters are subject to pass or fail criterion:

- [I] Sync. sequence
- [II] Unit ID and serial number

[III] Digital-B serial data verification

[IV] Reflector positions

[V] Radiometric data (Scene data) (Radiometric data will be limited to two channels only, channels 9 and 3. Channel 9 is physically located at the high bay of the sensor (A1-1 location) and channel 3 is located at the lower bay of the sensor (A1-2 location).

[VI] Temperature sensors.

For GSE 2 through 7, only the following parameters are subject to pass or fail criterion:

[IV] Reflector position.

[V] Radiometric data.

NOTE

Verification of GSE modes 2 through 7 are not required for the protoflight and flight instrument sensors since the modes are not used.

3.2.4.3.7.1 Equipment preparation and instrument turn-on procedure. To place instrument in GSE mode, proceed as follows:

1. Configure the test equipment as indicated in Figure 26.
2. Turn the unit on. Execute commands as necessary to obtain the following configuration:

COMMANDS			
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO [15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO [16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO [17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1 [18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO [19]
[14] ANTENNA WARM CAL POS =	NO	COLD CAL POSITION LSB =	ZERO [20]
POWER [4] ON		RETURN	[1]

Wait at least 18 seconds until the sending commands are acknowledged by the STE. At this point, the unit should be in the NO MODE with the STE collecting data.

3. Obtain a printout (9 pages) for all of the parameters ([I] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) On Commands Menu, press: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
 - (d) Select corresponding GSE mode under test.
 - (e) Press PRINT [3] FULL. The computer will start printing all 9 pages.

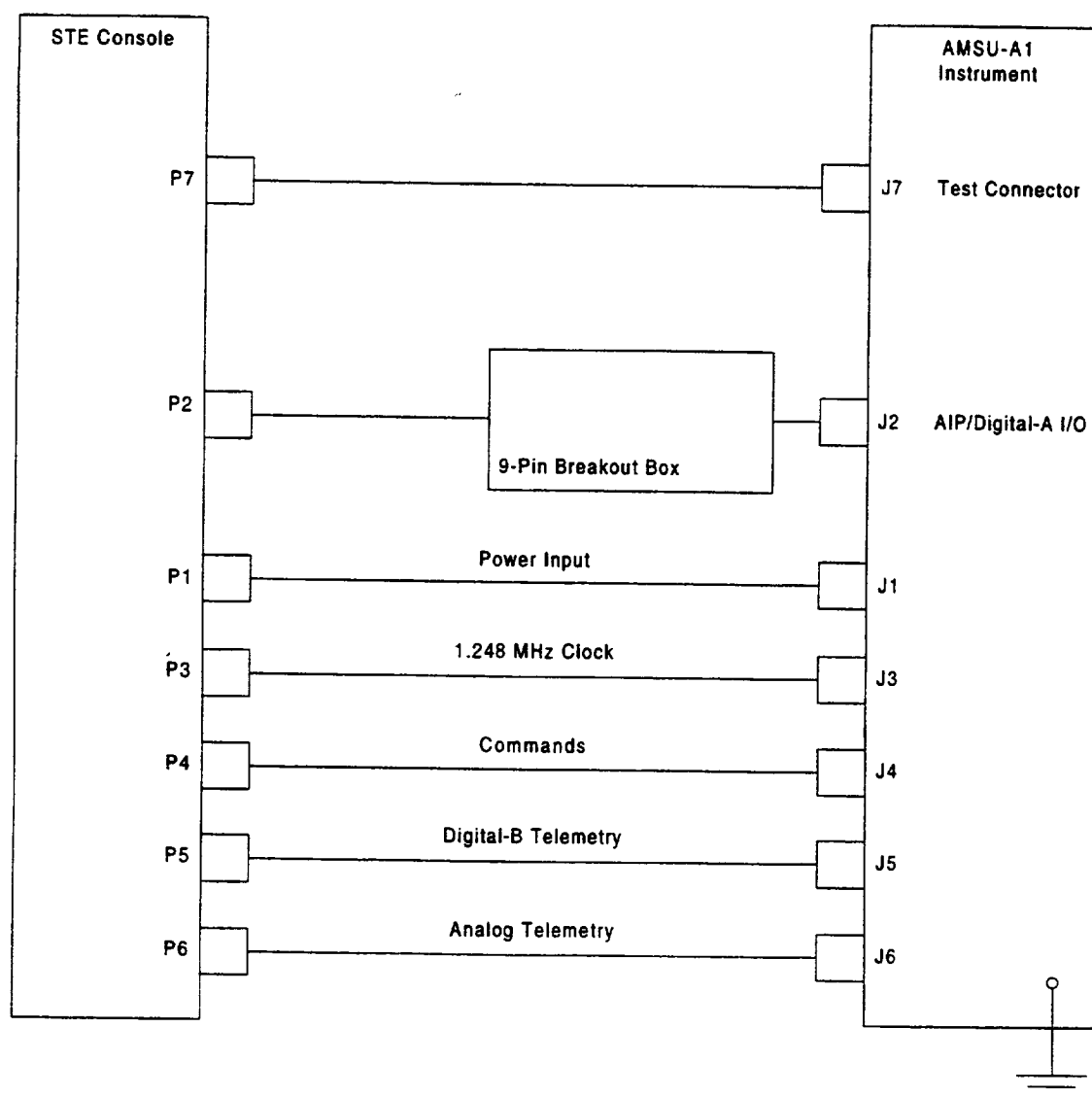


Figure 26. GSE Modes Verification Test

3.2.4.3.7.2 GSE Mode-1. The GSE mode-1 shall be tested as follows:

[II], [III], and [III] Sync, Unit ID, and Digital-B

1. Using the printout, verify that elements 1 through 8 are within the values specified on TDS 45. Record pass or fail.

NOTE

To verify the following steps, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS.

[IV] Reflector Positions

2. Using the individual printout, verify that the reflector positions are within the values specified in AE-26002/1, TDS 5 and 6. Record pass or fail on TDS 46.

[V] Radiometric Data

3. Using the individual printout, verify that the radiometric data are within the values specified on TDS 47.

[VI] Temperature Sensors

4. Using the individual printout, verify that elements 1090 through 1180 are within the values specified on **TDS 48 (sheets 1 and 2)**. Record pass or fail.

3.2.4.3.7.3 GSE Mode-2. The GSE Mode-2 shall be tested as follows:

1. Obtain a printout (9 pages) for all of the parameters ([II] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) Return to the Main Menu by pressing: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
 - (d) Select GSE mode 2 at the prompt.
 - (e) Press PRINT [3] FULL. The computer will start printing all 9 pages.

NOTE

To verify the following step, the operator may print out the individual parameters by using AE-26157 and attach the data to each TDS or the 9 full page printout may be used.

[IV] Reflector Positions

2. Using Pages 1 through 6 of the printout, verify that the reflector positions are within the values specified in AE-26002/1, TDS 5 and 6. Record pass or fail on TDS 46.

6Apr 99

3.2.4.3.7.4 GSE Mode-3. The GSE Mode-3 shall be tested as follows:

1. Obtain a printout (9 pages) for all of the parameters ([I] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) Return to the Main Menu by pressing: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
 - (d) Select GSE mode 3 at the prompt.

NOTE

To verify the following step, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS or the 9 full page printout may be used.

[IV] Reflector Positions

2. Verify that both A1-1 and A1-2 reflectors increment one step every eight seconds.

3.2.4.3.7.5 GSE Mode-4. The GSE Mode-4 shall be tested as follows:

1. Obtain a printout (9 pages) for all of the parameters ([I] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) Return to the Main Menu by pressing: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
 - (d) Select GSE mode 4 at the prompt.
 - (e) Press PRINT [3] FULL. The computer will start printing all 9 pages.

NOTE

To verify the following step, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS or the 9 full page printout may be used.

[IV] Reflector Positions

2. Using pages 1 through 6 of the printout, verify that the reflector positions are within the values specified in AE-26002/1, TDS 5 and 6. Record pass or fail on TDS 46.

3.2.4.3.7.6 GSE Mode-5. The GSE Mode-5 shall be tested as follows:

1. Obtain a printout (9 pages) for all of the parameters ([I] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) Return to the Main Menu by pressing: RETURN [1].

- (b) On Main Menu, select: [10] SELF TEST.
- (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
- (d) Select GSE mode 5 at the prompt.
- (e) Press PRINT [3] FULL. The computer will start printing all 9 pages.

NOTE

To verify the following step, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS or the 9 full page printout may be used.

[IV] Reflector Positions

- 2. Using pages 1 through 6 of the printout, verify that the reflector positions are within the values specified in AE-26002/1, TDS 5 and 6. Record pass or fail on TDS 46.

3.2.4.3.7.7 GSE Mode-7. The GSE Mode-7 shall be tested as follows:

- 1. Obtain a printout (9 pages) for all of the parameters ([I] through [VI]) described in 3.2.4.3.7 as follows:
 - (a) Return to the Main Menu by pressing: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)
 - (d) Select GSE mode 7 at the prompt.
 - (e) Press PRINT [3] FULL. The computer will start printing all 9 pages.

NOTE

To verify the following steps, the operator may printout the individual parameters by using AE-26157 and attach the data to each TDS or he may use the 9 page full printout.

[IV] Reflector Positions

- 2. Using pages 1 through 6 of the printout, verify that the reflector positions are within the values specified in AE-26002/1, TDS 5 and 6. Record pass or fail on TDS 46.
- 3. Set the STE to GSE MODE-0, failure to do so will cause the STE to produce faulty data when in normal mode. To enter GSE-MODE-0 into the computer:
 - (a) Return to the Main Menu by pressing: RETURN [1].
 - (b) On Main Menu, select: [10] SELF TEST.
 - (c) On Self Test Menu, select: [7] RUN GSE MODE.
(The computer will prompt: Enter GSE mode {0 to 15}.)

- (d) Select GSE mode 0.

3.2.4.4 Radiometer functional test. The purpose of the radiometer functional test is to verify the performance of the AMSU-A1 radiometer at the system level. This test shall consist of the following subtests:

- a. PLLO frequency measurements 3.2.4.4.1
- b. Relative NEΔT measurements 3.2.4.4.2

3.2.4.4.1 PLLO frequency measurements. Measure the PLLO frequencies as follows:

1. Prepare the unit and the test equipment as indicated in Figure 27. Frequency verification for the receiver shall be performed on the following frequency (see Figure 28 for sample plot):

(A1-1) Ch-9,10,11,12,13 and 14: 57.290344 GHz (PLLO No. 1 and PLLO No. 2)
2. Turn on the unit by using the procedure stated in 3.2.3.5. Allow not less than one hour for the equipment to warm-up and for the unit to stabilize.

On the Commands Menu, execute the following commands:

- (a) [14] ANTENNA WARM CAL POS = NO
 - (b) [15] ANTENNA COLD CAL POS = NO
 - (c) [16] ANTENNA NADIR POS = YES
 - (d) [17] ANTENNA FULL SCAN MODE = NO
3. Record the measured frequencies on TDS 49, and plotter data. Repeat step 2 for PLLO No. 2.
 4. Remove the test equipment but leave the unit on in preparation for the next test.

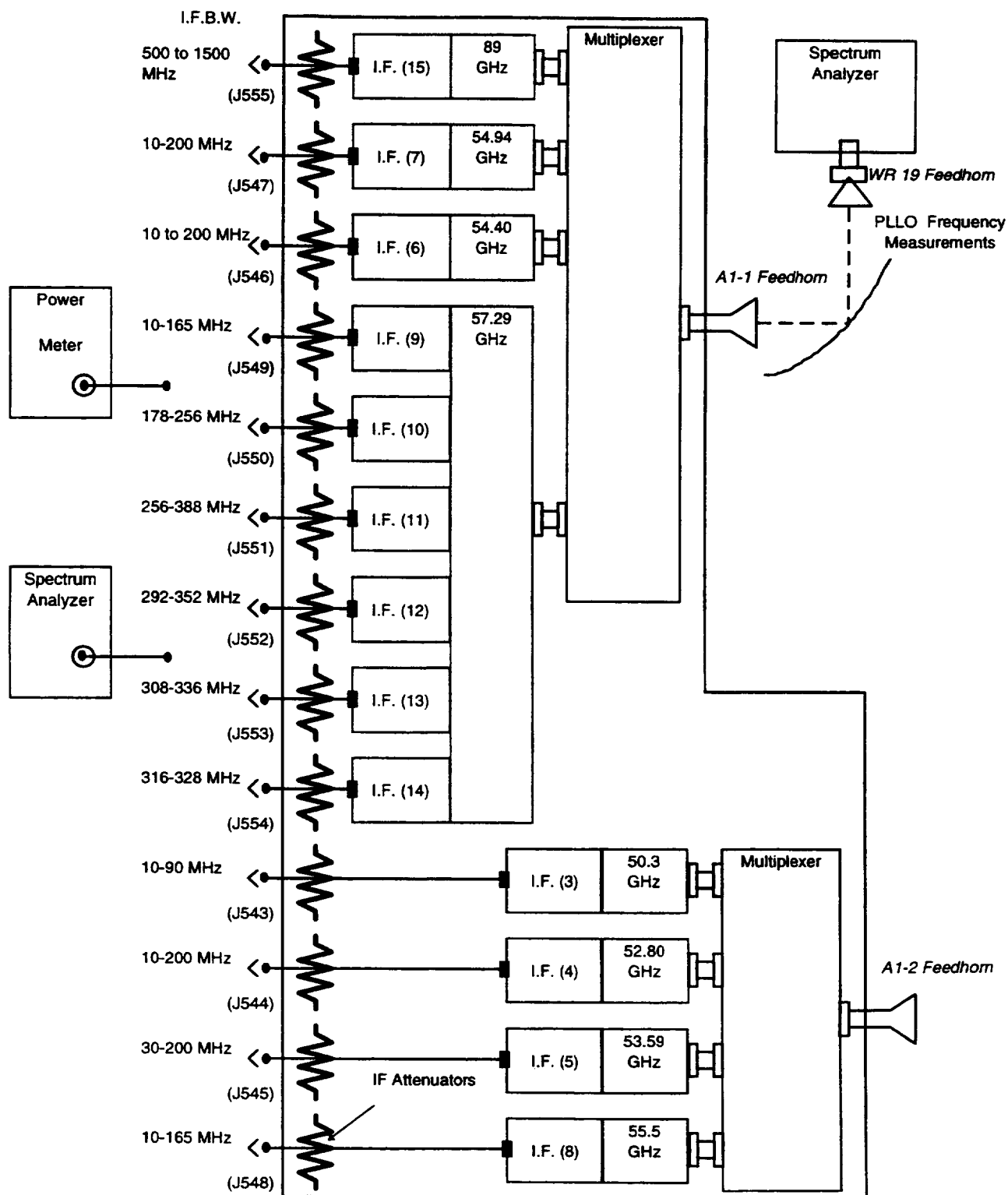


Figure 27. Configuration for RF Measurements

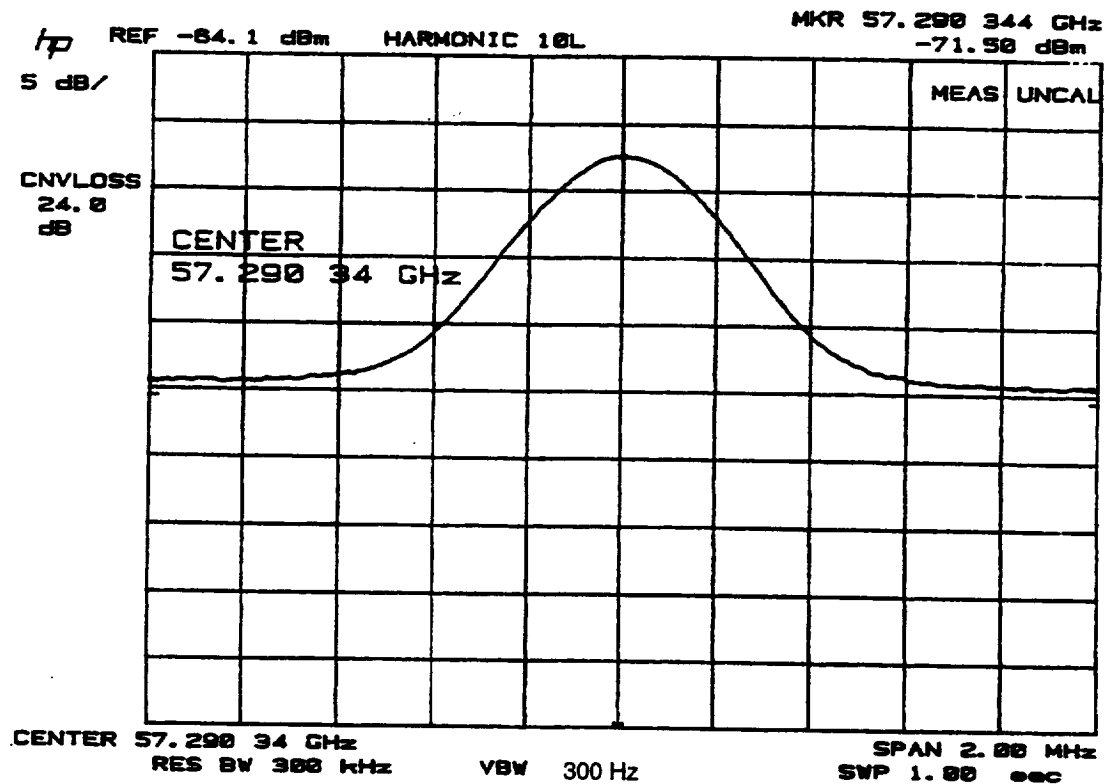


Figure 28. Sample Plot

3.2.4.4.2 Relative radiometer NEAT measurements. The purpose of this test is to perform a preliminary evaluation of the radiometer NEAT at a system level. Since the STE is not in the thermal-vacuum configuration, no temperature readings from the cold load are available. To compute the NEAT for this test, the temperature used for the cold load shall be LN₂ temperature.

The data obtained from this test are considered as relative NEAT and are to be used as a diagnostic tool to verify proper operation of the A/D converters and the spacecraft interface.

The equation to determine relative NEAT is as follows:

$$NEAT = \frac{[SD \times (Th - Tc)]}{M - N}$$

where:

- SD = Standard deviation of 120 samples at hot temperature (warm load)
- Th = Standard room temperature = 300 K
- Tc = Standard LN₂ temperature = 80 K
- M = Average of hot counts (120 samples)
- N = Average of cold counts (30 samples)

The sequence of testing shall be as follows:

- a. Equipment preparation and setup configuration
- b. Warm load radiometric data

- c. Cold load radiometric data
- d. Relative NEAT data collection

3.2.4.4.2.1 Equipment preparation and setup configuration. The equipment shall be set up as follows:

WARNING

The use of liquid nitrogen in a confined poorly ventilated area can cause asphyxiation and death due to a lack of oxygen (oxygen concentration below 20 percent). Accidental contact with liquid nitrogen will cause severe frostbite to the eyes or skin. When handling liquid nitrogen, personnel shall observe the following safety precautions:

- a. Ensure that the work area is well ventilated to prevent excessive gas buildup.
 - b. To protect your eyes always wear a face shield or safety goggles (safety glasses without side shields do not provide adequate protection).
 - c. To protect exposed skin, always wear an apron when pouring LN2 and whenever exposed to LN2, always wear a lab coat, gloves made for cryogenic work, cuffless trousers (worn outside the boots or shoes), and safety shoes.
 - d. Do not fill target fuller than 1.0 inch from the top. Fill target at the floor level, away from unit.
 - e. Do not move filled target without cover in place.
1. Configure the test equipment and the unit as indicated in Figure 29, except for the cold loads.
 2. Execute commands as necessary to obtain the following configuration:

COMMANDS				
[9] MODULE POWER =	CONNECT	ANTENNA IN COLD CAL POS =	NO	[15]
[10] SURVIVAL HTR PWR =	OFF	ANTENNA IN NADIR POS=	NO	[16]
[11] MODULE TOTALLY OFF =	ON	ANTENNA FULL SCAN MODE =	NO	[17]
[12] SCANNER A1-1 POWER =	ON	PLL POWER =	PLLO#1	[18]
[13] SCANNER A1-2 POWER =	ON	COLD CAL POSITION MSB =	ZERO	[19]
[14] ANTENNA WARM CAL POS =	YES	COLD CAL POSITION LSB =	ZERO	[20]
POWER [4] ON				

3. Allow 30 minutes for the unit to stabilize.

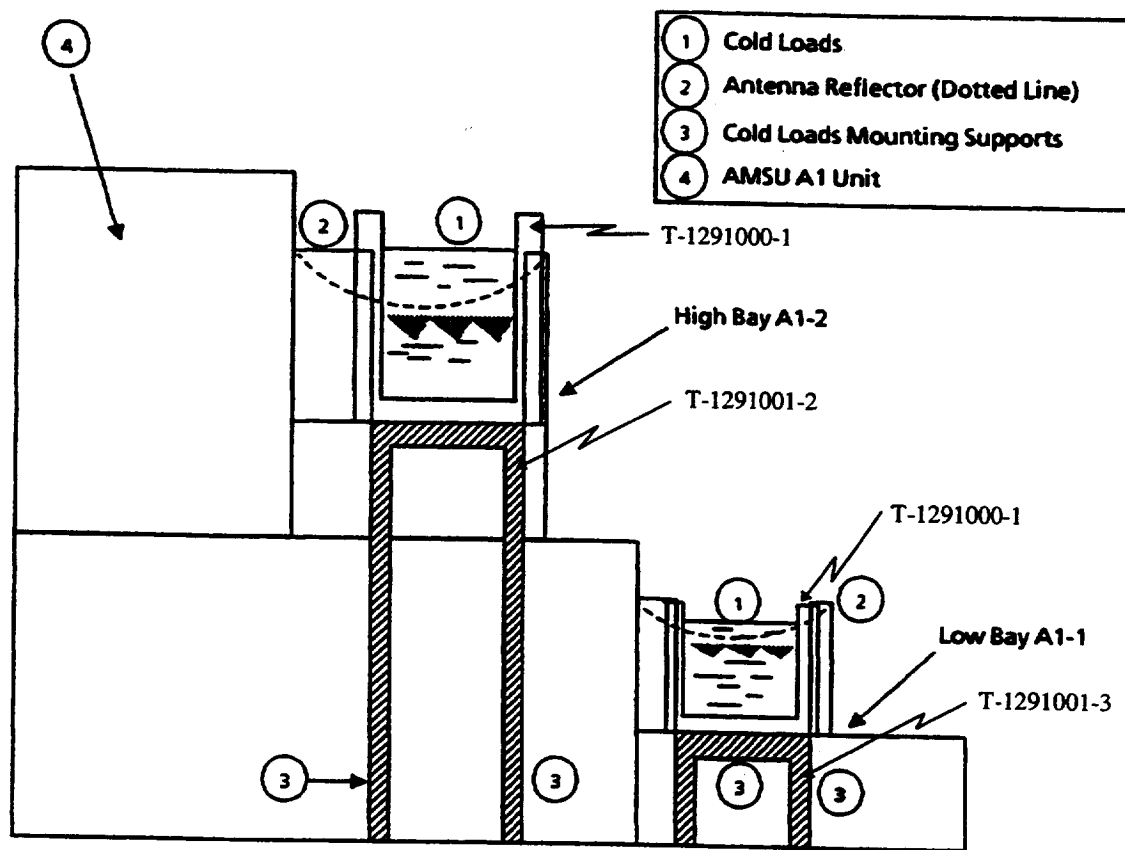


Figure 29. NEAT Setup Configuration

3.2.4.4.2.2 *Relative NEAT data collection*

1. Return to the Main Menu by pressing [1] RETURN.
2. On the Main Menu, select [13] FUNCTIONAL TEST. (The STE will automatically command the unit to position the antenna reflector to the warm and cold loads as it is taking data.)
3. Wait approximately one minute to verify that the NEAT results are displayed on the screen. Obtain a printout. Repeat step 2 four times and obtain four additional printouts. Average NEAT from these five data points. Enter the values on TDS 50. Attach the printout to the data sheet.
4. Repeat steps 1, 2, and 3 for the PLLO No. 2. Allow 30 minutes for the unit to stabilize after switching to PLLO No. 2.
5. Remove the cold loads and associated hardware.

3.2.4.5 Channel identification test. The purpose of the channel identification test is to verify the proper final configuration /assembly of each radiometer channel from antenna input to the spacecraft interface.

1. Configure the unit and test equipment as shown in Figures 26 and 32.
2. Connect the STE to instrument using the following STE interface cables.

- a. STE interface cable J1 (1356648-1)
 - b. STE interface cable J2 (1356648-2)
 - c. STE interface cable J3 (1356648-3)
 - d. STE interface cable J4 (1356648-4)
3. Follow the turn-on procedure per para. 3.2.3.5.
 4. Enter the STE command "SCANNER A1-1 POWER." Wait 18 seconds before issuing the next command.
 5. Enter the STE command "SCANNER A1-2 POWER." Wait 18 seconds before issuing the next command.
 6. Enter the STE command "ANTENNA COLD CAL." Wait 18 seconds before issuing the next command. Both reflectors should scan to the cold calibration beam position.
 7. Enter the STE command "[1] RETURN" to return to the monitor only screen.
 8. Enter the STE command "[10] DIGITAL-A." The STE should now display the digital-A data screen shown in Figure 30. From this screen enter the STE command "[9] BEAM POSITION NN-ALL CHANNELS."
 9. The STE then asks "ENTER BEAM POSITION NO (1 TO 30)." Enter "30" to show the radiometric counts data for channels 3-15. The STE should now display the radiometric data screen shown in Figure 31, except with a different set of count data.
 10. Allow the instrument to stabilize for approximately 20 minutes. Enter the STE command "[2]" to obtain a screen only printout.
 11. Configure the unit and test equipment as shown in Figure 32. Turn ON the sweeper and allow to warm up approximately 10 minutes. **Make sure** that the RF power is OFF during sweeper warm up.

CAUTION

Extreme care must be used when turning on RF power. When RF power is first applied the multiplier/gain horn should be approximately three to four feet from the unit. The RF power setting should be no greater than -20 dBm.

12. Set the sweeper frequency to 50.35 ± 0.01 GHz and set the RF power level to -20 dBm. Position the multiplier/gain horn three to four feet from the instrument so that the A1-2 antenna and gain horn are approximately aligned (see Figure 32). Rotate the gain horn, if needed, to the vertical polarization position.
13. Turn ON the RF power making sure the power level is set to -20 dBm. Allow the multiplier to warm up approximately five minutes.
14. At the STE screen compare the radiometric data counts of channel 3 to the counts printed out at step 10. Enter the STE command "[2]" to obtain a screen only printout.
15. From the printouts obtained in steps 10 and 14, verify that the radiometric data counts for channel 3 have increased significantly, approximately 1000 or more, and that the other channels' data counts have remained relatively unchanged, less than 300 counts.
16. Record the counts difference on TDS 52 of channel 3 from the printouts obtained in steps 10 and 14 and attach printouts to TDS 52.
17. Repeat steps 12 through 16 for the frequencies and polarizations listed on TDS 52.

18. After all A1 channels have been identified, turn OFF the RF power. Return the reflectors to the warm cal position.
19. Turn the STE Q/Main and N/Pulse switches to OFF.
20. Turn the STE power supply panel main power switch OFF.

EOS	A1-03 E1.EXE;31	COLD CAL MODE	P15-JUN-98	09:36:59	SCAN NUMBER	34
[5]	SCIENCE DATA	ELEMENT	0000			
[6]	CONTROL/STATUS	ELEMENT	00			
[7]	ENGINEERING	ELEMENT	00			
[8]	DATA STREAM (64 VALUES)					
[9]	BEAM POSITION NN-ALL CHANNELS					
[10]	CHANNEL NN -ALL BEAM POSITIONS					
[11]	WARM CALIBRATE					
[12]	COLD CALILBRATE					
[13]	REFLECTOR POSITIONS					
[14]	TEMPERATURE DATA (16 VALUES)					
ENGR OK	POWER	ON	CHECKSUM IN 15A1 SA28	34SA29	47	
SELECT BUTTON 2		SCREEN ONLY [2]	PRINT [3]	FULL	[1]	RETURN

Figure 30. Digital-A Data Screen

EOS	A1-03 E1.EXE;31	COLD CAL MODE	P15-JUN-98	09:49:07	SCAN NUMBER	11
[5]	SCIENCE DATA	ELEMENT	0000			
[6]	CONTROL/STATUS	ELEMENT	00			
[7]	ENGINEERING	ELEMENT	00			
RADIOMETRIC DATA						
BEAM POSITION						
	CH	DATA	CH	DATA	CH	DATA
	3	15798	8	15414	13	15811
	4	16252	9	16176	14	16029
	5	15661	10	16010	15	15102
	6	16413	11	15639		
	7	18044	12	15817		
[21]	UP		[22]	DOWN		
ENGR OK	POWER	ON	CHECKSUM	IN DF5D CALC	DFSD SA28	11 SA29 14
SELECT BUTTON 2						

Figure 31. Radiometric Data Screen

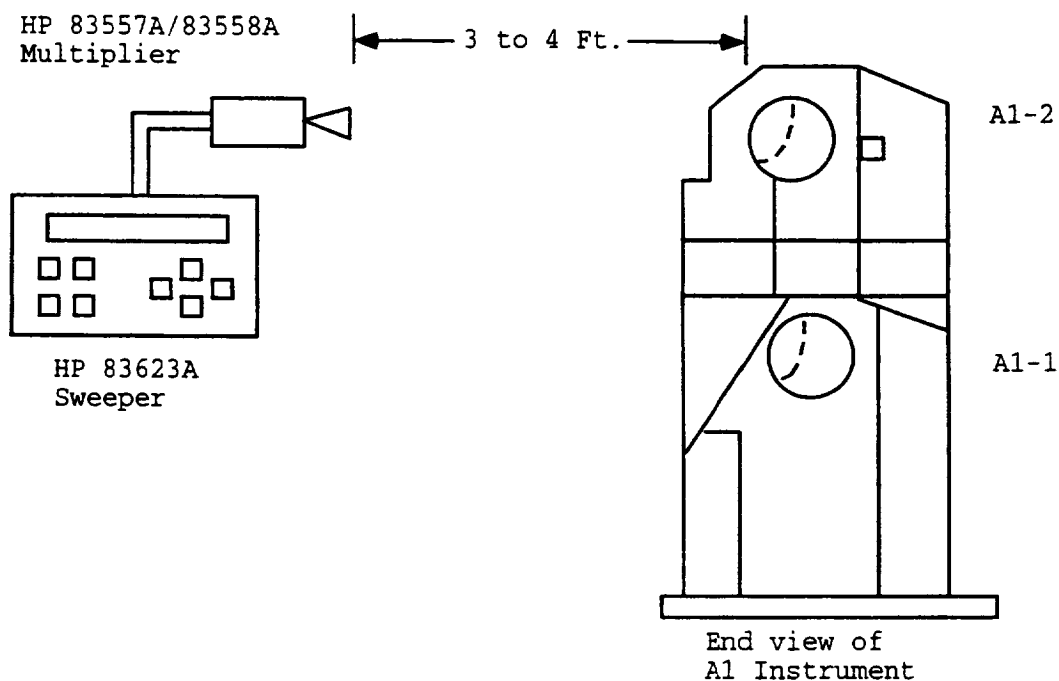


Figure 32. Channel Identification Setup

6Apr 99

4. QUALITY ASSURANCE PROVISIONS

4.1 Responsibility for inspection. Aerojet Quality Assurance shall inspect in accordance with the requirements of this test procedure and S-480-79 and S-480-80. Quality Control shall verify all test set-ups prior to start of test. Bonded software shall be used for all tests and shall be obtained from Quality Control. Quality Control shall review all test data for conformance to success criteria. The test data shall include test limits. For tests that satisfy requirements from S-480-80 on protoflight and flight units, customer representatives shall be invited to monitor tests and shall be invited to review the data and show approval on the test data sheets.

4.1.1 Test facilities. Unless otherwise specified, the examinations and tests described herein shall be conducted at GenCorp Aerojet, Azusa Operations, Azusa, CA.

4.1.2 Electrostatic Device (ESD) handling. All electronic hardware shall be handled in accordance with Aerojet Standard STD-2454.

4.2 Monitoring procedures. All tests in this procedure shall be monitored by quality control.

4.2.1 Test equipment. Test equipment calibration procedures shall comply with the requirements of MIL-STD-45662.

4.2.2 Software. Bonded software shall be used at all times.

4.3 Monitoring procedures for materials. Not applicable.

4.4 Certification. Certification for handling ESD-sensitive equipment is required for all personnel working on the assembly and test of the AMSU-A instrument, per STD-2454.

4.5 Test methods

4.5.1 Accept-reject criteria. The accept-reject criteria for each examination or test shall be as specified in the data sheets included in each phase of the applicable test procedure. The test results shall be recorded on the data sheets to demonstrate compliance with the applicable specification requirements. Methods of analysis shall be appropriate for the parameters being inspected. It shall be the responsibility of Aerojet to review the test data and determine conformance of the unit under test to the performance requirements contained in S-480-80 and this specification.

In the event of a failure during any phase of this test procedure, the test activity shall record the required information on the Test Anomaly Record (TAR) and alert the design assurance and quality engineers. Except for failures which only represent a limited out-of-tolerance condition for a particular parameter and are not expected to interfere with the balance of the testing and which are non-destructive, the testing must be stopped until a complete description of the observed anomaly failure is documented and a Failure Analysis Strategy (FAS) is formulated, documented, and implemented to preclude loss of information or evidence that may facilitate determining the failure cause. The full set of data from the referenced tests is required in order to formulate a plan of action. The cognizant reliability engineer, quality assurance engineer, and the system or responsible test engineer shall jointly develop the FAS which must be approved by Design Assurance and Quality Assurance. Analysis and reporting shall be performed per Aerojet procedures.

4.5.2 General. All data sheets associated with the tests on the unit plus the data reduction and analysis of specific parameters required by each applicable test procedure obtained from screen printouts and plots, oscilloscope photographs, or magnetic recordings shall be included with the associated shop order. During tests in which a CRT screen is to be printed or plotted and retained as a data sheet, the following annotation shall be applied:

Test/Systems Engineer: (Signature)	_____
Quality Control: (Signature)	_____
Customer Representative (Flight Hardware Only):	_____ (Signature)
Date:	_____
Test Paragraph No.:	_____
Subassembly/Assembly Serial No.:	_____
Shop Order No.:	_____

4.5.2.1 Test data. The test data shall be that which was obtained during performance of the tests specified and recorded on the Test Data Sheet(s) (TDS) (see Appendix A) and on printouts and plots and shall be attached to the shop order associated with the test.

6Apr 99

5. PREPARATION FOR DELIVERY

This section is not applicable to this specification.

6. NOTES**6.1 Acronyms and abbreviations**

AMSU	Advanced Microwave Sounding Unit
ATB	Analog telemetry bus
AWG	American Wire Gage
BP	Beam Position
CAL	Calibrate
CPT	Comprehensive performance test
d	delta
DC	Direct current
DVM	Digital volt meter
EMI	Electromagnetic interference
ESD	Electrostatic Sensitive Device
EXT	External
FAS	Failure analysis strategy
GHz	Gigahertz
GIIS	General Instrument Interface Specification
GND	Ground
GSE	Ground Support Equipment
HTR	Heater
kHz	Kilohertz
LPT	Limited performance test
LSB	Least significant bit
MA	Milliampere
METSAT	Meteorological Satellite
MLB	Main load bus
MFG	Manufacturer
MMW	Millimeter wave
MS, MSEC	Millisecond
MSB	Most significant bit
MV	Millivolt
NEAT	Noise equivalent delta temperature
PFM	Protoflight Model
PLB	Pulse load bus
PLL	Phase lock loop
PLLO	Phase lock loop oscillator

POS	Position
PWR	Power
RTN	Return
STE	Special Test Equipment
SW	
TAR	Test Anomaly Record
TDS	Test Data Sheet
TLM	Telemetry
TM	Instrument Temperature
UIIS	Unique Instrument Interface Specification
Vdc	Volts, direct current
μ s	Microsecond

6.2 Changes. The outside margins of this document have been marked to indicate where modifications, deletions, or additions have been made since the previous issue. This is done solely as a convenience to users, who are cautioned to evaluate the requirements of this document based on the entire content as written, regardless of the marginal notations and relationship to the previous issue.

APPENDIX A

TEST DATA SHEETS

10.1 Scope. This appendix contains the test data sheets for all tests and inspections listed in section 3.

TDS

	Page
1 Grounding System Test	A-2
2 +28 MLB During Turn-on Transient	A-11
3 +28 MLB Operating Power	A-12
4 +28 Pulse Load Bus	A-13
5 +28 V Analog Telemetry Bus	A-15
6 +10V Interface Bus Voltage	A-16
7 Power Input Test for LPT	A-17
8 1.248 MHz Clock Signal Verification	A-18
9 "C1" Shift Pulse Verification	A-19
10 "A1" Select Pulse Verification	A-20
11 "8 Seconds" Frame Sync Pulse	A-21
12 Synchronization Signals Relationship	A-22
13 Synchronization Signals Relationship	A-24
14 Commands and Digital-B Telemetry Verification	A-25
15 Scanner Commands Verification	A-26
16 Scanner Commands Verification	A-27
17 Scanner Commands Verification	A-28
18 Scanner Positions Commands	A-29
19 Digital-A Data Output Full Scan Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification	A-30
20 Reflector Positions Section [IV]	A-31
21 Digital-A Data Output Radiometer Data Section [V]	A-32
22 Full Scan Mode Temperature Sensors Section [VI]	A-33
23 Digital-A Data Output Warm Cal Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification	A-35
24 Reflector Position Warm Cal Mode Section [IV] and Reflector Position Nadir Mode Section [IV]	A-36
25 Digital-A Data Output Warm Cal Mode Radiometer Data Section [V]	A-37
26 Warm Cal Mode Temperature Sensors Section [VI]	A-38
27 Digital-A Data Output Cold Cal Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification	A-40
28 Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV]	A-41
29 Digital-A Data Output Cold Cal Mode Radiometer Data Section [V]	A-43
30 Cold Cal Mode Temperature Sensors Section [VI]	A-44
31 Digital-A Data Output Nadir Mode Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification	A-46
32 Digital-A Data Output Nadir Mode Radiometer Data Section [V]	A-47
33 Nadir Mode Temperature Sensors Section [VI]	A-48
34 Analog Telemetry Verification by Way of Connector J6	A-50
35 Analog Telemetry Signals by Way of the STE	A-51
36 Integrate/Hold and Dump Signal Verification	A-53
37 Integration Time (Analog Output) Verification	A-54
38 Integration Time (Analog Output) Verification	A-55
39 Integration Time (Analog Output) Verification	A-56
40 Integration Time (Analog Output) Verification	A-57
41 Integration Time (Analog Output) Verification	A-58
42 Integration Time (Analog Output) Verification	A-59
43 Integration Time (Analog Output) Verification	A-60
44 PLL0 No. 1 Verification and PLL0 No. 2 Verification	A-61
45 Digital-A/GSE Mode-1 Synch Sequence, Unit I.D./Serial Number and Digital-B Serial Data Verification	A-62
46 Reflector Position	A-63
47 Digital-A/GSE Mode-1 Radiometer Data Section [V]	A-65
48 Digital-A/GSE Mode-1 Temperature Sensors Section [VI]	A-66
49 Receiver Input Signals	A-68
50 Radiometer "Relative" NEDT Verification	A-69
51 Transient Susceptibility Test	A-71
52 Channel Identification Test	A-73

6 Apr 99

TEST DATA SHEET 1 (Sheet 1 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J1 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-1	+28 V MLB	> 100k		
J1-2	+28 V MLB	> 100k		
J1-3	+28 V MLB RTN	> 100k		
J1-4	+28 V MLB RTN	> 100k		
J1-5	+28 V PLB	> 100k		
J1-6	+28 V PLB	> 100k		
J1-7	+28 V PLB RTN	> 100k		
J1-8	+28 V PLB RTN	> 100k		
J1-9	+28 V TMB	> 100k		
J1-10	28 V TMB RTN	> 100k		
J1-11	NO CONNECTION	> 100k		
J1-12	NO CONNECTION	> 100k		
J1-13	CHASSIS GROUND (E1)	< 1		
J1-14	+28 V MLB	> 100k		
J1-15	+28 V MLB	> 100k		
J1-16	+28 V MLB RTN	> 100k		
J1-17	+28 V MLB RTN	> 100k		
J1-18	+28 V PLB	> 100k		
J1-19	+28 V PLB	> 100k		
J1-20	+28 V PLB RTN	> 100k		
J1-21	+28 V PLB RTN	> 100k		
J1-22	+28 V TMB	> 100k		
J1-23	28 V TMB RTN	> 100k		
J1-24	SAFETY HTR PWR	> 100k		
J1-25	SAFETY HTR RTN	> 100k		

TEST DATA SHEET 1 (Sheet 2 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

J2 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J2-1	Chassis Ground (E2)	< 1		
J2-2	DATA CLOCK (C1)	> 100k		
J2-3	Signal Return	> 100k		
J2-4	No Connection	> 100k		
J2-5	DIGITAL-A DATA OUT	> 100k		
J2-6	DATA ENABLE (A1)	> 100k		
J2-7	8 SEC SYNC PULSE	> 100k		
J2-8	No Connection	> 100k		
J2-9	No Connection	> 100k		

J3 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J3-1	1.248 MHz CLK	> 100k		
J3-2	1.248 MHz CLK RTN	> 100k		
J3-3	Chassis GND (E3)	< 1		

J5 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J5-1	Chassis Ground (E5)	< 1		
J5-2	MODULE PWR IND	> 100k		
J5-3	COLD CAL POS MSB (OUT)	> 100k		
J5-4	No Connection	> 100k		
J5-5	SCANNER A1-2 ON/OFF	> 100k		
J5-6	ANT IN COLD CAL POS	> 100k		
J5-7	PLL PRI/RED	> 100k		
J5-8	No Connection	> 100k		
J5-9	SURV HTR ON/OFF	> 100k		
J5-10	No Connection	> 100k		
J5-11	COLD CAL POS LSB (OUT)	> 100k		
J5-12	SCANNER A1-1 ON/OFF	> 100k		
J5-13	ANT IN WARM CAL POS	> 100k		
J5-14	ANT IN NADIR POS	> 100k		
J5-15	FULL SCAN MODE	> 100k		

6 Apr 99

TEST DATA SHEET 1 (Sheet 3 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J4 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J4-1	Chassis Ground (E4)	< 1		
J4-2	MODULE PWR DISCONN	> 100k		
J4-3	SURVIVAL HTR ON	> 100k		
J4-4	MODULE TOTALLY OFF	> 100k		
J4-5	SCANNER A1-2 ON/OFF	> 100k		
J4-6	ANT AT COLD CAL POS	> 100k		
J4-7	PLL SELECT	> 100k		
J4-8	ANT AT NADIR POS	> 100k		
J4-9	COLD CAL POS MSB (IN)	> 100k		
J4-10	No Connection	> 100k		
J4-11	No Connection	> 100k		
J4-12	+10 V INTERFACE BUS	> 100k		
J4-13	10 V INTERFACE BUS RTN	> 100k		
J4-14	MODULE PWR CONN	> 100k		
J4-15	SURVIVAL HTR OFF	> 100k		
J4-16	SCANNER AI-1 ON/OFF	> 100k		
J4-17	ANT AT WARM CAL POS	> 100k		
J4-18	FULL SCAN	> 100k		
J4-19	COLD CAL POS LSB (IN)	> 100k		
J4-20	No Connection	> 100k		
J4-21	No Connection	> 100k		
J4-22	No Connection	> 100k		
J4-23	No Connection	> 100k		
J4-24	+10 V INTERFACE BUS	> 100k		
J4-25	10 V INTERFACE BUS RTN	> 100k		

TEST DATA SHEET 1 (Sheet 4 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J6 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J6-1	Chassis GND (E6)	< 1		
J6-2	RF SHELF A1-1 TEMP	> 100k		
J6-3	A1-1 SCAN. MTR. TEMP	> 100k		
J6-4	WARM LOAD A1-1 TEMP	> 100k		
J6-5	No Connection	> 100k		
J6-6	PLLO RED LOCK DETECT	> 100k		
J6-7	No Connection	> 100k		
J6-8	A1-1 DRIVE MTR CURR	> 100k		
J6-9	+15 V ANT DR MON	> 100k		
J6-10	+5 V ANT DR MON	> 100k		
J6-11	+15 V SIG PROC MON	> 100k		
J6-12	+5 V SIG PROC MON	> 100k		
J6-13	L.O. VOLTAGE CH 3 MON	> 100k		
J6-14	L.O. VOLTAGE CH 5 MON	> 100k		
J6-15	L.O. VOLTAGE CH 7 MON	> 100k		
J6-16	+15 VDC PLL LO MON	> 100k		
J6-17	+10 V MIXER/AMP MON	> 100k		
J6-18	L.O. VOLTAGE CH 15 MON	> 100k		
J6-19	No Connection	> 100k		
J6-20	28 V TMB RTN	> 100k		
J6-21	RF SHELF A1-2 TEMP	> 100k		
J6-22	A1-2 SCAN MTR TEMP	> 100k		
J6-23	WARM LOAD A1-2 TEMP	> 100k		
J6-24	No Connection	> 100k		
J6-25	PLLO PRI LOCK DETECT	> 100k		
J6-26	No Connection	> 100k		
J6-27	A1-2 DRIVE MTR CURR	> 100k		
J6-28	-15 V ANT DR MON	> 100k		
J6-29	-15 V SIG PROC MON	> 100k		
J6-30	L.O. VOLTAGE CH 4 MON	> 100k		
J6-31	L.O. VOLTAGE CH 6 MON	> 100k		
J6-32	L.O. VOLTAGE CH 8 MON	> 100k		
J6-33	-15 VDC PLL LO MON	> 100k		
J6-34	+8 V IF AMP MON	> 100k		
J6-35	No Connection	> 100k		
J6-36	No Connection	> 100k		
J6-37	No Connection	> 100k		

6 Apr 99

TEST DATA SHEET 1 (Sheet 5 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J7 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J7-1	Chassis GND (E7)	< 1		
J7-2	No Connection	> 100k		
J7-3	REDUN PLO LOCK DET	> 100k		
J7-4	15 V RTN (2/3)	> 100k		
J7-5	15 V RTN (2/3)	> 100k		
J7-6	DUMP TEST POINT	> 100k		
J7-7	No Connection	> 100k		
J7-8	CH3 OUT TEST POINT	> 100k		
J7-9	CH4 OUT TEST POINT	> 100k		
J7-10	CH5 OUT TEST POINT	> 100k		
J7-11	CH6 OUT TEST POINT	> 100k		
J7-12	CH7 OUT TEST POINT	> 100k		
J7-13	CH8 OUT TEST POINT	> 100k		
J7-14	CH9 OUT TEST POINT	> 100k		
J7-15	No Connection	> 100k		
J7-16	No Connection	> 100k		
J7-17	GSE CMD LSB	> 100k		
J7-18	GSE CMD MSB-1	> 100k		
J7-19	+5 V GSE INTERLOCK A	> 100k		
J7-20	No Connection	> 100k		
J7-21	No Connection	> 100k		
J7-22	PRI PLO LOCK DET	> 100k		
J7-23	No Connection	> 100k		
J7-24	I/H TEST POINT	> 100k		
J7-25	No Connection	> 100k		
J7-26	15 V RTN (2/3)	> 100k		
J7-27	CH10 OUT TEST POINT	> 100k		
J7-28	CH11 OUT TEST POINT	> 100k		
J7-29	CH12 OUT TEST POINT	> 100k		
J7-30	CH13 OUT TEST POINT	> 100k		
J7-31	CH14 OUT TEST POINT	> 100k		
J7-32	CH15 OUT TEST POINT	> 100k		
J7-33	No Connection	> 100k		
J7-34	No Connection	> 100k		
J7-35	GSE CMD MSB	> 100k		
J7-36	5 V RTN (1)	> 100k		
J7-37	+5 V GSE INTERLOCK B	> 100k		

TEST DATA SHEET 1 (Sheet 6 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-1	J1-2	+28 V MLB	< 1		
J1-1	J1-14	+28 V MLB	< 1		
J1-1	J1-15	+28 V MLB	< 1		
J1-3	J1-4	28 V MLB RTN	< 1		
J1-3	J1-16	28 V MLB RTN	< 1		
J1-3	J1-17	28 V MLB RTN	< 1		
J1-5	J1-6	+28 V PLB	< 1		
J1-5	J1-18	+28 V PLB	< 1		
J1-5	J1-19	+28 V PLB	< 1		
J1-7	J1-8	28 V PLB RTN	< 1		
J1-7	J1-20	28 V PLB RTN	< 1		
J1-7	J1-21	28 V PLB RTN	< 1		
J1-9	J1-22	+28 V TMB	< 1		
J1-10	J1-23	28 V TMB RTN	< 1		
J1-10	J6-20	28 V TMB RTN	< 1		
J4-12	J4-24	+10 V INTERFACE BUS	< 1		
J4-13	J4-25	10 V INTERFACE BUS RTN	< 1		
J1-1	J1-3	+28 V MLB	> 100k		
J1-1	J1-5	+28 V MLB	> 100k		
J1-1	J1-7	+28 V MLB	> 100k		
J1-1	J1-9	+28 V MLB	> 100k		
J1-1	J1-10	+28 V MLB	> 100k		
J1-1	J1-24	+28 V MLB	> 100k		
J1-1	J1-25	+28 V MLB	> 100k		
J1-1	J2-3	+28 V MLB	> 100k		
J1-1	J4-12	+28 V MLB	> 100k		
J1-1	J4-13	+28 V MLB	> 100k		
J1-3	J1-5	28 V MLB RTN	> 100k		
J1-3	J1-7	28 V MLB RTN	> 100k		
J1-3	J1-9	28 V MLB RTN	> 100k		
J1-3	J1-10	28 V MLB RTN	> 100k		
J1-3	J1-24	28 V MLB RTN	> 100k		
J1-3	J1-25	28 V MLB RTN	> 100k		
J1-3	J2-3	28 V MLB RTN	> 100k		
J1-3	J4-12	28 V MLB RTN	> 100k		
J1-3	J4-13	28 V MLB RTN	> 100k		

TEST DATA SHEET 1 (Sheet 7 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-5	J1-7	+28 V PLB	> 100k		
J1-5	J1-9	+28 V PLB	> 100k		
J1-5	J1-10	+28 V PLB	> 100k		
J1-5	J1-24	+28 V PLB	> 100k		
J1-5	J1-25	+28 V PLB	> 100k		
J1-5	J2-3	+28 V PLB	> 100k		
J1-5	J4-12	+28 V PLB	> 100k		
J1-5	J4-13	+28 V PLB	> 100k		
J1-7	J1-9	28 V PLB RTN	> 100k		
J1-7	J1-10	28 V PLB RTN	> 100k		
J1-7	J1-24	28 V PLB RTN	> 100k		
J1-7	J1-25	28 V PLB RTN	> 100k		
J1-7	J2-3	28 V PLB RTN	> 100k		
J1-7	J4-12	28 V PLB RTN	> 100k		
J1-7	J4-13	28 V PLB RTN	> 100k		
J1-9	J1-10	+28 V TMB	> 100k		
J1-9	J1-24	+28 V TMB	> 100k		
J1-9	J1-25	+28 V TMB	> 100k		
J1-9	J2-3	+28 V TMB	> 100k		
J1-9	J4-12	+28 V TMB	> 100k		
J1-9	J4-13	+28 V TMB	> 100k		
J1-10	J1-24	28 V TMB RTN	> 100k		
J1-10	J1-25	28 V TMB RTN	> 100k		
J1-10	J2-3	28 V TMB RTN	> 100k		
J1-10	J4-12	28 V TMB RTN	> 100k		
J1-10	J4-13	28 V TMB RTN	> 100k		
J1-24	J1-25	SAFETY HTR PWR	> 100k		
J1-24	J2-3	SAFETY HTR PWR	> 100k		
J1-24	J4-12	SAFETY HTR PWR	> 100k		
J1-24	J4-13	SAFETY HTR PWR	> 100k		
J1-25	J2-3	SAFETY HTR PWR RTN	> 100k		
J1-25	J4-12	SAFETY HTR PWR RTN	> 100k		
J1-25	J4-13	SAFETY HTR PWR RTN	> 100k		
J2-3	J4-12	SIGNAL RTN	> 100k		
J2-3	J4-13	SIGNAL RTN	> 100k		
J4-12	J4-13	+10 V INTERFACE BUS	> 100k		

TEST DATA SHEET 1 (Sheet 8 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J2-2	J4-13	DATA CLOCK (C1)	> 2k		
J2-5	J4-13	DIGITAL-A DATA OUT	> 2k		
J2-6	J4-13	DATA ENABLE (A1)	> 2k		
J2-7	J4-13	8 SEC SYNC PULSE	> 2k		
J3-1	J4-13	1.248 MHZ CLK	> 2k		
J3-2	J4-13	1.248 MHZ CLK RTN	> 2k		
J4-2	J4-13	MODULE PWR DISCONN	> 2k		
J4-3	J4-13	SURVIVAL HTR ON	> 2k		
J4-4	J4-13	MODULE TOTALLY OFF	> 2k		
J4-5	J4-13	SCANNER A1-2 ON/OFF	> 2k		
J4-6	J4-13	ANT AT COLD CAL POS	> 2k		
J4-7	J4-13	PLL SELECT	> 2k		
J4-8	J4-13	ANT AT NADIR POS	> 2k		
J4-9	J4-13	COLD CAL POS MSB (IN)	> 2k		
J4-14	J4-13	MODULE PWR CONN	> 2k		
J4-15	J4-13	SURVIVAL HTR OFF	> 2k		
J4-16	J4-13	SCANNER A1-1 ON/OFF	> 2k		
J4-17	J4-13	ANT AT WARM CAL POS	> 2k		
J4-18	J4-13	FULL SCAN	> 2k		
J4-19	J4-13	COLD CAL POS LSB (IN)	> 2k		
J5-2	J4-13	MODULE PWR IND	> 2k		
J5-3	J4-13	COLD CAL POS MSB (OUT)	> 2k		
J5-5	J4-13	SCANNER A1-2 ON/OFF	> 2k		
J5-6	J4-13	ANT IN COLD CAL POS	> 2k		
J5-7	J4-13	PLL PRI/RED	> 2k		
J5-9	J4-13	SURV HTR ON/OFF	> 2k		
J5-11	J4-13	COLD CAL POS LSB (OUT)	> 2k		
J5-12	J4-13	SCANNER A1-1 ON/OFF	> 2k		
J5-13	J4-13	ANT IN WARM CAL POS	> 2k		
J5-14	J4-13	ANT IN NADIR POS	> 2k		
J5-15	J4-13	FULL SCAN MODE	> 2k		

TEST DATA SHEET 1 (Sheet 9 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J6-2	J1-10	RF SHELF A1-1 TEMP	> 2k		
J6-3	J1-10	A1-1 SCAN MTR.TEMP	> 2k		
J6-4	J1-10	WARM LOAD A1-1 TEMP	> 2k		
J6-6	J4-13	PLLO RED LOCK DETECT	> 2k		
J6-8	J4-13	A1-1 DRIVE MTR CVR	> 2k		
J6-9	J4-13	+15 VDC ANT DRIVE MON	> 2k		
J6-10	J4-13	+5 VDC ANT DRIVE MON	> 2k		
J6-11	J4-13	+15 VDC SIG PROC MON	> 2k		
J6-12	J4-13	+5VDC SIG PROC MON	> 2k		
J6-13	J4-13	L.O. VOLTAGE CH3 MON	> 2k		
J6-14	J4-13	L.O. VOLTAGE CH5 MON	> 2k		
J6-15	J4-13	L.O. VOLTAGE CH7 MON	> 2k		
J6-16	J4-13	+15 VDC PLL LO MON	> 2k		
J6-17	J4-13	+10 V MIXER/AMP MON	> 2k		
J6-18	J4-13	L.O. VOLTAGE CH15 MON	> 2k		
J6-21	J4-10	RF SHELF A1-2 TEMP	> 2k		
J6-22	J4-10	A1-2 SCAN MTR.TEMP	> 2k		
J6-23	J4-10	WARM LOAD A1-2 TEMP	> 2k		
J6-25	J4-13	PLLO PRI LOCK DETECT	> 2k		
J6-27	J4-13	A1-2 DRIVE MTR CURR	> 2k		
J6-28	J4-13	-15 VDC ANT DRIVE MON	> 2k		
J6-29	J4-13	-15 VDC SIG PROC MON	> 2k		
J6-30	J4-13	L.O. VOLTAGE CH4 MON	> 2k		
J6-31	J4-13	L.O. VOLTAGE CH6 MON	> 2k		
J6-32	J4-13	L.O. VOLTAGE CH8 MON	> 2k		
J6-33	J4-13	-15 VDC PLL LO MON	> 2k		
J6-34	J4-13	IF AMP MON	> 2k		

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Customer Representative
(Flight Hardware Only)

Date

Test Systems Engineer

Quality Control

Date

Date

TEST DATA SHEET 2
+28 MLB During Turn-on Transient (Paragraph 3.2.4.2.1.1)

At 28.56 Vdc:

Step	Parameter	Measured/ Calculated	Required*		
			S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	_____ms	20 ms max	300 ms max	
8	Peak Current	_____Amps	10.6 Amps	5.9 Amps	
10	Rate of Change (Slope): dI/dT	_____mA/μs	677 mA/μs	250 mA/μs	

At 27.44 Vdc:

Step	Parameter	Measured/ Calculated	Required*		
			S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	_____ms	20 ms max	300 ms max	
8	Peak Current	_____Amps	10.6 Amps	5.9 Amps	
10	Rate of Change (Slope): dI/dT	_____mA/μs	677 mA/μs	250 mA/μs	

At 28.00 Vdc:

Step	Parameter	Measured/ Calculated	Required*		
			S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	_____ms	20 ms max	300 ms max	
8	Peak Current	_____Amps	10.6 Amps	5.9 Amps	
10	Rate of Change (Slope): dI/dT	_____mA/μs	677 mA/μs	250 mA/μs	

Refer to Figure 5.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer _____ Date _____

Customer Representative _____ Date _____
(Flight Hardware Only)

Quality Control _____ Date _____

6 Apr 99

TEST DATA SHEET 3
+28 MLB Operating Power (Paragraph 3.2.4.2.1.2)

Step	+28V MLB at 27 Volts	Measured	Units	Required	Pass/Fail
2	+28 V MLB voltage at 27 V (V_b) (Measured)		Volts	27.0 ± 0.1	
3	Average Current (I_V) (PLLO#1)		Amps	N/A	N/A
4	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)		Watts	82 W max	
6	Average current (I_V) (PLLO#2)		Amps	N/A	N/A
7	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)		Watts	82 W max	
+28 V MLB at 28 Volts					
9	+28 V MLB bus voltage at 28 V (V_b) (Measured)		Volts	28.0 ± 0.1	
10	Average Current (I_V) (PLLO#1)		Amps	N/A	N/A
11	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)		Watts	82 W max	
13	Average current (I_V) (PLLO#2)		Amps	N/A	N/A
14	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)		Watts	82 W max	
+28 V MLB at 29 Volts					
16	+28 V MLB voltage at 29 V (V_b) (Measured)		Volts	29.0 ± 0.1	
17	Average Current (I_V) (PLLO#1)		Amps	N/A	N/A
18	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)		Watts	82 W max	
20	Average current (I_V) (PLLO#2)		Amps	N/A	N/A
21	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)		Watts	82 W max	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer _____ Date _____

Customer Representative _____ Date _____
(Flight Hardware Only)

Quality Control _____ Date _____

TEST DATA SHEET 4 (Sheet 1 of 2)
+28 Pulse Load Bus (Paragraph 3.2.4.2.2.1-3.2.4.2.2.6)

Paragraph	Parameter	Measured or Calculated	Required	Pass/ Fail
3.2.4.2.2.1	From -0.1 to two seconds			
	Peak Current = I_p	___Amps	1.3 amps max	
3.2.4.2.2.2	From 2 to 4 seconds			
	Peak Current = I_p	___Amps	1.3 amps max	
3.2.4.2.2.3	From 4 to 6 seconds			
	Peak Current = I_p	___Amps	1.3 amps max	
3.2.4.2.2.4	From 6 to 8 seconds			
	Peak Current = I_p	___Amps	1.3 amps max	
3.2.4.2.2.5	Eight Sec. Integrated Current Measurement:			
	Current	___mA	None	
3.2.4.2.2.6	Turn-on Transient:			
	dI/dT	___mA/ μ s	744 mA/ μ s *	
	Peak Current = I_p	___Amps	11.5 Amps	

* Refer to Figure 9.

Bus current during the I/H, D period

Paragraph	Parameter	Measured or Calculated	Pass/ Fail
3.2.4.2.2.1	From -0.1 to 2 secs	mA	N/A
3.2.4.2.2.2	From 2 to 4 secs	mA	N/A
3.2.4.2.2.3	From 4 to 6 secs	mA	N/A
3.2.4.2.2.5	From 6 to 8 secs	mA	N/A

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer _____ Date _____

Customer Representative _____ Date _____
(Flight Hardware Only)

Quality Control _____ Date _____

TEST DATA SHEET 4 (Sheet 2 of 2)
+28 Pulse Load Bus (Paragraph 3.2.4.2.2.7)

Bus current during warm cal, cold cal, & Nadir

Paragraph	Parameter	Measured or Calculated	Pass/ Fail
3.2.4.2.2.7 (2)	Warm cal	mA	N/A
3.2.4.2.2.7 (3)	Cold cal	mA	N/A
3.2.4.2.2.7 (4)	Nadir	mA	N/A
3.2.4.2.2.7 (5)	Warm cal (motors off)	mA	N/A

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

+28 V Analog Telemetry Bus (Paragraph 3.2.4.2.3)

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
3	+28 V ATB Bus Voltage (V_{at}) (Measured)	____ Volts	28.0 \pm 0.5	
4	Av. Current (I_a)	____ mA	7 mA max	
5	+28 V ATB Operating Power = $I_a \times V_{at}$	____ mW	200 mW max	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 6
+10 V Interface Bus Voltage (Paragraph 3.2.4.2.4)

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
3	Av. Current (I_a)	____ mA	10 mA max	
3	+10 V Interface Bus (V_{ib}) (Measured)	____ Volts	9.0 ± 1.0 V	
4	+10 V Interface Bus Power = $I_a \times V_{ib}$	____ mW	100 mW max	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

Power Input Test for LPT (Paragraph 3.2.4.2.5)

Step	Parameter	Measured	Units	Required	Pass/ Fail
3	+28 V MLB Voltage (Vb) (Measured at connector J1)		Volts	28 ±0.5	
3	Current		Amps	Between 0.5 and 4.3 Amps	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer
Date

Customer Representative _____ Date _____
 (Flight Hardware Only)

Quality Control
Date

TEST DATA SHEET 8
1.248 MHz Clock Signal Verification (Paragraph 3.2.4.3.2.1)

1.248 CLOCK SIGNAL
ATTACH PHOTOGRAPH OR PLOT HERE

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
5	Clock Frequency	____ MHz	1.248 \pm 10%	
	Clock Amplitude	____ Volts	9.0 \pm 1.0 V	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 9

"C1" Shift Pulse Verification (Paragraph 3.2.4.3.2.2)

“C1” SHIFT PULSE
Attach Photograph OR Plot Here

Parameter	Measured/ Calculated	Required	Pass/ Fail
Pulse Timing (A) *	____μs	48 μs ± 10%	
Pulse Timing (B) *	____μs	12 μs ± 10%	
Pulse Amplitude	____ Volts	9.0 ± 1.0 V	

* Refer to Figure 19 for location of the pulse timing A and B.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

"A1" Select Pulse Verification (Paragraph 3.2.4.3.2.3)

Parameter	Measured/ Calculated	Required	Pass/ Fail
Select Pulse Timing (F) *	___ μ s	961.5 μ s \pm 10%	
Select Pulse Amplitude	___ Volts	9.0 \pm 1.0 V	

* Refer to Figure 12.6 for timing details.

* Refer to Figure 13 for location of the pulse timing F

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative _____ Date _____
 (Flight Hardware Only)

Quality Control _____ Date _____

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
1*	Frame Sync Pulse Timing (G)*	___ Sec	8 Sec $\pm 10\%$	
	Frame Sync Pulse Timing (C)*	___ μ s	240.4 μ s $\pm 10\%$	
	Frame Sync Pulse Amplitude	___ Volts	9.0 ± 1.0 V	

A-21

A1 Select pulse and the 8 seconds Frame sync pulse.

ATTACH PHOTOGRAPH OR PLOT HERE

Verify that the sync pulse between H and C is as shown in Figure 19.

TIME MEASURED: _____

TIME REQUIRED: 1.2 ms $\pm 10\%$

PASS/FAIL _____

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer
Date

Customer Representative _____ Date _____
 (Flight Hardware Only)

Quality Control
Date

TEST DATA SHEET 12 (Sheet 2 of 2)
Synchronization Signals Relationship (Paragraph 3.2.4.3.2.5)

A1 Select pulse and the C1 Shift pulse.

ATTACH PHOTOGRAPH OR PLOT HERE

Verify that the sync pulse between I and E is as shown in Figure 19.

TIME MEASURED: _____

TIME REQUIRED: $24 \mu\text{s} \pm 1 \mu\text{s}$

PASS/FAIL _____

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 13

Synchronization Signals Relationship (Paragraph 3.2.4.3.2.5)

A1 Select pulse and the 1.248 MHz clock.

Verify that the sync pulse between I and J is as shown in Figure 19.

PASS/FAIL _____

ATTACH PHOTOGRAPH OR PLOT HERE

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer
Date

Customer Representative _____ Date _____
 (Flight Hardware Only)

Quality Control
Date

TEST DATA SHEET 14

Commands and Digital-B Telemetry Verification (Paragraphs 3.2.4.3.3.1, 3.2.4.3.3.2, 3.2.4.3.3.3, and 3.2.4.3.3.4)

Test	Digital-B Commands Verification Via STE			Visual Inspection		Pass/Fail
	Command	Observed	Required	Observed	Required	
3.2.4.3.3.1 Module Totally Off	Scanner A1-1		OFF		Antenna pointing to warm load.	
	Scanner A1-2		OFF		Antenna pointing to warm load.	
	Module Power		Disconnect	N/A	N/A	
	Survival Htr. Power.		OFF		28 V supply current=0	
3.2.4.3.3.2 Survival Heater Power	Survival Heater ON		ON	N/A	N/A	
	Survival Heater OFF		OFF	N/A	N/A	
3.2.4.3.3.3 Module Power Connect	Module Power		Connect		+28 V DC current is between 0.5 and 3.2 amps.	
3.2.4.3.3.4 PLL Power	PLLO#2		PLLO#2	N/A	N/A	
	PLLO#1		PLLO#1	N/A	N/A	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

6 Apr 99

TEST DATA SHEET 15
Scanner Commands Verification (Paragraph 3.2.4.3.3.5, Step 1)

Test	Digital "B" Verification			Pass/Fail
	Command	Observed	Required	
Full Scan	1 Module Power		CONNECT	
	2 Survival Heater		OFF	
	3 Scanner A1 Power		ON	
	4 Scanner A2 Power		ON	
	5 Antenna Warm Cal Pos.		NO	
	6 Antenna Cold Cal Pos.		NO	
	7 Antenna NADIR Position		NO	
	8 Antenna Full Scan		YES	
	9 PLL Power		PLL#1	
	10 Cold MSB		0	
	11 Cold LSB		0	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer_____
Date_____
Customer Representative
(Flight Hardware Only)_____
Date_____
Quality Control_____
Date

Scanner Commands Verification (Paragraph 3.2.4.3.3.5, Step 3)

Date _____

TEST DATA SHEET 18

Scanner Positions Commands (Paragraph 3.2.4.3.3.6)

Test	Digital "B" Verification			Pass/Fail	
	Step/Description		Observed		Required
Scanner Position Commands	1-Warm Cal.			YES	
	2-Cold Cal. Pos.	MSB		0	
		LSB		1	
	3-Cold Cal. Pos.	MSB		1	
		LSB		0	
	4-Cold Cal. Pos.	MSB		1	
		LSB		1	
	5-Cold Cal. Pos.	MSB		0	
		LSB		0	
6-NADIR			YES		
7-Warm Cal			YES		

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

TEST DATA SHEET 19
Digital-A Data Output Full Scan Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.1)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1		255	
	0002	Sync Sequence Byte 2		255	
	0003	Sync Sequence Byte 3		255	
[II]	0004	Unit I.D. and Serial N		*	
[III]	0005	Digital-B Data Byte 1		2	
	0006	Digital-B Data Byte 2		**	
	0007	Digital-B Data Byte 3		0	
	0008	Digital-B Data Byte 4		0	
* AMSU A1 Identification Words (data entered in decimal system)					
			Binary	Decimal	
		AMSU-A1 S/N 101	00000001	1	
		AMSU-A1 S/N 102	00000101	5	
		AMSU-A1 S/N 103	00001001	9	
		AMSU-A1 S/N 104	00001101	13	
		AMSU-A1 S/N 105	00010001	17	
		AMSU-A1 S/N 106	00010101	21	
		AMSU-A1 S/N 107	00011001	25	
		AMSU-A1 S/N 108	00011101	29	
		AMSU-A1 S/N 109	00100001	33	
** Required value = 14 when PLLO #1 is active; and = 6 when PLLO #2 is active.					
Circle Test: CPT LPT					
METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____					
_____ Test Systems Engineer					_____ Date
_____ Customer Representative (Flight Hardware Only)			_____ Date	_____ Quality Control	
_____ Date					

TEST DATA SHEET 20
Reflector Positions Section [IV] (Paragraph 3.2.4.3.4.1)

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0014				0016			
02	0048				0050			
03	0082				0084			
04	0116				0118			
05	0150				0152			
06	0184				0186			
07	0218				0220			
08	0252				0254			
09	0286				0288			
10	0320				0322			
11	0354				0356			
12	0388				0390			
13	0422				0424			
14	0456				0458			
15	0490				0492			
16	0524				0526			
17	0558				0560			
18	0592				0594			
19	0626				0628			
20	0660				0662			
21	0694				0696			
22	0728				0730			
23	0762				0764			
24	0796				0798			
25	0830				0832			
26	0864				0866			
27	0890				0900			
28	0932				0934			
29	0966				0968			
30	1000				1002			
CC	1034				1036			
WC	1186				1188			

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date Quality Control Date
(Flight Hardware Only)

6 Apr 99

TEST DATA SHEET 21

Digital-A Data Output Radiometer Data Section [V] (Paragraph 3.2.4.3.4.1)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
01	0018				0030			
02	0052				0064			
03	0086				0098			
04	0120				0132			
05	0154				0166			
06	0188				0200			
07	0222				0234			
08	0256				0268			
09	0290				0302			
10	0324				0336			
11	0356				0370			
12	0392				0404			
13	0426				0438			
14	0460				0472			
15	0494				0506			
16	0528				0540			
17	0562				0574			
18	0596				0608			
19	0630				0642			
20	0664				0676			
21	0698				0710			
22	0732				0744			
23	0766				0778			
24	0800				0812			
25	0834				0846			
26	0868				0880			
27	0902				0914			
28	0936				0948			
29	0970				0982			
30	1004				1016			
CC	1038				1050			
WC	1190				1202			

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = $16,500 \pm 4000$ counts.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer_____
Date_____
Customer Representative
(Flight Hardware Only)_____
Date_____
Quality Control_____
Date

TEST DATA SHEET 22 (Sheet 1 of 2)
Full Scan Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.1)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1		25 ± 15	
1092	A1-1 Warm Load 2		25 ± 15	
1094	A1-1 Warm Load 3		25 ± 15	
1096	A1-1 Warm Load 4		25 ± 15	
1098	A1-1 Warm Load Center		25 ± 15	
1100	A1-2 Warm Load 1		25 ± 15	
1102	A1-2 Warm Load 2		25 ± 15	
1104	A1-2 Warm Load 3		25 ± 15	
1106	A1-2 Warm Load 4		25 ± 15	
1108	A1-2 Warm Load Center		25 ± 15	
1110	Local Oscillator Channel 7		25 ± 15	
1112	Local Oscillator Channel 8		25 ± 15	
1114	Local Oscillator Channel 15		25 ± 15	
1116	PLL LO #2 Channels 9-14		25 ± 15	
1118	PLL LO #1 Channels 9-14		25 ± 15	
1120	PLLO (Reference Oscillator)**/ Not used ***		25 ± 15	
1122	Mixer I.F. Amp. Channel 3		25 ± 15	
1124	Mixer I.F. Amp. Channel 4		25 ± 15	
1126	Mixer I.F. Amp. Channel 5		25 ± 15	
1128	Mixer I.F. Amp. Channel 6		25 ± 15	
1130	Mixer I.F. Amp. Channel 7		25 ± 15	
1132	Mixer I.F. Amp. Channel 8		25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14		25 ± 15	
1136	Mixer I.F. Amp. Channel 15		25 ± 15	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** For S/N 101 through 104.

*** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 22 (Sheet 2 of 2)
Full Scan Mode Temperature Sensors Section [VI (Paragraph 3.2.4.3.4.1)]

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14		25 ± 15	
1140	I.F. Amp. Channel 9		25 ± 15	
1142	I.F. Amp. Channel 10		25 ± 15	
1144	I.F. Amp. Channel 11		25 ± 15	
1146	DC/DC Converter		25 ± 15	
1148	I.F. Amp. Channel 13		25 ± 15	
1150	I.F. Amp. Channel 14		25 ± 15	
1152	I.F. Amp. Channel 12		25 ± 15	
1154	RF Shelf A1-1		25 ± 15	
1156	RF Shelf A1-2		25 ± 15	
1158	Detector Preamp Assy.		25 ± 15	
1160	Scan Motor A1-1		25 ± 15	
1162	Scan Motor A1-2		25 ± 15	
1164	Feed Horn A1-1		25 ± 15	
1166	Feed Horn A1-2		25 ± 15	
1168	R.F. Mux A1-1		25 ± 15	
1170	R.F. Mux A1-2		25 ± 15	
1172	Local Oscillator Channel 3		25 ± 15	
1174	Local Oscillator Channel 4		25 ± 15	
1176	Local Oscillator Channel 5		25 ± 15	
1178	Local Oscillator Channel 6		25 ± 15	
1180	Temp Sensor Ref Voltage Count		**	

* Value is from the STE printout sheets. Copying data to this sheet is optional.
** = Count of 24,552 +1765,-1308.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 23
Digital-A Data Output Warm Cal Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.2)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1		255	
	0002	Sync Sequence Byte 2		255	
	0003	Sync Sequence Byte 3		255	
[II]	0004	Unit I.D. and Serial N		*	
[III]	0005	Digital-B Data Byte 1		4	
	0006	Digital-B Data Byte 2		14	
	0007	Digital-B Data Byte 3		0	
	0008	Digital-B Data Byte 4		0	
* AMSU A1 Identification Words (data entered in decimal system)					
			Binary	Decimal	
		AMSU-A1 S/N 101	00000001	1	
		AMSU-A1 S/N 102	00000101	5	
		AMSU-A1 S/N 103	00001001	9	
		AMSU-A1 S/N 104	00001101	13	
		AMSU-A1 S/N 105	00010001	17	
		AMSU-A1 S/N 106	00010101	21	
		AMSU-A1 S/N 107	00011001	25	
		AMSU-A1 S/N 108	00011101	29	
		AMSU-A1 S/N 109	00100001	33	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

TEST DATA SHEET 24

Reflector Position Warm Cal Mode Section [IV] and Reflector Position Nadir Mode Section [IV] (Paragraphs 3.2.4.3.4.2 and 3.2.4.3.4.4)

BP	A1-1 Reflector			
	Para No.	Position*	Required**	Pass/Fail
WC	3.2.4.3.4.2			
15	3.2.4.3.4.4			
WC = Warm Cal 15 = Nadir Position				
BP	A1-2 Reflector			
	Para No.	Position*	Required**	Pass/Fail
WC	3.2.4.3.4.2			
15	3.2.4.3.4.4			
WC = Warm Cal 15 = Nadir Position				
<p>* Actual counts from computer printout. Rewriting counts on this data sheet is optional.</p> <p>** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.</p>				

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer _____ Date _____

Customer Representative _____ Date _____
(Flight Hardware Only)

Quality Control _____ Date _____

TEST DATA SHEET 25
Digital-A Data Output Warm Cal Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.2)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0018				0030			
02	0052				0064			
03	0086				0098			
04	0120				0132			
05	0154				0166			
06	0188				0200			
07	0222				0234			
08	0256				0268			
09	0290				0302			
10	0324				0336			
11	0356				0370			
12	0392				0404			
13	0426				0438			
14	0460				0472			
15	0494				0506			
16	0528				0540			
17	0562				0574			
18	0596				0608			
19	0630				0642			
20	0664				0676			
21	0698				0710			
22	0732				0744			
23	0766				0778			
24	0800				0812			
25	0834				0846			
26	0868				0880			
27	0902				0914			
28	0936				0948			
29	0970				0982			
30	1004				1016			
CC	1038		0		1050		0	
WC	1190		0		1202		0	

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

TEST DATA SHEET 26 (Sheet 1 of 2)
Warm Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.2)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1		25 ± 15	
1092	A1-1 Warm Load 2		25 ± 15	
1094	A1-1 Warm Load 3		25 ± 15	
1096	A1-1 Warm Load 4		25 ± 15	
1098	A1-1 Warm Load Center		25 ± 15	
1100	A1-2 Warm Load 1		25 ± 15	
1102	A1-2 Warm Load 2		25 ± 15	
1104	A1-2 Warm Load 3		25 ± 15	
1106	A1-2 Warm Load 4		25 ± 15	
1108	A1-2 Warm Load Center		25 ± 15	
1110	Local Oscillator Channel 7		25 ± 15	
1112	Local Oscillator Channel 8		25 ± 15	
1114	Local Oscillator Channel 15		25 ± 15	
1116	PLL LO #2 Channels 9-14		25 ± 15	
1118	PLL LO #1 Channels 9-14		25 ± 15	
1120	PLLO (Reference Oscillator)**/ Not used ***			
1122	Mixer I.F. Amp. Channel 3		25 ± 15	
1124	Mixer I.F. Amp. Channel 4		25 ± 15	
1126	Mixer I.F. Amp. Channel 5		25 ± 15	
1128	Mixer I.F. Amp. Channel 6		25 ± 15	
1130	Mixer I.F. Amp. Channel 7		25 ± 15	
1132	Mixer I.F. Amp. Channel 8		25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14		25 ± 15	
1136	Mixer I.F. Amp. Channel 15		25 ± 15	

- * Value is from the STE printout sheets. Copying data to this sheet is optional.
 ** For S/N 101 through 104.
 *** For S/N 105 and up.

(Continued on Sheet 2)

Warm Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.2)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14		25 ± 15	
1140	I.F. Amp. Channel 9		25 ± 15	
1142	I.F. Amp. Channel 10		25 ± 15	
1144	I.F. Amp. Channel 11		25 ± 15	
1146	DC/DC Converter		25 ± 15	
1148	I.F. Amp. Channel 13		25 ± 15	
1150	I.F. Amp. Channel 14		25 ± 15	
1152	I.F. Amp. Channel 12		25 ± 15	
1154	RF Shelf A1-1		25 ± 15	
1156	RF Shelf A1-2		25 ± 15	
1158	Detector Preamp Assy.		25 ± 15	
1160	Scan Motor A1-1		25 ± 15	
1162	Scan Motor A1-2		25 ± 15	
1164	Feed Horn A1-1		25 ± 15	
1166	Feed Horn A1-2		25 ± 15	
1168	R.F. Mux A1-1		25 ± 15	
1170	R.F. Mux A1-2		25 ± 15	
1172	Local Oscillator Channel 3		25 ± 15	
1174	Local Oscillator Channel 4		25 ± 15	
1176	Local Oscillator Channel 5		25 ± 15	
1178	Local Oscillator Channel 6		25 ± 15	
1180	Temp Sensor Ref Voltage Count		**	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** = Count of 24,552 + 1765, - 1308.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 27
Digital-A Data Output Cold Cal Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.3)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1		255	
	0002	Sync Sequence Byte 2		255	
	0003	Sync Sequence Byte 3		255	
[II]	0004	Unit I.D. and Serial N		*	
[III]	0005	Digital-B Data Byte 1		8	
	0006	Digital-B Data Byte 2		14	
	0007	Digital-B Data Byte 3		0	
	0008	Digital-B Data Byte 4		0	
* AMSU A1 Identification Words (data entered in decimal system)					
			Binary	Decimal	
		AMSU-A1 S/N 101	00000001	1	
		AMSU-A1 S/N 102	00000101	5	
		AMSU-A1 S/N 103	00001001	9	
		AMSU-A1 S/N 104	00001101	13	
		AMSU-A1 S/N 105	00010001	17	
		AMSU-A1 S/N 106	00010101	21	
		AMSU-A1 S/N 107	00011001	25	
		AMSU-A1 S/N 108	00011101	29	
		AMSU-A1 S/N 109	00100001	33	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 28 (Sheet 1 of 2)

Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV] (Paragraphs 3.2.4.3.4.2, 3.2.4.3.4.3, and 3.2.4.3.4.4)

BP	A1-1 Reflector			
	Para No.	Position*	Required**	Pass/Fail
CC	3.2.4.3.4.3, Step 4			
	a.			
	b.			
	c.			
	d.			

CC = Cold Cal

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.

3.2.4.3.4.3, Step 4 Substep	MSB	LSB
a.	0	0
b.	0	1
c.	1	0
d.	1	1

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative (Flight Hardware Only) Date

Quality Control Date

TEST DATA SHEET 28 (Sheet 2 of 2)

Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV (Paragraphs 3.2.4.3.4.2, 3.2.4.3.4.3, and 3.2.4.3.4.4)

BP	A1-2 Reflector			
	Para No.	Position*	Required**	Pass/Fail
CC	3.2.4.3.4.3, Step 4			
	a.			
	b.			
	c.			
	d.			

CC = Cold Cal

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.

3.2.4.3.4.3, Step 4 Substep	MSB	LSB
a.	0	0
b.	0	1
c.	1	0
d.	1	1

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer
Date

Customer Representative _____ Date _____
 (Flight Hardware Only)

Quality Control
Date

TEST DATA SHEET 29

Digital-A Data Output Cold Cal Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.3)

Condition: Cold Cal Position MSB=0 and Cold Cal Position LSB=0

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0018				0030			
02	0052				0064			
03	0086				0098			
04	0120				0132			
05	0154				0166			
06	0188				0200			
07	0222				0234			
08	0256				0268			
09	0290				0302			
10	0324				0336			
11	0356				0370			
12	0392				0404			
13	0426				0438			
14	0460				0472			
15	0494				0506			
16	0528				0540			
17	0562				0574			
18	0596				0608			
19	0630				0642			
20	0664				0676			
21	0698				0710			
22	0732				0744			
23	0766				0778			
24	0800				0812			
25	0834				0846			
26	0868				0880			
27	0902				0914			
28	0936				0948			
29	0970				0982			
30	1004				1016			
CC	1038		0		1050		0	
WC	1190		0		1202		0	

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

TEST DATA SHEET 30 (Sheet 1 of 2)
Cold Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.3)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1		25 ± 15	
1092	A1-1 Warm Load 2		25 ± 15	
1094	A1-1 Warm Load 3		25 ± 15	
1096	A1-1 Warm Load 4		25 ± 15	
1098	A1-1 Warm Load Center		25 ± 15	
1100	A1-2 Warm Load 1		25 ± 15	
1102	A1-2 Warm Load 2		25 ± 15	
1104	A1-2 Warm Load 3		25 ± 15	
1106	A1-2 Warm Load 4		25 ± 15	
1108	A1-2 Warm Load Center		25 ± 15	
1110	Local Oscillator Channel 7		25 ± 15	
1112	Local Oscillator Channel 8		25 ± 15	
1114	Local Oscillator Channel 15		25 ± 15	
1116	PLL LO #2 Channels 9-14		25 ± 15	
1118	PLL LO #1 Channels 9-14		25 ± 15	
1120	PLLO (Reference Oscillator)**/ Not used ***			
1122	Mixer I.F. Amp. Channel 3		25 ± 15	
1124	Mixer I.F. Amp. Channel 4		25 ± 15	
1126	Mixer I.F. Amp. Channel 5		25 ± 15	
1128	Mixer I.F. Amp. Channel 6		25 ± 15	
1130	Mixer I.F. Amp. Channel 7		25 ± 15	
1132	Mixer I.F. Amp. Channel 8		25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14		25 ± 15	
1136	Mixer I.F. Amp. Channel 15		25 ± 15	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** For S/N 101 through 104.

*** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 30 (Sheet 2 of 2)
Cold Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.3)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14		25 ± 15	
1140	I.F. Amp. Channel 9		25 ± 15	
1142	I.F. Amp. Channel 10		25 ± 15	
1144	I.F. Amp. Channel 11		25 ± 15	
1146	DC/DC Converter		25 ± 15	
1148	I.F. Amp. Channel 13		25 ± 15	
1150	I.F. Amp. Channel 14		25 ± 15	
1152	I.F. Amp. Channel 12		25 ± 15	
1154	RF Shelf A1-1		25 ± 15	
1156	RF Shelf A1-2		25 ± 15	
1158	Detector Preamp Assy.		25 ± 15	
1160	Scan Motor A1-1		25 ± 15	
1162	Scan Motor A1-2		25 ± 15	
1164	Feed Horn A1-1		25 ± 15	
1166	Feed Horn A1-2		25 ± 15	
1168	R.F. Mux A1-1		25 ± 15	
1170	R.F. Mux A1-2		25 ± 15	
1172	Local Oscillator Channel 3		25 ± 15	
1174	Local Oscillator Channel 4		25 ± 15	
1176	Local Oscillator Channel 5		25 ± 15	
1178	Local Oscillator Channel 6		25 ± 15	
1180	Temp Sensor Ref Voltage Count		**	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** = Count of 24,552 +1765,-1308.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 31
Digital-A Data Output Nadir Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.4)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1		255	
	0002	Sync Sequence Byte 2		255	
	0003	Sync Sequence Byte 3		255	
[II]	0004	Unit I.D. and Serial N		*	
[III]	0005	Digital-B Data Byte 1		16	
	0006	Digital-B Data Byte 2		14	
	0007	Digital-B Data Byte 3		0	
	0008	Digital-B Data Byte 4		0	

* AMSU A1 Identification Words
(data entered in decimal system)

	Binary	Decimal
AMSU-A1 S/N 101	00000001	1
AMSU-A1 S/N 102	00000101	5
AMSU-A1 S/N 103	00001001	9
AMSU-A1 S/N 104	00001101	13
AMSU-A1 S/N 105	00010001	17
AMSU-A1 S/N 106	00010101	21
AMSU-A1 S/N 107	00011001	25
AMSU-A1 S/N 108	00011101	29
AMSU-A1 S/N 109	00100001	33

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

TEST DATA SHEET 32
Digital-A Data Output Nadir Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.4)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
01	0018				0030			
02	0052				0064			
03	0086				0098			
04	0120				0132			
05	0154				0166			
06	0188				0200			
07	0222				0234			
08	0256				0268			
09	0290				0302			
10	0324				0336			
11	0356				0370			
12	0392				0404			
13	0426				0438			
14	0460				0472			
15	0494				0506			
16	0528				0540			
17	0562				0574			
18	0596				0608			
19	0630				0642			
20	0664				0676			
21	0698				0710			
22	0732				0744			
23	0766				0778			
24	0800				0812			
25	0834				0846			
26	0868				0880			
27	0902				0914			
28	0936				0948			
29	0970				0982			
30	1004				1016			
CC	1038		0		1050		0	
WC	1190		0		1202		0	

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.
 ** Required = 16,500 ± 4000 counts (Unless otherwise indicated).

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 33 (Sheet 1 of 2)
Nadir Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.4)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1		25 ± 15	
1092	A1-1 Warm Load 2		25 ± 15	
1094	A1-1 Warm Load 3		25 ± 15	
1096	A1-1 Warm Load 4		25 ± 15	
1098	A1-1 Warm Load Center		25 ± 15	
1100	A1-2 Warm Load 1		25 ± 15	
1102	A1-2 Warm Load 2		25 ± 15	
1104	A1-2 Warm Load 3		25 ± 15	
1106	A1-2 Warm Load 4		25 ± 15	
1108	A1-2 Warm Load Center		25 ± 15	
1110	Local Oscillator Channel 7		25 ± 15	
1112	Local Oscillator Channel 8		25 ± 15	
1114	Local Oscillator Channel 15		25 ± 15	
1116	PLL LO #2 Channels 9-14		25 ± 15	
1118	PLL LO #1 Channels 9-14		25 ± 15	
1120	PLLO (Reference Oscillator)**/ Not used ***		25 ± 15	
1122	Mixer I.F. Amp. Channel 3		25 ± 15	
1124	Mixer I.F. Amp. Channel 4		25 ± 15	
1126	Mixer I.F. Amp. Channel 5		25 ± 15	
1128	Mixer I.F. Amp. Channel 6		25 ± 15	
1130	Mixer I.F. Amp. Channel 7		25 ± 15	
1132	Mixer I.F. Amp. Channel 8		25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14		25 ± 15	
1136	Mixer I.F. Amp. Channel 15		25 ± 15	

- * Value is from the STE printout sheets. Copying data to this sheet is optional.
 ** For S/N 101 through 104.
 *** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 33 (Sheet 2 of 2)
Nadir Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.4)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14		25 ± 15	
1140	I.F. Amp. Channel 9		25 ± 15	
1142	I.F. Amp. Channel 10		25 ± 15	
1144	I.F. Amp. Channel 11		25 ± 15	
1146	DC/DC Converter		25 ± 15	
1148	I.F. Amp. Channel 13		25 ± 15	
1150	I.F. Amp. Channel 14		25 ± 15	
1152	I.F. Amp. Channel 12		25 ± 15	
1154	RF Shelf A1-1		25 ± 15	
1156	RF Shelf A1-2		25 ± 15	
1158	Detector Preamp Assy.		25 ± 15	
1160	Scan Motor A1-1		25 ± 15	
1162	Scan Motor A1-2		25 ± 15	
1164	Feed Horn A1-1		25 ± 15	
1166	Feed Horn A1-2		25 ± 15	
1168	R.F. Mux A1-1		25 ± 15	
1170	R.F. Mux A1-2		25 ± 15	
1172	Local Oscillator Channel 3		25 ± 15	
1174	Local Oscillator Channel 4		25 ± 15	
1176	Local Oscillator Channel 5		25 ± 15	
1178	Local Oscillator Channel 6		25 ± 15	
1180	Temp Sensor Ref Voltage Count		**	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** = Count of 24,552 +1765,-1308.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

6 Apr 99

TEST DATA SHEET 34

Analog Telemetry Verification by Way of Connector J6 (Paragraph 3.2.4.3.5.1)

	From	Description	To	Measured (volts)	Required (volts)	Pass/Fail
03	J6-02	RF Shelf A1-1 Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
01	J6-03	A1-1 Scan Motor Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
05	J6-04	Warm Load A1-1 Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
04	J6-21	RF Shelf A1-2 Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
02	J6-22	A1-2 Scan Motor Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
06	J6-23	Warm Load A1-2 Temp.	J1-10	_____	$3.5 \pm 2 \text{ V}$	_____
25	J6-06	PLLO No. 2 Lock detect	J2-03	_____	***	_____
07	J6-08	A1-1 Drive Motor Curr.	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
10	J6-09	+15 V Antenna Drive	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
15	J6-10	+5 V Antenna Drive	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
09	J6-11	+15 V Signal Processing	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
14	J6-12	+5 V Signal Processing	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
22	J6-13	L.O. Voltage Channel 3	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
24	J6-14	L.O. Voltage Channel 5	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
20	J6-15	L.O. Voltage Channel 7	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
16	J6-16	+15 V PLL LO Ch 9-14	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
17	J6-17	*	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
27	J6-18	L.O. Voltage Channel 15	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
26	J6-25	PLLO No. 1 Lock detect	J2-03	_____	***	_____
08	J6-27	A1-2 Drive Motor Curr.	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
12	J6-28	-15 V Antenna Drive	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
11	J6-29	-15 V Signal Processing	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
23	J6-30	L.O. Voltage Channel 4	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
21	J6-31	L.O. Voltage Channel 6	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
19	J6-32	L.O. Voltage Channel 8	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
18	J6-33	-15 V PLL LO Ch 9-14	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____
13	J6-34	**	J2-03	_____	$3.5 \pm 2 \text{ V}$	_____

* +8.5 V PLL LO Ch 9-14 for S/N 101-104, +10V Mixer Amp for S/N 105 and above.

** +8 V Receiver for S/N 101-104, +8 V IF Amp for S/N 105 and above.

*** 4.5 ± 0.5 when locked, 0.5 ± 0.5 when unlocked or OFF. One must be locked.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer_____
Date_____
Customer Representative
(Flight Hardware Only)_____
Date_____
Quality Control_____
Date

Analog Telemetry Signals by Way of the STE (Paragraph 3.2.4.3.5.2)

	Description	(*)	Measured (Deg. C)	Required (Deg. C)	Pass/Fail
01	A1-1 Scanner Motor	Temp	_____	25 ± 15	_____
02	A1-2 Scanner Motor	Temp	_____	25 ± 15	_____
03	A1-1 RF Shelf	Temp	_____	25 ± 15	_____
04	A1-2 RF Shelf	Temp	_____	25 ± 15	_____
05	A1-1 Warm Load	Temp	_____	25 ± 15	_____
06	A1-2 Warm Load	Temp	_____	25 ± 15	_____
			(mAmps)	(mAmps)	
07	Ant A1-1 Drv Motor Current		_____	125 mA (Max)	_____
08	Ant A1-2 Drv Motor Current		_____	125 mA (Max)	_____

(*) Data from the printout sheet. Rewriting data on this space is optional.

(Continued on sheet 2)

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 35 (Sheet 2 of 2)
Analog Telemetry Signals by Way of the STE (Paragraph 3.2.4.3.5.2)

	Description	(*)	Measured (volts)	Required (volts)	Pass/ Fail
09	Signal Processing	+15 V	_____	15.0 ± 0.5 V	_____
10	Antenna Drive	+15 V	_____	15.0 ± 0.5 V	_____
11	Signal Processing	-15 V	_____	-15.0 ± 0.5 V	_____
12	Antenna Drive	-15 V	_____	-15.0 ± 0.5 V	_____
13	Receiver	+8 V	_____	8.0 ± 0.5 V	_____
14	Sig Processing	+5 V	_____	5.0 ± 0.5 V	_____
15	Antenna Drive	+5 V	_____	5.0 ± 0.5 V	_____
16	Phase Lock Loop Ch 9-14 (a)/ Receiver/Mixer IF (b)	+8.5 V +10 V	_____ _____	8.5 ± 0.5 V 10.0 ± 0.5 V	_____ _____
17	Phase Lock Loop Ch 9-14	+15 V	_____	15.0 ± 0.5 V	_____
18	Phase Lock Loop Ch 9-14	-15 V	_____	-15.0 ± 0.5 V	_____
19	L.O. #8	Ch-8	_____	(**)__ ± 0.5 V	_____
20	L.O. #7	Ch-7	_____	(**)__ ± 0.5 V	_____
21	L.O. #6	Ch-6	_____	(**)__ ± 0.5 V	_____
22	L.O. #3	Ch-3	_____	(**)__ ± 0.5 V	_____
23	L.O. #4	Ch-4	_____	(**)__ ± 0.5 V	_____
24	L.O. #5	Ch-5	_____	(**)__ ± 0.5 V	_____
25	PLLO No. 2 Lock Detect		_____	(***)	_____
26	PLLO No. 1 Lock Detect		_____	(***)	_____
27	L.O. #15	Ch-15	_____	(**)__ ± 0.5 V	_____

(*) Data from the printout sheet. Rewriting data on this space is optional.

(**) GDO voltages from the manufacturer data sheet for S/N 101-104; DRO CH3-8 10V, GDO CH15 15V for S/N 105 and above.

(***) Locked PLO voltage 0 to +15 V, other PLO voltage ±15.0 V; one must be locked for S/N 101-104. Locked PLO voltage 4.0 ± 1.0 V, other PLO voltage 0.0 ± 0.2 V, one must be locked for S/N 105 and above.

(a) For S/N 101 through 104. (b) For S/N 105 and up.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

Parameter	Measured	Required	Pass/ Fail
Scope Channel-1: Integration/Hold			
Time Measured (A)*	ms	165 ms \pm 10%	
Time Measured (B)*	ms	35 ms \pm 10%	
Amplitude Measured	V	5.0 \pm 0.2 V	
Scope Channel-2: Dump Signal			
Time Measured (D)*	ms	9 ms to 15 ms	
Amplitude Measured	V	5.0 \pm 0.2 V	

Quality Control
Date

TEST DATA SHEET 37
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____03
Frequency: _____50.3 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____04
Frequency: _____52.8 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 38
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____05
Frequency: _____53.596 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____06
Frequency: _____54.4 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 39

Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel 07

Frequency: _____ 54.94 GHz

INTEGRATION (X) *

Measured _____ ms

Required 165 ms \pm 10%

Pass/Fail _____

HOLD (B-D) *

Measured _____ ms

Required 25 ms \pm 10%

Pass/Fail _____

DUMP (D) *

Measured _____ ms

Required 9 ms to 15 ms

Pass/Fail _____

Channel 08

Frequency: _____ 55.5 GHz

INTEGRATION (X) *

Measured _____ ms

Required 165 ms \pm 10%

Pass/Fail _____

HOLD (B-D) *

Measured _____ ms

Required 25 ms \pm 10%

Pass/Fail _____

DUMP (D) *

Measured _____ ms

Required 9 ms to 15 ms

Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 40
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____09
Frequency: _____57.2903 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____10
Frequency: _____57.2903 GHz

INTEGRATION (X) *
Measured _____ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ms
Required 9 ms to 15 ms
Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

6 Apr 99

TEST DATA SHEET 41

Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____ 11

Frequency: _____ 57.3903 GHz

INTEGRATION (X) *

Measured _____ ms

Required 165 ms \pm 10%

Pass/Fail _____

HOLD (B-D) *

Measured _____ ms

Required 25 ms \pm 10%

Pass/Fail _____

DUMP (D) *

Measured _____ ms

Required 9 ms to 15 ms

Pass/Fail _____

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____ 12

Frequency: _____ 57.3903 GHz

INTEGRATION (X) *

Measured _____ ms

Required 165 ms \pm 10%

Pass/Fail _____

HOLD (B-D) *

Measured _____ ms

Required 25 ms \pm 10%

Pass/Fail _____

DUMP (D) *

Measured _____ ms

Required 9 ms to 15 ms

Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer_____
Date_____
Customer Representative
(Flight Hardware Only)_____
Date_____
Quality Control_____
Date

TEST DATA SHEET 42
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____ 13
Frequency: _____ 57.3903 GHz

INTEGRATION (X) *
Measured _____ ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ ms
Required 9 ms to 15 ms
Pass/Fail _____

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____ 14
Frequency: _____ 57.3903 GHz

INTEGRATION (X) *
Measured _____ ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ ms
Required 9 ms to 15 ms
Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 43
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

ATTACH PHOTOGRAPH OR PLOT HERE

Channel _____ 15
Frequency: _____ 89 GHz

INTEGRATION (X) *
Measured _____ ms
Required 165 ms \pm 10%
Pass/Fail _____

HOLD (B-D) *
Measured _____ ms
Required 25 ms \pm 10%
Pass/Fail _____

DUMP (D) *
Measured _____ ms
Required 9 ms to 15 ms
Pass/Fail _____

* Refer to Figure 2 for waveform configuration.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 44
PLLO No. 1 Verification (Paragraph 3.2.4.3.6.3)
PLLO No. 2 Verification (Paragraph 3.2.4.3.6.4)

PLLO NO. 1

PLLO No. 1 dc Level _____

Required: *

Pass/Fail _____

PLLO NO. 2

PLLO No. 2 dc Level _____

Required: *

Pass/Fail _____

* -15 to +15 V dc level for S/N 101 - S/N 104, 4.0 \pm 1.0 V for S/N 105 and above.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____

S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

6 Apr 99

TEST DATA SHEET 45

Digital-A/GSE Mode-1 Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.7.2)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1		255	
	0002	Sync Sequence Byte 2		255	
	0003	Sync Sequence Byte 3		255	
[II]	0004	Unit I.D. and Serial N		*	
[III]	0005	Digital-B Data Byte 1		0	
	0006	Digital-B Data Byte 2		14	
	0007	Digital-B Data Byte 3		0	
	0008	Digital-B Data Byte 4		0	
* AMSU A1 Identification Words (data entered in decimal system)					
			Binary	Decimal	
AMSU-A1 S/N 101			00000001	1	
AMSU-A1 S/N 102			00000101	5	
AMSU-A1 S/N 103			00001001	9	
AMSU-A1 S/N 104			00001101	13	
AMSU-A1 S/N 105			00010001	17	
AMSU-A1 S/N 106			00010101	21	
AMSU-A1 S/N 107			00011001	25	
AMSU-A1 S/N 108			00011101	29	
AMSU-A1 S/N 109			00100001	33	

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date

(Flight Hardware Only) _____
Quality Control Date

TEST DATA SHEET 46 (Sheet 1 of 2)
Reflector Position (Paragraphs 3.2.4.3.7.2 - 3.2.4.3.7.7)

3.2.4.3.7.2 Digital-A/GSE Mode-1 Reflector Position Section [IV] ***

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
06	0184				0186			
CC	354				356			
WC	694				696			

3.2.4.3.7.3 Digital-A/GSE Mode-2 Reflector Position Section [IV] ***

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
01	0014				0016			

3.2.4.3.7.4 Digital-A/GSE Mode-3 Reflector Position Section [IV] ***

A1-1 Reflector			A1-2 Reflector		
Observed	Required**	Pass/Fail	Observed	Required**	Pass/Fail
	****			****	

- * Actual counts from computer printout. Rewriting counts on this data sheet is optional.
 ** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.
 *** GSE Modes do not require verification or testing for PFM & FM modules
 **** Observe that both A1-1 and A1-2 reflectors increment one step every 8 seconds.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 46 (Sheet 2 of 2)
Reflector Position (Paragraphs 3.2.4.3.7.2 - 3.2.4.3.7.7)

3.2.4.3.7.5 Digital-A/GSE Mode-4 Reflector Position Section [IV] ***

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
30	1000				1002			

3.2.4.3.7.6 Digital-A/GSE Mode-5 Reflector Position Section [IV] ***

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
06	0184				0186			

3.2.4.3.7.7 Digital-A/GSE Mode-7 Reflector Position Section [IV] ***

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
06	0184				0186			

- * Actual counts from computer printout. Rewriting counts on this data sheet is optional.
 ** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.
 *** GSE Modes do not require verification or testing for PFM & FM modules

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 47

Digital-A/GSE Mode-1 Radiometer Data Section [V] (Paragraph 3.2.4.3.7.2)

BP	A1-1 Reflector			A1-2 Reflector		
	Channel-3*	Required**	Pass/Fail	Channel-9*	Required**	Pass/Fail
01						
02						
03						
04						
05						
06						
07						
08						
09						
10						
11						
12						
13						
14						
15						
16						
17						
18						
19						
20						
21						
22						
23						
24						
25						
26						
27						
28						
29						
30						

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

**** Required = 16,500 ± 4000 counts.**

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 48 (Sheet 1 of 2)
Digital-A/GSE Mode-1 Temperature Sensors Section [VI] (Paragraph 3.2.4.3.7.2)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1		25 ± 15	
1092	A1-1 Warm Load 2		25 ± 15	
1094	A1-1 Warm Load 3		25 ± 15	
1096	A1-1 Warm Load 4		25 ± 15	
1098	A1-1 Warm Load Center		25 ± 15	
1100	A1-2 Warm Load 1		25 ± 15	
1102	A1-2 Warm Load 2		25 ± 15	
1104	A1-2 Warm Load 3		25 ± 15	
1106	A1-2 Warm Load 4		25 ± 15	
1108	A1-2 Warm Load Center		25 ± 15	
1110	Local Oscillator Channel 7		25 ± 15	
1112	Local Oscillator Channel 8		25 ± 15	
1114	Local Oscillator Channel 15		25 ± 15	
1116	PLL LO #2 Channels 9-14		25 ± 15	
1118	PLL LO #1 Channels 9-14		25 ± 15	
1120	PLLO (Reference Oscillator)**/ Not used ***			
1122	Mixer I.F. Amp. Channel 3		25 ± 15	
1124	Mixer I.F. Amp. Channel 4		25 ± 15	
1126	Mixer I.F. Amp. Channel 5		25 ± 15	
1128	Mixer I.F. Amp. Channel 6		25 ± 15	
1130	Mixer I.F. Amp. Channel 7		25 ± 15	
1132	Mixer I.F. Amp. Channel 8		25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14		25 ± 15	
1136	Mixer I.F. Amp. Channel 15		25 ± 15	

- * Value is from the STE printout sheets. Copying data to this sheet is optional.
 ** For S/N 101 through 104.
 *** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 48 (Sheet 2 of 2)

Digital-A/GSE Mode-1 Temperature Sensors Section [VI] (Paragraph 3.2.4.3.7.2)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14		25 ± 15	
1140	I.F. Amp. Channel 9		25 ± 15	
1142	I.F. Amp. Channel 10		25 ± 15	
1144	I.F. Amp. Channel 11		25 ± 15	
1146	DC/DC Converter		25 ± 15	
1148	I.F. Amp. Channel 13		25 ± 15	
1150	I.F. Amp. Channel 14		25 ± 15	
1152	I.F. Amp. Channel 12		25 ± 15	
1154	RF Shelf A1-1		25 ± 15	
1156	RF Shelf A1-2		25 ± 15	
1158	Detector Preamp Assy.		25 ± 15	
1160	Scan Motor A1-1		25 ± 15	
1162	Scan Motor A1-2		25 ± 15	
1164	Feed Horn A1-1		25 ± 15	
1166	Feed Horn A1-2		25 ± 15	
1168	R.F. Mux A1-1		25 ± 15	
1170	R.F. Mux A1-2		25 ± 15	
1172	Local Oscillator Channel 3		25 ± 15	
1174	Local Oscillator Channel 4		25 ± 15	
1176	Local Oscillator Channel 5		25 ± 15	
1178	Local Oscillator Channel 6		25 ± 15	
1180	Temp Sensor Ref Voltage Count		**	

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** = Count of 24,552 +1765,-1308.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 49
Receiver Input Signals (Paragraph 3.2.4.4.1)

CH 9 through 14 PLLO	PRT Temp (°C)		Measured *	Requirements **	Pass/ Fail
PLLO No. 1	PLO No. 1	Xtal *** Osc.		57290.334 MHz ± 50 kHz	
PLLO No. 2	PLO No. 2	Xtal *** Osc.		57290.334 MHz ± 50 kHz	

* Attach spectrum analyzer plots.

** = At 18°C

*** PRT not connected on S/N 105 and above.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

Radiometer "Relative" NEAT Verification* (Paragraph 3.2.4.4.2.2)

Channels 3, 4, 5, 6, 7, 8, and 15. PLL0 No. 1 (Channels 9 through 14)

Channel Number>	3	4	5	6
NEAT (Average of 5 data)	_____	_____	_____	_____
Pass/Fail	_____	_____	_____	_____
NEAT (Specified) K **	0.40	0.25	0.25	0.25
Channel Number>	7	8	9	10
NEAT (Average of 5 data)	_____	_____	_____	_____
Pass/Fail	_____	_____	_____	_____
NEAT (Specified) K **	0.25	0.25	0.25	0.40
Channel Number>	11	12	13	14
NEAT (Average of 5 data)	_____	_____	_____	_____
Pass/Fail	_____	_____	_____	_____
NEAT (Specified) K **	0.40	0.60	0.80	1.20
Channel Number>	15			
NEAT (Average of 5 data)	_____			
Pass/Fail	_____			
NEAT (Specified) K **	0.50			

* Baseline data for acceptance tests. Use first CPT or first LPT data along with specification value for pass/fail criteria

**** For reference only**

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer

Date _____

**Customer Representative
(Flight Hardware Only)**

Date _____

Quality Control

Date _____

TEST DATA SHEET 50 (Sheet 2 of 2)
Radiometer "Relative" NEΔT Verification* (Paragraph 3.2.4.4.2.2)

PLLO No. 2 (Channels 9 through 14)

Channel Number>	9	10	11	12
NEΔT (Average of 5 data)	_____	_____	_____	_____
Pass/Fail	_____	_____	_____	_____
NEΔT (Specified) K **	0.25	0.40	0.40	0.60
Channel Number>	13	14		
NEΔT (Average of 5 data)	_____	_____		
Pass/Fail	_____	_____		
NEΔT (Specified) K **	0.80	1.20		

* Baseline data for acceptance tests. Use first CPT or first LPT data along with specification value for pass/fail criteria

** For reference only

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 51 (Sheet 1 of 2)
Transient Susceptibility Test (Paragraph 3.2.4.2.1.4, 3.2.4.2.2.9, 3.2.4.2.3.3)

Test Setup Verified: _____
Signature

3.2.4.2.1.4: +28V Main Bus Load-Induced Transient Test

Subpara	Step	Load Induced Transient	Functional Performance Results/Deviations	Comments/ Observations
3.2.4.2.1.4.2	8	Low frequency in accordance with Figure 8		
3.2.4.2.1.4.3	10	High frequency 1.43 Hz 200 mV p-p		
3.2.4.2.1.4.3	10	High frequency 2.86 Hz 1.00 V p-p		
3.2.4.2.1.4.3	10	High frequency 6.67 Hz 1.50 V p-p		

NOTE: Attach all backup data generated during the test (photos, printouts, plots, test logs, additional comments or observations, etc.) to this data sheet.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

TEST DATA SHEET 51 (Sheet 2 of 2)
Transient Susceptibility Test (Paragraph 3.2.4.2.1.4, 3.2.4.2.2.9, 3.2.4.2.3.3)

Test Setup Verified: _____
Signature

3.2.4.2.2.9: +28V Pulse Load Bus Load-Induced Transient Test

Subpara	Step	Load Induced Transient	Functional Performance Results/Deviations	Comments/ Observations
3.2.4.2.2.9.2	8	Low frequency in accordance with Figure 13		
3.2.4.2.2.9.3	10	High frequency 1.43 Hz 200 mV p-p		
3.2.4.2.2.9.3	10	High frequency 2.86 Hz 1.00 V p-p		
3.2.4.2.2.9.3	10	High frequency 6.67 Hz 1.50 V p-p		

3.2.4.2.3.3: +28V Analog Telemetry Bus Load-Induced Transient Test

Subpara	Step	Load Induced Transient	Functional Performance Results/Deviations	Comments/ Observations
3.2.4.2.3.3.2	8	Low frequency in accordance with Figure 16		
3.2.4.2.3.3.3	10	High frequency 1.43 Hz 200 mV p-p		
3.2.4.2.3.3.3	10	High frequency 2.86 Hz 1.00 V p-p		
3.2.4.2.3.3.3	10	High frequency 6.67 Hz 1.50 V p-p		

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

Channel Number	Antenna Location	Sweeper Freq. Setting (GHz)	Polarization (H/V)	Radiometric Data Counts Δ Counts	Channel Verified (Yes/No)
3	A1-2	50.35	V		
4	A1-2	52.85	V		
5	A1-2	53.70	H		
6	A1-1	54.45	H		
7	A1-1	54.99	V		
8	A1-2	55.55	H		
9	A1-1	57.34	H		
10	A1-1	57.50	H		
11	A1-1	57.564	H		
12	A1-1	57.59	H		
13	A1-1	57.602	H		
14	A1-1	57.608	H		
15	A1-1	89.55	V		

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: _____ S/N: _____

Test Systems Engineer Date

Customer Representative Date
(Flight Hardware Only)

Quality Control Date

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DOCUMENT APPROVAL SHEET

TITLE Process Specification METSAT/KLM/AMSU-A1, System Comprehensive and Limited Performance Tests Test Procedure				DOCUMENT NO. AE-26156/3C 6 April 1999	
INPUT FROM:		DATE	CDRL:	SPECIFICATION ENGINEER: <i>James A. Gomer</i>	DATE 4-5-99
CHECKED BY: <i>EM W. Sch.</i>		DATE 99-04-06	JOB NUMBER: N/A		DATE
APPROVED SIGNATURES				DEPT. NO.	DATE
System Safety (W. Neighbors) <i>W. Neighbors</i>				8331	4/7/99
Product Team Leader (A. Nieto) <i>A. Nieto</i>				8341	4/7/99
Systems Engineer (R. Platt) <i>P. R. Patel</i>				8341	4/7/99
Design Assurance (E. Lorenz) <i>D. Woon (for E. Lorenz)</i>				8331	4/8/99
Quality Assurance (R. Taylor) <i>My N/A (for R. Taylor)</i>				7831	4/8/99
Technical Director/PMO (P. Patel) <i>P. R. Patel</i>				8341	4/7/99
Released: Configuration Management (J. Cavanaugh) <i>J. Cavanaugh</i>				8361	4/12/99
Approved as Final per customer's letter dated 12 April 1999 (ECN CAMSU-2091, CAMSU-2101 and CAMSU-2104)					
By my signature, I certify the above document has been reviewed by me and concurs with the technical requirements related to my area of responsibility.					
(Data Center) FINAL					
<i>Ramona Cornejo</i> 4-13-99					

6 Apr 99

TEST DATA SHEET 1 (Sheet 1 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J1 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-1	+28 V MLB	> 100k	OL	Pass
J1-2	+28 V MLB	> 100k	OL	Pass
J1-3	+28 V MLB RTN	> 100k	OL	Pass
J1-4	+28 V MLB RTN	> 100k	OL	Pass
J1-5	+28 V PLB	> 100k	OL	Pass
J1-6	+28 V PLB	> 100k	OL	Pass
J1-7	+28 V PLB RTN	> 100k	OL	Pass
J1-8	+28 V PLB RTN	> 100k	OL	Pass
J1-9	+28 V TMB	> 100k	OL	Pass
J1-10	28 V TMB RTN	> 100k	OL	Pass
J1-11	NO CONNECTION	> 100k	OL	Pass
J1-12	NO CONNECTION	> 100k	OL	Pass
J1-13	CHASSIS GROUND (E1)	< 1	0.21 Ω	Pass
J1-14	+28 V MLB	> 100k	OL	Pass
J1-15	+28 V MLB	> 100k	OL	Pass
J1-16	+28 V MLB RTN	> 100k	OL	Pass
J1-17	+28 V MLB RTN	> 100k	OL	Pass
J1-18	+28 V PLB	> 100k	OL	Pass
J1-19	+28 V PLB	> 100k	OL	Pass
J1-20	+28 V PLB RTN	> 100k	OL	Pass
J1-21	+28 V PLB RTN	> 100k	OL	Pass
J1-22	+28 V TMB	> 100k	OL	Pass
J1-23	28 V TMB RTN	> 100k	OL	Pass
J1-24	SAFETY HTR PWR	> 100k	OL	Pass
J1-25	SAFETY HTR RTN	> 100k	OL	Pass

OL = $\geq 40 \text{ Meg } \Omega$

TEST DATA SHEET 1 (Sheet 2 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

J2 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J2-1	Chassis Ground (E2)	< 1	.18 Ω	P
J2-2	DATA CLOCK (C1)	> 100k	OL	P
J2-3	Signal Return	> 100k	OL	P
J2-4	No Connection	> 100k	OL	P
J2-5	DIGITAL-A DATA OUT	> 100k	OL	P
J2-6	DATA ENABLE (A1)	> 100k	OL	P
J2-7	8 SEC SYNC PULSE	> 100k	OL	P
J2-8	No Connection	> 100k	OL	P
J2-9	No Connection	> 100k	OL	P

J3 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J3-1	1.248 MHz CLK	> 100k	OL	P
J3-2	1.248 MHz CLK RTN	> 100k	OL	P
J3-3	Chassis GND (E3)	< 1	.18 Ω	P

J5 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J5-1	Chassis Ground (E5)	< 1	.15 Ω	P
J5-2	MODULE PWR IND	> 100k	OL	P
J5-3	COLD CAL POS MSB (OUT)	> 100k	OL	P
J5-4	No Connection	> 100k	OL	P
J5-5	SCANNER A1-2 ON/OFF	> 100k	OL	P
J5-6	ANT IN COLD CAL POS	> 100k	OL	P
J5-7	PLL PRI/RED	> 100k	OL	P
J5-8	No Connection	> 100k	OL	P
J5-9	SURV HTR ON/OFF	> 100k	OL	P
J5-10	No Connection	> 100k	OL	P
J5-11	COLD CAL POS LSB (OUT)	> 100k	OL	P
J5-12	SCANNER A1-1 ON/OFF	> 100k	OL	P
J5-13	ANT IN WARM CAL POS	> 100k	OL	P
J5-14	ANT IN NADIR POS	> 100k	OL	P
J5-15	FULL SCAN MODE	> 100k	OL	P

OL = $\geq 40 \text{ Meg } \Omega$

TEST DATA SHEET 1 (Sheet 3 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J4 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J4-1	Chassis Ground (E4)	< 1	0.15 Ω	P
J4-2	MODULE PWR DISCONN	> 100k	OL	P
J4-3	SURVIVAL HTR ON	> 100k	OL	P
J4-4	MODULE TOTALLY OFF	> 100k	OL	P
J4-5	SCANNER A1-2 ON/OFF	> 100k	OL	P
J4-6	ANT AT COLD CAL POS	> 100k	OL	P
J4-7	PLL SELECT	> 100k	OL	P
J4-8	ANT AT NADIR POS	> 100k	OL	P
J4-9	COLD CAL POS MSB (IN)	> 100k	OL	P
J4-10	No Connection	> 100k	OL	P
J4-11	No Connection	> 100k	OL	P
J4-12	+10 V INTERFACE BUS	> 100k	OL	P
J4-13	10 V INTERFACE BUS RTN	> 100k	14.3 M	P
J4-14	MODULE PWR CONN	> 100k	OL	P
J4-15	SURVIVAL HTR OFF	> 100k	OL	P
J4-16	SCANNER A1-1 ON/OFF	> 100k	OL	P
J4-17	ANT AT WARM CAL POS	> 100k	OL	P
J4-18	FULL SCAN	> 100k	OL	P
J4-19	COLD CAL POS LSB (IN)	> 100k	OL	P
J4-20	No Connection	> 100k	OL	P
J4-21	No Connection	> 100k	OL	P
J4-22	No Connection	> 100k	OL	P
J4-23	No Connection	> 100k	OL	P
J4-24	+10 V INTERFACE BUS	> 100k	OL	P
J4-25	10 V INTERFACE BUS RTN	> 100k	OL	P

OL = $\geq 40 \text{ MEG } \Omega$

TEST DATA SHEET 1 (Sheet 4 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J6 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J6-1	Chassis GND (E6)	< 1	0.15 Ω	P
J6-2	RF SHELF A1-1 TEMP	> 100k	OL	P
J6-3	A1-1 SCAN. MTR. TEMP	> 100k	OL	P
J6-4	WARM LOAD A1-1 TEMP	> 100k	OL	P
J6-5	No Connection	> 100k	OL	P
J6-6	PLLO RED LOCK DETECT	> 100k	OL	P
J6-7	No Connection	> 100k	OL	P
J6-8	A1-1 DRIVE MTR CURR	> 100k	OL	P
J6-9	+15 V ANT DR MON	> 100k	OL	P
J6-10	+5 V ANT DR MON	> 100k	OL	P
J6-11	+15 V SIG PROC MON	> 100k	OL	P
J6-12	+5 V SIG PROC MON	> 100k	OL	P
J6-13	L.O. VOLTAGE CH 3 MON	> 100k	OL	P
J6-14	L.O. VOLTAGE CH 5 MON	> 100k	OL	P
J6-15	L.O. VOLTAGE CH 7 MON	> 100k	OL	P
J6-16	+15 VDC PLL LO MON	> 100k	OL	P
J6-17	+10 V MIXER/AMP MON	> 100k	OL	P
J6-18	L.O. VOLTAGE CH 15 MON	> 100k	OL	P
J6-19	No Connection	> 100k	OL	P
J6-20	28 V TMB RTN	> 100k	OL	P
J6-21	RF SHELF A1-2 TEMP	> 100k	OL	P
J6-22	A1-2 SCAN MTR TEMP	> 100k	OL	P
J6-23	WARM LOAD A1-2 TEMP	> 100k	OL	P
J6-24	No Connection	> 100k	OL	P
J6-25	PLLO PRI LOCK DETECT	> 100k	OL	P
J6-26	No Connection	> 100k	OL	P
J6-27	A1-2 DRIVE MTR CURR	> 100k	OL	P
J6-28	-15 V ANT DR MON	> 100k	OL	P
J6-29	-15 V SIG PROC MON	> 100k	OL	P
J6-30	L.O. VOLTAGE CH 4 MON	> 100k	OL	P
J6-31	L.O. VOLTAGE CH 6 MON	> 100k	OL	P
J6-32	L.O. VOLTAGE CH 8 MON	> 100k	OL	P
J6-33	-15 VDC PLL LO MON	> 100k	OL	P
J6-34	+8 V IF AMP MON	> 100k	OL	P
J6-35	No Connection	> 100k	OL	P
J6-36	No Connection	> 100k	OL	P
J6-37	No Connection	> 100k	OL	P

OL \Rightarrow 90 MEG Ω

6 Apr 99

TEST DATA SHEET 1 (Sheet 5 of 9)
Grounding System Test (Paragraph 3.2.4.1)

J7 of Spacecraft Interface				
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J7-1	Chassis GND (E7)	< 1	0.17 Ω	P
J7-2	No Connection	> 100k	OL	P
J7-3	REDUN PLO LOCK DET	> 100k	OL	P
J7-4	15 V RTN (2/3)	> 100k	OL	P
J7-5	15 V RTN (2/3)	> 100k	OL	P
J7-6	DUMP TEST POINT	> 100k	OL	P
J7-7	No Connection	> 100k	OL	P
J7-8	CH3 OUT TEST POINT	> 100k	OL	P
J7-9	CH4 OUT TEST POINT	> 100k	OL	P
J7-10	CH5 OUT TEST POINT	> 100k	OL	P
J7-11	CH6 OUT TEST POINT	> 100k	OL	P
J7-12	CH7 OUT TEST POINT	> 100k	OL	P
J7-13	CH8 OUT TEST POINT	> 100k	OL	P
J7-14	CH9 OUT TEST POINT	> 100k	OL	P
J7-15	No Connection	> 100k	OL	P
J7-16	No Connection	> 100k	OL	P
J7-17	GSE CMD LSB	> 100k	OL	P
J7-18	GSE CMD MSB-1	> 100k	OL	P
J7-19	+5 V GSE INTERLOCK A	> 100k	OL	P
J7-20	No Connection	> 100k	OL	P
J7-21	No Connection	> 100k	OL	P
J7-22	PRI PLO LOCK DET	> 100k	OL	P
J7-23	No Connection	> 100k	OL	P
J7-24	I/H TEST POINT	> 100k	OL	P
J7-25	No Connection	> 100k	OL	P
J7-26	15 V RTN (2/3)	> 100k	OL	P
J7-27	CH10 OUT TEST POINT	> 100k	OL	P
J7-28	CH11 OUT TEST POINT	> 100k	OL	P
J7-29	CH12 OUT TEST POINT	> 100k	OL	P
J7-30	CH13 OUT TEST POINT	> 100k	OL	P
J7-31	CH14 OUT TEST POINT	> 100k	OL	P
J7-32	CH15 OUT TEST POINT	> 100k	OL	P
J7-33	No Connection	> 100k	OL	P
J7-34	No Connection	> 100k	OL	P
J7-35	GSE CMD MSB	> 100k	OL	P
J7-36	5 V RTN (1)	> 100k	OL	P
J7-37	+5 V GSE INTERLOCK B	> 100k	OL	P

TEST DATA SHEET 1 (Sheet 6 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-1	J1-2	+28 V MLB	< 1	.25 Ω	P
J1-1	J1-14	+28 V MLB	< 1	.37 Ω	P
J1-1	J1-15	+28 V MLB	< 1	.35 Ω	P
J1-3	J1-4	28 V MLB RTN	< 1	.23 Ω	P
J1-3	J1-16	28 V MLB RTN	< 1	.29 Ω	P
J1-3	J1-17	28 V MLB RTN	< 1	.25 Ω	P
J1-5	J1-6	+28 V PLB	< 1	.26 Ω	P
J1-5	J1-18	+28 V PLB	< 1	.35 Ω	P
J1-5	J1-19	+28 V PLB	< 1	.29 Ω	P
J1-7	J1-8	28 V PLB RTN	< 1	.21 Ω	P
J1-7	J1-20	28 V PLB RTN	< 1	.27 Ω	P
J1-7	J1-21	28 V PLB RTN	< 1	.32 Ω	P
J1-9	J1-22	+28 V TMB	< 1	.17 Ω	P
J1-10	J1-23	28 V TMB RTN	< 1	.25 Ω	P
J1-10	J6-20	28 V TMB RTN	< 1	.43 Ω	P
J4-12	J4-24	+10 V INTERFACE BUS	< 1	.31 Ω	P
J4-13	J4-25	10 V INTERFACE BUS RTN	< 1	.37 Ω	P
J1-1	J1-3	+28 V MLB	> 100k	8.6 M Ω	P
J1-1	J1-5	+28 V MLB	> 100k	16.2 M Ω	P
J1-1	J1-7	+28 V MLB	> 100k	17.2 M Ω	P
J1-1	J1-9	+28 V MLB	> 100k	OL	P
J1-1	J1-10	+28 V MLB	> 100k	OL	P
J1-1	J1-24	+28 V MLB	> 100k	OL	P
J1-1	J1-25	+28 V MLB	> 100k	OL	P
J1-1	J2-3	+28 V MLB	> 100k	15.6 M Ω	P
J1-1	J4-12	+28 V MLB	> 100k	OL	P
J1-1	J4-13	+28 V MLB	> 100k	OL	P
J1-3	J1-5	28 V MLB RTN	> 100k	1.3 M Ω	P
J1-3	J1-7	28 V MLB RTN	> 100k	393 K Ω	P
J1-3	J1-9	28 V MLB RTN	> 100k	OL	P
J1-3	J1-10	28 V MLB RTN	> 100k	OL	P
J1-3	J1-24	28 V MLB RTN	> 100k	OL	P
J1-3	J1-25	28 V MLB RTN	> 100k	OL	P
J1-3	J2-3	28 V MLB RTN	> 100k	192 K Ω	P
J1-3	J4-12	28 V MLB RTN	> 100k	2.6 M Ω	P
J1-3	J4-13	28 V MLB RTN	> 100k	2.5 M Ω	P

OL = $\geq 40 \text{ MEG } \Omega$

TEST DATA SHEET 1 (Sheet 7 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-5	J1-7	+28 V PLB	> 100k	5.4 MΩ	P
J1-5	J1-9	+28 V PLB	> 100k	OL	P
J1-5	J1-10	+28 V PLB	> 100k	OL	P
J1-5	J1-24	+28 V PLB	> 100k	OL	P
J1-5	J1-25	+28 V PLB	> 100k	OL	P
J1-5	J2-3	+28 V PLB	> 100k	7.6 MΩ	P
J1-5	J4-12	+28 V PLB	> 100k	20.7 MΩ	P
J1-5	J4-13	+28 V PLB	> 100k	22.3 MΩ	P
J1-7	J1-9	28 V PLB RTN	> 100k	OL	P
J1-7	J1-10	28 V PLB RTN	> 100k	OL	P
J1-7	J1-24	28 V PLB RTN	> 100k	OL	P
J1-7	J1-25	28 V PLB RTN	> 100k	OL	P
J1-7	J2-3	28 V PLB RTN	> 100k	182 KΩ 2.6 MΩ	P
J1-7	J4-12	28 V PLB RTN	> 100k	2.6 MΩ	P
J1-7	J4-13	28 V PLB RTN	> 100k	2.5 MΩ	P
J1-9	J1-10	+28 V TMB	> 100k	7.7 MΩ	P
J1-9	J1-24	+28 V TMB	> 100k	OL	P
J1-9	J1-25	+28 V TMB	> 100k	OL	P
J1-9	J2-3	+28 V TMB	> 100k	OL	P
J1-9	J4-12	+28 V TMB	> 100k	OL	P
J1-9	J4-13	+28 V TMB	> 100k	OL	P
J1-10	J1-24	28 V TMB RTN	> 100k	OL	P
J1-10	J1-25	28 V TMB RTN	> 100k	OL	P
J1-10	J2-3	28 V TMB RTN	> 100k	OL	P
J1-10	J4-12	28 V TMB RTN	> 100k	OL	P
J1-10	J4-13	28 V TMB RTN	> 100k	OL	P
J1-24	J1-25	SAFETY HTR PWR	> 100k	OL	P
J1-24	J2-3	SAFETY HTR PWR	> 100k	OL	P
J1-24	J4-12	SAFETY HTR PWR	> 100k	OL	P
J1-24	J4-13	SAFETY HTR PWR	> 100k	OL	P
J1-25	J2-3	SAFETY HTR PWR RTN	> 100k	OL	P
J1-25	J4-12	SAFETY HTR PWR RTN	> 100k	OL	P
J1-25	J4-13	SAFETY HTR PWR RTN	> 100k	OL	P
J2-3	J4-12	SIGNAL RTN	> 100k	2.4 MΩ	P
J2-3	J4-13	SIGNAL RTN	> 100k	2.3 MΩ	P
J4-12	J4-13	+10 V INTERFACE BUS	> 100k	4.7 MΩ	P

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OL = ≥ 40 MEG Ω

TEST DATA SHEET 1 (Sheet 8 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J2-2	J4-13	DATA CLOCK (C1)	> 2k	10.9 MΩ	P
J2-5	J4-13	DIGITAL-A DATA OUT	> 2k	55.5 MΩ	P
J2-6	J4-13	DATA ENABLE (A1)	> 2k	12.8 MΩ	P
J2-7	J4-13	8 SEC SYNC PULSE	> 2k	14.3 MΩ	P
J3-1	J4-13	1.248 MHZ CLK	> 2k	13.2 MΩ	P
J3-2	J4-13	1.248 MHZ CLK RTN	> 2k	OL	P
J4-2	J4-13	MODULE PWR DISCONN	> 2k	13.1 MΩ	P
J4-3	J4-13	SURVIVAL HTR ON	> 2k	14.7 MΩ	P
J4-4	J4-13	MODULE TOTALLY OFF	> 2k	14.7 MΩ	P
J4-5	J4-13	SCANNER A1-2 ON/OFF	> 2k	17.7 MΩ	P
J4-6	J4-13	ANT AT COLD CAL POS	> 2k	19.1 MΩ	P
J4-7	J4-13	PLL SELECT	> 2k	19.8 MΩ	P
J4-8	J4-13	ANT AT NADIR POS	> 2k	21.1 MΩ	P
J4-9	J4-13	COLD CAL POS MSB (IN)	> 2k	22.2 MΩ	P
J4-14	J4-13	MODULE PWR CONN	> 2k	21.4 MΩ	P
J4-15	J4-13	SURVIVAL HTR OFF	> 2k	22.2 MΩ	P
J4-16	J4-13	SCANNER A1-1 ON/OFF	> 2k	21.7 MΩ	P
J4-17	J4-13	ANT AT WARM CAL POS	> 2k	22.8 MΩ	P
J4-18	J4-13	FULL SCAN	> 2k	23.7 MΩ	P
J4-19	J4-13	COLD CAL POS LSB (IN)	> 2k	24.5 MΩ	P
J5-2	J4-13	MODULE PWR IND	> 2k	37.3 MΩ	P
J5-3	J4-13	COLD CAL POS MSB (OUT)	> 2k	38.0 MΩ	P
J5-5	J4-13	SCANNER A1-2 ON/OFF	> 2k	37.7 MΩ	P
J5-6	J4-13	ANT IN COLD CAL POS	> 2k	37.5 MΩ	P
J5-7	J4-13	PLL PRI/RED	> 2k	38.0 MΩ	P
J5-9	J4-13	SURV HTR ON/OFF	> 2k	37.5 MΩ	P
J5-11	J4-13	COLD CAL POS LSB (OUT)	> 2k	37.1 MΩ	P
J5-12	J4-13	SCANNER A1-1 ON/OFF	> 2k	37.6 MΩ	P
J5-13	J4-13	ANT IN WARM CAL POS	> 2k	37.1 MΩ	P
J5-14	J4-13	ANT IN NADIR POS	> 2k	36.8 MΩ	P
J5-15	J4-13	FULL SCAN MODE	> 2k	37.4 MΩ	P

OL = ≥ 40 MEG Ω

TEST DATA SHEET 1 (Sheet 9 of 9)
Grounding Interface Test (Paragraph 3.2.4.1)

Source Pin	Destination Pin	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J6-2	J1-10	RF SHELF A1-1 TEMP	> 2k	14.7 KΩ	P
J6-3	J1-10	A1-1 SCAN MTR.TEMP	> 2k	14.7 KΩ	P
J6-4	J1-10	WARM LOAD A1-1 TEMP	> 2k	14.6 KΩ	P
J6-6	J4-13	PLLO RED LOCK DETECT	> 2k	2.5 MΩ	P
J6-8	J4-13	A1-1 DRIVE MTR CVR	> 2k	3.5 MΩ	P
J6-9	J4-13	+15 VDC ANT DRIVE MON	> 2k	2.9 MΩ	P
J6-10	J4-13	+5 VDC ANT DRIVE MON	> 2k	2.8 MΩ	P
J6-11	J4-13	+15 VDC SIG PROC MON	> 2k	2.5 MΩ	P
J6-12	J4-13	+5VDC SIG PROC MON	> 2k	2.5 MΩ	P
J6-13	J4-13	L.O. VOLTAGE CH3 MON	> 2k	2.5 MΩ	P
J6-14	J4-13	L.O. VOLTAGE CH5 MON	> 2k	2.5 MΩ	P
J6-15	J4-13	L.O. VOLTAGE CH7 MON	> 2k	2.5 MΩ	P
J6-16	J4-13	+15 VDC PLL LO MON	> 2k	2.5 MΩ	P
J6-17	J4-13	+10 V MIXER/AMP MON	> 2k	2.5 MΩ	P
J6-18	J4-13	L.O. VOLTAGE CH15 MON	> 2k	2.5 MΩ	P
J6-21	J4-10	RF SHELF A1-2 TEMP	> 2k	OL	P
J6-22	J4-10	A1-2 SCAN MTR.TEMP	> 2k	OL	P
J6-23	J4-10	WARM LOAD A1-2 TEMP	> 2k	OL	P
J6-25	J4-13	PLLO PRI LOCK DETECT	> 2k	2.5 MΩ	P
J6-27	J4-13	A1-2 DRIVE MTR CURR	> 2k	3.5 MΩ	P
J6-28	J4-13	-15 VDC ANT DRIVE MON	> 2k	2.8 MΩ	P
J6-29	J4-13	-15 VDC SIG PROC MON	> 2k	2.5 MΩ	P
J6-30	J4-13	L.O. VOLTAGE CH4 MON	> 2k	2.5 MΩ	P
J6-31	J4-13	L.O. VOLTAGE CH6 MON	> 2k	2.5 MΩ	P
J6-32	J4-13	L.O. VOLTAGE CH8 MON	> 2k	2.5 MΩ	P
J6-33	J4-13	-15 VDC PLL LO MON	> 2k	2.5 MΩ	P
J6-34	J4-13	IF AMP MON	> 2k	2.5 MΩ	P

OL = $\geq 40 \text{ MEG } \Omega$

Circle Test: **FINAL**
CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford 7-13-00
Customer Representative (Flight Hardware Only) Date

Ray Hubbard 7-14-00
Test Systems Engineer Date
Chad Morgan 7-14-00
Quality Control Date

TEST DATA SHEET 2
+28 MLB During Turn-on Transient (Paragraph 3.2.4.2.1.1)

At 28.56 Vdc:

Required*					
Step	Parameter	Measured/ Calculated	S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	230.5 ms	20 ms max	300 ms max	P
8	Peak Current	4.75 Amps	10.6 Amps	5.9 Amps	P
10	Rate of Change (Slope): dI/dT	81.4 mA/μs	677 mA/μs	250 mA/μs	P

At 27.44 Vdc:

Required*					
Step	Parameter	Measured/ Calculated	S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	245.4 ms	20 ms max	300 ms max	P
8	Peak Current	4.65 Amps	10.6 Amps	5.9 Amps	P
10	Rate of Change (Slope): dI/dT	116.2 mA/μs	677 mA/μs	250 mA/μs	P

At 28.00 Vdc:

Required*					
Step	Parameter	Measured/ Calculated	S/N 101-104	S/N 105 & up	Pass/ Fail
7	Time to reach steady state current	229.1 ms	20 ms max	300 ms max	P
8	Peak Current	4.69 Amps	10.6 Amps	5.9 Amps	P
10	Rate of Change (Slope): dI/dT	136.6 mA/μs	677 mA/μs	250 mA/μs	P

* Refer to Figure 5.

Circle Test: **FINAL** CPT LPT

OP : 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N 10109

Customer Representative
(Flight Hardware Only)

7-15-00

Date

Test Systems Engineer

Quality Control

7-14-00

7A 194 7/14/00

Date

$$\Delta Y = 47.47 \text{ mV}$$

10. DE >
/ 0 i /

$$\text{PEAK CURRENT} = \frac{1.0 \text{ A}}{100 \text{ V}} \times 47,470 \text{ V} = 4747 \text{ mA}$$
$$= 4.75 \text{ Amperes}$$
$$\text{STEADY STATE TIME} = \underline{2300 \text{ s}} \cdot \text{sec}$$

100

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1.04/10.22

□₂

0.0.0.YXRXE

MCB
TUB-V-ON

Sec + 28.56 V

300m

S/O: 787920 CP: 0830

P/N: 1331720-3-75T SN: 109

3.2.4.2.1.1

FINAL CPT

T25-2

TEST ENG:

DA 17V.

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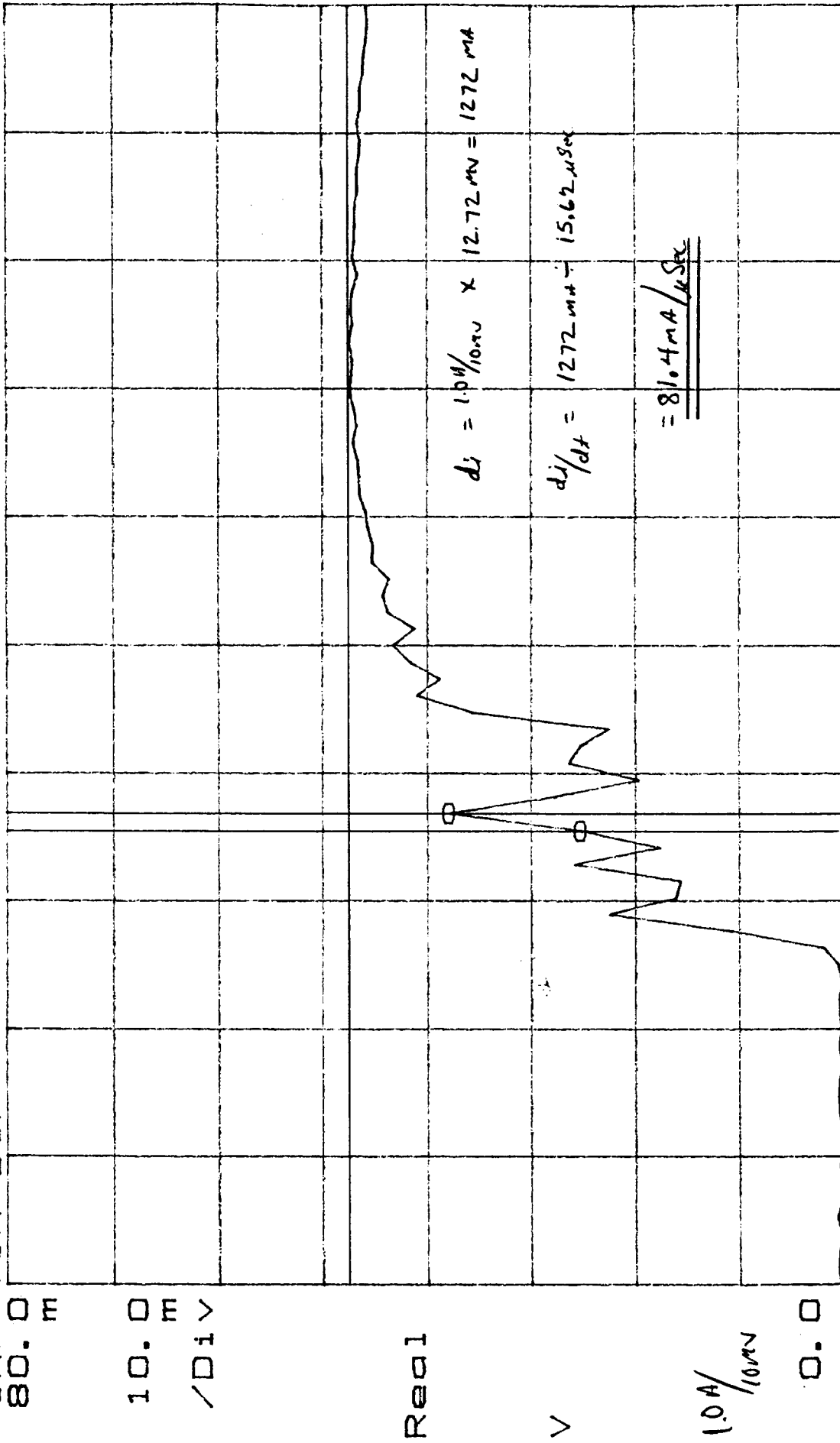
DATE: 7-14-00

7A
194

7/14/00

$X=8.609\text{ms}$ $\Delta X=15.62\mu\text{s}$ $Y=0.0$ $\Delta Y=47.47\text{mV}$
 $Y_0=25.1818\text{m}$ $\Delta Y_0=12.72\text{mV}$

CAP TIM BUF



FXdXY 8.19m Sec + 28.56 9.37m

S/O: 787920 CP: 0830 TDS-2 TEST ENG: DATE: 7-14-00
 PN: 1331720-3-TST SN: 109 QUALITY: (7A 7/4/00) (794)

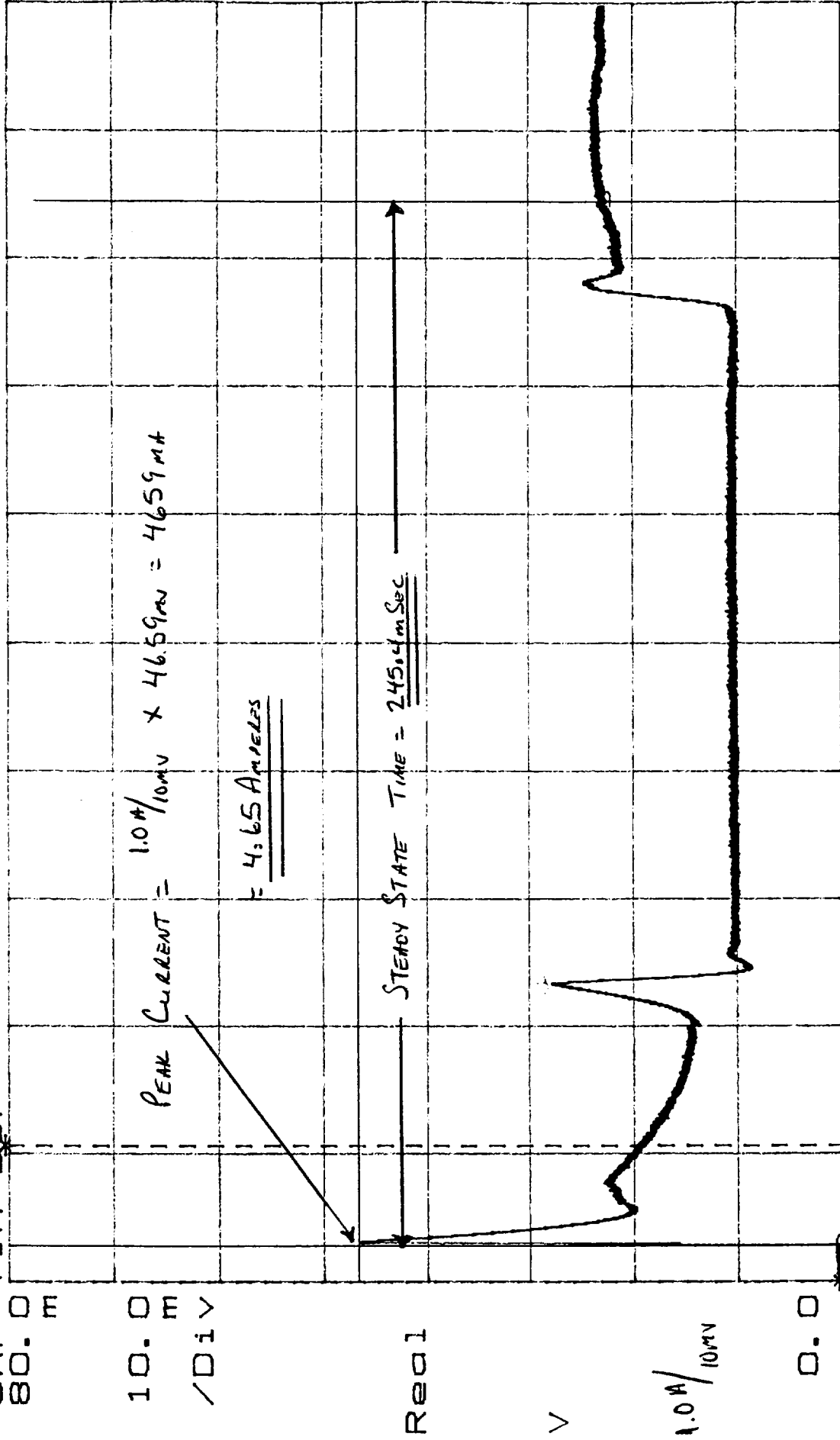
X=8.297ms
Y=0.0

ΔX=245.4ms
ΔY=22.46mV

Y=0.0

ΔY=46.59mV

CAP TIM BUF



FxdXY 0.0

MLB TURN-ON

Sec +27.44 V

300m

TDS-2

S/O: 787920 OP: 0830

3242.11.

TEST ENG:

DATE: 7-14-00

P/N: 1331720-3-TST SN: 109

FINAL CPT

QUALITY:

7-14-00

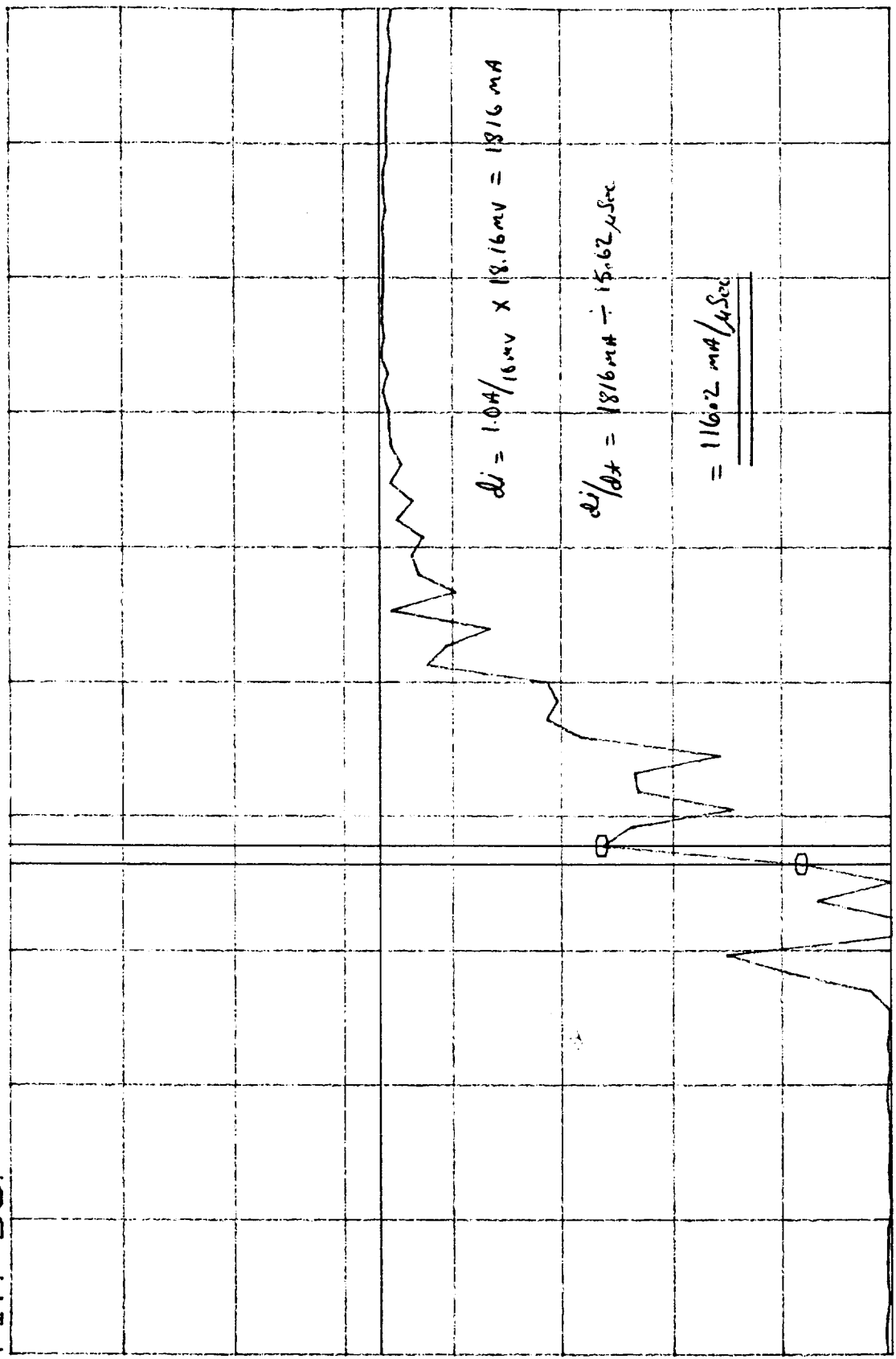
X=8.906ms
Y=8.1033m

$\Delta X = 15.62 \mu s$
 $\Delta Y = 18.16 mV$

Y=0.0

$\Delta Y = 46.59 mV$

CAP TIM BUF



80.0 m

10.0 m

/Div

Real

V

1.0A / 10mV

0.0

FxdXY 8.48m

ms di/dt

Sec + 27.44 V

9.64m

S/O: 787920 CP: 0830

P/N: 1331720-3-TST SN: 109

3242.11

FINAL CPT

TDS-2

TEST ENG:

QUALITY:

139
T

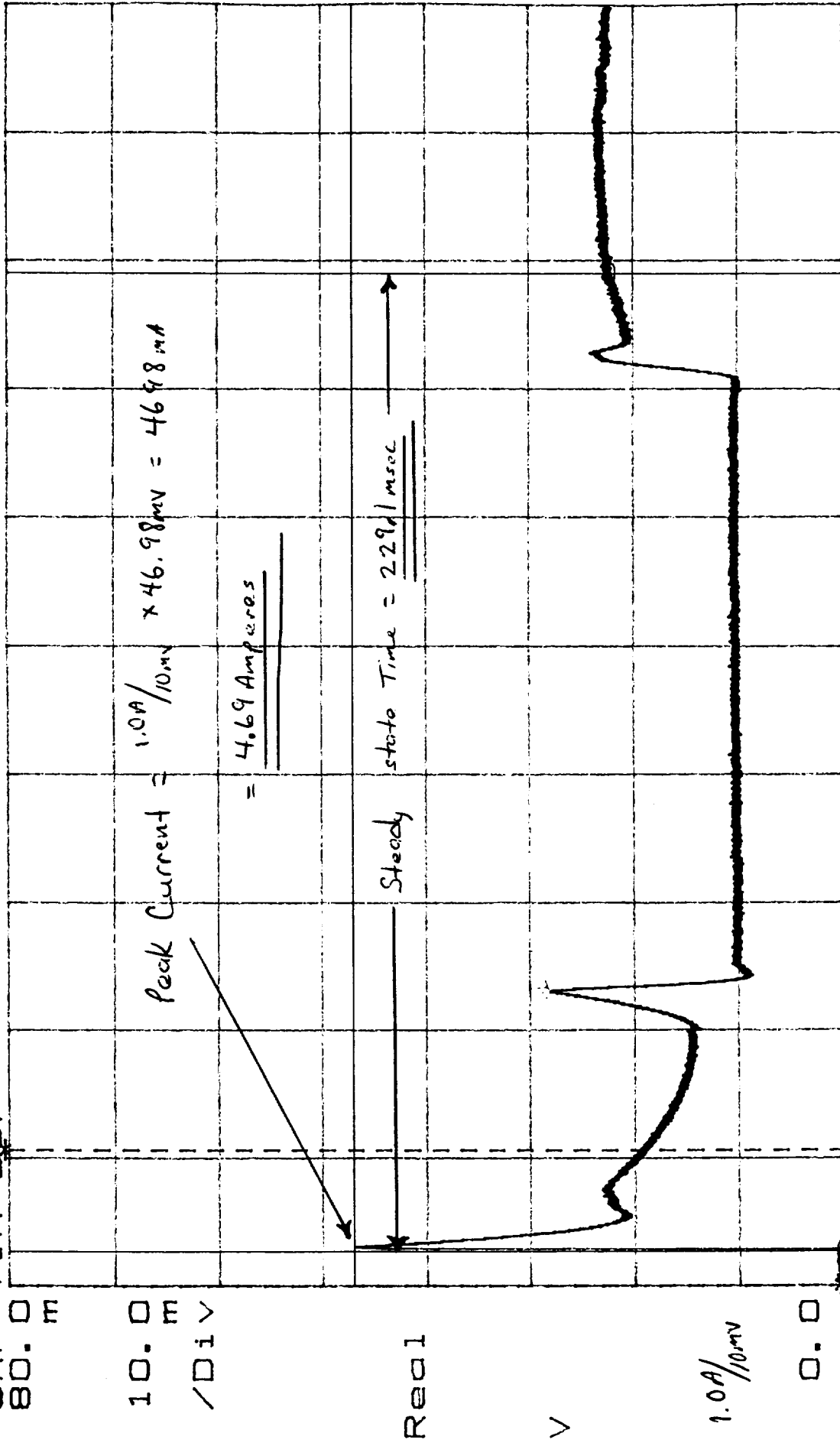
DATE: 7-14-00

74
194

7-14-00

$X=8.406\text{ms}$ $\Delta X=229.1\text{ms}$ $Y=47.0303\text{m}$ $\Delta Y=46.98\text{mV}$
 $Y_0=102.573\mu$ $\Delta Y_0=22.05\text{mV}$

CAP TIM BUF



FXDXY 0.0 MLB TURN-ON Sec +28.0V

S/O: 787920 OP: 0830 TDS-2 TEST ENG: DATE: 7-14-00
 P/N: 1331720-3-TST SN: 109 QUALITY: 7-14-00
 (139 / T) (7A / 194)

X=8.734mS ΔX=15.62μS Y=47.0303m ΔY=46.98mV

CAP TIM BUF

80.0 M

10.0 M

/Div

Real

V

1.0V/10mV

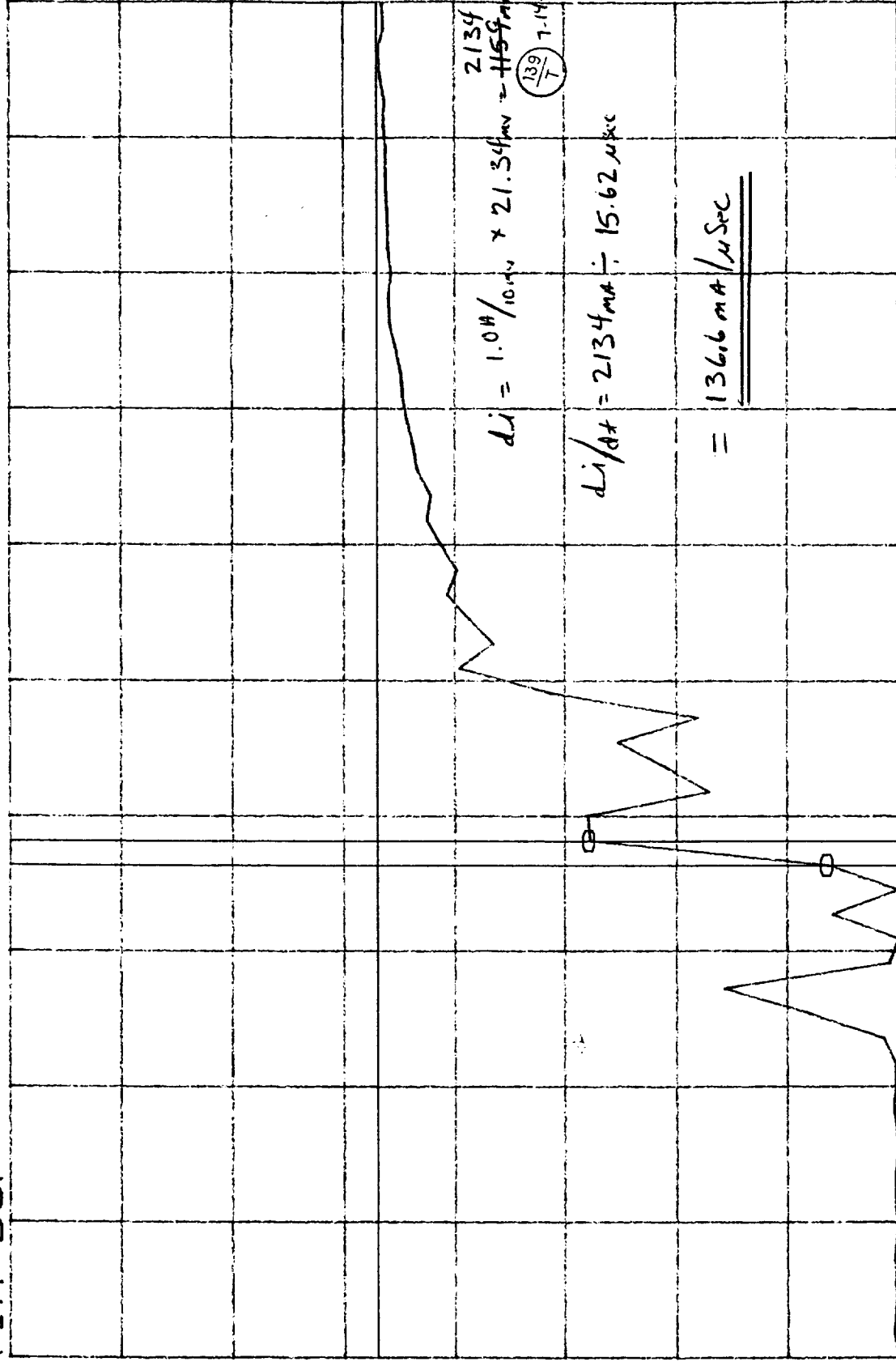
0.0

FxdXY 8.41m

mb di/dt

Sec +28.0V

9.27m



(139/T)

TDS-2

3.2.4.2.1.1

FINAL CPT

S/O: 787920 OP: 0830

PN: 1331720-3-TST SN: 109

TEST ENG:

DATE: 7-14-00

QUALITY:

(184) 7-14-00

TEST DATA SHEET 3
+28 MLB Operating Power (Paragraph 3.2.4.2.1.2)

Step	+28V MLB at 27 Volts	Measured	Units	Required	Pass/Fail
2	+28 V MLB voltage at 27 V (V_b) (Measured)	27.03 V	Volts	27.0 \pm 0.1	P
3	Average Current (I_V) (PLLO#1)	2.34 A	Amps	N/A	N/A
4	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)	63.2 W	Watts	82 W max	P
6	Average current (I_V) (PLLO#2)	2.48 A	Amps	N/A	N/A
7	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)	67.03 W	Watts	82 W max	P
+28 V MLB at 28 Volts					
9	+28 V MLB bus voltage at 28 V (V_b) (Measured)	28.00 V	Volts	28.0 \pm 0.1	P
10	Average Current (I_V) (PLLO#1)	2.26 A	Amps	N/A	N/A
11	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)	63.3 W	Watts	82 W max	P
13	Average current (I_V) (PLLO#2)	2.37 A	Amps	N/A	N/A
14	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)	66.36 W	Watts	82 W max	P
+28 V MLB at 29 Volts					
16	+28 V MLB voltage at 29 V (V_b) (Measured)	29.00 V	Volts	29.0 \pm 0.1	P
17	Average Current (I_V) (PLLO#1)	2.20 A	Amps	N/A	N/A
18	+28 V MLB operating power = $I_V \times V_b$ (PLLO#1)	63.8 W	Watts	82 W max	P
20	Average current (I_V) (PLLO#2)	2.30 A	Amps	N/A	N/A
21	+28 V MLB operating power = $I_V \times V_b$ (PLLO#2)	66.7 W	Watts	82 W max	P

Circle Test: **FINAL** CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

OP: 0830

Test Systems Engineer

Quality Control

Customer Representative
(Flight Hardware Only)

Date

Date

Date

6 Apr 99

TEST DATA SHEET 4 (Sheet 1 of 2)
+28 Pulse Load Bus (Paragraph 3.2.4.2.2.1-3.2.4.2.2.6)

Paragraph	Parameter	Measured or Calculated	Required	Pass/ Fail
3.2.4.2.2.1	From -0.1 to two seconds			
	Peak Current = I_p	<u>1.05</u> Amps	1.3 amps max	<i>P</i>
3.2.4.2.2.2	From 2 to 4 seconds			
	Peak Current = I_p	<u>1.05</u> Amps	1.3 amps max	<i>P</i>
3.2.4.2.2.3	From 4 to 6 seconds			
	Peak Current = I_p	<u>1.02</u> Amps	1.3 amps max	<i>P</i>
3.2.4.2.2.4	From 6 to 8 seconds			
	Peak Current = I_p	<u>1.04</u> Amps	1.3 amps max	<i>P</i>
3.2.4.2.2.5	Eight Sec. Integrated Current Measurement:			
	Current	<u>111.4</u> mA	None	<i>N/A</i>
3.2.4.2.2.6	Turn-on Transient:			
	dI/dT	<u>285.6</u> mA/ μ s	744 mA/ μ s *	<i>P</i>
	Peak Current = I_p	<u>6.75</u> Amps	11.5 Amps	

* Refer to Figure 9.

Bus current during the I/H, D period

Paragraph	Parameter	Measured or Calculated	Pass/ Fail
3.2.4.2.2.1	From -0.1 to 2 secs	<u>25.2</u> mA	N/A
3.2.4.2.2.2	From 2 to 4 secs	<u>19.4</u> mA	N/A
3.2.4.2.2.3	From 4 to 6 secs	<u>21.34</u> mA	N/A
3.2.4.2.2.5	From 6 to 8 secs	<u>16.48</u> mA	N/A

Circle Test: CPT LPTFINAL

METSAT/AMSU-AI System P/N IS-1331720

Shop Order: 787920S/N: 109

[Signature]
 Customer Representative
 (Flight Hardware Only)

7-15-00
 Date

[Signature]
 Test Systems Engineer

7-14-00
 Date

[Signature]
 Quality Control

7-15-00
 Date

TEST DATA SHEET 4 (Sheet 2 of 2)
+28 Pulse Load Bus (Paragraph 3.2.4.2.2.7)

Bus current during warm cal, cold cal, & Nadir

Paragraph	Parameter	Measured or Calculated	Pass/ Fail
3.2.4.2.2.7 (2)	Warm cal	12.1 mA	N/A
3.2.4.2.2.7 (3)	Cold cal	13.9 mA	N/A
3.2.4.2.2.7 (4)	Nadir	19.3 mA	N/A
3.2.4.2.2.7 (5)	Warm cal (motors off)	.002 mA	N/A

Circle Test:



LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order:

0P: 0830

707920

S/N:

109

J. Sanford
Customer Representative
(Flight Hardware Only)



7-15-00
Date

Test Systems/Engineer

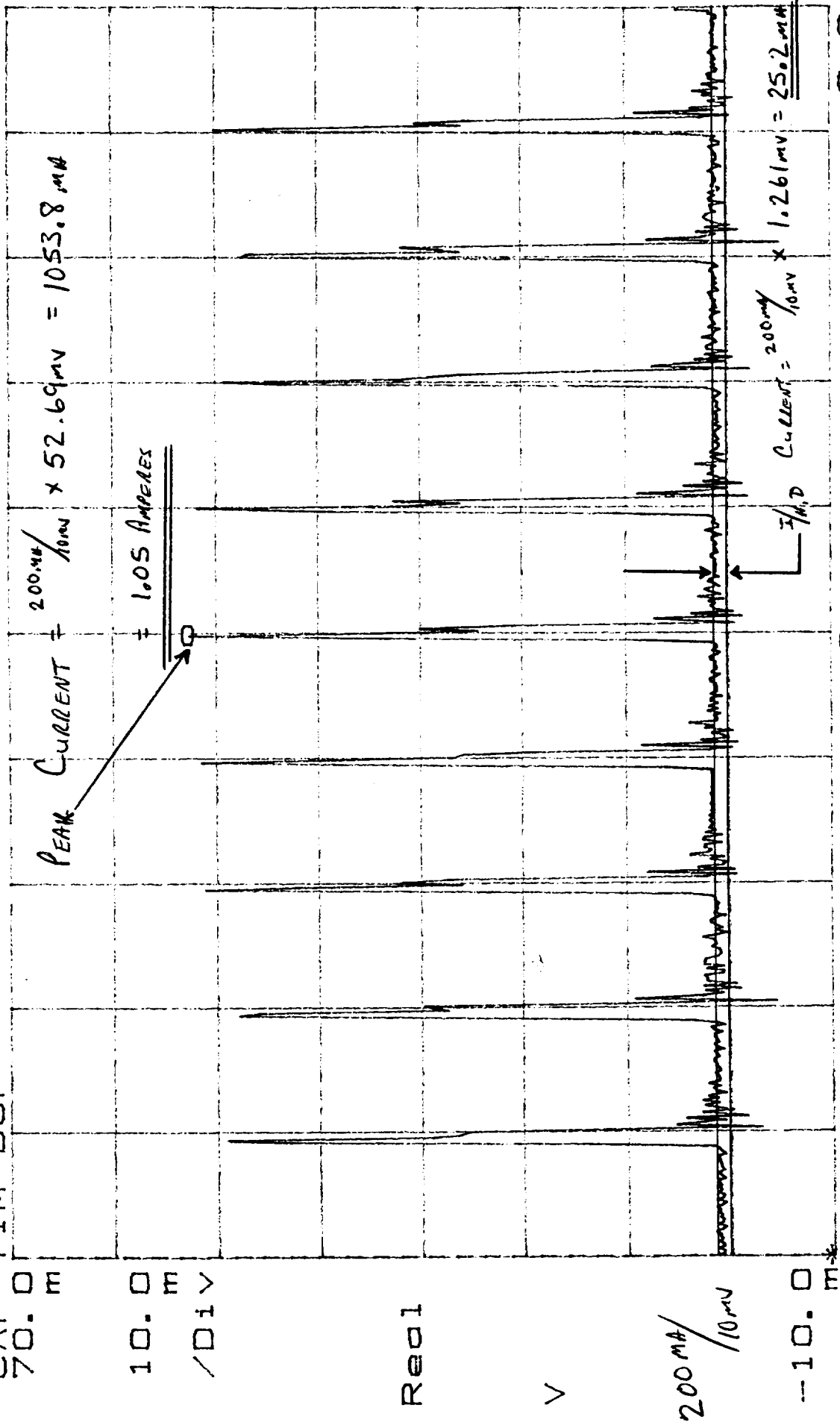
Date

Quality Control

Date

X=996. 1mSec
Y=52. 6957mV
CAP TIM BUF
70.0 m

Y=181. 817 μ $\Delta Y=1. 261mV$



0-2 sec TDS -4

S/O: 787920 CP: 0830
PN: 1331720-3-TST SN: 109
TEST ENG: QUALITY CONTROL
DATE: 7/14/00
139
T
7/15/00

X=3.9586 Sec
Y=52.612mV

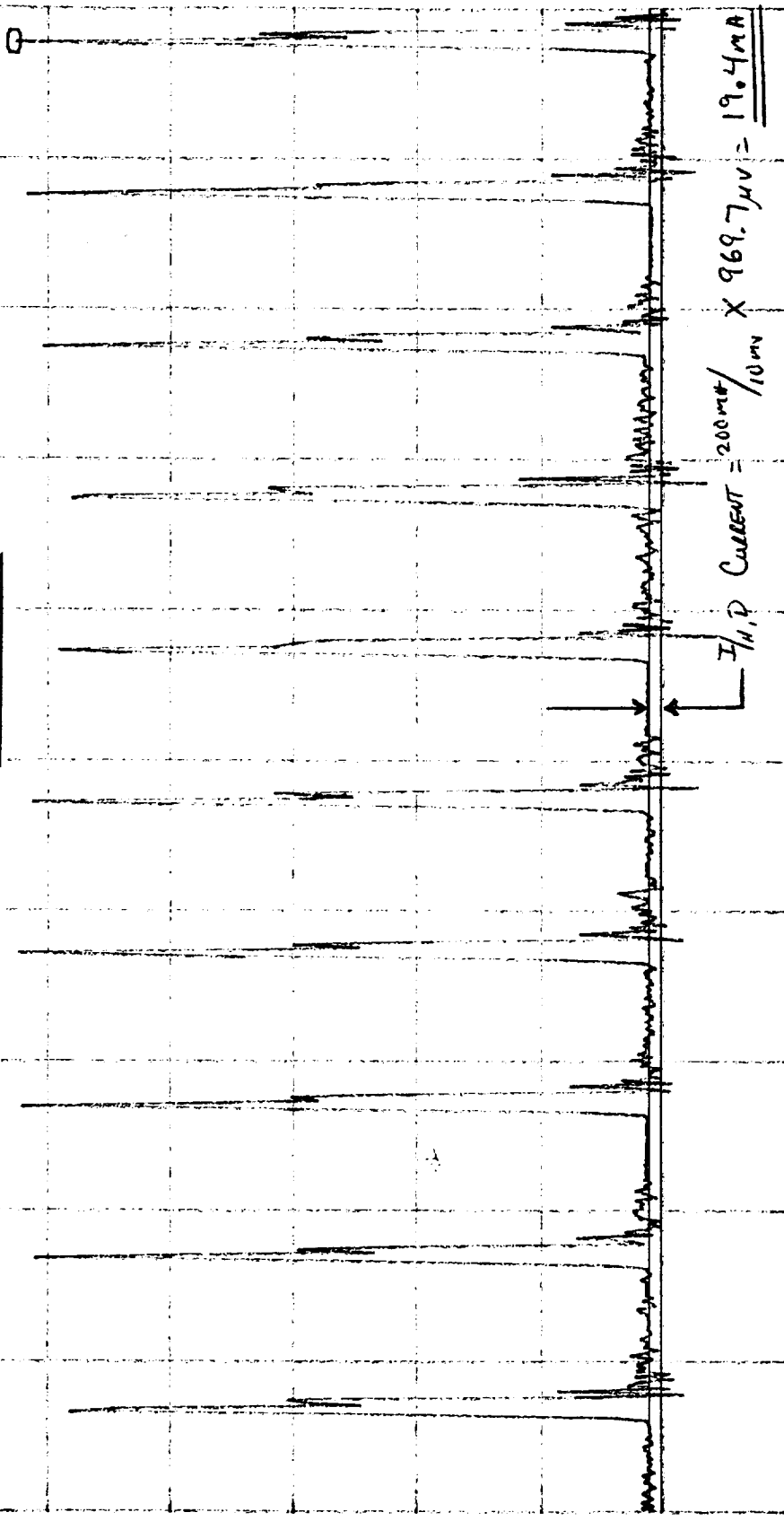
Y=181.817μ ΔY=969.7μV

CAP TIM REC
70.0 M

10.0 M
/DIV

PEAK CURRENT = $\frac{200mA}{10mV} \times 52.612mV = 1052.2 mA$

= 1.05 AMPERES



200mA/10mV

-10.0 M

EXDXY 2.0

2-4 sec

Sec PLB CURRENT

4.0

TDS 4

$\frac{13.9}{T}$

TEST END:

DATE: 7/14/00

3.2.4.2.2.2
FINAL (PT)

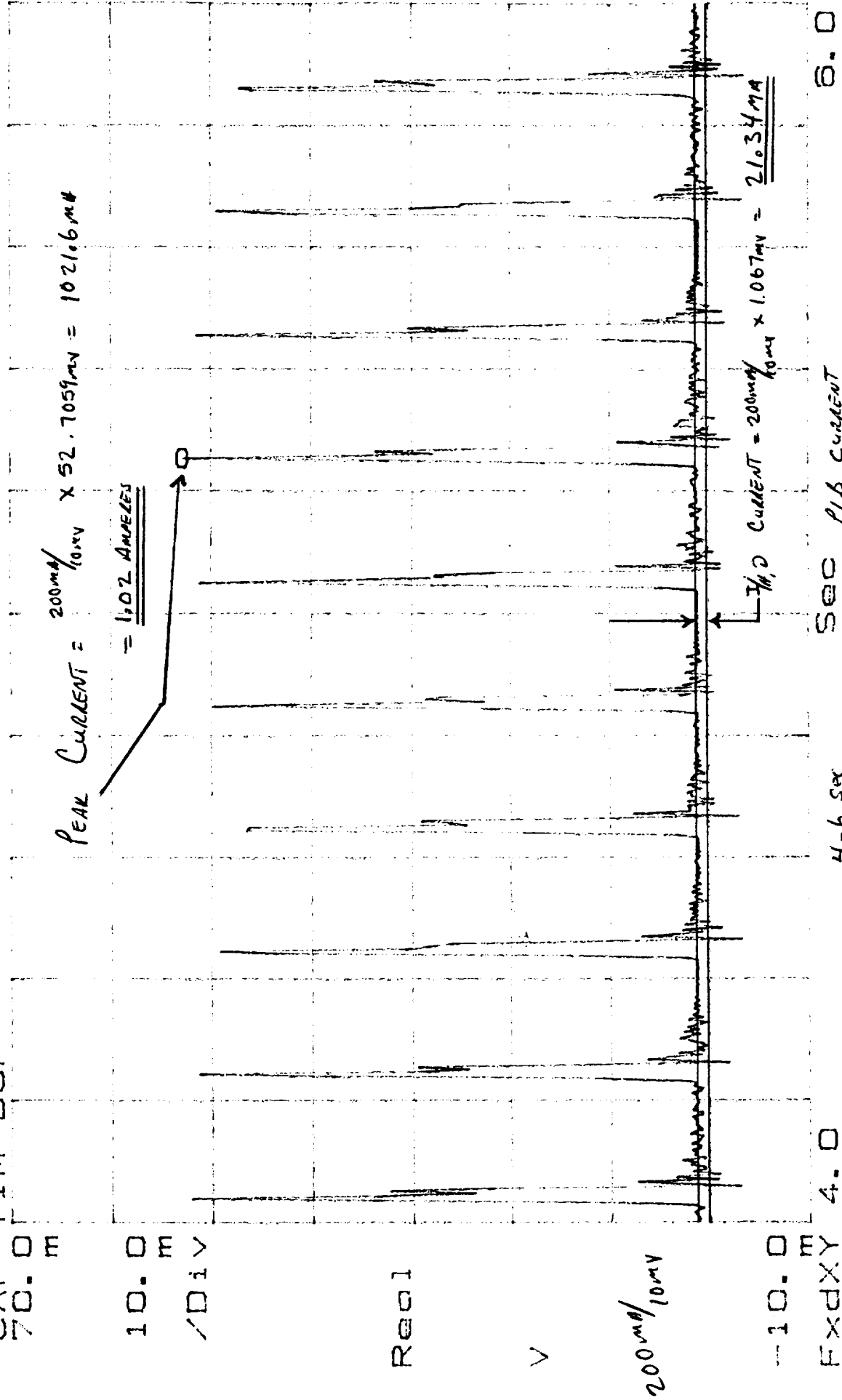
SN: 109

S/O: 787920 CP: 0830

DIAGNOSTIC MONITOR (200 7/15/00)

X=5.2555 Sec
Y=52.7059mV
CAP TIM BUF
70.0 m

Y=181.817 μ $\Delta Y=1.067mV$



S/O: 787920 CP: 0830
PAN: 1331720-3-TST SN: 109
TEST ENG: DATE: 7/14/00
DIA IT Audia Donnell 297/15/

X=6.6812 Sec
Y=52.1649mV

CAP TIM BUF
70.01

Y=181.817μ ΔY=824.2μV

PEAK CURRENT = $\frac{200mV}{10mV} \times 52.1649mV \sim 1043.3mA$

= 1.04 AMPERES

10.0 m
/DIV

Real

200mV/10mV

--10.0 m

FxdXY 6.0

6-8 sec

Sec

PLB CURRENT

8.0

$I_{H,D} \text{ CURRENT} = \frac{200mV}{10mV} \times 824.2\mu V = 16.48mA$

705-4

$\frac{139}{T}$

TEST ENDS:

DATE: 7/14/00

3.2.4.2.2.4

FINAL - PT

SN: 109

PN: 13 720-3-TST

QUALITY ~~Audio~~ Jerry 6.8.7/15/00

S/O: 787920 CP: 0830

X=7.9961 Sec
Y=44.5912mV

M:CAP TIM REC

70.0 m

10.0 m

/Div

Real

V

200mV/10mV

--10.0 m

FxdY 0.0

8sec Ave CURRENT

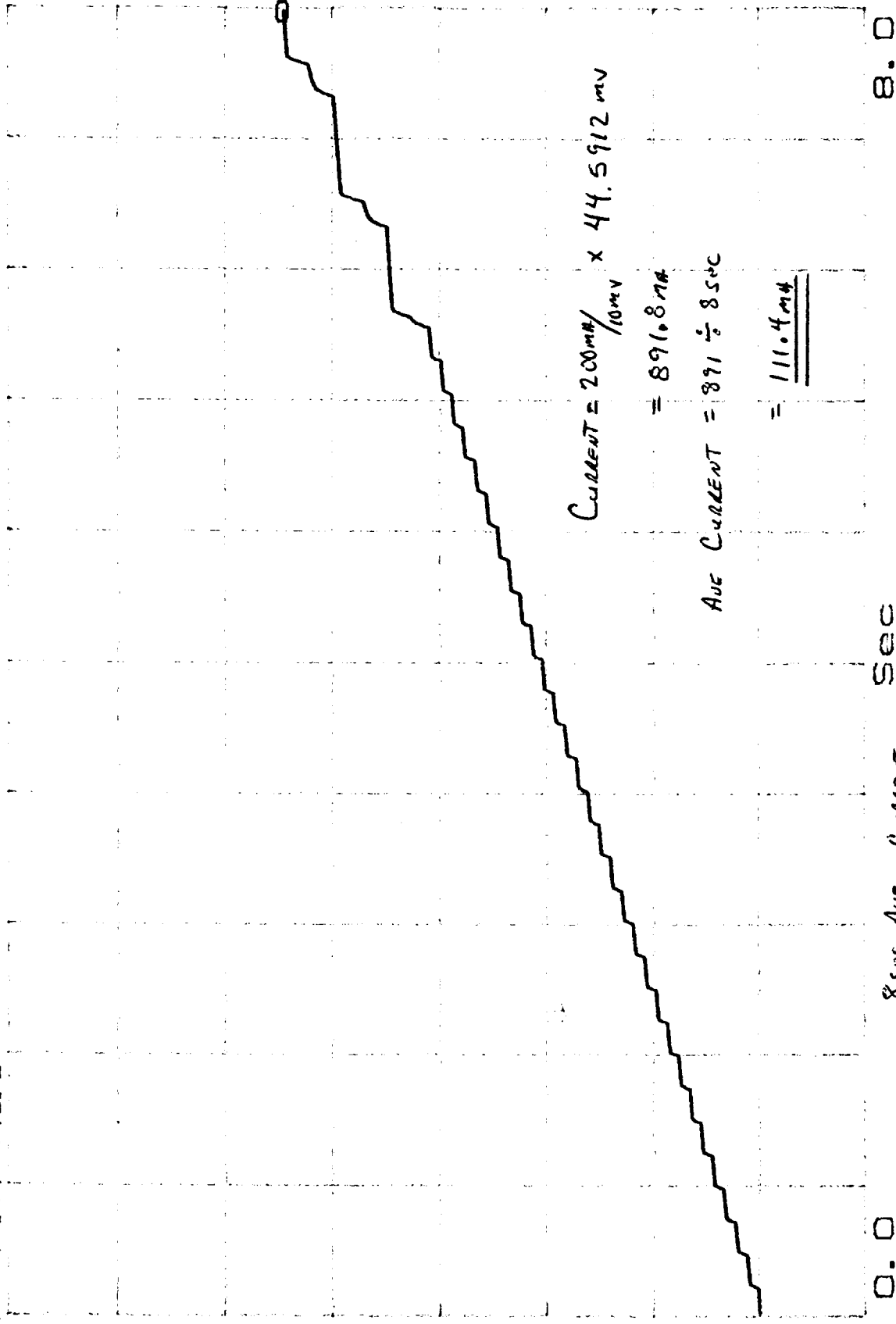
Sec

TDS-4

S/O: 787920 CP: 0830

PN: 13720-3-TST SN: 109

3.2.4.2.2.5
FINAL PT



$$\text{CURRENT} = 200 \text{ mV} / 10 \text{ mV} \times 44.5912 \text{ mV}$$

$$= 891.8 \text{ mV}$$

$$\text{Ave CURRENT} = 891 \div 8 \text{ sec}$$

$$= 111.4 \text{ mV}$$

139

TEST ENG:

DATE: 7/14/00
QUALITY CONTROL

X=20.31ms ΔX=11.84ms Y=336.764m ΔY=337.7mV
 Y0=43.3373m ΔY0=43.18mV

CAP TIM BUF
 470m

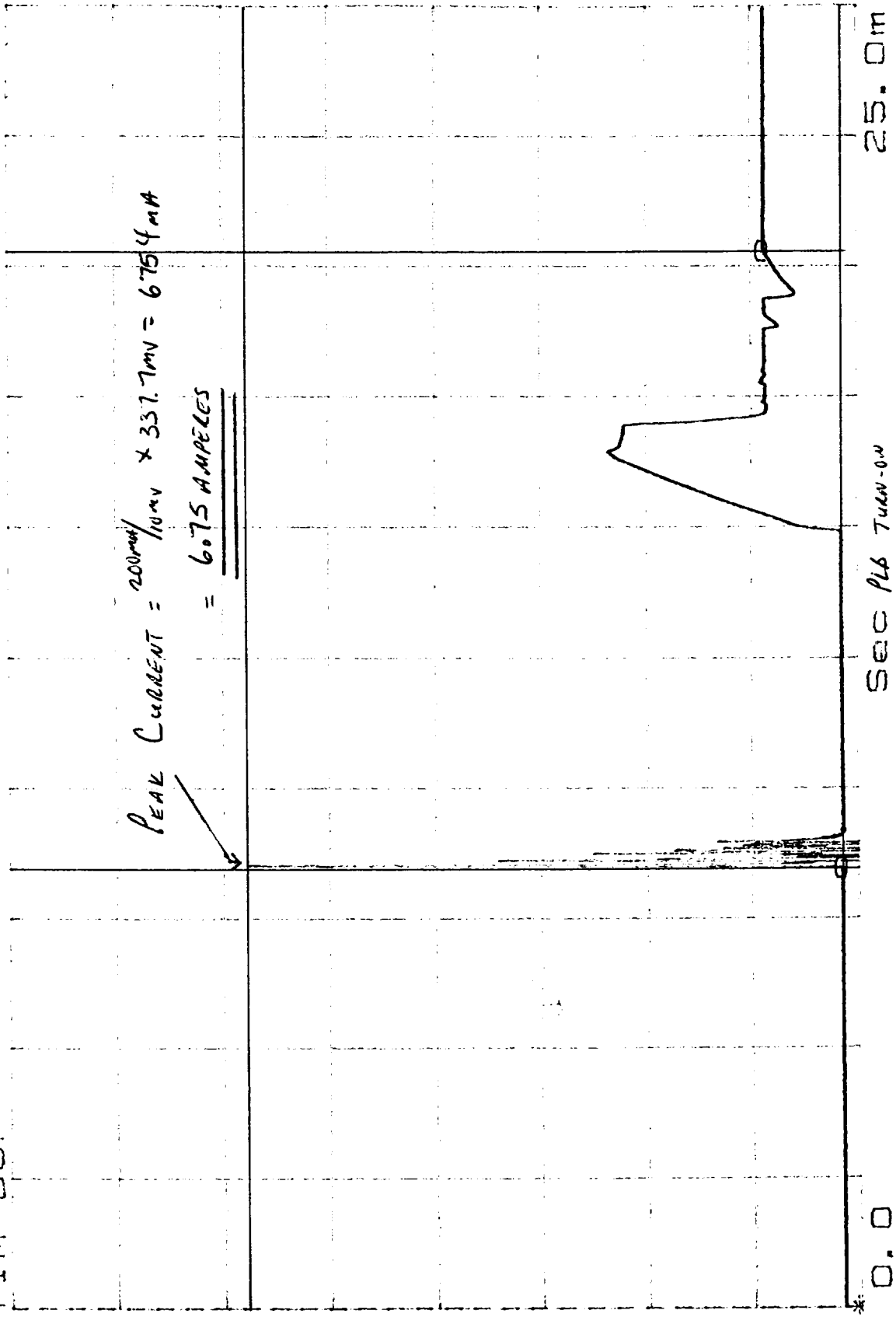
60.0m
 /Div

Real

V

200mV/10mV

-10.0m
 EXDXY 0.0



139

TEST ENG:

DATE: 7/14/00
 QUALITY: 7/15/00

TDS-4

3.2.4.2.2.6

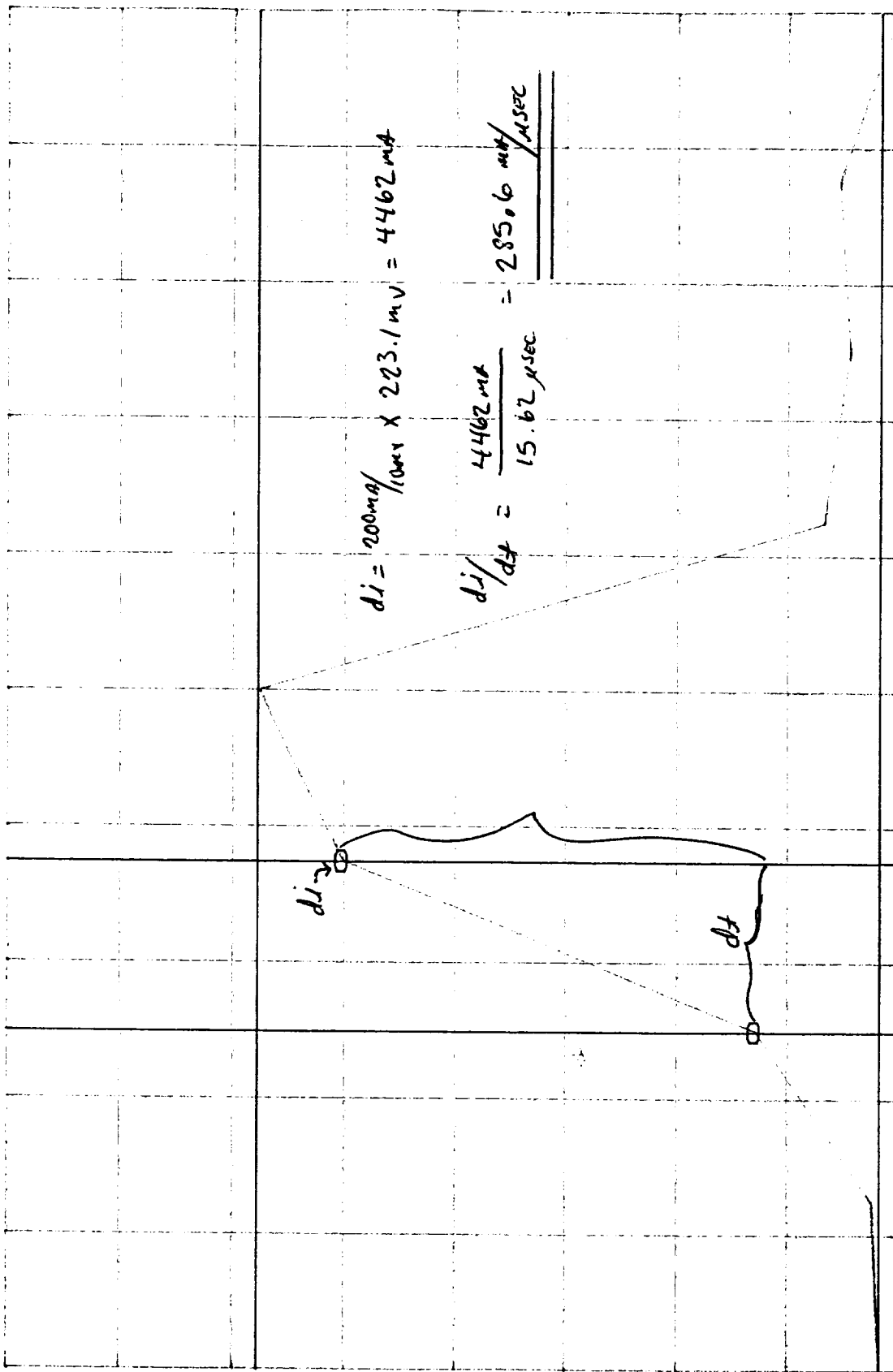
FINAL CPT

S/O: 787920 OP: 0830

PAN: 1331720-3-TST SN: 109

X=8.516ms ΔX=15.62μs Y=336.764m ΔY=337.7mV
 Y0=290.95m ΔY0=223.1mV

CAP TIM BUF



$$di = 200 \text{ mA} / 10 \text{ V} \times 223.1 \text{ mV} = 4462 \text{ mA}$$

$$di/dt = \frac{4462 \text{ mA}}{15.62 \text{ μsec}} = 285.6 \text{ mA/μsec}$$

200mA/10V

-10.0m

FXdXY 8.47m

PLB di/dt SEC TDS-Y

8.59m

S/O: 787920 OP: 0830 3.2.4.2.2.6 DATE: 7/14/00
 PN: 137720-3-TST SN: 109 [FINAL - PT] QUALITY India Honey 7/15/00
 TEST ENG: 139/T

TEST DATA SHEET 5
+28 V Analog Telemetry Bus (Paragraph 3.2.4.2.3)

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
3	+28 V ATB Bus Voltage (V_{at}) (Measured)	<u>27.96</u> Volts	28.0 \pm 0.5	P
4	Av. Current (I_a)	<u>1.8</u> mA	7 mA max	P
5	+28 V ATB Operating Power = $I_a \times V_{at}$	<u>50.36</u> mW	200 mW max	P

Circle Test: FINAL CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-15-00
Date

Ray Hefberg
Test Systems Engineer

7-15-00
Date

Indira Henry
Quality Control

7-15-00
Date

TEST DATA SHEET 6
+10 V Interface Bus Voltage (Paragraph 3.2.4.2.4)

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
3	Av. Current (I_a)	<u>6.3</u> mA	10 mA max	P
3	+10 V Interface Bus (V_{ib}) (Measured)	<u>8.34</u> Volts	9.0 \pm 1.0 V	P
4	+10 V Interface Bus Power = $I_a \times V_{ib}$	<u>52.5</u> mW	100 mW max	P

Circle Test: FINAL CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)



7-15-00
Date

Ray Herberg
Test Systems Engineer

7-15-00
Date

Judith Hervey
Quality Control

7/15/00
Date

$$Y = 324.847 \mu \quad \Delta Y = 315.2 \mu V$$

CAP TIM BUF

4.0 m

1.0 m
DIV

$$\text{Current} = 200 \text{ mA} / 10 \text{ mV} \times 315.2 \mu V = 6304 \mu A$$

$$= \underline{\underline{6.3 \text{ mA}}}$$

Real

V

200 mA / 10 mV

4.0 m

FIXED 0.0

500

+10V INTERFACE BUS

S/O: 787920 OP: 0830

PAN: 1331720-3-TEST SN: 109

3.2.4.2.4.1

FINAL COPY

TDS-6

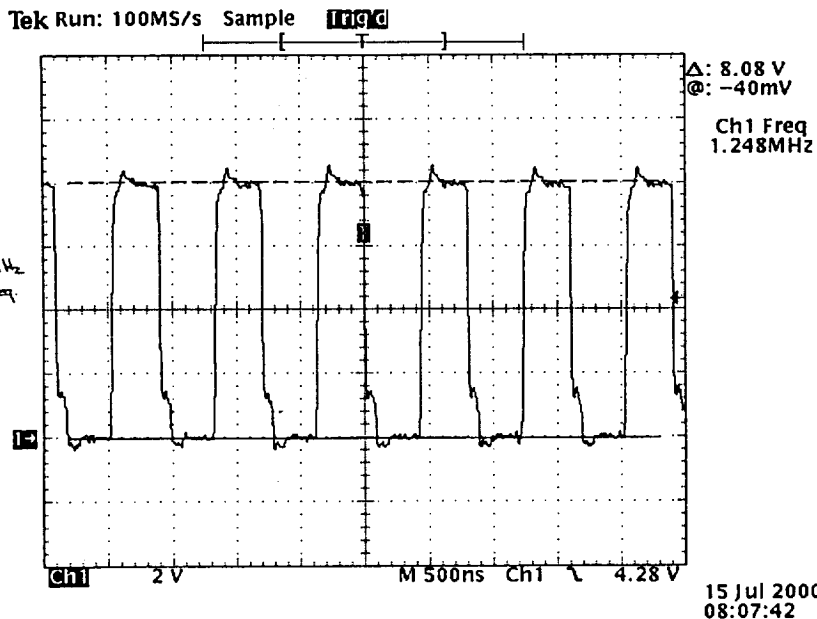
TEST END

DATE: 7-15-00

7/15/00

8.0

TEST DATA SHEET 8 1.248 MHz Clock Signal Verification (Paragraph 3.2.4.3.2.1)



S/O: 787720 OP: 0830 TDS B TEST ENG: *[Signature]* DATE: 7-15-00
P/N: 1331720-3-TST SN: 109 FINAL CPT QUALITY: *[Signature]* DATE: 7-15-00

Step	Parameter	Measured/ Calculated	Required	Pass/ Fail
5	Clock Frequency	1.248 MHz	1.248 \pm 10%	PASS
	Clock Amplitude	8.08 Volts	9.0 \pm 1.0 V	PASS

Circle Test: FINAL CPT LPT

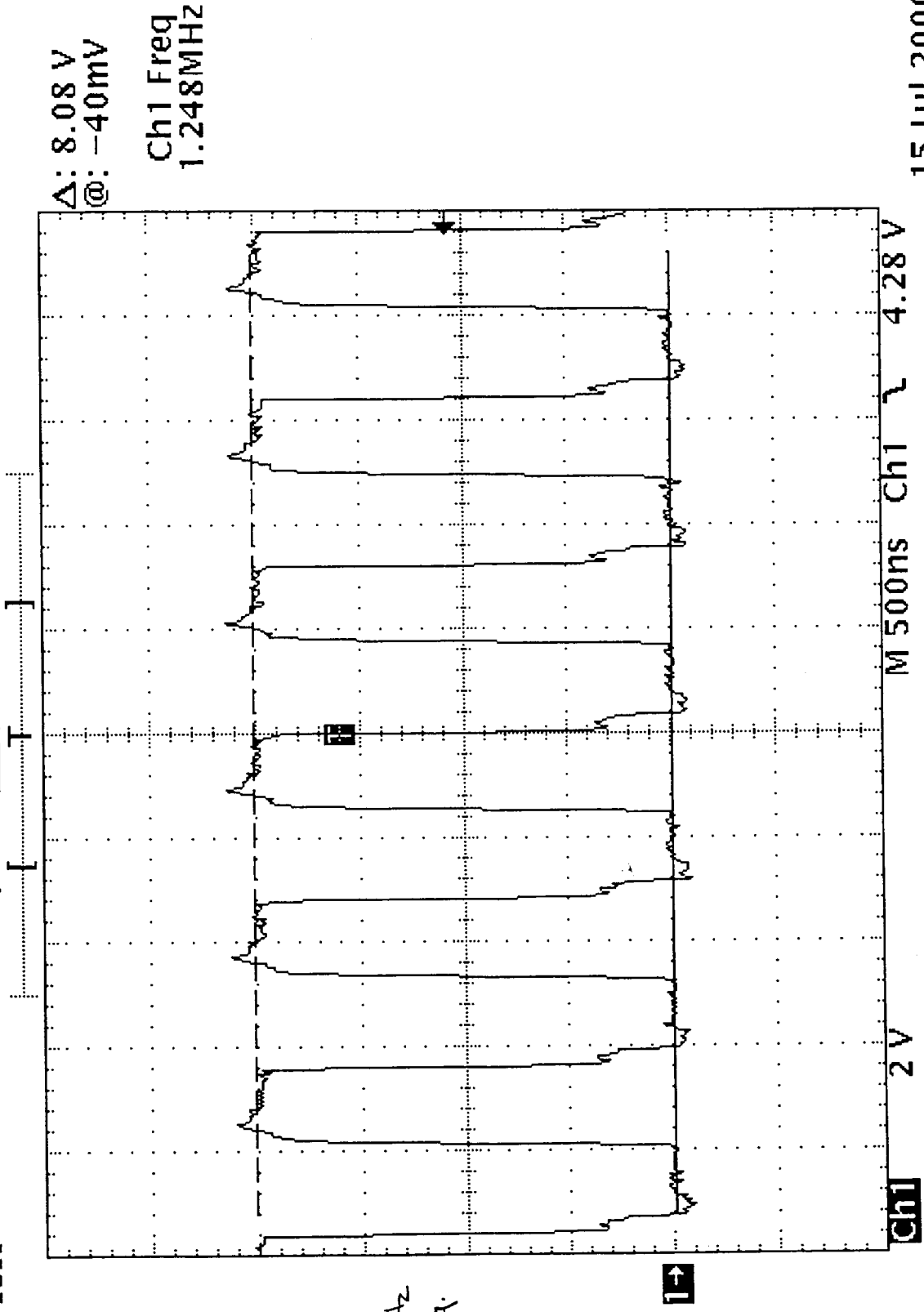
METSAT/AMSU-A1 System P/N IS-1331720

Op 0830
Shop Order: 787920 SN: 109

[Signature] 7-15-00
Customer Representative (Flight Hardware Only) Date

[Signature] 7-15-00
Test Systems Engineer Date
[Signature] 7-15-00
Quality Control Date

Tek Run: 100MS/s Sample Trig'd



1.248MHz
Clock Freq.

15 Jul 2000
08:07:42

3,243,21

TDSB

TEST ENG: Sanjay Kulkarni DATE: 7-15-00
QUALITY: Sanjay Kulkarni 7-15-00

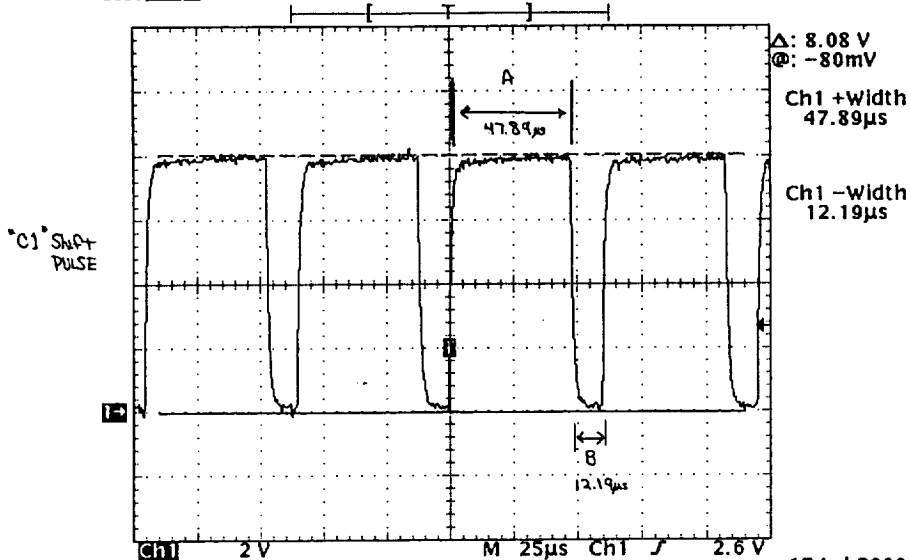
S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

FINAL CPT

TEST DATA SHEET 9

"C1" Shift Pulse Verification (Paraoranh 3 2 4 3 2 2)
Tek Stop 2MS/s 3278 Acqs



S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

3.24322 TDS 9

FINAL CPT

TEST ENG: Ray H. H. DATE: 7-15-00

QUALITY: J. G. H. DATE: 7-15-00 s/ Fail

Pulse Timing (A) *	47.89μs	48 μs ± 10%	PASS
Pulse Timing (B) *	12.19μs	12 μs ± 10%	PASS
Pulse Amplitude	8.08 Volts	9.0 ± 1.0 V	PASS

* Refer to Figure 19 for location of the pulse timing A and B.

Circle Test: **FINAL CPT** LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

Op. 0830

Test Systems Engineer

7-15-00 Date

Customer Representative
(Flight Hardware Only)

7-15-00 Date

Quality Control

7-15-00 Date

3278 Acqs

3278 Acqs

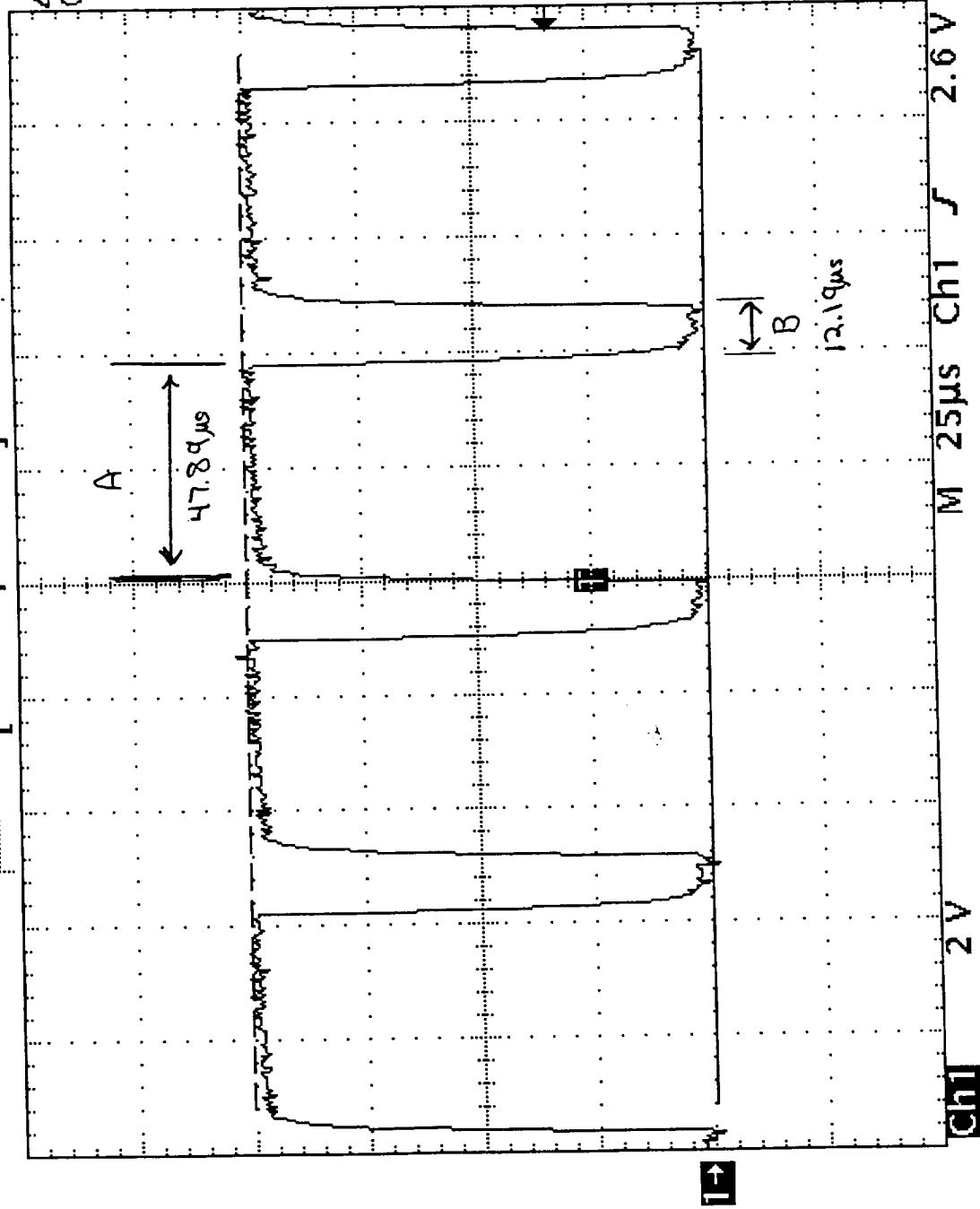
$$[\cdots]$$

Δ : 8.08 V
@: -80mV

Ch1 +width
47.89 μ s

Ch1 -width
12.19 μ s

"C1" Shift
PULSE



15 Jul 2000
08:55:17

324322

TDS 9

S/O: 787920 CP: 0830

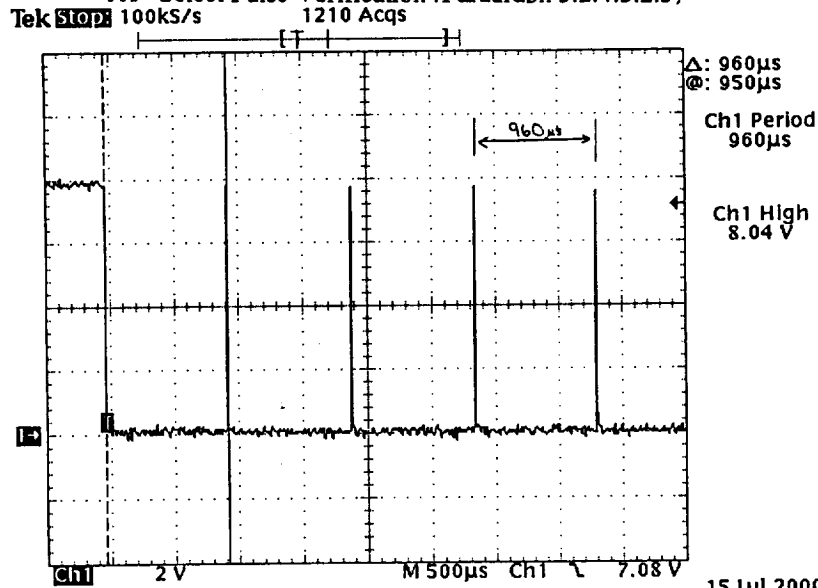
PAI. 1331720-3-TST SN. 109

FINAL CPT

TEST ENG. *For Analysis DATE: 7-15-00*
QUALITY CONTROL *DATE: 7-15-00*

TEST DATA SHEET 10

"A1" Select Pulse Verification (Paragraph 3.2.4.3.2.3)



15 Jul 2000
09:03:45

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

3.24323

TDS 10

FINAL CPT

TEST ENG: Ray Huthberg Date: 7-15-00

QUALITY: Judy Hervey Date: 7-15-00

	Calculated		
Select Pulse Timing (F) *	960.0 µs	961.5 µs ± 10%	P
Select Pulse Amplitude	8.04 Volts	9.0 ± 1.0 V	P

* Refer to Figure 13 for location of the pulse timing F

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 SN: 109

Op 0830

Ray Huthberg
Test Systems Engineer

7-15-00
Date

J. Sanford
Customer Representative
(Flight Hardware Only)

7-15-00
Date

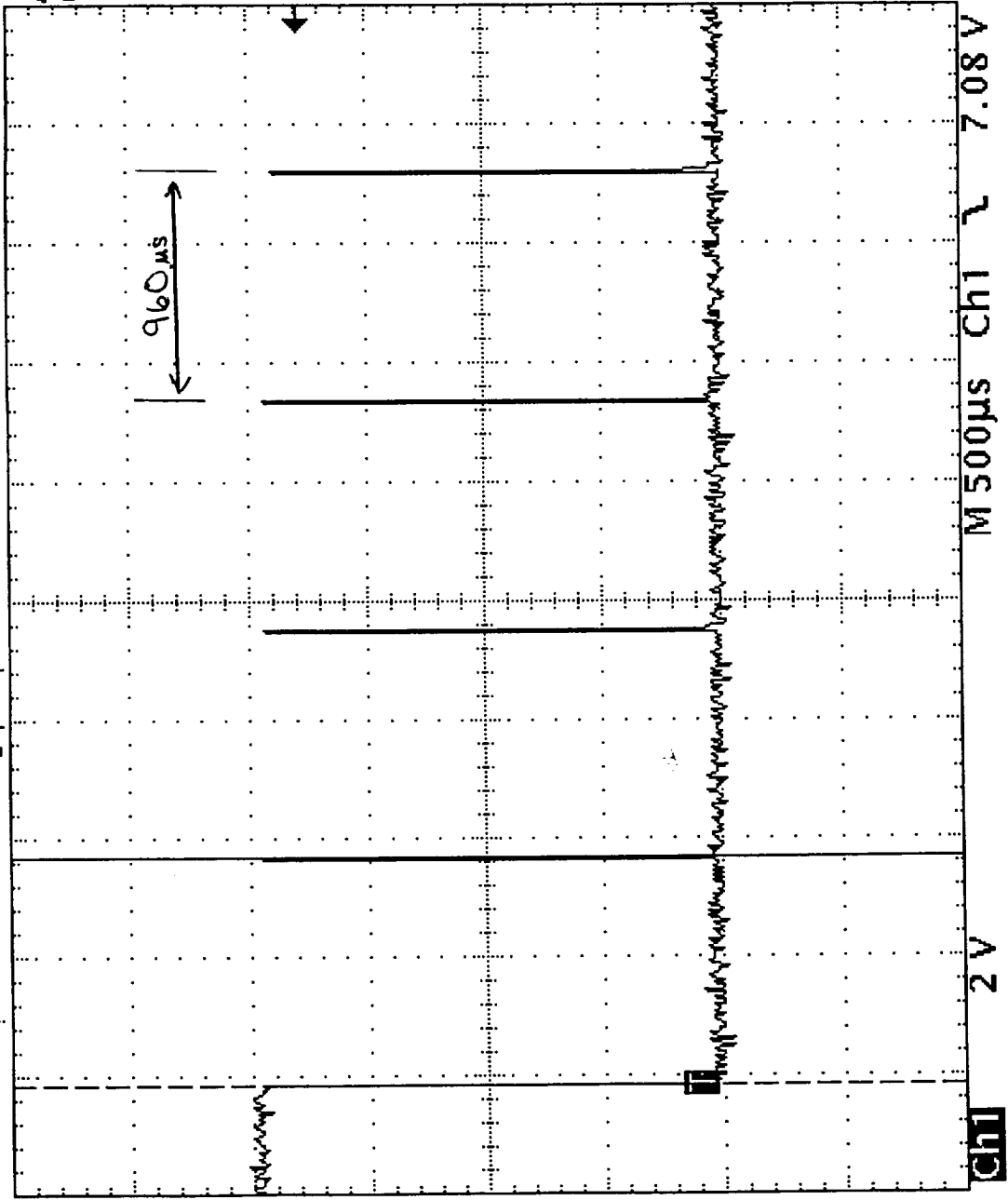
Judy Hervey
Quality Control

7-15-00
Date

Tek Stop: 100KS/s

1210 Acqs

[T]



A1
Select
Pulse

15 Jul 2000
09:03:45

324323

TDS10

S/O: 787920 CP: 0830

P/N: 1331720-3-TST SN: 109

FINAL CPT

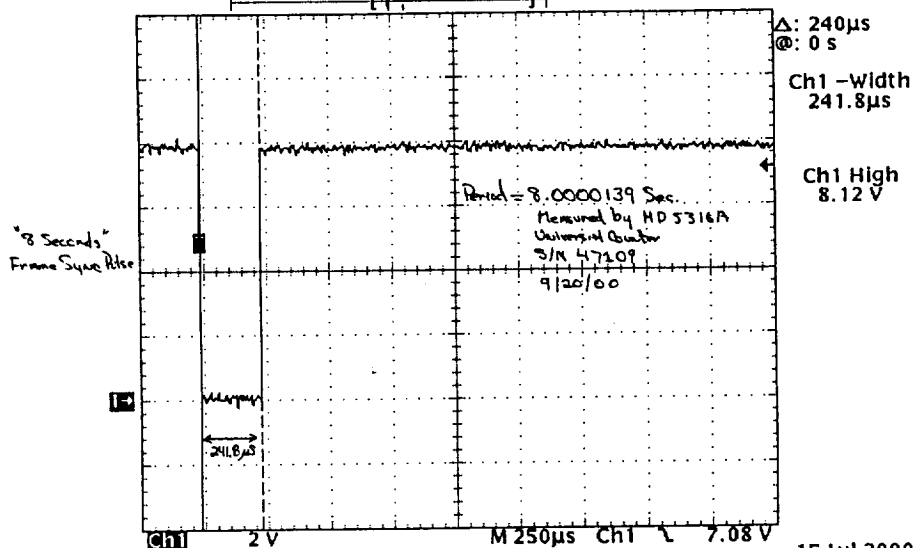
TEST ENG: *[Signature]* DATE: 7-15-00
DUAL IT: *[Signature]* DATE: 7-15-00

TEST DATA SHEET 11

"8 Seconds" Frame Sync Pulse (Para 3.2.4 3 2 4)

Tek Stop 200K5/s

384 Acqs



3.24324

S/O: 787920 OP: 0830

TDS II

TEST ENG: Ray Huthrey Date: 7-15-00

Step

P/N: 1331720-3-TST SN: 109

FINAL CPT

QUALITY: 7-15-00

ss/ Fail

		Calculated		
1*	Frame Sync Pulse Timing (G)*	8.0 Sec	8 Sec ±10%	PASS
	Frame Sync Pulse Timing (C)*	241.8 μs	240.4 μs ±10%	PASS
	Frame Sync Pulse Amplitude	8.12 Volts	9.0 ±1.0 V	PASS

* Refer to Figure 13 for location of the timing pulses for G and C.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

Op. 0830

Test Systems Engineer

7-15-00

Date

Customer Representative
(Flight Hardware Only)



7-15-00

Date

Quality Control



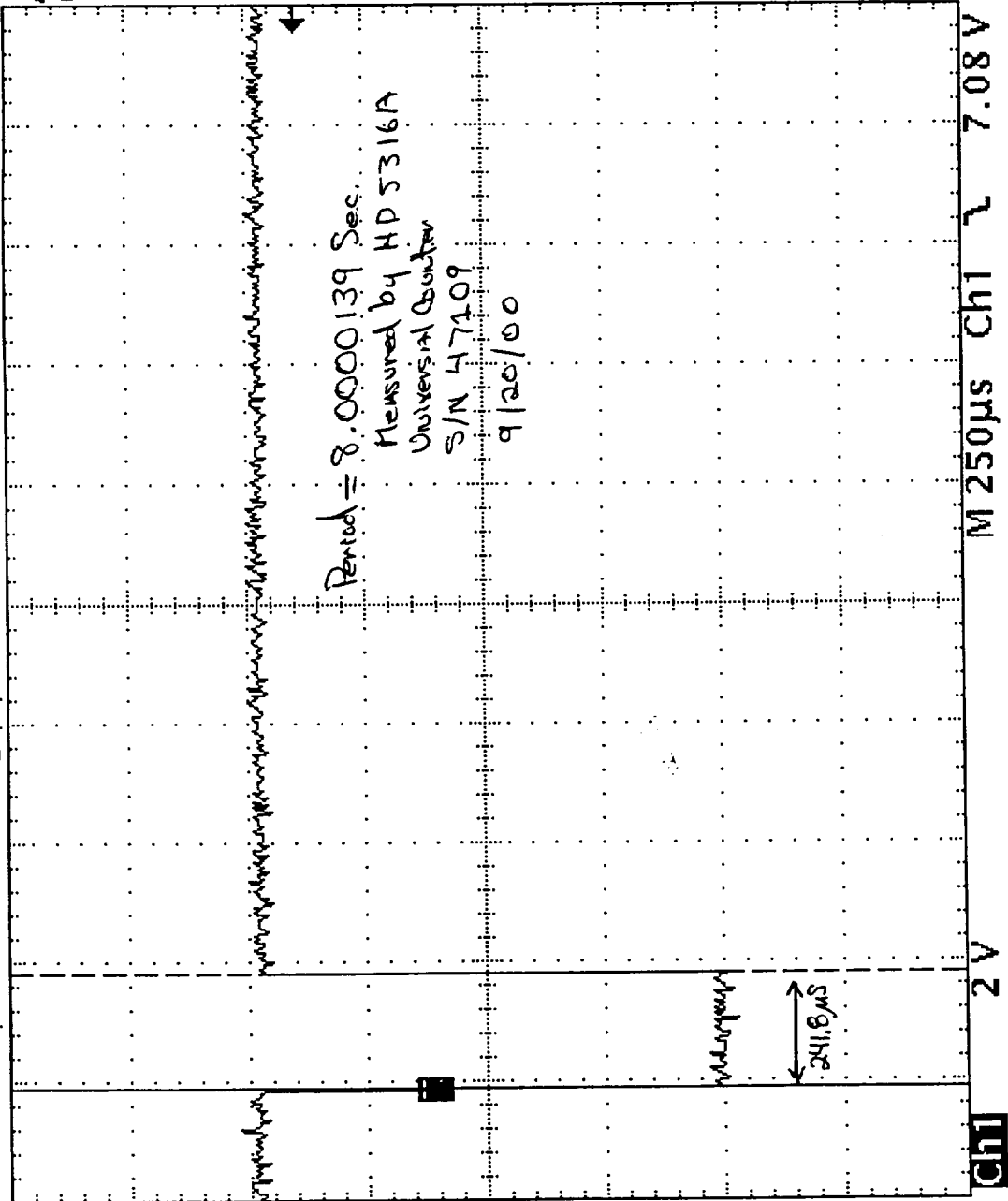
7-15-00

Date

Tek Stop 200ks/s

384 Acqs

[T] []



Δ : 240µs
@: 0 s

Ch1 - width
241.8µs

Ch1 High
8.12 V

15 Jul 2000
09:12:44

3.24324

S/O: 787920 OP: 0830

DATE: 15 JUL 2000 09:12:44

TOS 11

FILE: 001

TEST ENG: [Signature]
DATE: 7-15-00
TIME: 11:17:24
PAGE: 1

—

—

—

6 Apr 99

TEST DATA SHEET 12 (Sheet 1 of 2)
Synchronization Signals Relationship (Paragraph 3.2.4.3.2.5)

A1 Select pulse and the 8 seconds Frame sync pulse.

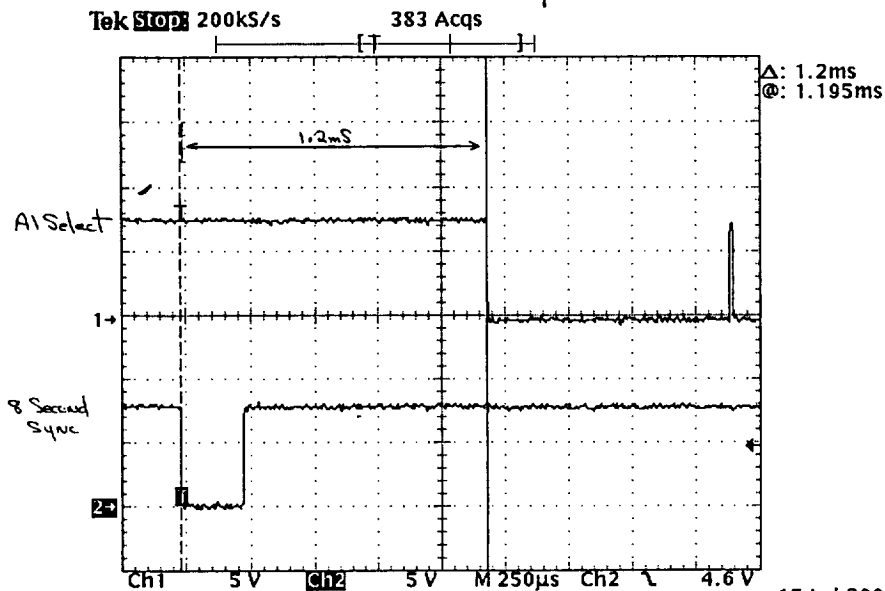
Verify that the sync pulse between H and C is as shown in Figure 19.

TIME MEASURED: 1.2ms

TIME REQUIRED: 1.2 ms \pm 10%

PASS/FAIL PASS

ATTACH PHOTOGRAPH OR PLOT HERE



15 Jul 2000
09:58:21

3.2.4.3.2.5

S/O: 787920 OP: 0830 TDS 12 (Sheet 1 of 2) TEST ENG: Ray Burberg DATE: 7-15-00
P/N: 1331720-3-TST SN: 109 FINAL CPT QUALITY: Indira Hervey DATE: 7-15-00

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-15-00
Date

Ray Burberg
Test Systems Engineer

7-15-00
Date

Indira Hervey
Quality Control

7-15-00
Date

—

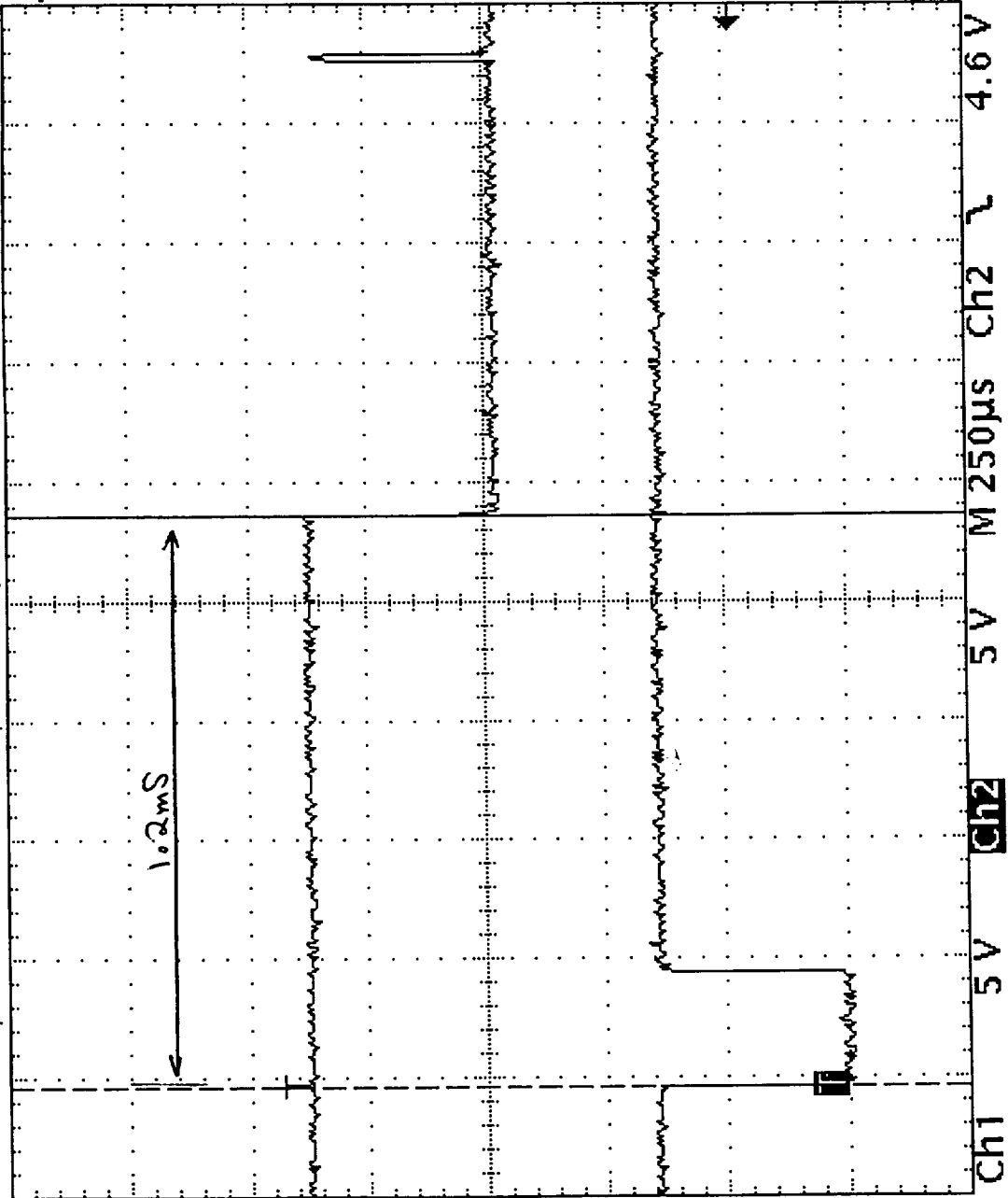
—

—

Tek Scope 200KS/s

383 Acqs

[T]



15 Jul 2000
09:58:21

3.2.4.3.25

S/O: 787920 CP: 0830

P/N: 1331720-3-TST SN: 109

TDS 12 (Sheet 1 of 2) TEST ENG: *[Signature]* DATE: 7-15-00

QUALITY CONTROL *[Signature]* 7-15-00

6 Apr 99

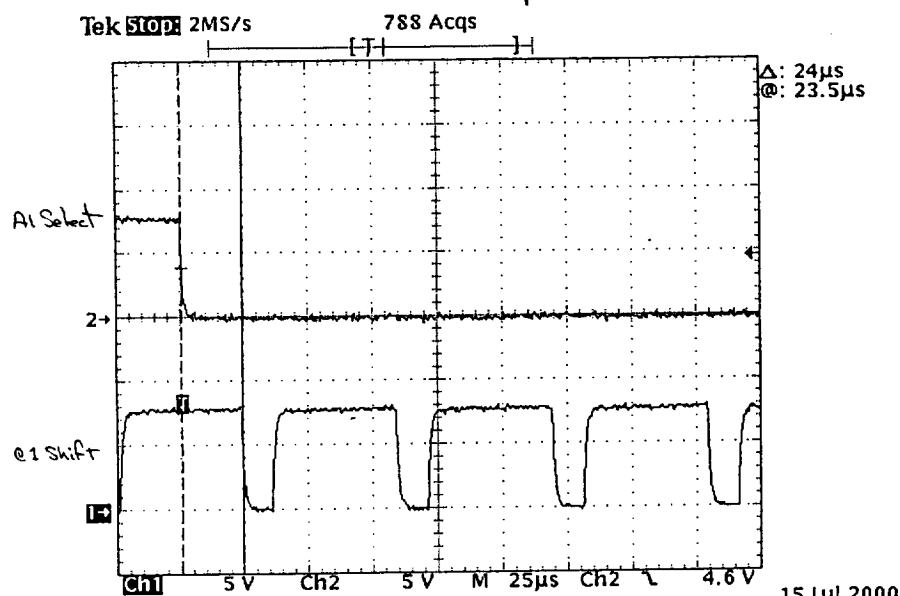
TEST DATA SHEET 12 (Sheet 2 of 2)
Synchronization Signals Relationship (Paragraph 3.2.4.3.2.5)

A1 Select pulse and the C1 Shift pulse.

Verify that the sync pulse between I and E is as shown in Figure 19.

TIME MEASURED: 24.0 μ sTIME REQUIRED: 24 μ s \pm 1 μ sPASS/FAIL PASS

ATTACH PHOTOGRAPH OR PLOT HERE

15 Jul 2000
10:07:27

S/O: 787920 OP: 0830 TDS12 (Sheet 2 of 2) TEST ENG: [Signature] DATE: 7-15-00
 P/N: 1331720-3-TST SN: 109 FINAL CPT QUALITY: [Signature] DATE: 7-15-00

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op 0830
Shop Order: 787920 S/N: 109

[Signature]
 Customer Representative
 (Flight Hardware Only)

26 7-15-00
 Date

[Signature]
 Test Systems Engineer

7-15-00
 Date

[Signature]
 Quality Control

26 7-15-00
 Date

[illegible]

@1 shift

3.24.3.2.5

DAI. 1331720-3-75T SN. 109

FINAL CPT

FINAL CPT

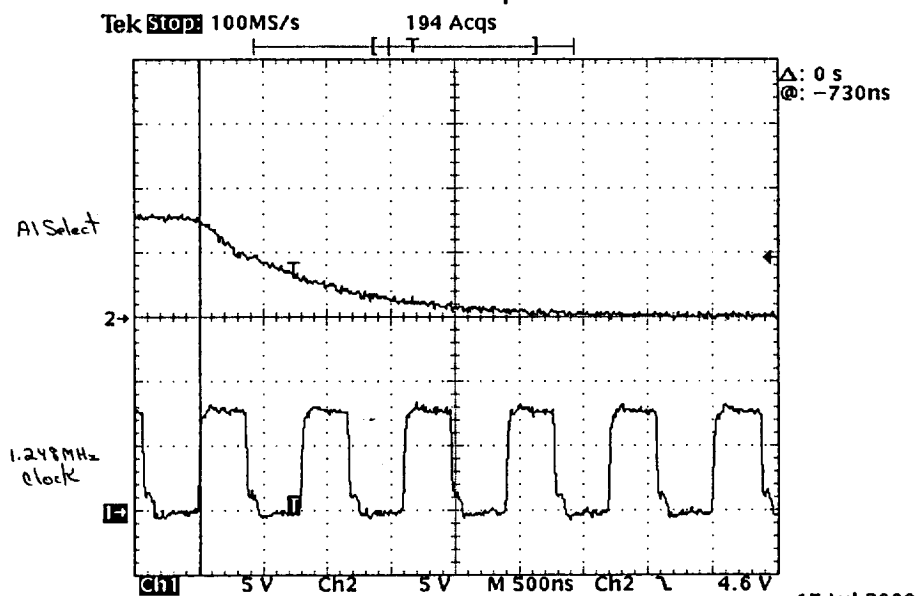
TEST DATA SHEET 13
Synchronization Signals Relationship (Paragraph 3.2.4.3.2.5)

A1 Select pulse and the 1.248 MHz clock.

Verify that the sync pulse between I and J is as shown in Figure 19.

PASS/FAIL PASS

ATTACH PHOTOGRAPH OR PLOT HERE



15 Jul 2000
10:20:20

3.24.3.2.5

S/O: 787920 OP: 0830

TDS 13

TEST ENG: [Signature] DATE: 7-15-00

P/N: 1331720-3-TST SN: 109

FINAL CPT

QUALITY: [Signature] 7-15-00

Circle Test: Final CPT LPT

Op 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 SN: 109

[Signature]
Customer Representative
(Flight Hardware Only)

7-15-00
Date

[Signature]
Test Systems Engineer

7-15-00
Date

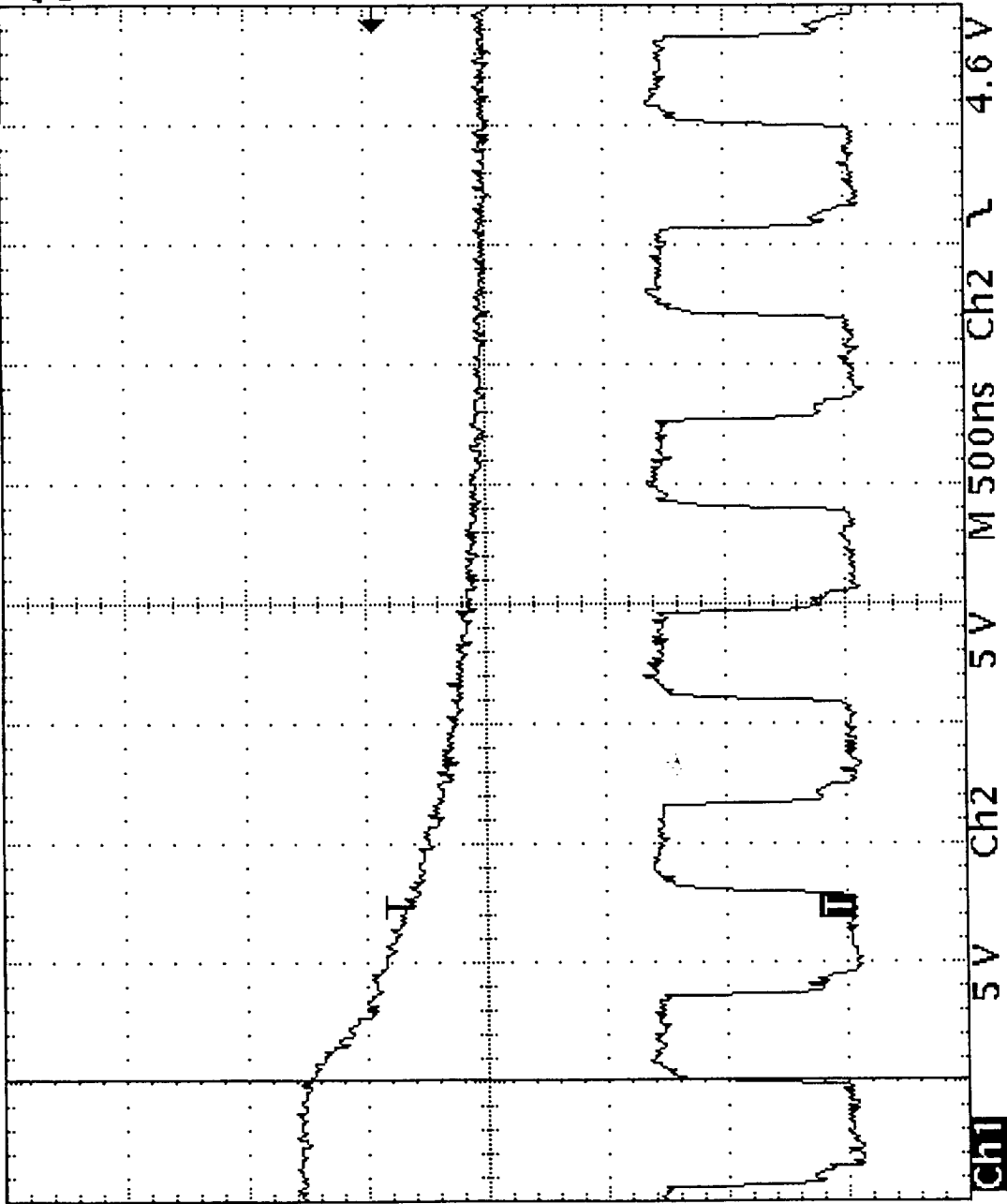
[Signature]
Quality Control

7-15-00
Date

Tek Stop: 100MS/s

194 Acqs

[T]



$\Delta: 0\text{ s}$
@: -730ns

AI Select

1.248MHz
clock

15 Jul 2000
10:20:20

3.24, 3.25

S/O: 787920 CP: 0830

P/N: 1331720-3-TST SN: 109

TDS 13

FINAL CPT

TEST ENG: *[Signature]* DATE: 7-15-00
QUALITY: *[Signature]* 7-15-00

6 Apr 99

TEST DATA SHEET 14

Commands and Digital-B Telemetry Verification (Paragraphs 3.2.4.3.3.1, 3.2.4.3.3.2, 3.2.4.3.3.3, and 3.2.4.3.3.4)

Test	Digital-B Commands Verification Via STE			Visual Inspection		Pass/Fail
	Command	Observed	Required	Observed	Required	
3.2.4.3.3.1 Module Totally Off	Scanner A1-1	OFF	OFF	TO WARM LOAD	Antenna pointing to warm load.	P
	Scanner A1-2	OFF	OFF	TO WARM LOAD	Antenna pointing to warm load.	P
	Module Power	DISCONNECT	Disconnect	N/A	N/A	P
	Survival Htr. Power.	OFF	OFF	0.0 A	28 V supply current=0	P
3.2.4.3.3.2 Survival Heater Power	Survival Heater ON	ON	ON	N/A	N/A	P
	Survival Heater OFF	OFF	OFF	N/A	N/A	P
3.2.4.3.3.3 Module Power Connect	Module Power	CONNECT	Connect	2.2 A	+28 V DC current is between 0.5 and 3.2 amps.	P
3.2.4.3.3.4 PLL Power	PLLO#2	PLLO#2	PLLO#2	N/A	N/A	P
	PLLO#1	PLLO#1	PLLO#1	N/A	N/A	P

Circle Test: Final CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787922

S/N: 609

Customer Representative
(Flight Hardware Only)7-15-00
Date

Test Systems Engineer

Quality Control

7-15-00
Date7-15-00
Date

TEST DATA SHEET 15
Scanner Commands Verification (Paragraph 3.2.4.3.3.5, Step 1)


Test	Digital "B" Verification			Pass/Fail
	Command	Observed	Required	
Full Scan	1 Module Power	CONNECT	CONNECT	P
	2 Survival Heater	OFF	OFF	P
	3 Scanner A1 Power	ON	ON	P
	4 Scanner A2 Power	ON	ON	P
	5 Antenna Warm Cal Pos.	NO	NO	P
	6 Antenna Cold Cal Pos.	NO	NO	P
	7 Antenna NADIR Position	NO	NO	P
	8 Antenna Full Scan	YES	YES	P
	9 PLL Power	PLL#1	PLL#1	P
	10 Cold MSB	0	0	P
	11 Cold LSB	0	0	P

Circle Test: Final
CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 767920 S/N: 109

J. Sanford  7-15-00
Customer Representative (Flight Hardware Only) Date

Raymond Bue 7-15-00
Test Systems Engineer Date
Judith Hervey 7-15-00
Quality Control Date

TEMPERATURE DEG C

VALUE

DESCRIPTION

1090	SCAN MOTOR A1-1	18410	24.34
1092	SCAN MOTOR A1-2	19700	25.62
1094	FEEDHORN A1-1	20371	27.50
1096	FEEDHORN A1-2	21410	29.50
1098	RF MUX A1-1	22213	30.58
1100	RF MUX A1-2	23318	32.71
1102	LOCAL OSCILLATOR CHANNEL 3	24451	34.85
1104	LOCAL OSCILLATOR CHANNEL 4	24509	34.63
1106	LOCAL OSCILLATOR CHANNEL 5	23898	33.97
1108	LOCAL OSCILLATOR CHANNEL 6	21687	29.94
1110	LOCAL OSCILLATOR CHANNEL 7	22808	31.86
1112	LOCAL OSCILLATOR CHANNEL 8	24094	34.07
1114	LOCAL OSCILLATOR CHANNEL 15	23966	33.54
1116	PLL LO #2 CHANNELS 9 THROUGH 14	15071	30.37
1118	PLL LO #1 CHANNELS 9 THROUGH 14	25202	36.37
1120	SPARE (NOT USED)	32767	51.27
1122	MIXER/IF AMPLIFIER CHANNEL 3	24022	33.19
1124	MIXER/IF AMPLIFIER CHANNEL 4	24023	33.41
1126	MIXER/IF AMPLIFIER CHANNEL 5	23595	32.97
1128	MIXER/IF AMPLIFIER CHANNEL 6	22418	32.05
1130	MIXER/IF AMPLIFIER CHANNEL 7	22600	31.64
1132	MIXER/IF AMPLIFIER CHANNEL 8	23860	33.41
1134	MIXER/IF AMPLIFIER CH 9 THRU 14	22380	30.48
1136	MIXER/IF AMPLIFIER CHANNEL 15	23395	33.40
1138	MIXER/IF AMPLIFIER CHANNEL 11 THRU 14	23474	33.12
1140	IF AMPLIFIER CHANNEL 9	23501	33.27
1142	IF AMPLIFIER CHANNEL 10	23672	33.28
1144	IF AMPLIFIER CHANNEL 11	22240	30.81
1146	DC/DC CONVERTER	23627	33.31
1148	IF AMPLIFIER CHANNEL 13	22231	30.75
1150	IF AMPLIFIER CHANNEL 14	22329	30.72
1152	IF AMPLIFIER CHANNEL 12	22141	30.72
1154	RF SHELF A1-1	22258	31.82
1156	RF SHELF A1-2	23088	32.57
1158	DETECTOR/PREAMPLIFIER ASSEMBLY	20714	28.21
1160	A1-1 WARM LOAD 1	24232	24.77
1162	A1-1 WARM LOAD 2	24012	24.75
1164	A1-1 WARM LOAD 3	24217	24.78
1166	A1-1 WARM LOAD 4	24179	24.79
1168	A1-1 WARM LOAD CENTER	24252	24.90
1170	A1-2 WARM LOAD 1	24937	26.55
1172	A1-2 WARM LOAD 2	25078	26.48
1174	A1-2 WARM LOAD 3	25246	26.62
1176	A1-2 WARM LOAD 4	25013	26.63
1178	A1-2 WARM LOAD CENTER	24932	26.58
1180	TEMP SENSOR REFERENCE VOLTAGE	25331	

PRT TEMPERATURES

VARIABLE TARGET

NO.	DEG K	A1-1	DEG K	A1-2	DEG K
601	42.00			601	14.00
616	43.00			602	15.00
617	44.00			603	16.00
618	45.00			604	17.00
619	46.00			605	18.00
620	47.00			606	19.00
621	48.00			607	20.00
622	49.00			608	21.00
623	50.00			609	22.00
624	51.00			610	23.00
625	52.00			611	24.00
626	53.00			612	25.00
627	67.00			613	69.00
628	68.00			614	70.00
629	71.00			630	72.00
631	26.00			632	27.00

FIXED TARGET

BASEPLATE

THERMOCOUPLE TEMPERATURES

FIXED TARGET SHROUD

VARIABLE TARGET SHROUD

FIXED TARGET N2

VARIABLE TARGET N2

HEATER N2

FIXED TARGET FLOW METER

VARIABLE TARGET FLOW METER

BASEPLATE HEATER N2

BASEPLATE N2

BASEPLATE FLOW METER

ADJUNCT RADIATORS

NO.	DEG K	A1-1	DEG K	A1-2	DEG K
558	5.00			537	34.00
559	6.00			538	35.00
550	7.00			524	36.00
551	8.00			525	37.00
506	57.00			502	30.00
507	58.00			503	31.00
516	59.00			511	32.00
517	60.00			512	33.00
514	1.00			509	38.00
515	2.00			510	39.00
508	63.00			504	61.00
518	64.00			513	62.00
519	3.00			520	4.00
521	9.00			522	10.00
523	65.00				
575	73.00			577	74.00
579	75.00			581	76.00

AMSU A1-33 A1 EXE:62 FULL SCAN MODE P1 15-JUL-00 15:28:32 SCAN NUMBER 39
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
		1		2		1		2	
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	23	23	9	1233	1236	17	2444	2449	25
2	167	174	10	1384	1389	18	2597	2602	26
3	324	327	11	1534	1539	19	2749	2753	27
4	474	479	12	1686	1691	20	2899	2905	28
5	624	632	13	1839	1843	21	3052	3057	29
6	778	782	14	1989	1995	22	3202	3206	30
7	930	934	15	2143	2146	23	3350	3357	CC
8	1079	1085	16	2294	2299	24	3504	3509	WC
[21] UP		[22] DOWN							

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

BP	REFLECTOR POSITIONS										
	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2
1	16224	16224	9	1050	1054	17	2259	2265	25	3475	3479
2	16371	16374	10	1201	1204	18	2415	2419	26	3626	3632
3	142	147	11	1354	1356	19	2569	2569	27	3781	3785
4	296	298	12	1506	1508	20	2718	2721	28	3936	3936
5	444	447	13	1656	1660	21	2870	2873	29	4082	4087
6	594	598	14	1806	1811	22	3023	3026	30	4236	4238
7	746	748	15	1962	1963	23	3172	3176	CC	5832	5833
8	899	902	16	2111	2114	24	3326	3328	WC	10232	10232
[21]	UP			[22]	DOWN						

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62 FULL SCAN MODE P1 15-JUL-00 15:29:36 SCAN NUMBER 47
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA					
CHANNEL 3					
BP	DATA	BP	DATA	BP	DATA
1	16744	9	16792	17	16823
2	16751	10	16832	18	16763
3	16751	11	16845	19	16755
4	16754	12	16813	20	16753
5	16748	13	16809	21	16750
6	16746	14	16818	22	16750
7	16748	15	16848	23	16747
8	16749	16	16850	24	16750
	[22]	DOWN			WC

[21] UP

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL
SELECT TOUCHSCREEN BUTTON 2 [1] RETURN

AMSU	A1-33	A1.EXE:62	FULL SCAN MODE	P1	15-JUL-00	15:30:01	SCAN NUMBER	50
[5]		DIGITAL A DATA	ELEMENT 0000					

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

BP	AGGREGATED CHANNEL DATA			
	DATA	BP	DATA	BP
1	16319	9	16317	17
2	16313	10	16316	18
3	16310	11	16316	19
4	16316	12	16325	20
5	16322	13	16323	21
6	16326	14	16332	22
7	16331	15	16329	23
8	16329	16	16326	24
		22	DOWN	

[21] UP

POWER [4] ON

	POWER [1]	SCREEN ONLY [2]	PRINT [3]	FULL [1]	RETURN
SELECT TOUCHSCREEN BUTTON 2					

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 1 TO 16			
	DATA	TEMP C	NO	DATA
1	SCAN MOTOR A1-1	18430	9	LO CHANNEL 5
2	SCAN MOTOR A1-2	19723	10	LO CHANNEL 6
3	FEEDHORN A1-1	20397	11	LO CHANNEL 7
4	FEEDHORN A1-2	21426	12	LO CHANNEL 8
5	RF MUX A1-1	22267	13	LO CHANNEL 15
6	RF MUX A1-2	23384	14	PLLO #2 CH 9/14
7	LO CHANNEL 3	24523	15	PLLO #1 CH 9/14
8	LO CHANNEL 4	24584	16	PLLO REFERENCE
[21]	UP			
		[22]	DOWN	
				23960
				21731
				22862
				24164
				24040
				15108
				25294
				32767
				34.09
				30.02
				31.96
				34.21
				33.68
				30.44
				36.54
				51.27

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
 SELECT TOUCHSCREEN BUTTON 2

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

DIGITAL A TEMPERATURES 17 TO 32									
NO	DATA	TEMP C	NO	IF AMP	CH	DATA	TEMP C		
17 MIXER IF CH 3	24095	33.33	25 IF AMP	CH 11/14	23548	33.29			
18 MIXER IF CH 4	24097	33.55	26 IF AMP	CH 9	23578	33.42			
19 MIXER IF CH 5	23662	33.09	27 IF AMP	CH 10	23750	33.43			
20 MIXER IF CH 6	22474	31.16	28 IF AMP	CH 11	22292	30.91			
21 MIXER IF CH 7	22663	31.76	29 DC/DC CONVERTER		23695	33.44			
22 MIXER IF CH 8	23936	33.56	30 IF AMP	CH 13	22281	30.88			
23 MIXER IF CH 9/14	22434	30.59	31 IF AMP	CH 14	22379	30.85			
24 MIXER IF CH 15	23463	33.53	32 IF AMP	CH 12	22191	30.82			
[21] UP		[22] DOWN							

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
 SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 FULL SCAN MODE P1 15-JUL-00 15:30:55 SCAN NUMBER 57
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 31 TO 46				DATA	TEMP C
	DATA	TEMP C	NO			
31	IF AMP CH 14	22381	30.85	39 A1-1 WARM LOAD	4 24196	24.82
32	IF AMP CH 12	22195	30.82	40 A1-1 WARM LOAD	C 24257	24.91
33	RF SHELF A1-1	22323	31.94	41 A1-2 WARM LOAD	1 24956	26.59
34	RF SHELF A1-2	23156	32.71	42 A1-2 WARM LOAD	2 25099	26.52
35	DETECTOR/PREAMP	20755	28.28	43 A1-2 WARM LOAD	3 25273	26.67
36	A1-1 WARM LOAD	24242	24.79	44 A1-2 WARM LOAD	4 25036	26.67
37	A1-1 WARM LOAD	24017	24.76	45 A1-2 WARM LOAD	C 24948	26.61
38	A1-1 WARM LOAD	24226	24.80	THERMAL REFERENCE	25331	
[21]	UP		[22]	DOWN		

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 FULL SCAN MODE P1 15-JUL-00 15:31:31 SCAN NUMBER 61
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 1 TO 18

1	A1-1	SCANR MOTOR	216	25.90	DEG C	10	ANTENNA DRIVE	15VDC	15.07
2	A1-2	SCANR MOTOR	217	27.06	DEG C	11	SIGNAL PROCESSING	-15VDC	-15.01
3	A1-1	RF SHELF	219	29.10	DEG C	12	ANTENNA DRIVE	-15VDC	-14.97
4	A1-2	RF SHELF	220	31.21	DEG C	13	RECEIVER AMPLIFIER	8VDC	8.00
5	A1-1	WARM LOAD	215	25.75	DEG C	14	SIGNAL PROCESSOR	5 VDC	4.99
6	A1-2	WARM LOAD	217	28.41	DEG C	15	ANTENNA DRIVE	5 VDC	5.02
7	ANT A1-1	DRIVE MOTOR CURRENT		40.27		16	RECEIVER MIXER/IF	10VDC	9.91
8	ANT A1-2	DRIVE MOTOR CURRENT		39.65		17	PHASE LOCK LOOP CH9/14	15VDC	15.02
9	SIGNAL PROCESSING	+15VDC		15.03		18	PHASE LOCK LOOP CH9/14	-15VDC	-14.97
[21]	UP	[22]	DOWN						

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 10 TO 27

10 ANTENNA DRIVE	15VDC	15.17	19 L.O.	VOLTAGE	CH 8	10.00
11 SIGNAL PROCESSING	-15VDC	-15.01	20 L.O.	VOLTAGE	CH 7	9.98
12 ANTENNA DRIVE	-15VDC	-14.90	21 L.O.	VOLTAGE	CH 6	10.00
13 RECEIVER AMPLIFIER	8VDC	8.00	22 L.O.	VOLTAGE	CH 3	9.99
14 SIGNAL PROCESSOR	5 VDC	4.99	23 L.O.	VOLTAGE	CH 4	10.02
15 ANTENNA DRIVE	5 VDC	5.07	24 L.O.	VOLTAGE	CH 5	10.00
16 RECEIVER MIXER/IF	10VDC	9.90	25 PLLO # 2	LOCK DETECT		0.08
17 PHASE LOCK LOOP CH9/14	15VDC	15.04	26 PLLO # 1	LOCK DETECT		4.40
18 PHASE LOCK LOOP CH9/14	-15VDC	-14.97	27 L.O.	VOLTAGE	CH15	15.00
[21] UP	[22] DOWN					

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN

SELECT TOUCHSCREEN BUTTON 2

TEST DATA SHEET 16
Scanner Commands Verification (Paragraph 3.2.4.3.3.5, Step 2)

Test	Digital "B" Verification			Pass/Fail
	Command	Observed	Required	
Full Scan	1 Module Power	CONNECT	CONNECT	P
	2 Survival Heater	OFF	OFF	P
	3 Scanner A1 Power	OFF	OFF	P
	4 Scanner A2 Power	OFF	OFF	P
	5 Antenna Warm Cal Pos.	NO	NO	P
	6 Antenna Cold Cal Pos.	NO	NO	P
	7 Antenna NADIR Position	NO	NO	P
	8 Antenna Full Scan	YES	YES	P
	9 PLL Power	PLLO#1	PLLO#1	P
	10 Cold MSB	0	0	P
	11 Cold LSB	0	0	P

Circle Test: Final CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-15-00
Date

Ray Burkley
Test Systems Engineer

7-15-00
Date

Judith Hervey
Quality Control

7-15-00
Date

TEST DATA SHEET 17
Scanner Commands Verification (Paragraph 3.2.4.3.3.5, Step 3)

Test	Digital "B" Verification			Pass/Fail
	Command	Observed	Required	
Full Scan	1 Module Power	CONNECT	CONNECT	P
	2 Survival Heater	OFF	OFF	P
	3 Scanner A1 Power	ON	ON	P
	4 Scanner A2 Power	ON	ON	P
	5 Antenna Warm Cal Pos.	NO	NO	P
	6 Antenna Cold Cal Pos.	NO	NO	P
	7 Antenna NADIR Position	NO	NO	P
	8 Antenna Full Scan	YES	YES	P
	9 PLL Power	PLLO#1	PLLO#1	P
	10 Cold MSB	0	0	P
	11 Cold LSB	0	-0	P

Circle Test: Final CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

Test Systems Engineer

Date

Customer Representative
(Flight Hardware Only)

Date

Quality Control

Date

6 Apr 99

TEST DATA SHEET 18
Scanner Positions Commands (Paragraph 3.2.4.3.3.6)

Test	Digital "B" Verification			Pass/Fail
	Step/Description	Observed	Required	
Scanner Position Commands	1-Warm Cal.		YES	P
	2-Cold Cal.	MSB	0	P
	Pos.	LSB	1	P
	3-Cold Cal.	MSB	1	P
	Pos.	LSB	0	P
	4-Cold Cal.	MSB	1	P
	Pos.	LSB	1	P
	5-Cold Cal.	MSB	0	P
	Pos.	LSB	0	P
	6-NADIR		YES	P
	7-Warm Cal		YES	P

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 0A: 0830 767920 S/N: 109

[Signature]
Customer Representative
(Flight Hardware Only)

7-15-00
Date

[Signature]
Test Systems Engineer

7-15-00
Date

[Signature]
Quality Control

7-15-00
Date

TEST DATA SHEET 19
Digital-A Data Output Full Scan Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.1)


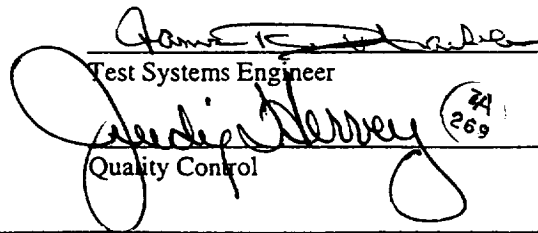
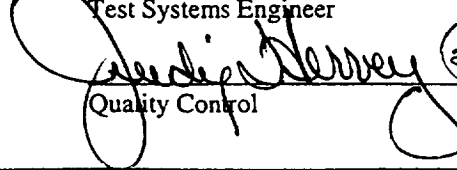
Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1	255	255	Pass
	0002	Sync Sequence Byte 2	255	255	
	0003	Sync Sequence Byte 3	255	255	
[II]	0004	Unit I.D. and Serial N	33	*	
[III]	0005	Digital-B Data Byte 1	2	2	
	0006	Digital-B Data Byte 2	14	**	
	0007	Digital-B Data Byte 3	0	0	
	0008	Digital-B Data Byte 4	0	0	Pass

* AMSU A1 Identification Words (data entered in decimal system)	Binary	Decimal
AMSU-A1 S/N 101	00000001	1
AMSU-A1 S/N 102	00000101	5
AMSU-A1 S/N 103	00001001	9
AMSU-A1 S/N 104	00001101	13
AMSU-A1 S/N 105	00010001	17
AMSU-A1 S/N 106	00010101	21
AMSU-A1 S/N 107	00011001	25
AMSU-A1 S/N 108	00011101	29
AMSU-A1 S/N 109	00100001	33

** Required value = 14 when PLLO #1 is active; and = 6 when PLLO #2 is active.

Circle Test: CPT LPT op: 0830

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: 787920 SN: 109

 Customer Representative (Flight/Hardware Only)	7-17-00 Date	 Test Systems Engineer	7-15-00 Date
		 Quality Control	7/15/00 Date

TEST DATA SHEET 20
Reflector Positions Section [IV] (Paragraph 3.2.4.3.4.1)

BP	A1-1 Reflector				A1-2 Reflector			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0014	23	23	P	0016	16224	16225	P
02	0048	174	175		0050	16374	16377	
03	0082	326	327		0084	148	145	
04	0116	478	478		0118	299	296	
05	0150	632	630		0152	447	448	
06	0184	781	782		0186	596	600	
07	0218	933	933		0220	748	751	
08	0252	1085	1085		0254	902	903	
09	0286	1236	1237		0288	1054	1055	
10	0320	1389	1388		0322	1204	1206	
11	0354	1534	1540		0356	1356	1358	
12	0388	1691	1692		0390	1509	1510	
13	0422	1843	1843		0424	1660	1661	
14	0456	1994	1995		0458	1812	1813	
15	0490	2147	2147		0492	1963	1965	
16	0524	2299	2298		0526	2119	2116	
17	0558	2450	2450		0560	2265	2268	
18	0592	2602	2602	P	0594	2418	2420	P
19	0626	2753	2753		0628	2569	2571	
20	0660	2905	2905		0662	2721	2723	
21	0694	3056	3057		0696	2873	2875	
22	0728	3206	3208		0730	3026	3026	
23	0762	3357	3360		0764	3177	3178	
24	0796	3509	3512		0798	3328	3330	
25	0830	3660	3663		0832	3479	3481	
26	0864	3812	3815		0866	3631	3633	
27	0890	3971	3967		0900	3784	3785	
28	0932	4115	4118		0934	3935	3936	
29	0966	4267	4270		0968	4087	4088	
30	1000	4423	4422		1002	4238	4240	
CC	1034	6017	6016		1036	5833	5834	
WC	1186	10416	10415	P	1188	10232	10233	P

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.

Circle Test: CPT LPT

op: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 SN: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

James P. [Signature]
Test Systems Engineer

[Signature]
Quality Control

7-15-00

Date

7-15-00

Date

6 Apr 99

TEST DATA SHEET 21
Digital-A Data Output Radiometer Data Section [V] (Paragraph 3.2.4.3.4.1)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
01	0018	16750	16500 ± 4000	P	0030	16326	16500 ± 4000	P
02	0052	16750			0064	16316		
03	0086	16751			0098	16315		
04	0120	16752			0132	16323		
05	0154	16747			0166	16326		
06	0188	16759			0200	16329		
07	0222	16750			0234	16335		
08	0256	16758			0268	16337		
09	0290	16790			0302	16324		
10	0324	16838			0336	16328		
11	0356	16846			0370	16324		
12	0392	16821			0404	16324		
13	0426	16810			0438	16326		
14	0460	16815			0472	16333		
15	0494	16852			0506	16333		
16	0528	16854			0540	16329		
17	0562	16815		P	0574	16330		
18	0596	16768			0608	16333		
19	0630	16754			0642	16330		
20	0664	16751			0676	16328		
21	0698	16754			0710	16327		
22	0732	16754			0744	16328		
23	0766	16751			0778	16328		
24	0800	16750			0812	16325		
25	0834	16758			0846	16325		
26	0868	16751			0880	16326		
27	0902	16753			0914	16329		
28	0936	16752			0948	16319		
29	0970	16749			0982	16322		
30	1004	16754			1016	16324		
CC	1038	16744			1050	16372		
WC	1190	16798	16500 ± 4000	Pass	1202	16343	16500 ± 4000	P

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts.

Circle Test: CPT LPT
Final

METSAT/AMSU-AT System P/N IS-1331720

CP: 0830

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

James J. Starnes
Test Systems Engineer

Judith Harvey
Quality Control

7-15-00

Date

(289) 7/15/00

Date

TEST DATA SHEET 22 (Sheet 1 of 2)
Full Scan Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.1)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1	24.34	25 ± 15	Pass
1092	A1-1 Warm Load 2	25.62	25 ± 15	↑
1094	A1-1 Warm Load 3	27.52	25 ± 15	
1096	A1-1 Warm Load 4	29.50	25 ± 15	
1098	A1-1 Warm Load Center	30.58	25 ± 15	
1100	A1-2 Warm Load 1	32.71	25 ± 15	
1102	A1-2 Warm Load 2	34.85	25 ± 15	
1104	A1-2 Warm Load 3	34.63	25 ± 15	
1106	A1-2 Warm Load 4	33.97	25 ± 15	
1108	A1-2 Warm Load Center	29.94	25 ± 15	
1110	Local Oscillator Channel 7	31.86	25 ± 15	
1112	Local Oscillator Channel 8	34.07	25 ± 15	
1114	Local Oscillator Channel 15	33.54	25 ± 15	
1116	PLL LO #2 Channels 9-14	30.37	25 ± 15	
1118	PLL LO #1 Channels 9-14	36.37	25 ± 15	Pass
1120	PLLO (Reference Oscillator)**/ Not used ***	N/A	25 ± 15	N/A
1122	Mixer I.F. Amp. Channel 3	33.19	25 ± 15	Pass
1124	Mixer I.F. Amp. Channel 4	33.41	25 ± 15	
1126	Mixer I.F. Amp. Channel 5	32.97	25 ± 15	
1128	Mixer I.F. Amp. Channel 6	31.05	25 ± 15	
1130	Mixer I.F. Amp. Channel 7	31.64	25 ± 15	
1132	Mixer I.F. Amp. Channel 8	35.41	25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14	30.48	25 ± 15	
1136	Mixer I.F. Amp. Channel 15	33.40	25 ± 15	Pass

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** For S/N 101 through 104.

*** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 22 (Sheet 2 of 2)
Full Scan Mode Temperature Sensors Section [VI (Paragraph 3.2.4.3.4.1)]

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14	33.15	25 ± 15	Pass
1140	I.F. Amp. Channel 9	33.27	25 ± 15	↑
1142	I.F. Amp. Channel 10	33.28	25 ± 15	
1144	I.F. Amp. Channel 11	30.81	25 ± 15	
1146	DC/DC Converter	33.31	25 ± 15	
1148	I.F. Amp. Channel 13	30.78	25 ± 15	
1150	I.F. Amp. Channel 14	30.75	25 ± 15	
1152	I.F. Amp. Channel 12	30.72	25 ± 15	
1154	RF Shelf A1-1	31.82	25 ± 15	
1156	RF Shelf A1-2	32.57	25 ± 15	
1158	Detector Preamp Assy.	28.21	25 ± 15	
1160	Scan Motor A1-1	24.77	25 ± 15	
1162	Scan Motor A1-2	24.75	25 ± 15	
1164	Feed Horn A1-1	24.78	25 ± 15	
1166	Feed Horn A1-2	24.79	25 ± 15	
1168	R.F. Mux A1-1	24.90	25 ± 15	
1170	R.F. Mux A1-2	26.55	25 ± 15	
1172	Local Oscillator Channel 3	26.48	25 ± 15	
1174	Local Oscillator Channel 4	26.62	25 ± 15	
1176	Local Oscillator Channel 5	26.63	25 ± 15	
1178	Local Oscillator Channel 6	26.58	25 ± 15	↓
1180	Temp Sensor Ref Voltage Count	25331	**	Pass

* Value is from the STE printout sheets. Copying data to this sheet is optional.
** = Count of 24,552 +1765,-1308.

Circle Test: Final
CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

Customer Representative
(Flight Hardware Only)

7-17-00
Date

Test Systems Engineer

Quality Control

7-15-00

Date

Date

AMSU A1-33 A1.EXE:62 FULL SCAN MODE P1 15-JUL-00 15:24:53 SCAN NUMBER 12
[5] DIGITAL A DATA ELEMENT 0000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

COMMANDS
[9] MODULE POWER = CONNECT ANTENNA IN COLD CAL POSIT = NO [15]
[10] SURVIVAL HEATER POWER = OFF ANTENNA IN NADIR POSITION = NO [16]
[11] MODULE TOTALLY OFF = ON ANTENNA IN FULL SCAN MODE = YES [17]
[12] SCANNER A1 - 1 POWER = ON PLL POWER = PLLO # 1 [18]
[13] SCANNER A1 - 2 POWER = ON COLD CAL POSITION MSB = ZERO [19]
[14] ANTENNA IN WARM CAL POSIT = NO COLD CAL POSITION LSB = ZERO [20]

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 3

Para: 3.2, 4.3, 4.1
TDS: 19 → 22

S/O: 787920 CP: 0830
P/N: 1331720-3-TST SN: 109 FINAL CPT

TEST ENG: ~~Amesbury~~ DATE: 7-15-00
QUALITY: ~~Amesbury~~ 7-15-00

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
1	SYNC SEQUENCE BYTE 1	11111111	572	SCENE DATA BP 17	CH 8
2	SYNC SEQUENCE BYTE 2	11111111	574		CH 9
3	SYNC SEQUENCE BYTE 3	11111111	576		CH 10
4	UNIT ID AND SERIAL NO	00100001	578		CH 11
5	DIGITAL B DATA BYTE 1	00000010	580		CH 12
6	DIGITAL B DATA BYTE 2	00001110	582		CH 13
7	DIGITAL B DATA BYTE 3	00000000	584		CH 14
8	DIGITAL B DATA BYTE 4	00000000	586		CH 15
10	REFLECTOR 1 POSITION	16224	588	REFLECTOR 1 POSITION 18	2597
12	REFLECTOR 2 POSITION	16224	590	REFLECTOR 2 POSITION 18	2415
14	REFL 1 POS 1 2ND LOOK	16224	592	REFL 1 POS 18 2ND LOOK	2602
16	REFL 2 POS 1 2ND LOOK	16224	594	REFL 2 POS 18 2ND LOOK	2418
18	SCENE DATA BP 1	16750	596	SCENE DATA BP 18	16768
20		16225	598		16249
22		16802	600		16821
24		16762	602		16766
26		16474	604		16479
28		14966	606		14976
30		16326	608		16333
32		16030	610		16034
34		16234	612		16248
36		16659	614		16665
38		16418	616		16438
40		16837	618		16835
42		14887	620		14890
44	REFLECTOR 1 POSITION	167	622	REFLECTOR 1 POSITION 19	2749
46	REFLECTOR 2 POSITION	16372	624	REFLECTOR 2 POSITION 19	2568
48	REFL 1 POS 2 2ND LOOK	174	626	REFL 1 POS 19 2ND LOOK	2753
50	REFL 2 POS 2 2ND LOOK	16374	628	REFL 2 POS 19 2ND LOOK	2569
52	SCENE DATA BP 2	16750	630	SCENE DATA BP 19	16754
54		16227	632		16235
56		16805	634		16812
58		16755	636		16763
60		16468	638		16477
62		14961	640		14966
64		16316	642		16330
66		16235	644		16033
68		16661	646		16245
70		16436	648		16673
72		16856	650		16448
74		14886	652		16851
76	REFLECTOR 1 POSITION	323	654	REFLECTOR 1 POSITION 20	14890
78	REFLECTOR 2 POSITION	143	656	REFLECTOR 2 POSITION 20	2899
80	REFL 1 POS 3 2ND LOOK	326	658	REFL 1 POS 20 2ND LOOK	2718
82	REFL 2 POS 3 2ND LOOK	148	660	REFL 2 POS 20 2ND LOOK	2905
84	SCENE DATA BP 3	16751	662	SCENE DATA BP 20	2721
86		16227	664		16751
88		16810	666		16233
90		16750	668		16808
92			670		16768

FULL SCAN MODE

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
94	CH	16466	672	CH	16478
96	CH	14961	674	CH	14969
98	CH	16315	676	CH	16328
100	CH	16025	678	CH	16040
102	CH	16234	680	CH	16249
104	CH	16663	682	CH	16668
106	CH	16407	684	CH	16449
108	CH	16817	686	CH	16837
110	CH	14885	688	CH	14891
112	REFLECTOR 1 POSITION	474	690	REFLECTOR 1 POSITION	3053
114	REFLECTOR 2 POSITION	295	692	REFLECTOR 2 POSITION	2868
116	REFL 1 POS	478	694	REFL 1 POS	3056
118	REFL 2 POS	299	696	REFL 2 POS	2873
120	SCENE DATA	16752	698	SCENE DATA	16754
122	CH	16225	700	CH	16230
124	CH	16811	702	CH	16808
126	CH	16758	704	CH	16761
128	CH	16472	706	CH	16476
130	CH	14967	708	CH	14968
132	CH	16323	710	CH	16327
134	CH	16027	712	CH	16033
136	CH	16240	714	CH	16237
138	CH	16663	716	CH	16671
140	CH	16444	718	CH	16428
142	CH	16831	720	CH	16831
144	CH	14889	722	CH	14889
146	REFLECTOR 1 POSITION	625	724	REFLECTOR 1 POSITION	3202
148	REFLECTOR 2 POSITION	443	726	REFLECTOR 2 POSITION	3203
150	REFL 1 POS	632	728	REFL 1 POS	3206
152	REFL 2 POS	447	730	REFL 2 POS	3026
154	SCENE DATA	16747	732	SCENE DATA	16754
156	CH	16232	734	CH	16229
158	CH	16810	736	CH	16806
160	CH	16763	738	CH	16766
162	CH	16474	740	CH	16478
164	CH	14965	742	CH	14967
166	CH	16326	744	CH	16328
168	CH	16036	746	CH	16033
170	CH	16244	748	CH	16248
172	CH	16664	750	CH	16671
174	CH	16421	752	CH	16445
176	CH	16837	754	CH	16841
178	CH	14889	756	CH	14891
180	REFLECTOR 1 POSITION	777	758	REFLECTOR 1 POSITION	3350
182	REFLECTOR 2 POSITION	593	760	REFLECTOR 2 POSITION	3173
184	REFL 1 POS	781	762	REFL 1 POS	3357
186	REFL 2 POS	598	764	REFL 2 POS	3177
188	SCENE DATA	16759	766	SCENE DATA	16751
190	CH	16232	768	CH	16228
192	CH	16807	770	CH	16807

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
194	CH 6	16762	772	REFLECTOR 1 POSITION 24	3504
196	CH 7	16477	774	REFLECTOR 2 POSITION 24	3324
198	CH 8	14964	776	REFL 1 POS 24	3509
200	CH 9	16329	778	REFL 2 POS 24	3328
202	CH 10	16037	780	SCENE DATA BP 24	16750
204	CH 11	16246	782	CH 3	16228
206	CH 12	16674	784	CH 4	16808
208	CH 13	16442	786	CH 5	16761
210	CH 14	16833	788	CH 6	16476
212	CH 15	14889	790	CH 7	14968
214	REFLECTOR 1 POSITION 7	929	792	CH 8	16325
216	REFLECTOR 2 POSITION 7	745	794	CH 9	16037
218	REFL 1 POS 7	933	796	CH 10	16248
220	REFL 2 POS 7	748	798	CH 11	16675
222	SCENE DATA BP 7	16750	800	CH 12	16675
224	CH 3	16231	802	CH 13	16434
226	CH 4	16806	804	CH 14	16840
228	CH 5	16770	806	CH 15	14889
230	CH 6	16485	808	REFLECTOR 1 POSITION 25	3654
232	CH 7	14963	810	REFLECTOR 2 POSITION 25	3477
234	CH 8	16335	812	REFL 1 POS 25	3660
236	CH 9	16040	814	REFL 2 POS 25	3479
238	CH 10	16256	816	SCENE DATA BP 25	16758
240	CH 11	16676	818	CH 3	16230
242	CH 12	16447	820	CH 4	16811
244	CH 13	16846	822	CH 5	16758
246	CH 14	14893	824	CH 6	16475
248	CH 15	1079	826	CH 7	14973
250	REFLECTOR 1 POSITION 8	1098	828	CH 8	16033
252	REFLECTOR 2 POSITION 8	1085	830	CH 9	16246
254	REFL 1 POS 8	1902	832	CH 10	16660
256	REFL 2 POS 8	16758	834	CH 11	16437
258	SCENE DATA BP 8	16230	836	CH 12	16828
260	CH 3	16810	838	CH 13	14889
262	CH 4	16770	840	CH 14	3806
264	CH 5	16485	842	CH 15	3626
266	CH 6	14968	844	REFLECTOR 1 POSITION 26	3812
268	CH 7	16337	846	REFLECTOR 2 POSITION 26	3631
270	CH 8	16040	848	REFL 1 POS 26	16751
272	CH 9	16250	850	REFL 2 POS 26	16229
274	CH 10	16673	852	SCENE DATA BP 26	
276	CH 11	16433	854	CH 3	
278	CH 12	16855	856	CH 4	
280	CH 13	14890	858		
282	CH 14	12322	860		
284	CH 15	1049	862		
286	REFLECTOR 1 POSITION 9	1236	864		
288	REFLECTOR 2 POSITION 9	1054	866		
290	REFL 1 POS 9	16790	868		
292	REFL 2 POS 9	16266	870		
	SCENE DATA BP 9				
	CH 3				
	CH 4				

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
294	CH 5	16844	872	CH 5	16807
296	CH 6	16758	874	CH 6	16763
298	CH 7	16473	876	CH 7	16471
300	CH 8	14988	878	CH 8	14969
302	CH 9	16324	880	CH 9	16326
304	CH 10	16029	882	CH 10	16029
306	CH 11	16239	884	CH 11	16240
308	CH 12	16667	886	CH 12	16659
310	CH 13	16427	888	CH 13	16437
312	CH 14	16850	890	CH 14	16825
314	CH 15	14887	892	CH 15	14887
316	REFLECTOR 1 POSITION 10	1385	894	REFLECTOR 1 POSITION 27	3970
318	REFLECTOR 2 POSITION 10	1202	896	REFLECTOR 2 POSITION 27	3780
320	REFL 1 POS 10 2ND LOOK	1389	898	REFL 1 POS 27 2ND LOOK	3971
322	REFL 2 POS 10 2ND LOOK	1204	900	REFL 2 POS 27 2ND LOOK	3784
324	SCENE DATA BP 10	16838	902	SCENE DATA BP 27	16753
326	CH 3	16301	904	CH 3	16229
328	CH 4	16936	906	CH 4	16807
330	CH 5	16761	908	CH 5	16763
332	CH 6	16471	910	CH 6	16473
334	CH 7	15023	912	CH 7	14962
336	CH 8	16328	914	CH 8	16324
338	CH 9	16028	916	CH 9	16028
340	CH 10	16240	918	CH 10	16238
342	CH 11	16662	920	CH 11	16659
344	CH 12	16430	922	CH 12	16435
346	CH 13	16857	924	CH 13	16848
348	CH 14	14888	926	CH 14	14888
350	CH 15	1534	928	CH 15	4111
352	REFLECTOR 1 POSITION 11	1353	930	REFLECTOR 1 POSITION 28	3937
354	REFLECTOR 2 POSITION 11	1353	932	REFLECTOR 2 POSITION 28	4115
356	REFL 1 POS 11 2ND LOOK	1356	934	REFL 1 POS 28 2ND LOOK	3935
358	REFL 2 POS 11 2ND LOOK	16846	936	REFL 2 POS 28 2ND LOOK	16752
360	SCENE DATA BP 11	16316	938	SCENE DATA BP 28	16225
362	CH 3	16956	940	CH 3	16809
364	CH 4	16764	942	CH 4	16758
366	CH 5	16469	944	CH 5	16473
368	CH 6	15030	946	CH 6	14964
370	CH 7	16324	948	CH 7	16319
372	CH 8	16027	950	CH 8	16028
374	CH 9	16242	952	CH 9	16238
376	CH 10	16666	954	CH 10	16656
378	CH 11	16431	956	CH 11	16425
380	CH 12	16830	958	CH 12	16827
382	CH 13	14887	960	CH 13	14886
384	CH 14	1686	962	CH 14	4260
386	CH 15	1506	964	CH 15	4084
388	REFLECTOR 1 POSITION 12	1691	966	REFLECTOR 1 POSITION 29	4267
390	REFLECTOR 2 POSITION 12	1509	968	REFLECTOR 2 POSITION 29	4087
392	REFL 1 POS 12 2ND LOOK	16821	970	REFL 1 POS 29 2ND LOOK	16749
	REFL 2 POS 12 2ND LOOK			REFL 2 POS 29 2ND LOOK	
	SCENE DATA BP 12			SCENE DATA BP 29	

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
394	CH 4	16309	972	CH 4	16229
396	CH 5	16933	974	CH 5	16808
398	CH 6	16764	976	CH 6	16757
400	CH 7	16478	978	CH 7	16474
402	CH 8	15027	980	CH 8	14966
404	CH 9	16324	982	CH 9	16322
406	CH 10	16029	984	CH 10	16024
408	CH 11	16238	986	CH 11	16238
410	CH 12	16671	988	CH 12	16668
412	CH 13	16432	990	CH 13	16437
414	CH 14	16838	992	CH 14	16825
416	CH 15	14889	994	CH 15	14888
418	REFLECTOR 1 POSITION 13	18339	996	REFLECTOR 1 POSITION 30	4419
420	REFLECTOR 2 POSITION 13	1657	998	REFLECTOR 2 POSITION 30	4235
422	REFL 1 POS 13 2ND LOOK	1843	1000	REFL 1 POS 30 2ND LOOK	4423
424	REFL 2 POS 13 2ND LOOK	1660	1002	REFL 2 POS 30 2ND LOOK	4238
426	SCENE DATA BP 13	16810	1004	SCENE DATA BP 30	16754
428	CH 4	16312	1006	CH 4	16232
430	CH 5	16907	1008	CH 5	16808
432	CH 6	16766	1010	CH 6	16761
434	CH 7	16479	1012	CH 7	16473
436	CH 8	15025	1014	CH 8	14968
438	CH 9	16326	1016	CH 9	16324
440	CH 10	16035	1018	CH 10	16029
442	CH 11	16246	1020	CH 11	16240
444	CH 12	16674	1022	CH 12	16667
446	CH 13	16445	1024	CH 13	16440
448	CH 14	16830	1026	CH 14	16834
450	CH 15	14890	1028	CH 15	14889
452	REFLECTOR 1 POSITION 14	1989	1030	REFLECTOR 1 COLD CAL POS	6017
454	REFLECTOR 2 POSITION 14	1807	1032	REFLECTOR 2 COLD CAL POS	5833
456	REFL 1 POS 14 2ND LOOK	1994	1034	REFL 1 COLD CAL 2ND LOOK	6017
458	REFL 2 POS 14 2ND LOOK	1812	1036	REFL 2 COLD CAL 2ND LOOK	5833
460	SCENE DATA BP 14	16815	1038	COLD CAL DATA 1	16744
462	CH 4	16309	1040	CH 4	16224
464	CH 5	16886	1042	CH 5	16803
466	CH 6	16776	1044	CH 6	16809
468	CH 7	16487	1046	CH 7	16533
470	CH 8	15017	1048	CH 8	14960
472	CH 9	16333	1050	CH 9	16372
474	CH 10	16040	1052	CH 10	16076
476	CH 11	16249	1054	CH 11	16286
478	CH 12	16670	1056	CH 12	16716
480	CH 13	16441	1058	CH 13	16486
482	CH 14	16847	1060	CH 14	16889
484	CH 15	14890	1062	CH 15	14926
486	REFLECTOR 1 POSITION 15	2143	1064	COLD CAL DATA 2	16747
488	REFLECTOR 2 POSITION 15	1961	1066	CH 3	16225
490	REFL 1 POS 15 2ND LOOK	2147	1068	CH 4	16804
492	REFL 2 POS 15 2ND LOOK	1963	1070	CH 5	16808

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
494	SCENE DATA BP 15	3	1072	CH	16852
496		4	1074	CH	16343
498		5	1076	CH	16901
500		6	1078	CH	16768
502		7	1080	CH	16482
504		8	1082	CH	15031
506		9	1084	CH	16333
508		10	1086	CH	16035
510		11	1088	CH	16246
512		12	1182	REFLECTOR 1 WARM CAL POS	16669
514		13	1184	REFLECTOR 2 WARM CAL POS	16448
516		14	1186	REFL 1 WARM CAL 2ND LOOK	16853
518		15	1188	REFL 2 WARM CAL 2ND LOOK	14891
520	REFLECTOR 1 POSITION 16	16	1190	WARM CAL DATA 1	2294
522	REFLECTOR 2 POSITION 16	16	1192		2111
524	REFL 1 POS 16	16	1194		2299
526	REFL 2 POS 16	16	1196		2114
528	SCENE DATA BP 16	16	1198		16854
530		3	1200		16348
532		4	1202		16902
534		5	1204		16765
536		6	1206		16477
538		7	1208		15027
540		8	1210		16329
542		9	1212		16032
544		10	1214		16241
546		11	1216		16674
548		12	1218		16455
550		13	1220		16830
552		14	1222		14891
554	REFLECTOR 1 POSITION 17	17	1224		2444
556	REFLECTOR 2 POSITION 17	17	1226		2261
558	REFL 1 POS 17	17	1228		2450
560	REFL 2 POS 17	17	1230		2265
562	SCENE DATA BP 17	17	1232		16815
564		3	1234		16317
566		4	1236		16863
568		5	1238		16765
570		6	1240		16480
572		7			
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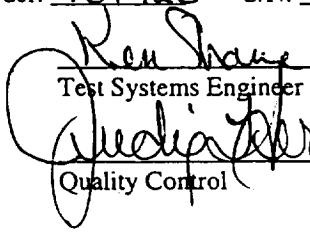
TEST DATA SHEET 23
Digital-A Data Output Warm Cal Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.2)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1	255	255	PASS
	0002	Sync Sequence Byte 2	255	255	↓
	0003	Sync Sequence Byte 3	255	255	
[II]	0004	Unit I.D. and Serial N	33	*	
[III]	0005	Digital-B Data Byte 1	4	4	
	0006	Digital-B Data Byte 2	14	14	
	0007	Digital-B Data Byte 3	0	0	
	0008	Digital-B Data Byte 4	0	0	PASS

* AMSU A1 Identification Words (data entered in decimal system)	Binary	Decimal
AMSU-A1 S/N 101	00000001	1
AMSU-A1 S/N 102	00000101	5
AMSU-A1 S/N 103	00001001	9
AMSU-A1 S/N 104	00001101	13
AMSU-A1 S/N 105	00010001	17
AMSU-A1 S/N 106	00010101	21
AMSU-A1 S/N 107	00011001	25
AMSU-A1 S/N 108	00011101	29
AMSU-A1 S/N 109	00100001	33

Circle Test: **CPT** LPT Op. 0830

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: 787920 S/N: 109

			<u>7/15/00</u>
Customer Representative (Flight Hardware Only)	<u>7-17-00</u> Date	Test Systems Engineer Quality Control	Date

TEST DATA SHEET 24

Reflector Position Warm Cal Mode Section [IV] and Reflector Position Nadir Mode Section [IV] (Paragraphs 3.2.4.3.4.2 and 3.2.4.3.4.4)


BP	A1-1 Reflector			
	Para No.	Position*	Required**	Pass/Fail
WC	3.2.4.3.4.2	10413	10415	P
15	3.2.4.3.4.4	2147	2147	P
WC = Warm Cal 15 = Nadir Position				
BP	A1-2 Reflector			
	Para No.	Position*	Required**	Pass/Fail
WC	3.2.4.3.4.2	10226	10233	P
15	3.2.4.3.4.4	1963	1965	P
WC = Warm Cal 15 = Nadir Position				
<p>* Actual counts from computer printout. Rewriting counts on this data sheet is optional.</p> <p>** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.</p>				


Circle Test: **FINAL**
CPT LPT

Op 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 SN: 109

J. Sanford  7-17-00
Customer Representative (Flight Hardware Only) Date

R. Shaw 7/15/00
Test Systems Engineer Date
Judith Hervey  7/15/00
Quality Control Date

TEST DATA SHEET 25
Digital-A Data Output Warm Cal Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.2)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0018	16795	16,500 ± 4000	PASS	0030	16337	16,500 ± 4000	PASS
02	0052	16798			0064	16333		
03	0086	16793			0098	16333		
04	0120	16793			0132	16338		
05	0154	16790			0166	16336		
06	0188	16798			0200	16337		
07	0222	16795			0234	16335		
08	0256	16795			0268	14993		
09	0290	16797			0302	16336		
10	0324	16802			0336	16339		
11	0356	16795			0370	16336		
12	0392	16797			0404	16334		
13	0426	16793			0438	16333		
14	0460	16789			0472	16337		
15	0494	16796			0506	16338		
16	0528	16795			0540	16336		
17	0562	16795			0574	16335		
18	0596	16795			0608	16336		
19	0630	16794			0642	16335		
20	0664	16797			0676	16337		
21	0698	16794			0710	16334		
22	0732	16793			0744	16336		
23	0766	16801			0778	16339		
24	0800	16797			0812	16335		
25	0834	16799			0846	16336		
26	0868	16796			0880	16335		
27	0902	16792			0914	16335		
28	0936	16794			0948	16339		
29	0970	16795			0982	16338		
30	1004	16794	16,500 ± 4000		1016	16335	16,500 ± 4000	
CC	1038	0	0		1050	0	0	
WC	1190	0	0	PASS	1202	0	0	PASS

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts.

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op. 0830
Shop Order: 787920 SN: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Ken Shaw
Test Systems Engineer




Judith Kennedy
Quality Control

7-15-00

Date

7/15/00
Date

TEST DATA SHEET 26 (Sheet 1 of 2)
Warm Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.2)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1	24.43	25 ± 15	PASS
1092	A1-1 Warm Load 2	25.74	25 ± 15	
1094	A1-1 Warm Load 3	27.60	25 ± 15	
1096	A1-1 Warm Load 4	29.60	25 ± 15	
1098	A1-1 Warm Load Center	30.77	25 ± 15	
1100	A1-2 Warm Load 1	32.95	25 ± 15	
1102	A1-2 Warm Load 2	35.11	25 ± 15	
1104	A1-2 Warm Load 3	34.89	25 ± 15	
1106	A1-2 Warm Load 4	34.19	25 ± 15	
1108	A1-2 Warm Load Center	30.09	25 ± 15	
1110	Local Oscillator Channel 7	32.05	25 ± 15	
1112	Local Oscillator Channel 8	34.33	25 ± 15	
1114	Local Oscillator Channel 15	33.79	25 ± 15	
1116	PLL LO #2 Channels 9-14	30.52	25 ± 15	
1118	PLL LO #1 Channels 9-14	36.68	25 ± 15	PASS
1120	PLLO (Reference Oscillator)**/ Not used ***	N/A 	N/A 	N/A 
1122	Mixer I.F. Amp. Channel 3	33.44	25 ± 15	PASS
1124	Mixer I.F. Amp. Channel 4	33.67	25 ± 15	
1126	Mixer I.F. Amp. Channel 5	33.20	25 ± 15	
1128	Mixer I.F. Amp. Channel 6	31.24	25 ± 15	
1130	Mixer I.F. Amp. Channel 7	31.86	25 ± 15	
1132	Mixer I.F. Amp. Channel 8	33.67	25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14	30.67	25 ± 15	
1136	Mixer I.F. Amp. Channel 15	33.63	25 ± 15	PASS

- * Value is from the STE printout sheets. Copying data to this sheet is optional.
 ** For S/N 101 through 104.
 *** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 26 (Sheet 2 of 2)
Warm Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.2)

Thermistor Sensors		Recorded Value*	Required Value	Pass/Fail
Element	Description	(deg. C)	(deg. C)	
1138	I.F. Amp. Channel 11-14	33.41	25 ± 15	PASS
1140	I.F. Amp. Channel 9	33.52	25 ± 15	
1142	I.F. Amp. Channel 10	33.54	25 ± 15	
1144	I.F. Amp. Channel 11	30.99	25 ± 15	
1146	DC/DC Converter	33.56	25 ± 15	
1148	I.F. Amp. Channel 13	30.96	25 ± 15	
1150	I.F. Amp. Channel 14	30.92	25 ± 15	
1152	I.F. Amp. Channel 12	30.90	25 ± 15	
1154	RF Shelf A1-1	32.03	25 ± 15	
1156	RF Shelf A1-2	32.80	25 ± 15	
1158	Detector Preamp Assy.	28.34	25 ± 15	
1160	Scan Motor A1-1	24.80	25 ± 15	
1162	Scan Motor A1-2	24.78	25 ± 15	
1164	Feed Horn A1-1	24.82	25 ± 15	
1166	Feed Horn A1-2	24.82	25 ± 15	
1168	R.F. Mux A1-1	24.92	25 ± 15	
1170	R.F. Mux A1-2	26.62	25 ± 15	
1172	Local Oscillator Channel 3	26.56	25 ± 15	
1174	Local Oscillator Channel 4	26.70	25 ± 15	
1176	Local Oscillator Channel 5	26.70	25 ± 15	
1178	Local Oscillator Channel 6	26.64	25 ± 15	↓
1180	Temp Sensor Ref Voltage Count	25331	**	PASS

* Value is from the STE printout sheets. Copying data to this sheet is optional.
** = Count of 24,552 +1765, -1308.

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op. 0830
Shop Order: 787920 SN: 109

J. Sanford 7-17-00
Customer Representative (Flight Hardware Only) Date

Ken Shane 7/15/00
Test Systems Engineer Date
Judith Harvey 7/15/00
Quality Control Date

[5] DIGITAL A DATA ELEMENT 0000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

COMMANDS
[9] MODULE POWER = CONNECT ANTENNA IN COLD CAL POSIT = NO [15]
[10] SURVIVAL HEATER POWER = OFF ANTENNA IN NADIR POSITION = NO [16]
[11] MODULE TOTALLY OFF = ON ANTENNA IN FULL SCAN MODE = NO [17]
[12] SCANNER A1 - 1 POWER = ON PLL POWER = PLLO # 1 [18]
[13] SCANNER A1 - 2 POWER = ON COLD CAL POSITION MSB = ZERO [19]
[14] ANTENNA IN WARM CAL POSIT = YES COLD CAL POSITION LSB = ZERO [20]

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 3

3.2.43.42 and 3.2.43.44

TDS 23 to 26

TEST ENG: KUSHAN DATE: 7/15/00
QUALITY: Xudip Noddy 7/15/00

S/O: 787920 CP: 0830
P/N: 1331720-3-TST SN: 109

FINAL CPT

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
1	SYNC SEQUENCE	11111111	572	WARM CAL SAMPLE 17	14987
2	SYNC SEQUENCE	11111111	574		16335
3	SYNC SEQUENCE	11111111	576		16039
4	UNIT ID AND SERIAL NO	00100001	578		16239
5	DIGITAL B DATA BYTE 1	00000100	580		16664
6	DIGITAL B DATA BYTE 2	00000110	582		16441
7	DIGITAL B DATA BYTE 3	00000000	584		16826
8	DIGITAL B DATA BYTE 4	00000000	586		14890
10	REFLECTOR 1 POSITION	10413	588	REFLECTOR 1 POSITION 18	10413
12	REFLECTOR 2 POSITION	10226	590	REFLECTOR 2 POSITION 18	10226
14	REFL 1 POS 1 2ND LOOK	10413	592	REFL 1 POS 18 2ND LOOK	10413
16	REFL 2 POS 1 2ND LOOK	10226	594	REFL 2 POS 18 2ND LOOK	10226
18	WARM CAL SAMPLE 1	16795	596	WARM CAL SAMPLE 18	16795
20		16268	598		16263
22		16854	600		16852
24		16779	602		16776
26		16486	604		16487
28		14990	606		14993
30		16337	608		16336
32		16040	610		16043
34		16241	612		16241
36		16672	614		16673
38		16443	616		16435
40		16834	618		16840
42		14890	620		14891
44	REFLECTOR 1 POSITION	10413	622	REFLECTOR 1 POSITION 19	10413
46	REFLECTOR 2 POSITION	10226	624	REFLECTOR 2 POSITION 19	10226
48	REFL 1 POS 2 2ND LOOK	10413	626	REFL 1 POS 19 2ND LOOK	10413
50	REFL 2 POS 2 2ND LOOK	10226	628	REFL 2 POS 19 2ND LOOK	10226
52	WARM CAL SAMPLE 2	16798	630	WARM CAL SAMPLE 19	16794
54		16264	632		16266
56		16853	634		16856
58		16775	636		16779
60		16486	638		16486
62		14992	640		14996
64		16333	642		16335
66		16039	644		16040
68		16239	646		16247
70		16679	648		16670
72		16428	650		16447
74		16825	652		16847
76		14891	654		14890
78	REFLECTOR 1 POSITION	10413	656	REFLECTOR 1 POSITION 20	10413
80	REFLECTOR 2 POSITION	10226	658	REFLECTOR 2 POSITION 20	10226
82	REFL 1 POS 3 2ND LOOK	10413	660	REFL 1 POS 20 2ND LOOK	10413
84	REFL 2 POS 3 2ND LOOK	10226	662	REFL 2 POS 20 2ND LOOK	10226
86	WARM CAL SAMPLE 3	16793	664	WARM CAL SAMPLE 20	16797
88		16268	666		16265
90		16854	668		16850
92		16774	670		16776

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
94	CH	16485	672	CH	16484
96	CH	14992	674	CH	14990
98	CH	16333	676	CH	16337
100	CH	16042	678	CH	16041
102	CH	16241	680	CH	16242
104	CH	16670	682	CH	16670
106	CH	16433	684	CH	16446
108	CH	16848	686	CH	16834
110	CH	14891	688	CH	14891
112	REFLECTOR 1 POSITION	10413	690	REFLECTOR 1 POSITION 21	10413
114	REFLECTOR 2 POSITION	10226	692	REFLECTOR 2 POSITION 21	10226
116	REFL 1 POS 4 2ND LOOK	10413	694	REFL 1 POS 21 2ND LOOK	10413
118	REFL 2 POS 4 2ND LOOK	10226	696	REFL 2 POS 21 2ND LOOK	10226
120	WARM CAL SAMPLE 4	16793	698	WARM CAL SAMPLE 21	16794
122	CH	16265	700	CH	16268
124	CH	16852	702	CH	16850
126	CH	16778	704	CH	16778
128	CH	16486	706	CH	16486
130	CH	14995	708	CH	14992
132	CH	16338	710	CH	16334
134	CH	16043	712	CH	16036
136	CH	16245	714	CH	16239
138	CH	16666	716	CH	16668
140	CH	16436	718	CH	16451
142	CH	16832	720	CH	16837
144	CH	14891	722	CH	14890
146	REFLECTOR 1 POSITION	10413	724	REFLECTOR 1 POSITION 22	10413
148	REFLECTOR 2 POSITION	10226	726	REFLECTOR 2 POSITION 22	10226
150	REFL 1 POS 5 2ND LOOK	10413	728	REFL 1 POS 22 2ND LOOK	10413
152	REFL 2 POS 5 2ND LOOK	10226	730	REFL 2 POS 22 2ND LOOK	10226
154	WARM CAL SAMPLE 5	16790	732	WARM CAL SAMPLE 22	16793
156	CH	16266	734	CH	16268
158	CH	16856	736	CH	16853
160	CH	16778	738	CH	16774
162	CH	16487	740	CH	16485
164	CH	14994	742	CH	14993
166	CH	16336	744	CH	16336
168	CH	16043	746	CH	16038
170	CH	16239	748	CH	16241
172	CH	16661	750	CH	16665
174	CH	16431	752	CH	16436
176	CH	16849	754	CH	16825
178	CH	14891	756	CH	14892
180	REFLECTOR 1 POSITION	10413	758	REFLECTOR 1 POSITION 23	10413
182	REFLECTOR 2 POSITION	10226	760	REFLECTOR 2 POSITION 23	10226
184	REFL 1 POS 6 2ND LOOK	10413	762	REFL 1 POS 23 2ND LOOK	10413
186	REFL 2 POS 6 2ND LOOK	10226	764	REFL 2 POS 23 2ND LOOK	10226
188	WARM CAL SAMPLE 6	16798	766	WARM CAL SAMPLE 23	16801
190	CH	16266	768	CH	16268
192	CH	16853	770	CH	16851

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
194	CH 6	16775	772	CH 6	16773
196	CH 7	16487	774	CH 7	16486
198	CH 8	14990	776	CH 8	14993
200	CH 9	16337	778	CH 9	16339
202	CH 10	16039	780	CH 10	16042
204	CH 11	16241	782	CH 11	16246
206	CH 12	16671	784	CH 12	16663
208	CH 13	16435	786	CH 13	16444
210	CH 14	16828	788	CH 14	16834
212	CH 15	14890	790	CH 15	14892
214	REFLECTOR 1 POSITION	10413	792	REFLECTOR 1 POSITION 24	10413
216	REFLECTOR 2 POSITION	10226	794	REFLECTOR 2 POSITION 24	10226
218	REFL 1 POS 7 2ND LOOK	10413	796	REFL 1 POS 24 2ND LOOK	10413
220	REFL 2 POS 7 2ND LOOK	10226	798	REFL 2 POS 24 2ND LOOK	10226
222	WARM CAL SAMPLE 7	16795	800	WARM CAL SAMPLE 24	16797
224	CH 3	16264	802	CH 3	16263
226	CH 4	16855	804	CH 4	16854
228	CH 5	16778	806	CH 5	16773
230	CH 6	16486	808	CH 6	16485
232	CH 7	14995	810	CH 7	14993
234	CH 8	16335	812	CH 8	16335
236	CH 9	16046	814	CH 9	16041
238	CH 10	16246	816	CH 10	16248
240	CH 11	16667	818	CH 11	16669
242	CH 12	16446	820	CH 12	16436
244	CH 13	16837	822	CH 13	16816
246	CH 14	14890	824	CH 14	14891
248	CH 15	10413	826	CH 15	10413
250	REFLECTOR 1 POSITION	10226	828	REFLECTOR 1 POSITION 25	10226
252	REFLECTOR 2 POSITION	10413	830	REFLECTOR 2 POSITION 25	10413
254	REFL 1 POS 8 2ND LOOK	10226	832	REFL 1 POS 25 2ND LOOK	10226
256	REFL 2 POS 8 2ND LOOK	16795	834	REFL 2 POS 25 2ND LOOK	16799
258	WARM CAL SAMPLE 8	16267	836	WARM CAL SAMPLE 25	16268
260	CH 3	16853	838	CH 3	16852
262	CH 4	16777	840	CH 4	16775
264	CH 5	16488	842	CH 5	16486
266	CH 6	14993	844	CH 6	14994
268	CH 7	16335	846	CH 7	16336
270	CH 8	16041	848	CH 8	16044
272	CH 9	16244	850	CH 9	16243
274	CH 10	16668	852	CH 10	16674
276	CH 11	16441	854	CH 11	16444
278	CH 12	16823	856	CH 12	16833
280	CH 13	14891	858	CH 13	14891
282	CH 14	10413	860	CH 14	10413
284	REFLECTOR 1 POSITION	10226	862	REFLECTOR 1 POSITION 26	10226
286	REFLECTOR 2 POSITION	10413	864	REFLECTOR 2 POSITION 26	10413
288	REFL 1 POS 9 2ND LOOK	10226	866	REFL 1 POS 26 2ND LOOK	10226
290	REFL 2 POS 9 2ND LOOK	16797	868	REFL 2 POS 26 2ND LOOK	16796
292	WARM CAL SAMPLE 9	16269	870	WARM CAL SAMPLE 26	16268

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
294	CH	16854	872	CH	16849
296	CH	16775	874	CH	16778
298	CH	16485	876	CH	16487
300	CH	14993	878	CH	14988
302	CH	16336	880	CH	16335
304	CH	16041	882	CH	16044
306	CH	16244	884	CH	16245
308	CH	16671	886	CH	16660
310	CH	16443	888	CH	16438
312	CH	16829	890	CH	16842
314	CH	14890	892	CH	14890
316	REFLECTOR 1 POSITION 10	10413	894	REFLECTOR 1 POSITION 27	10413
318	REFLECTOR 2 POSITION 10	10226	896	REFLECTOR 2 POSITION 27	10226
320	REFL 1 POS 10 2ND LOOK	10413	898	REFL 1 POS 27 2ND LOOK	10413
322	REFL 2 POS 10 2ND LOOK	10226	900	REFL 2 POS 27 2ND LOOK	10226
324	WARM CAL SAMPLE 10	16802	902	WARM CAL SAMPLE 27	16792
326	CH	16265	904	CH	16263
328	CH	16847	906	CH	16849
330	CH	16773	908	CH	16776
332	CH	16486	910	CH	16485
334	CH	14991	912	CH	14994
336	CH	16339	914	CH	16335
338	CH	16045	916	CH	16044
340	CH	16240	918	CH	16244
342	CH	16674	920	CH	16667
344	CH	16427	922	CH	16426
346	CH	16838	924	CH	16830
348	CH	14890	926	CH	14889
350	REFLECTOR 1 POSITION 11	10413	928	REFLECTOR 1 POSITION 28	10413
352	REFLECTOR 2 POSITION 11	10226	930	REFLECTOR 2 POSITION 28	10226
354	REFL 1 POS 11 2ND LOOK	10413	932	REFL 1 POS 28 2ND LOOK	10413
356	REFL 2 POS 11 2ND LOOK	10226	934	REFL 2 POS 28 2ND LOOK	10226
358	WARM CAL SAMPLE 11	16795	936	WARM CAL SAMPLE 28	16794
360	CH	16265	938	CH	16267
362	CH	16851	940	CH	16851
364	CH	16781	942	CH	16771
366	CH	16483	944	CH	16484
368	CH	14992	946	CH	14989
370	CH	16336	948	CH	16339
372	CH	16040	950	CH	16040
374	CH	16242	952	CH	16243
376	CH	16668	954	CH	16667
378	CH	16439	956	CH	16434
380	CH	16838	958	CH	16823
382	CH	14889	960	CH	14890
384	REFLECTOR 1 POSITION 12	10413	962	REFLECTOR 1 POSITION 29	10413
386	REFLECTOR 2 POSITION 12	10226	964	REFLECTOR 2 POSITION 29	10226
388	REFL 1 POS 12 2ND LOOK	10413	966	REFL 1 POS 29 2ND LOOK	10413
390	REFL 2 POS 12 2ND LOOK	10226	968	REFL 2 POS 29 2ND LOOK	10226
392	WARM CAL SAMPLE 12	16797	970	WARM CAL SAMPLE 29	16795

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
394	CH 4	16266	972	CH 4	16267
396	CH 5	16856	974	CH 5	16853
398	CH 6	16774	976	CH 6	16776
400	CH 7	16485	978	CH 7	16484
402	CH 8	14993	980	CH 8	14992
404	CH 9	16334	982	CH 9	16338
406	CH 10	16047	984	CH 10	16040
408	CH 11	16244	986	CH 11	16245
410	CH 12	16667	988	CH 12	16673
412	CH 13	16434	990	CH 13	16442
414	CH 14	16833	992	CH 14	16831
416	CH 15	14891	994	CH 15	14891
418	REFLECTOR 1 POSITION 13	10413	996	REFLECTOR 1 POSITION 30	10413
420	REFLECTOR 2 POSITION 13	10226	998	REFLECTOR 2 POSITION 30	10226
422	REFL 1 POS 13 2ND LOOK	10413	1000	REFL 1 POS 30 2ND LOOK	10413
424	REFL 2 POS 13 2ND LOOK	10226	1002	REFL 2 POS 30 2ND LOOK	10226
426	WARM CAL SAMPLE 13	16793	1004	WARM CAL SAMPLE 30	16794
428	CH 3	16269	1006	CH 3	16265
430	CH 4	16850	1008	CH 4	16853
432	CH 5	16775	1010	CH 5	16774
434	CH 6	16485	1012	CH 6	16486
436	CH 7	14992	1014	CH 7	14991
438	CH 8	16333	1016	CH 8	16335
440	CH 9	16038	1018	CH 9	16042
442	CH 10	16242	1020	CH 10	16243
444	CH 11	16667	1022	CH 11	16665
446	CH 12	16447	1024	CH 12	16439
448	CH 13	16836	1026	CH 13	16823
450	CH 14	14890	1028	CH 14	14890
452	CH 15	10413	1030	CH 15	0E
454	REFLECTOR 1 POSITION 14	10226	1032	REFLECTOR 1 COLD CAL POS	0E
456	REFLECTOR 2 POSITION 14	10226	1034	REFLECTOR 2 COLD CAL POS	0E
458	REFL 1 POS 14 2ND LOOK	10413	1036	REFL 1 COLD CAL 2ND LOOK	0E
460	REFL 2 POS 14 2ND LOOK	10226	1038	REFL 2 COLD CAL 2ND LOOK	0E
462	WARM CAL SAMPLE 14	16789	1040	COLD CAL DATA 1	0
464	CH 3	16270	1042	CH 3	0
466	CH 4	16850	1044	CH 4	0
468	CH 5	16776	1046	CH 5	0
470	CH 6	16482	1048	CH 6	0
472	CH 7	14994	1050	CH 7	0
474	CH 8	16337	1052	CH 8	0
476	CH 9	16043	1054	CH 9	0
478	CH 10	16240	1056	CH 10	0
480	CH 11	16673	1058	CH 11	0
482	CH 12	16432	1060	CH 12	0
484	CH 13	16844	1062	CH 13	0
486	CH 14	14892	1064	CH 14	0
488	CH 15	10413	1066	COLD CAL DATA 2	0
490	REFLECTOR 1 POSITION 15	10226	1068	CH 15	0
492	REFLECTOR 2 POSITION 15	10226	1070	CH 15	0
	REFL 1 POS 15 2ND LOOK	10413		CH 3	0
	REFL 2 POS 15 2ND LOOK	10226		CH 4	0
				CH 5	0

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
494	WARM CAL SAMPLE 15	16796	1072		0
496		16265	1074		0
498		16853	1076		0
500		16774	1078		0
502		16484	1080		0
504		14992	1082		0
506		16338	1084		0
508		16047	1086		0
510		16245	1088		0
512		16675	1182	REFLECTOR 1 WARM CAL POS	0E
514		16437	1184	REFLECTOR 2 WARM CAL POS	0E
516		16819	1186	REFL 1 WARM CAL 2ND LOOK	0E
518		14890	1188	REFL 2 WARM CAL 2ND LOOK	0E
520	REFLECTOR 1 POSITION 16	10413	1190	WARM CAL DATA 1	0
522	REFLECTOR 2 POSITION 16	10226	1192		0
524	REFL 1 POS 16 2ND LOOK	10413	1194		0
526	REFL 2 POS 16 2ND LOOK	10226	1196		0
528	WARM CAL SAMPLE 16	16795	1198		0
530		16265	1200		0
532		16850	1202		0
534		16777	1204		0
536		16485	1206		0
538		14990	1208		0
540		16336	1210		0
542		16041	1212		0
544		16244	1214		0
546		16674	1216		0
548		16428	1218		0
550		16835	1220	WARM CAL DATA 2	0
552		14891	1222		0
554	REFLECTOR 1 POSITION 17	10413	1224		0
556	REFLECTOR 2 POSITION 17	10226	1226		0
558	REFL 1 POS 17 2ND LOOK	10413	1228		0
560	REFL 2 POS 17 2ND LOOK	10226	1230		0
562	WARM CAL SAMPLE 17	16795	1232		0
564		16269	1234		0
566		16851	1236		0
568		16774	1238		0
570		16486	1240		0

WARM CAL MODE

ELEMENT	DESCRIPTION	VALUE	TEMPERATURE	DEG C
1090	SCAN MOTOR A1-1	18458	24.43	
1092	SCAN MOTOR A1-2	19763	25.74	
1094	FEEDHORN A1-1	20417	27.60	
1096	FEEDHORN A1-2	21461	29.60	
1098	RF MUX A1-1	22313	30.77	
1100	RF MUX A1-2	23443	32.95	
1102	LOCAL OSCILLATOR CHANNEL 3	24586	35.11	
1104	LOCAL OSCILLATOR CHANNEL 4	24646	34.89	
1106	LOCAL OSCILLATOR CHANNEL 5	24011	34.19	
1108	LOCAL OSCILLATOR CHANNEL 6	21767	30.09	
1110	LOCAL OSCILLATOR CHANNEL 7	22908	32.05	
1112	LOCAL OSCILLATOR CHANNEL 8	24226	34.33	
1114	LOCAL OSCILLATOR CHANNEL 15	24096	33.79	
1116	PLL LO #2 CHANNELS 9 THROUGH 14	15146	30.52	
1118	PLL LO #1 CHANNELS 9 THROUGH 14	25362	36.68	
1120	SPARE (NOT USED)	32767	51.27	
1122	MIXER/IF AMPLIFIER CHANNEL 3	24152	33.44	
1124	MIXER/IF AMPLIFIER CHANNEL 4	24157	33.67	
1126	MIXER/IF AMPLIFIER CHANNEL 5	23716	33.20	
1128	MIXER/IF AMPLIFIER CHANNEL 6	22518	31.24	
1130	MIXER/IF AMPLIFIER CHANNEL 7	22713	31.86	
1132	MIXER/IF AMPLIFIER CHANNEL 8	23996	33.67	
1134	MIXER/IF AMPLIFIER CH 9 THRU 14	22479	30.67	
1136	MIXER/IF AMPLIFIER CHANNEL 15	23514	33.63	
1138	IF AMPLIFIER CHANNEL 11 THRU 14	23606	33.41	
1140	IF AMPLIFIER CHANNEL 9	23634	33.52	
1142	IF AMPLIFIER CHANNEL 10	23807	33.54	
1144	IF AMPLIFIER CHANNEL 11	22333	30.99	
1146	DC/DC CONVERTER	23756	33.56	
1148	IF AMPLIFIER CHANNEL 13	22324	30.96	
1150	IF AMPLIFIER CHANNEL 14	22418	30.92	
1152	IF AMPLIFIER CHANNEL 12	22234	30.90	
1154	RF SHELF A1-1	22370	32.03	
1156	RF SHELF A1-2	23205	32.80	
1158	DETECTOR/PREAMPLIFIER ASSEMBLY	20782	28.34	
1160	A1-1 WARM LOAD 1	24248	24.80	
1162	A1-1 WARM LOAD 2	24025	24.78	
1164	A1-1 WARM LOAD 3	24234	24.82	
1166	A1-1 WARM LOAD 4	24197	24.92	
1168	A1-1 WARM LOAD CENTER	24266	26.62	
1170	A1-2 WARM LOAD 1	24973	26.56	
1172	A1-2 WARM LOAD 2	25119	26.70	
1174	A1-2 WARM LOAD 3	25286	26.70	
1176	A1-2 WARM LOAD 4	25051	26.64	
1178	A1-2 WARM LOAD CENTER	24967		
1180	TEMP SENSOR REFERENCE VOLTAGE	25331		

DESCRIPTION	STATUS	STATUS	STATUS
SCANNER A1-1 POWER	ON	ON	ON
SCANNER A1-2 POWER	ON	ON	ON
PLL POWER	PLLO # 1	PLLO # 1	PLLO # 1
ANTENNA IN WARM CAL POSITION MODE	YES	YES	YES
ANTENNA IN COLD CAL POSITION MODE	NO	NO	NO
ANTENNA IN NADIR POSITION MODE	NO	NO	NO
ANTENNA IN FULL SCAN MODE	NO	NO	NO
SURVIVAL HEATER POWER	OFF	OFF	OFF
MODULE POWER	CONNECT	CONNECT	CONNECT
COLD CAL POSITION MSB	ZERO	ZERO	ZERO
COLD CAL POSITION LSB	ZERO	ZERO	ZERO

DESCRIPTION	ANALOG DATA			
	VALUE	DEG C	VALUE	DEG C
A1-1 SCANNER MOTOR TEMPERATURE	216	25.1	216	25.1
A1-2 SCANNER MOTOR TEMPERATURE	217	26.2	217	26.2
A1-1 RF SHELF TEMPERATURE	218	27.5	219	28.8
A1-2 RF SHELF TEMPERATURE	220	30.1	220	30.1
A1-1 WARM LOAD TEMPERATURE	215	24.7	215	24.7
A1-2 WARM LOAD TEMPERATURE	217	27.7	217	27.7

DESCRIPTION	VALUE	AMPS/ VOLTS	VALUE	AMPS/ VOLTS	VALUE	AMPS/ VOLTS
A1-1 ANTENNA DRIVE MOTOR	55	25.63	48	22.37	42	AMPS/ VOLTS
A1-2 ANTENNA DRIVE MOTOR	54	25.16	47	21.90	40	19.57
SIGNAL PROCESSING +15 VDC	170	14.99	170	14.99	170	18.64
SIGNAL DRIVE +15 VDC	170	14.99	170	14.99	169	14.99
SIGNAL PROCESSING -15 VDC	147	-15.05	147	-15.05	147	14.90
SIGNAL DRIVE -15 VDC	147	-15.00	147	-15.00	147	-15.05
ANTENNA DRIVE +8 VDC	156	8.00	156	8.00	156	-15.00
RECEIVER AMPLIFIER +5 VDC	144	4.99	144	4.99	144	8.00
SIGNAL PROCESSOR +5 VDC	145	4.99	145	4.99	145	4.99
ANTENNA DRIVE +5 VDC	168	9.89	168	9.89	168	4.99
RECEIVER MIXER/IF +10 VDC	168	14.99	168	14.99	168	9.89
PHASE LOCK LOOP { CHANNEL 9/14 }	143	-15.00	143	-15.00	143	14.99
PHASE LOCK LOOP { CHANNEL 9/14 }	171	-9.99	171	-9.99	171	-15.00
{ CHANNEL 8 }	170	9.94	170	9.94	170	-9.99
{ CHANNEL 7 }	171	9.97	171	9.97	171	9.94
{ CHANNEL 6 }	171	9.98	171	9.98	171	9.97
{ CHANNEL 3 }	171	9.99	171	9.99	171	9.98
{ CHANNEL 4 }	171	9.99	171	9.99	171	9.99
{ CHANNEL 5 }	170	9.95	170	9.95	170	9.95
L.O. VOLTAGE	4	0.08	4	0.08	4	0.08
L.O. VOLTAGE	219	4.38	219	4.38	219	0.38
L.O. VOLTAGE	169	14.90	169	14.90	169	14.90
L.O. VOLTAGE	169	14.90	169	14.90	169	14.90
PLLO # 2 LOCK DETECT						
PLLO # 1 LOCK DETECT						
VOLTAGE (CHANNEL 15)						

PRT TEMPERATURES

VARIABLE TARGET

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
615	42.00	601	14.00
616	43.00	602	15.00
617	44.00	603	16.00
618	45.00	604	17.00
619	46.00	605	18.00
620	47.00	606	19.00
621	48.00	607	20.00
622	49.00	608	21.00
623	50.00	609	22.00
624	51.00	610	23.00
625	52.00	611	24.00
626	53.00	612	25.00
627	67.00	613	69.00
628	68.00	614	70.00
629	71.00	630	72.00
631	26.00	632	27.00

FIXED TARGET

BASEPLATE

THERMOCOUPLE TEMPERATURES

FIXED TARGET SHROUD

VARIABLE TARGET SHROUD

FIXED TARGET N2

VARIABLE TARGET N2

HEATER N2

FIXED TARGET FLOW METER
VARIABLE TARGET FLOW METER
BASEPLATE HEATER N2
BASEPLATE N2
BASEPLATE FLOW METER
ADJUNCT RADIATORS

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
558	5.00	537	34.00
559	6.00	538	35.00
550	7.00	524	36.00
551	8.00	525	37.00
506	57.00	502	30.00
507	58.00	503	31.00
516	59.00	511	32.00
517	60.00	512	33.00
514	1.00	509	38.00
515	2.00	510	39.00
508	63.00	504	61.00
518	64.00	513	62.00
519	3.00	520	4.00
521	9.00	522	10.00
523	65.00		
575	73.00	577	74.00
579	75.00	581	76.00

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

BP	DATA	BP	DATA	BP	DATA	BP	DATA
1	16795	9	16799	17	16795	25	16790
2	16795	10	16796	18	16791	26	16798
3	16789	11	16803	19	16795	27	16798
4	16797	12	16800	20	16793	28	16794
5	16793	13	16793	21	16796	29	16793
6	16798	14	16797	22	16796	30	16799
7	16795	15	16798	23	16793	CC	0
8	16798	16	16798	24	16791	WC	0

[21] UP

POWER [4] ON

SCREEN ONLY [2] PRINT [3] FULL
SELECT TOUCHSCREEN BUTTON 2

[1] RETURN

AMSU A1-33 A1.EXE:62 WARM CAL MODE P1 15-JUL-00 15:39:05 SCAN NUMBER 117
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

		CHANNEL		9	
BP	DATA	BP	DATA	BP	DATA
1	16331	9	16333	17	16333
2	16333	10	16336	18	16336
3	16334	11	16335	19	16335
4	16335	12	16334	20	16334
5	16333	13	16332	21	16333
6	16332	14	16336	22	16334
7	16332	15	16335	23	16331
8	16335	16	16332	24	16336
		[22] DOWN			

[21] UP

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL
SELECT TOUCHSCREEN BUTTON 2 [1] RETURN

AMSU A1-33 A1.EXE:62 WARM CAL MODE P1 15-JUL-00 15:39:38 SCAN NUMBER 121

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
		1		2		3		4	
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	10413	10413	9	10413	10413	17	10413	10413	25
2	10413	10413	10	10413	10413	18	10413	10413	26
3	10413	10413	11	10413	10413	19	10413	10413	27
4	10413	10413	12	10413	10413	20	10413	10413	28
5	10413	10413	13	10413	10413	21	10413	10413	29
6	10413	10413	14	10413	10413	22	10413	10413	30
7	10413	10413	15	10413	10413	23	10413	10413	CC
8	10413	10413	16	10413	10413	24	10413	10413	WC
[21] UP		[22] DOWN							

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 WARM CAL MODE P1 15-JUL-00 15:39:56 SCAN NUMBER 123
[5] DIGITAL A DATA ELEMENT 0000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

BP	REFLECTOR POSITIONS					
	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	10226	10226	9	10226	10226	25
2	10226	10226	10	10226	10226	26
3	10226	10226	11	10226	10226	27
4	10226	10226	12	10226	10226	28
5	10226	10226	13	10226	10226	29
6	10226	10226	14	10226	10226	30
7	10226	10226	15	10226	10226	CC
8	10226	10226	16	10226	10226	WC
[21] UP		[22] DOWN				

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 1 TO 16				DATA	TEMP C	TEMP C
	DATA	TEMP C	NO				
1	SCAN MOTOR A1-1	18432	24.38	9 LO CHANNEL 5	24060	34.28	
2	SCAN MOTOR A1-2	19745	25.70	10 LO CHANNEL 6	21795	30.15	
3	FEEDHORN A1-1	20479	27.72	11 LO CHANNEL 7	22949	32.13	
4	FEEDHORN A1-2	21541	29.75	12 LO CHANNEL 8	24275	34.42	
5	RF MUX A1-1	22356	30.85	13 LO CHANNEL 15	24142	33.88	
6	RF MUX A1-2	23495	33.05	14 PLL0 #2 CH 9/14	15183	30.59	
7	LO CHANNEL 3	24626	35.19	15 PLL0 #1 CH 9/14	25414	36.78	
8	LO CHANNEL 4	24699	35.00	16 PLL0 REFERENCE	32767	51.27	
[21] UP		[22] DOWN					

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU	A1-33 A1.EXE;62	WARM CAL MODE	P1	15-JUL-00	15:40:25	SCAN NUMBER	127
[5]	DIGITAL A DATA	ELEMENT 0000					
[6]	DIGITAL B DATA	ELEMENT 00					
[7]	ANALOG DATA	ELEMENT 00					

[illegible]

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POWER [ 4 ] ON SCREEN ONLY [ 2 ] PRINT [ 3 ] FULL [ 1 ] RETURN
SELECT TOUCHSCREEN BUTTON 2
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P1 15-JUL-00 15:40:37 SCAN NUMBER 128

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AMSU  A1-33 A1.EXE:62 WARM CAL MODE
[ 5 ] DIGITAL A DATA ELEMENT 0000

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[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

	DIGITAL A	TEMPERATURES	NO	DATA	TEMP C	TEMP C
NO						
31	IF AMP CH 14	22458	39	A1-1 WARM LOAD	31.00	24.85
32	IF AMP CH 12	22275	40	A1-1 WARM LOAD	30.98	24.96
33	RF SHELF A1-1	22415	41	A1-2 WARM LOAD	32.12	26.67
34	RF SHELF A1-2	23256	42	A1-2 WARM LOAD	32.90	26.61
35	DETECTOR/PREAMP	20812	43	A1-2 WARM LOAD	28.39	26.75
36	A1-1 WARM LOAD 1	24265	44	A1-2 WARM LOAD	24.83	26.76
37	A1-1 WARM LOAD 2	24039	45	A1-2 WARM LOAD	24.81	26.70
38	A1-1 WARM LOAD 3	24250	THERMAL REFERENCE		24.85	
[21] UP						
[22] DOWN						

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 WARM CAL MODE P1 15-JUL-00 15:41:05 SCAN NUMBER 132

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 1 TO 18

1	A1-1	SCANR MOTOR	216	25.38	DEG C	10	ANTENNA DRIVE	15VDC	14.96
2	A1-2	SCANR MOTOR	216	26.14	DEG C	11	SIGNAL PROCESSING	-15VDC	-15.01
3	A1-1	RF SHELF	219	29.08	DEG C	12	ANTENNA DRIVE	-15VDC	-15.01
4	A1-2	RF SHELF	220	30.87	DEG C	13	RECEIVER AMPLIFIER	8VDC	8.00
5	A1-1	WARM LOAD	216	26.23	DEG C	14	SIGNAL PROCESSOR	5 VDC	5.00
6	A1-2	WARM LOAD	217	28.66	DEG C	15	ANTENNA DRIVE	5 VDC	4.99
7	ANT A1-1	DRIVE MOTOR CURRENT				16	RECEIVER MIXER/IF	10VDC	9.92
8	ANT A1-2	DRIVE MOTOR CURRENT				17	PHASE LOCK LOOP CH9/14	15VDC	15.03
9	SIGNAL PROCESSING	+15VDC				18	PHASE LOCK LOOP CH9/14	-15VDC	-14.97
[21]		UP				[22]	DOWN		

POWER [4] ON

SCREEN ONLY [2] PRINT [3] FULL

SELECT TOUCHSCREEN BUTTON 2

[1] RETURN

AMSU A1-33 A1.EXE:62 WARM CAL MODE P1 15-JUL-00 15:41:17 SCAN NUMBER 133
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 10 TO 27

10 ANTENNA DRIVE	15VDC	14.93	19 L.O.	VOLTAGE	CH 8	10.00
11 SIGNAL PROCESSING	-15VDC	-15.01	20 L.O.	VOLTAGE	CH 7	9.97
12 ANTENNA DRIVE	-15VDC	-15.01	21 L.O.	VOLTAGE	CH 6	10.00
13 RECEIVER AMPLIFIER	8VDC	8.00	22 L.O.	VOLTAGE	CH 3	10.00
14 SIGNAL PROCESSOR	5 VDC	5.00	23 L.O.	VOLTAGE	CH 4	10.02
15 ANTENNA DRIVE	5 VDC	4.98	24 L.O.	VOLTAGE	CH 5	10.00
16 RECEIVER MIXER/IF	10VDC	9.91	25 PLLO # 2	LOCK DETECT		0.08
17 PHASE LOCK LOOP CH9/14	15VDC	15.04	26 PLLO # 1	LOCK DETECT		4.39
18 PHASE LOCK LOOP CH9/14	-15VDC	-14.96	27 L.O.	VOLTAGE	CH15	14.96
[21] UP	[22] DOWN					

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

TEST DATA SHEET 27
Digital-A Data Output Cold Cal Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.3)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1	255	255	P
	0002	Sync Sequence Byte 2	255	255	P
	0003	Sync Sequence Byte 3	255	255	P
[II]	0004	Unit I.D. and Serial N	33	*	P
[III]	0005	Digital-B Data Byte 1	8	8	P
	0006	Digital-B Data Byte 2	14	14	P
	0007	Digital-B Data Byte 3	0	0	P
	0008	Digital-B Data Byte 4	0	0	P
* AMSU A1 Identification Words (data entered in decimal system)					
			Binary	Decimal	
	AMSU-A1 S/N 101		00000001	1	
	AMSU-A1 S/N 102		00000101	5	
	AMSU-A1 S/N 103		00001001	9	
	AMSU-A1 S/N 104		00001101	13	
	AMSU-A1 S/N 105		00010001	17	
	AMSU-A1 S/N 106		00010101	21	
	AMSU-A1 S/N 107		00011001	25	
	AMSU-A1 S/N 108		00011101	29	
	AMSU-A1 S/N 109		00100001	33	

Circle Test: CPT LPT
FINAL

METSAT/AMSU-A1 System P/N IS-1331720 Shop Order: 787920 S/N: 109 OP: 0830

Joseph Sampson 1-17-00 Date
Customer Representative
(Flight Hardware Only)

Ray Hubbard 7-15-00 Date
Test Systems Engineer

Judith Harvey 7/15/00 Date
Quality Control

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TEST DATA SHEET 28 (Sheet 1 of 2)
Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV] (Paragraphs 3.2.4.3.4.2, 3.2.4.3.4.3, and 3.2.4.3.4.4)

BP	A1-1 Reflector			
	Para No.	Position*	Required**	Pass/Fail
CC	3.2.4.3.4.3, Step 4			
	a.	6012	6016	P
	b.	5941	5940	P
	c.	5864	5864	P
	d.	5713	5713	P
CC = Cold Cal				
* Actual counts from computer printout. Rewriting counts on this data sheet is optional.				
** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.				

3.2.4.3.4.3, Step 4 Substep	MSB	LSB
a.	0	0
b.	0	1
c.	1	0
d.	1	1

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

OP: 0830

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Ray H. Hefner
Test Systems Engineer
Judith Hervey
Quality Control

7-15-00
Date

7-15-00
Date

TEST DATA SHEET 28 (Sheet 2 of 2)

Reflector Position Warm Cal Mode Section [IV], Reflector Position Cold Cal Mode Section [IV], and Reflector Position Nadir Mode Section [IV] (Paragraphs 3.2.4.3.4.2, 3.2.4.3.4.3, and 3.2.4.3.4.4)

BP	A1-2 Reflector			
	Para No.	Position*	Required**	Pass/Fail
CC	3.2.4.3.4.3, Step 4			
	a.	5826	5834	P
	b.	5758	5758	P
	c.	5682	5682	P
	d.	5532	5531	P
CC = Cold Cal				
<p>* Actual counts from computer printout. Rewriting counts on this data sheet is optional.</p> <p>** Required range for instrument serial number from TDS 6 of AE-26002/1 ± 10 counts. Rewriting range on this data sheet is optional.</p>				

3.2.4.3.4.3, Step 4 Substep	MSB	LSB
a.	0	0
b.	0	1
c.	1	0
d.	1	1

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

SN: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Ray Hubbug
Test Systems Engineer

7-15-00
Date

Judith Morrey
Quality Control

7-15-00
Date

P1 15-JUL-00 15:43:46 SCAN NUMBER 152

AMSU A1-33 A1.EXE;62 COLD CAL MODE
[5] DIGITAL A DATA ELEMENT 0000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

COMMANDS
[9] MODULE POWER = CONNECT ANTENNA IN COLD CAL POSIT = YES [15]
[10] SURVIVAL HEATER POWER = OFF ANTENNA IN NADIR POSITION = NO [16]
[11] MODULE TOTALLY OFF = ON ANTENNA IN FULL SCAN MODE = NO [17]
[12] SCANNER A1 - 1 POWER = ON PLL POWER = PLLO # 1 [18]
[13] SCANNER A1 - 2 POWER = ON COLD CAL POSITION MSB = ZERO [19]
[14] ANTENNA IN WARM CAL POSIT = NO COLD CAL POSITION LSB = ZERO [20]
POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
ERROR ... BUTTON ENTRY OUT OF RANGE ... TRY AGAIN (0001 TO 23) 3

COLD CAL

32.4.34.3

FINAL CPT

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

TEST END: 7-15-00
QUALITY: 7/15/00

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
1	SYNC SEQUENCE BYTE 1	11111111	572	COLD CAL SAMPLE 17	CH 8
2	SYNC SEQUENCE BYTE 2	11111111	574		CH 9
3	SYNC SEQUENCE BYTE 3	11111111	576		CH 10
4	UNIT ID AND SERIAL NO	00100001	578		CH 11
5	DIGITAL B DATA BYTE 1	00001000	580		CH 12
6	DIGITAL B DATA BYTE 2	00001110	582		CH 13
7	DIGITAL B DATA BYTE 3	00000000	584		CH 14
8	DIGITAL B DATA BYTE 4	00000000	586		CH 15
10	REFLECTOR 1 POSITION	6012	588	REFLECTOR 1 POSITION 18	6012
12	REFLECTOR 2 POSITION	5826	590	REFLECTOR 2 POSITION 18	5826
14	REFL 1 POS 1 2ND LOOK	6012	592	REFL 1 POS 18 2ND LOOK	6012
16	REFL 2 POS 1 2ND LOOK	5826	594	REFL 2 POS 18 2ND LOOK	5826
18	COLD CAL SAMPLE 1	16738	596	COLD CAL SAMPLE 18	16743
20		16203	598		16205
22		16788	600		16788
24		16738	602		16737
26		16446	604		16448
28		14940	606		14942
30		16300	608		16298
32		16000	610		16003
34		16201	612		16191
36		16617	614		16620
38		16393	616		16387
40		16766	618		16795
42		14865	620		14866
44	REFLECTOR 1 POSITION	6012	622	REFLECTOR 1 POSITION 19	6012
46	REFLECTOR 2 POSITION	5826	624	REFLECTOR 2 POSITION 19	5826
48	REFL 1 POS 2 2ND LOOK	6012	626	REFL 1 POS 19 2ND LOOK	6011
50	REFL 2 POS 2 2ND LOOK	5826	628	REFL 2 POS 19 2ND LOOK	5826
52	COLD CAL SAMPLE 2	16744	630	COLD CAL SAMPLE 19	16741
54		16204	632		16205
56		16787	634		16786
58		16739	636		16733
60		16448	638		16446
62		14938	640		14943
64		16298	642		16300
66		16002	644		15999
68		16195	646		16191
70		16620	648		16619
72		16374	650		16397
74		16809	652		16789
76		14864	654		14866
78	REFLECTOR 1 POSITION	6011	656	REFLECTOR 1 POSITION 20	6012
80	REFLECTOR 2 POSITION	5826	658	REFLECTOR 2 POSITION 20	5826
82	REFL 1 POS 3 2ND LOOK	6012	660	REFL 1 POS 20 2ND LOOK	6012
84	REFL 2 POS 3 2ND LOOK	5826	662	REFL 2 POS 20 2ND LOOK	5826
86	COLD CAL SAMPLE 3	16741	664	COLD CAL SAMPLE 20	16740
88		16206	666		16205
90		16789	668		16786
92		16735	670		16737

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
94	CH	16450	672	CH	16448
96	CH	14940	674	CH	14941
98	CH	16297	676	CH	16300
100	CH	16005	680	CH	15999
102	CH	16196	682	CH	16199
104	CH	16626	684	CH	16625
106	CH	16389	686	CH	16391
108	CH	16787	688	CH	16783
110	CH	14866	690	CH	14865
112	REFLECTOR 1 POSITION	6012	692	REFLECTOR 1 POSITION	6012
114	REFLECTOR 2 POSITION	5826	694	REFLECTOR 2 POSITION	5826
116	REFL 1 POS 4 2ND LOOK	6012	696	REFL 1 POS 21 2ND LOOK	6012
118	REFL 2 POS 4 2ND LOOK	5826	698	REFL 2 POS 21 2ND LOOK	5826
120	COLD CAL SAMPLE 4	16748	700	COLD CAL SAMPLE 21	16739
122	CH	16205	702	CH	16205
124	CH	16786	704	CH	16789
126	CH	16737	706	CH	16734
128	CH	16448	708	CH	16448
130	CH	14945	710	CH	14943
132	CH	16296	712	CH	16297
134	CH	16002	714	CH	16006
136	CH	16194	716	CH	16200
138	CH	16620	718	CH	16621
140	CH	16386	720	CH	16381
142	CH	16756	722	CH	16770
144	CH	14866	724	CH	14866
146	REFLECTOR 1 POSITION	6012	726	REFLECTOR 1 POSITION	6012
148	REFLECTOR 2 POSITION	5826	728	REFLECTOR 2 POSITION	5826
150	REFL 1 POS 5 2ND LOOK	6012	730	REFL 1 POS 22 2ND LOOK	6012
152	REFL 2 POS 5 2ND LOOK	5826	732	REFL 2 POS 22 2ND LOOK	5826
154	COLD CAL SAMPLE 5	16749	734	COLD CAL SAMPLE 22	16735
156	CH	16205	736	CH	16201
158	CH	16786	738	CH	16788
160	CH	16740	740	CH	16738
162	CH	16447	742	CH	16447
164	CH	14941	744	CH	14943
166	CH	16298	746	CH	16295
168	CH	15999	748	CH	15998
170	CH	16199	750	CH	16202
172	CH	16623	752	CH	16202
174	CH	16390	754	CH	16390
176	CH	16785	756	CH	16764
178	CH	14865	758	CH	14866
180	REFLECTOR 1 POSITION	6012	760	REFLECTOR 1 POSITION	6012
182	REFLECTOR 2 POSITION	5826	762	REFLECTOR 2 POSITION	5826
184	REFL 1 POS 6 2ND LOOK	6011	764	REFL 1 POS 23 2ND LOOK	6012
186	REFL 2 POS 6 2ND LOOK	5826	766	REFL 2 POS 23 2ND LOOK	5826
188	COLD CAL SAMPLE 6	16740	768	COLD CAL SAMPLE 23	16740
190	CH	16206	770	CH	16206
192	CH	16787		CH	16789

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
194	CH 6	16736	772	CH 6	16735
196	CH 7	16449	774	CH 7	16448
198	CH 8	14945	776	CH 8	14941
200	CH 9	16296	778	CH 9	16299
202	CH 10	16004	780	CH 10	16007
204	CH 11	16197	782	CH 11	16200
206	CH 12	16616	784	CH 12	16606
208	CH 13	16383	786	CH 13	16384
210	CH 14	16779	788	CH 14	16793
212	CH 15	14868	790	CH 15	14864
214	REFLECTOR 1 POSITION	6012	792	REFLECTOR 1 POSITION 24	6012
216	REFLECTOR 2 POSITION	5826	794	REFLECTOR 2 POSITION 24	5826
218	REFL 1 POS 7	6012	796	REFL 1 POS 24 2ND LOOK	6012
220	REFL 2 POS 7	5826	798	REFL 2 POS 24 2ND LOOK	5826
222	COLD CAL SAMPLE 7	16742	800	COLD CAL SAMPLE 24	16741
224	CH 3	16202	802	CH 3	16204
226	CH 4	16786	804	CH 4	16787
228	CH 5	16737	806	CH 5	16735
230	CH 6	16449	808	CH 6	16448
232	CH 7	14948	810	CH 7	14939
234	CH 8	16300	812	CH 8	16295
236	CH 9	16003	814	CH 9	16004
238	CH 10	16198	816	CH 10	16197
240	CH 11	16629	818	CH 11	16618
242	CH 12	16397	820	CH 12	16384
244	CH 13	16787	822	CH 13	16797
246	CH 14	14867	824	CH 14	14866
248	CH 15	6011	826	CH 15	6012
250	REFLECTOR 1 POSITION	5826	828	REFLECTOR 1 POSITION 25	5826
252	REFLECTOR 2 POSITION	6012	830	REFLECTOR 2 POSITION 25	6012
254	REFL 1 POS 8	5826	832	REFL 1 POS 25 2ND LOOK	5826
256	REFL 2 POS 8	16742	834	REFL 2 POS 25 2ND LOOK	16742
258	COLD CAL SAMPLE 8	16205	836	COLD CAL SAMPLE 25	16207
260	CH 3	16787	838	CH 3	16786
262	CH 4	16738	840	CH 4	16734
264	CH 5	16447	842	CH 5	16449
266	CH 6	14944	844	CH 6	14944
268	CH 7	16302	846	CH 7	16297
270	CH 8	16001	848	CH 8	16000
272	CH 9	16198	850	CH 9	16199
274	CH 10	16619	852	CH 10	16628
276	CH 11	16393	854	CH 11	16396
278	CH 12	16782	856	CH 12	16785
280	CH 13	14866	858	CH 13	14867
282	CH 14	6012	860	CH 14	6012
284	CH 15	5826	862	CH 15	5826
286	REFLECTOR 1 POSITION	6011	864	REFLECTOR 1 POSITION 26	6012
288	REFLECTOR 2 POSITION	5826	866	REFLECTOR 2 POSITION 26	5826
290	REFL 1 POS 9	16738	868	REFL 1 POS 26 2ND LOOK	16742
292	REFL 2 POS 9	16202	870	REFL 2 POS 26 2ND LOOK	16206
	COLD CAL SAMPLE 9			COLD CAL SAMPLE 26	
	CH 3			CH 3	
	CH 4			CH 4	

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
294	CH 5	16789	872	REFLECTOR 1 POSITION 27	6012
296	CH 6	16736	874	REFLECTOR 2 POSITION 27	5826
298	CH 7	16445	876	REFL 1 POS 27 2ND LOOK	6012
300	CH 8	14940	878	REFL 2 POS 27 2ND LOOK	5826
302	CH 9	16301	880	COLD CAL SAMPLE 27	16743
304	CH 10	16002	882	CH 3	16203
306	CH 11	16201	884	CH 4	16787
308	CH 12	16617	886	CH 5	16738
310	CH 13	16388	888	CH 6	16447
312	CH 14	16771	890	CH 7	14943
314	CH 15	14866	892	CH 8	16296
316	REFLECTOR 1 POSITION 10	6012	894	CH 9	16000
318	REFLECTOR 2 POSITION 10	5826	896	CH 10	16195
320	REFL 1 POS 10 2ND LOOK	6012	898	CH 11	16622
322	REFL 2 POS 10 2ND LOOK	5826	900	CH 12	16386
324	COLD CAL SAMPLE 10	16745	902	CH 13	16797
326	CH 3	16205	904	CH 14	14867
328	CH 4	16785	906	CH 15	6012
330	CH 5	16734	908	REFLECTOR 1 POSITION 28	5826
332	CH 6	16444	910	REFLECTOR 2 POSITION 28	6012
334	CH 7	14940	912	REFL 1 POS 28 2ND LOOK	6012
336	CH 8	16295	914	REFL 2 POS 28 2ND LOOK	5826
338	CH 9	16006	916	COLD CAL SAMPLE 28	16745
340	CH 10	16193	918	CH 3	16208
342	CH 11	16620	920	CH 4	16788
344	CH 12	16382	922	CH 5	16734
346	CH 13	16793	924	CH 6	16445
348	CH 14	14864	926	CH 7	14939
350	CH 15	6011	928	CH 8	16300
352	REFLECTOR 1 POSITION 11	5826	930	CH 9	16000
354	REFLECTOR 2 POSITION 11	6012	932	CH 10	16194
356	REFL 1 POS 11 2ND LOOK	5826	934	CH 11	16620
358	REFL 2 POS 11 2ND LOOK	16740	936	CH 12	16387
360	COLD CAL SAMPLE 11	16206	938	CH 13	16797
362	CH 3	16789	940	CH 14	14866
364	CH 4	16740	942	CH 15	6012
366	CH 5	16447	944	REFLECTOR 1 POSITION 29	5826
368	CH 6	14942	946	REFLECTOR 2 POSITION 29	6012
370	CH 7	16297	948	REFL 1 POS 29 2ND LOOK	5826
372	CH 8	16003	950	REFL 2 POS 29 2ND LOOK	16744
374	CH 9	16196	952	CH 3	
376	CH 10	16612	954	CH 4	
378	CH 11	16390	956	CH 5	
380	CH 12	16766	958	CH 6	
382	CH 13	14865	960	CH 7	
384	CH 14	6012	962	CH 8	
386	CH 15	5826	964	CH 9	
388	REFLECTOR 1 POSITION 12	6012	966	CH 10	
390	REFLECTOR 2 POSITION 12	6012	968	CH 11	
392	REFL 1 POS 12 2ND LOOK	5826	970	CH 12	
	REFL 2 POS 12 2ND LOOK	16742		CH 13	
	COLD CAL SAMPLE 12			CH 14	
	CH 3			CH 15	

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
394	CH 4	16206	972	CH 4	16202
396	CH 5	16789	974	CH 5	16788
398	CH 6	16734	976	CH 6	16734
400	CH 7	16449	978	CH 7	16449
402	CH 8	14937	980	CH 8	14941
404	CH 9	16296	982	CH 9	16297
406	CH 10	16005	984	CH 10	16002
408	CH 11	16196	986	CH 11	16199
410	CH 12	16624	988	CH 12	16619
412	CH 13	16375	990	CH 13	16389
414	CH 14	16772	992	CH 14	16769
416	CH 15	14865	994	CH 15	14865
418	REFLECTOR 1 POSITION 13	6011	996	REFLECTOR 1 POSITION 30	6012
420	REFLECTOR 2 POSITION 13	5826	998	REFLECTOR 2 POSITION 30	5826
422	REFL 1 POS 13 2ND LOOK	6012	1000	REFL 1 POS 30 2ND LOOK	6012
424	REFL 2 POS 13 2ND LOOK	5826	1002	REFL 2 POS 30 2ND LOOK	5826
426	COLD CAL SAMPLE 13	16742	1004	COLD CAL SAMPLE 30	16742
428	CH 3	16205	1006	CH 3	16201
430	CH 4	16788	1008	CH 4	16790
432	CH 5	16738	1010	CH 5	16735
434	CH 6	16451	1012	CH 6	16448
436	CH 7	14941	1014	CH 7	14941
438	CH 8	16299	1016	CH 8	16301
440	CH 9	16002	1018	CH 9	16006
442	CH 10	16198	1020	CH 10	16199
444	CH 11	16624	1022	CH 11	16618
446	CH 12	16388	1024	CH 12	16382
448	CH 13	16793	1026	CH 13	16788
450	CH 14	14865	1028	CH 14	14865
452	CH 15	6012	1030	CH 15	6012
454	REFLECTOR 1 POSITION 14	5826	1032	REFLECTOR 1 COLD CAL POS	0E
456	REFLECTOR 2 POSITION 14	6012	1034	REFLECTOR 2 COLD CAL POS	0E
458	REFL 1 POS 14 2ND LOOK	5826	1036	REFL 1 COLD CAL 2ND LOOK	0E
460	REFL 2 POS 14 2ND LOOK	16737	1038	REFL 2 COLD CAL 2ND LOOK	0E
462	COLD CAL SAMPLE 14	16202	1040	COLD CAL DATA 1	0
464	CH 3	16788	1042	CH 3	0
466	CH 4	16735	1044	CH 4	0
468	CH 5	16448	1046	CH 5	0
470	CH 6	14938	1048	CH 6	0
472	CH 7	16299	1050	CH 7	0
474	CH 8	16002	1052	CH 8	0
476	CH 9	16190	1054	CH 9	0
478	CH 10	16616	1056	CH 10	0
480	CH 11	16377	1058	CH 11	0
482	CH 12	16790	1060	CH 12	0
484	CH 13	14865	1062	CH 13	0
486	CH 14	6012	1064	CH 14	0
488	CH 15	5826	1066	CH 15	0
490	REFLECTOR 1 POSITION 15	6012	1068	REFLECTOR 2 POSITION 15	0
492	REFLECTOR 2 POSITION 15	5826	1070	REFL 1 POS 15 2ND LOOK	0
	REFL 2 POS 15 2ND LOOK			REFL 2 POS 15 2ND LOOK	0
				COLD CAL DATA 2	

AMSU A1_33 A1.EXE;62		DIGITAL A DATA		15-JUL-00		15:43:49		PAGE		6	
		COLD CAL MODE									
ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE						
494	COLD CAL SAMPLE 15	16737	1072			CH	7				
496		16206	1074			CH	8				
498		16788	1076			CH	9				
500		16741	1078			CH	10				
502		16450	1080			CH	11				
504		14941	1082			CH	12				
506		16295	1084			CH	13				
508		16002	1086			CH	14				
510		16194	1088			CH	15				
512		16618	1182	REFLECTOR 1 WARM CAL POS		CH	POS			OE	
514		16385	1184	REFLECTOR 2 WARM CAL POS		CH	POS			OE	
516		16794	1186	REFL 1 WARM CAL 2ND LOOK		CH	2ND LOOK			OE	
518		14865	1188	REFL 2 WARM CAL 2ND LOOK		CH	2ND LOOK			OE	
520	REFLECTOR 1 POSITION 16	6012	1190	WARM CAL DATA 1		CH	3				
522	REFLECTOR 2 POSITION 16	5826	1192			CH	4				
524	REFL 1 POS 16 2ND LOOK	6012	1194			CH	5				
526	REFL 2 POS 16 2ND LOOK	5826	1196			CH	6				
528	COLD CAL SAMPLE 16	16745	1198			CH	7				
530		16205	1200			CH	8				
532		16787	1202			CH	9				
534		16739	1204			CH	10				
536		16449	1206			CH	11				
538		14938	1208			CH	12				
540		16296	1210			CH	13				
542		16005	1212			CH	14				
544		16192	1214			CH	15				
546		16620	1216	WARM CAL DATA 2		CH	3				
548		16372	1218			CH	4				
550		16780	1220			CH	5				
552		14865	1222			CH	6				
554	REFLECTOR 1 POSITION 17	6012	1224			CH	7				
556	REFLECTOR 2 POSITION 17	5826	1226			CH	8				
558	REFL 1 POS 17 2ND LOOK	6012	1228			CH	9				
560	REFL 2 POS 17 2ND LOOK	5826	1230			CH	10				
562	COLD CAL SAMPLE 17	16740	1232			CH	11				
564		16205	1234			CH	12				
566		16789	1236			CH	13				
568		16736	1238			CH	14				
570		16449	1240			CH	15				

ELEMENT	DESCRIPTION	VALUE	TEMPERATURE	DEG C
1090	SCAN MOTOR A1-1	18432	24.38	8
1092	SCAN MOTOR A1-2	19733	25.68	
1094	FEEDHORN A1-1	20500	27.76	
1096	FEEDHORN A1-2	21566	29.80	
1098	RF MUX A1-1	22385	30.91	
1100	RF MUX A1-2	23532	33.12	
1102	LOCAL OSCILLATOR CHANNEL 3	24654	35.24	
1104	LOCAL OSCILLATOR CHANNEL 4	24734	35.06	
1106	LOCAL OSCILLATOR CHANNEL 5	24079	34.32	
1108	LOCAL OSCILLATOR CHANNEL 6	21815	30.19	
1110	LOCAL OSCILLATOR CHANNEL 7	22977	32.19	
1112	LOCAL OSCILLATOR CHANNEL 8	24308	34.48	
1114	LOCAL OSCILLATOR CHANNEL 15	24175	33.95	
1116	PLL LO #2 CHANNELS 9 THROUGH 14	15212	30.64	
1118	PLL LO #1 CHANNELS 9 THROUGH 14	25447	36.84	
1120	SPARE (NOT USED)	32767	51.27	
1122	MIXER/IF AMPLIFIER CHANNEL 3	24237	33.61	
1124	MIXER/IF AMPLIFIER CHANNEL 4	24243	33.83	
1126	MIXER/IF AMPLIFIER CHANNEL 5	23799	33.36	
1128	MIXER/IF AMPLIFIER CHANNEL 6	22587	31.37	
1130	MIXER/IF AMPLIFIER CHANNEL 7	22785	32.00	
1132	MIXER/IF AMPLIFIER CHANNEL 8	24084	33.84	
1134	MIXER/IF AMPLIFIER CH 9 THRU 14	22546	30.80	
1136	MIXER/IF AMPLIFIER CHANNEL 15	23590	33.77	
1138	IF AMPLIFIER CHANNEL 11 THRU 14	23684	33.56	
1140	IF AMPLIFIER CHANNEL 9	23714	33.68	
1142	IF AMPLIFIER CHANNEL 10	23887	33.70	
1144	IF AMPLIFIER CHANNEL 11	22399	31.12	
1146	DC/DC CONVERTER	23821	33.68	
1148	IF AMPLIFIER CHANNEL 13	22388	31.09	
1150	IF AMPLIFIER CHANNEL 14	22484	31.05	
1152	IF AMPLIFIER CHANNEL 12	22298	31.02	
1154	RF SHELF A1-1	22443	32.17	
1156	RF SHELF A1-2	23289	32.96	
1158	DETECTOR/PREAMPLIFIER ASSEMBLY	20832	28.43	
1160	A1-1 WARM LOAD 1	24272	24.85	
1162	A1-1 WARM LOAD 2	24051	24.83	
1164	A1-1 WARM LOAD 3	24259	24.87	
1166	A1-1 WARM LOAD 4	24222	24.87	
1168	A1-1 WARM LOAD CENTER	24291	24.97	
1170	A1-2 WARM LOAD 1	25011	26.70	
1172	A1-2 WARM LOAD 2	25158	26.64	
1174	A1-2 WARM LOAD 3	25324	26.77	
1176	A1-2 WARM LOAD 4	25088	26.78	
1178	A1-2 WARM LOAD CENTER	25003	26.72	
1180	TEMP SENSOR REFERENCE VOLTAGE	25331		

DESCRIPTION	STATUS	STATUS	STATUS
SCANNER A1-1 POWER	ON	ON	ON
SCANNER A1-2 POWER	ON	ON	ON
PLL POWER	PLLO # 1	PLLO # 1	PLLO # 1
ANTENNA IN WARM CAL POSITION MODE	NO	NO	NO
ANTENNA IN COLD CAL POSITION MODE	YES	YES	YES
ANTENNA IN NADIR POSITION MODE	NO	NO	NO
ANTENNA IN FULL SCAN MODE	NO	NO	NO
SURVIVAL HEATER POWER	OFF	OFF	OFF
MODULE POWER	CONNECT	CONNECT	CONNECT
COLD CAL POSITION MSB	ZERO	ZERO	ZERO
COLD CAL POSITION LSB	ZERO	ZERO	ZERO

DESCRIPTION	VALUE	DEG C	VALUE	DEG C	VALUE	DEG C
A1-1 SCANNER MOTOR TEMPERATURE	216	25.1	216	25.1	216	25.1
A1-2 SCANNER MOTOR TEMPERATURE	217	26.2	217	26.2	217	26.2
A1-1 RF SHELF TEMPERATURE	219	28.8	219	28.8	219	28.8
A1-2 RF SHELF TEMPERATURE	221	31.5	221	31.5	221	31.5
A1-1 WARM LOAD TEMPERATURE	215	24.7	215	24.7	215	24.7
A1-2 WARM LOAD TEMPERATURE	217	27.7	217	27.7	217	27.7

DESCRIPTION	VALUE	AMPS/ VOLTS	VALUE	AMPS/ VOLTS	VALUE	AMPS/ VOLTS
A1-1 ANTENNA DRIVE MOTOR CURRENT (AVRG)	4	1.86	4	1.86	4	1.86
A1-2 ANTENNA DRIVE MOTOR CURRENT (AVRG)	4	1.86	4	1.86	4	1.86
SIGNAL PROCESSING +15 VDC	170	14.99	170	14.99	170	14.99
ANTENNA DRIVE +15 VDC	169	14.90	169	14.90	169	14.90
SIGNAL PROCESSING -15 VDC	147	-15.05	147	-15.05	147	-15.05
ANTENNA DRIVE -15 VDC	146	-15.05	146	-15.05	146	-15.05
RECEIVER AMPLIFIER +8 VDC	156	8.00	156	8.00	156	8.00
SIGNAL PROCESSOR +5 VDC	144	4.96	144	4.96	144	4.96
ANTENNA DRIVE +5 VDC	144	4.96	144	4.96	144	4.96
RECEIVER MIXER/IF +10 VDC	168	9.89	168	9.89	168	9.89
PHASE LOCK LOOP (CHANNEL 9/14)	168	14.99	168	14.99	168	14.99
PHASE LOCK LOOP (CHANNEL 9/14)	143	-15.00	143	-15.00	143	-15.00
PHASE LOCK LOOP (CHANNEL 8)	171	9.99	171	9.99	171	9.99
L.O. VOLTAGE (CHANNEL 7)	170	9.94	170	9.94	170	9.94
L.O. VOLTAGE (CHANNEL 6)	171	9.97	171	9.97	171	9.97
L.O. VOLTAGE (CHANNEL 3)	171	9.98	171	9.98	171	9.98
L.O. VOLTAGE (CHANNEL 4)	171	9.98	171	9.98	171	9.98
L.O. VOLTAGE (CHANNEL 5)	170	9.95	170	9.95	170	9.95
PLLO # 2 LOCK DETECT	4	0.08	4	0.08	4	0.08
PLLO # 1 LOCK DETECT	219	4.38	219	4.38	219	4.38
L.O. VOLTAGE (CHANNEL 15)	169	14.90	169	14.90	169	14.90

PRT TEMPERATURES

VARIABLE TARGET

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
615	42.00	601	14.00
616	43.00	602	15.00
617	44.00	603	16.00
618	45.00	604	17.00
619	46.00	605	18.00
620	47.00	606	19.00
621	48.00	607	20.00
622	49.00	608	21.00
623	50.00	609	22.00
624	51.00	610	23.00
625	52.00	611	24.00
626	53.00	612	25.00
627	67.00	613	69.00
628	68.00	614	70.00
629	71.00	630	72.00
631	26.00	632	27.00

FIXED TARGET

BASEPLATE

THERMOCOUPLE TEMPERATURES

FIXED TARGET SHROUD

VARIABLE TARGET SHROUD

FIXED TARGET N2

VARIABLE TARGET N2

HEATER N2

FIXED TARGET FLOW METER

VARIABLE TARGET FLOW METER

BASEPLATE HEATER N2

BASEPLATE N2

BASEPLATE FLOW METER

ADJUNCT RADIATORS

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
558	5.00	537	34.00
559	6.00	538	35.00
550	7.00	524	36.00
551	8.00	525	37.00
506	57.00	502	30.00
507	58.00	503	31.00
516	59.00	511	32.00
517	60.00	512	33.00
514	1.00	509	38.00
515	2.00	510	39.00
508	63.00	504	61.00
518	64.00	513	62.00
519	3.00	520	4.00
521	9.00	522	10.00
523	65.00		
575	73.00	577	74.00
579	75.00	581	76.00

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
		1		2		1		2	
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	6011	6012	9	6012	6012	17	6011	6012	25
2	6012	6012	10	6011	6012	18	6012	6012	26
3	6012	6012	11	6012	6012	19	6012	6012	27
4	6012	6012	12	6012	6012	20	6012	6012	28
5	6012	6012	13	6012	6011	21	6011	6012	29
6	6012	6011	14	6012	6012	22	6012	6012	30
7	6012	6012	15	6012	6011	23	6012	6012	CC
8	6011	6012	16	6012	6012	24	6012	6012	WC
[21] UP		[22] DOWN							

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

M5B 45B
O O

AMSU A1-33 A1.EXE;62 COLD CAL MODE P1 15-JUL-00 15:46:13 SCAN NUMBER 170
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	5826	5826	9	5826	5826	17	5826	5826	25
2	5826	5826	10	5826	5826	18	5826	5826	26
3	5826	5826	11	5826	5826	19	5826	5826	27
4	5826	5826	12	5826	5826	20	5826	5826	28
5	5826	5826	13	5826	5826	21	5826	5826	29
6	5826	5826	14	5826	5826	22	5826	5826	30
7	5826	5826	15	5826	5826	23	5826	5826	CC
8	5826	5826	16	5826	5826	24	5826	5826	WC
[21] UP				[22] DOWN					

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

M 0
0 0

AMSU A1-33 A1 EXE:62 COLD CAL MODE P1 15-JUL-00 15:49:20 SCAN NUMBER 194
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
1									
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	5941	5941	9	5941	5941	17	5941	5942	25
2	5941	5941	10	5941	5942	18	5942	5942	26
3	5941	5942	11	5942	5942	19	5942	5941	27
4	5941	5942	12	5942	5941	20	5941	5941	28
5	5941	5942	13	5941	5941	21	5941	5941	29
6	5942	5941	14	5941	5941	22	5941	5942	30
7	5941	5941	15	5941	5941	23	5941	5942	CC
8	5941	5941	16	5941	5942	24	5941	5941	WC
[21] UP				[22] DOWN					

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSA 0 / 256

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:49:30 SCAN NUMBER 195
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

		REFLECTOR POSITIONS									
		2				2					
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2
1	5758	5758	9	5758	5758	17	5759	5758	25	5758	5759
2	5758	5758	10	5758	5758	18	5759	5758	26	5758	5759
3	5758	5758	11	5758	5759	19	5758	5758	27	5758	5759
4	5759	5758	12	5758	5759	20	5758	5758	28	5758	5758
5	5759	5758	13	5758	5758	21	5758	5758	29	5758	5758
6	5758	5758	14	5758	5758	22	5758	5758	30	5758	5758
7	5758	5758	15	5758	5758	23	5758	5758	CC	0	0
8	5758	5758	16	5758	5758	24	5758	5758	WC	0	0
[21] UP				[22] DOWN							

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

M L
D)

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS									
		1		2		1		2	
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	5864	5864	9	5864	5864	17	5864	5864	25
2	5864	5864	10	5864	5864	18	5864	5864	26
3	5864	5864	11	5864	5865	19	5865	5864	27
4	5864	5864	12	5864	5864	20	5864	5865	28
5	5864	5865	13	5865	5864	21	5864	5864	29
6	5864	5864	14	5864	5865	22	5865	5864	30
7	5864	5864	15	5864	5864	23	5864	5864	CC
8	5864	5865	16	5865	5864	24	5864	5864	WC
[21] UP		[22] DOWN							

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSB / LSB
0

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:50:52 SCAN NUMBER 205
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

		REFLECTOR POSITIONS 2									
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2
1	5681	5681	9	5681	5682	17	5681	5681	25	5681	5681
2	5681	5682	10	5681	5681	18	5682	5681	26	5681	5681
3	5681	5681	11	5681	5681	19	5682	5682	27	5681	5681
4	5681	5681	12	5681	5681	20	5681	5681	28	5682	5682
5	5681	5681	13	5682	5681	21	5681	5681	29	5681	5681
6	5681	5681	14	5681	5682	22	5681	5681	30	5681	5681
7	5682	5681	15	5681	5681	23	5682	5682	CC	0	0
8	5682	5682	16	5681	5681	24	5681	5681	WC	0	0

[21] UP [22] DOWN

POWER [4] ON
SELECT TOUCHSCREEN BUTTON 2 SCREEN ONLY [2] PRINT [3] FULL [1] RETURN

M L
1 0

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:52:32 SCAN NUMBER 218

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

REFLECTOR POSITIONS ¹									
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP
1	5714	5713	9	5714	5713	17	5713	5714	25
2	5713	5713	10	5714	5713	18	5713	5714	26
3	5713	5714	11	5713	5714	19	5713	5714	27
4	5713	5713	12	5714	5713	20	5713	5713	28
5	5713	5713	13	5714	5713	21	5713	5714	29
6	5713	5713	14	5713	5713	22	5713	5713	30
7	5713	5714	15	5713	5713	23	5713	5714	CC
8	5713	5714	16	5713	5714	24	5713	5714	WC
[21] UP [22] DOWN									

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSB

LSB

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AMSU	A1-33 A1.EXE:62	COLD CAL MODE	P1	15-JUL-00	15:52:44	SCAN NUMBER
[5]	DIGITAL A DATA	ELEMENT 0000				219

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

BP	REFLECTOR POSITIONS										
	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2
1	532	532	9	532	532	17	532	532	25	532	532
2	532	532	10	532	532	18	532	532	26	532	532
3	532	532	11	532	532	19	532	532	27	532	532
4	532	532	12	532	532	20	532	532	28	532	532
5	532	532	13	532	532	21	532	532	29	532	532
6	532	532	14	532	532	22	532	532	30	532	532
7	532	532	15	532	532	23	532	532	CC	0	0
8	532	532	16	532	532	24	532	532	WC	0	0
[21]	UP			[22]	DOWN						

[21] UP [22] DOWN

POWER [4] ON

POWER [1] ON SCREEN ONLY [2] PRINT [3] FULL
SELECT TOUCHSCREEN BUTTON 2

[1] RETURN

SELECT TOUCHSCREEN BUTTON 2

MSB 1
LSB 1

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:47:34 SCAN NUMBER 181
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 1 TO 18

1	A1-1	SCANR MOTOR	216	25.73	DEG C	10	ANTENNA DRIVE	15VDC	14.96
2	A1-2	SCANR MOTOR	217	26.50	DEG C	11	SIGNAL PROCESSING	-15VDC	-15.00
3	A1-1	RF SHELF	219	29.02	DEG C	12	ANTENNA DRIVE	-15VDC	-15.01
4	A1-2	RF SHELF	221	31.53	DEG C	13	RECEIVER AMPLIFIER	8VDC	8.00
5	A1-1	WARM LOAD	215	25.97	DEG C	14	SIGNAL PROCESSOR	5 VDC	5.01
6	A1-2	WARM LOAD	217	29.00	DEG C	15	ANTENNA DRIVE	5 VDC	4.99
7	ANT A1-1	DRIVE MOTOR CURRENT			1.88	16	RECEIVER MIXER/IF	10VDC	9.92
8	ANT A1-2	DRIVE MOTOR CURRENT			1.88	17	PHASE LOCK LOOP CH9/14	15VDC	15.05
9	SIGNAL PROCESSING	+15VDC			15.01	18	PHASE LOCK LOOP CH9/14	-15VDC	-14.97
[21]	UP	[22]	DOWN						

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

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AMSU A1-33 A1-EXE:62 COLD CAL MODE P1 15-JUL-00 15:47:47 SCAN NUMBER 182
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 10 TO 27

10 ANTENNA DRIVE	15VDC	14.93	19 L.O. VOLTAGE	CH 8	10.00
11 SIGNAL PROCESSING	-15VDC	-15.00	20 L.O. VOLTAGE	CH 7	9.98
12 ANTENNA DRIVE	-15VDC	-15.01	21 L.O. VOLTAGE	CH 6	9.99
13 RECEIVER AMPLIFIER	8VDC	8.00	22 L.O. VOLTAGE	CH 3	10.00
14 SIGNAL PROCESSOR	5 VDC	5.01	23 L.O. VOLTAGE	CH 4	10.01
15 ANTENNA DRIVE	5 VDC	4.98	24 L.O. VOLTAGE	CH 5	10.00
16 RECEIVER MIXER/IF	10VDC	9.91	25 PLLO # 2 LOCK DETECT		0.08
17 PHASE LOCK LOOP CH9/14	15VDC	15.06	26 PLLO # 1 LOCK DETECT		4.39
18 PHASE LOCK LOOP CH9/14	-15VDC	-14.96	27 L.O. VOLTAGE	CH15	14.96
[21] UP	[22] DOWN				

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSB

/

LSB

/

TEST DATA SHEET 29

Digital-A Data Output Cold Cal Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.3)

Condition: Cold Cal Position MSB=0 and Cold Cal Position LSB=0

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Measured*	Required**	Pass/Fail	Element (For Ref)	Measured*	Required**	Pass/Fail
01	0018	16746	16500±4000	PASS	0030	16292	16500±4000	PASS
02	0052	16735			0064	16295		
03	0086	16736			0098	16298		
04	0120	16737			0132	16298		
05	0154	16741			0166	16296		
06	0188	16736			0200	16298		
07	0222	16745			0234	16294		
08	0256	16739			0268	16298		
09	0290	16734			0302	16296		
10	0324	16739			0336	16295		
11	0356	16739			0370	16294		
12	0392	16742			0404	16299		
13	0426	16742			0438	16296		
14	0460	16742			0472	16299		
15	0494	16738			0506	16297		
16	0528	16737			0540	16298		
17	0562	16744			0574	16301		
18	0596	16740			0608	16297		
19	0630	16746			0642	16296		
20	0664	16737			0676	16300		
21	0698	16742			0710	16299		
22	0732	16738			0744	16299		
23	0766	16739			0778	16296		
24	0800	16740			0812	16295		
25	0834	16745			0846	16294		
26	0868	16743			0880	16295		
27	0902	16739			0914	16300		
28	0936	16741			0948	16294		
29	0970	16741	↓		0982	16295	↓	
30	1004	16738	16500±4000		1016	16296	16500±4000	
CC	1038	0	0	↓	1050	0	0	↓
WC	1190	0	0	PASS	1202	0	0	PASS

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts.

Circle Test: CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op. 0830

Shop Order: 767920 S/N: 109

Customer Representative
(Flight Hardware Only)

7-17-00
Date



Test Systems Engineer

Quality Control

7/15/00

Date

7/15/00
Date

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

		CHANNEL 9			
BP	DATA	BP	DATA	BP	DATA
1	16292	9	16296	17	16301
2	16295	10	16295	18	16297
3	16298	11	16294	19	16296
4	16298	12	16299	20	16300
5	16296	13	16296	21	16299
6	16298	14	16299	22	16299
7	16294	15	16297	23	16296
8	16298	16	16298	24	16295
		[22] DOWN			

[21] UP

POWER [4] ON

SCREEN ONLY [2] PRINT [3] FULL
SELECT TOUCHSCREEN BUTTON 2

[1] RETURN

AMSU A1-33 A1.EXE;62 COLD CAL MODE P1 15-JUL-00 15:45:20 SCAN NUMBER 164
[5] DIGITAL A DATA ELEMENT 0000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA
CHANNEL 3
BP DATA BP DATA BP DATA BP DATA
1 16746 9 16734 17 16744 25 16745
2 16735 10 16739 18 16740 26 16743
3 16736 11 16739 19 16746 27 16739
4 16737 12 16742 20 16737 28 16741
5 16741 13 16742 21 16742 29 16741
6 16736 14 16742 22 16738 30 16738
7 16745 15 16738 23 16739 CC 0
8 16739 16 16737 24 16740 WC 0
[22] DOWN

[21] UP

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

TEST DATA SHEET 30 (Sheet 1 of 2)
Cold Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.3)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1	24.85	25 ± 15	Pass
1092	A1-1 Warm Load 2	24.84	25 ± 15	↑
1094	A1-1 Warm Load 3	24.88	25 ± 15	
1096	A1-1 Warm Load 4	24.88	25 ± 15	
1098	A1-1 Warm Load Center	24.98	25 ± 15	
1100	A1-2 Warm Load 1	26.72	25 ± 15	
1102	A1-2 Warm Load 2	26.66	25 ± 15	
1104	A1-2 Warm Load 3	26.79	25 ± 15	
1106	A1-2 Warm Load 4	26.80	25 ± 15	
1108	A1-2 Warm Load Center	26.74	25 ± 15	
1110	Local Oscillator Channel 7	32.04	25 ± 15	
1112	Local Oscillator Channel 8	33.89	25 ± 15	
1114	Local Oscillator Channel 15	33.82	25 ± 15	
1116	PLL LO #2 Channels 9-14	30.69 ^{***}	25 ± 15	
1118	PLL LO #1 Channels 9-14	36.89	25 ± 15	Pass
1120	PLLO (Reference Oscillator)**/ Not used ***	N/A		N/A
1122	Mixer I.F. Amp. Channel 3	33.67	25 ± 15	Pass
1124	Mixer I.F. Amp. Channel 4	33.89	25 ± 15	↑
1126	Mixer I.F. Amp. Channel 5	33.41	25 ± 15	
1128	Mixer I.F. Amp. Channel 6	31.42	25 ± 15	
1130	Mixer I.F. Amp. Channel 7	32.04	25 ± 15	
1132	Mixer I.F. Amp. Channel 8	33.89	25 ± 15	
1134	Mixer I.F. Amp. Channels 9-14	30.85	25 ± 15	
1136	Mixer I.F. Amp. Channel 15	33.82	25 ± 15	Pass

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** For S/N 101 through 104.

*** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 30 (Sheet 2 of 2)
Cold Cal Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.3)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14	33.61	25 ± 15	Pass
1140	I.F. Amp. Channel 9	33.73	25 ± 15	↑
1142	I.F. Amp. Channel 10	33.75	25 ± 15	
1144	I.F. Amp. Channel 11	31.16	25 ± 15	
1146	DC/DC Converter	33.75	25 ± 15	
1148	I.F. Amp. Channel 13	31.13	25 ± 15	
1150	I.F. Amp. Channel 14	31.09	25 ± 15	
1152	I.F. Amp. Channel 12	31.07	25 ± 15	
1154	RF Shelf A1-1	32.22	25 ± 15	
1156	RF Shelf A1-2	33.01	25 ± 15	
1158	Detector Preamp Assy.	28.46	25 ± 15	
1160	Scan Motor A1-1	24.36	25 ± 15	
1162	Scan Motor A1-2	25.66	25 ± 15	
1164	Feed Horn A1-1	27.72	25 ± 15	
1166	Feed Horn A1-2	29.72	25 ± 15	
1168	R.F. Mux A1-1	30.95	25 ± 15	
1170	R.F. Mux A1-2	33.17	25 ± 15	
1172	Local Oscillator Channel 3	35.32	25 ± 15	
1174	Local Oscillator Channel 4	35.12	25 ± 15	
1176	Local Oscillator Channel 5	34.37	25 ± 15	
1178	Local Oscillator Channel 6	30.22	25 ± 15	↓
1180	Temp Sensor Ref Voltage Count	25331	**	Pass

* Value is from the STE printout sheets. Copying data to this sheet is optional.
** = Count of 24,552 + 1765, -1308.

Circle Test: CPT LPT

Final

CP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109



Customer Representative
(Flight Hardware Only)

7-17-00

Date

Test Systems Engineer

7-15-00

Date

Quality Control

7-15-00

Date

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:46:36 SCAN NUMBER 173
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 1 TO 16			
	DATA	TEMP C	NO	DATA
1	SCAN MOTOR A1-1	18419	9	LO CHANNEL 5
2	SCAN MOTOR A1-2	19724	10	LO CHANNEL 6
3	FEEDHORN A1-1	20477	11	LO CHANNEL 7
4	FEEDHORN A1-2	21536	12	LO CHANNEL 8
5	RF MUX A1-1	22404	13	LO CHANNEL 15
6	RF MUX A1-2	23558	14	PLLO #2 CH 9/14
7	LO CHANNEL 3	24697	15	PLLO #1 CH 9/14
8	LO CHANNEL 4	24761	16	PLLO REFERENCE
[21] UP [22] DOWN				

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSB

LSB

0

1

Para: 3.2.4.3 4.3

TDS:

AMSU A1-33 A1.EXE:62 COLD CAL MODE P1 15-JUL-00 15:46:45 SCAN NUMBER 174

[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

DIGITAL A TEMPERATURES 17 TO 32			
NO	DATA	TEMP C	NO
17 MIXER IF CH 3	24268	33.67	25 IF AMP CH 11/14
18 MIXER IF CH 4	24272	33.89	26 IF AMP CH 9
19 MIXER IF CH 5	23828	33.41	27 IF AMP CH 10
20 MIXER IF CH 6	22609	31.42	28 IF AMP CH 11
21 MIXER IF CH 7	22809	32.04	29 DC/DC CONVERTER
22 MIXER IF CH 8	24110	33.89	30 IF AMP CH 13
23 MIXER IF CH 9/14	22570	30.85	31 IF AMP CH 14
24 MIXER IF CH 15	23616	33.82	32 IF AMP CH 12
[21] UP		[22] DOWN	

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

M
O
L
I

AMSU A1-33 A1.EXE;62 COLD CAL MODE P1 15-JUL-00 15:47:01 SCAN NUMBER 176
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 31 TO 46			
	DATA	TEMP C	NO	DATA
31 IF AMP CH 14	22507	31.10	39 A1-1 WARM LOAD 4	24226
32 IF AMP CH 12	22324	31.07	40 A1-1 WARM LOAD C	24297
33 RF SHELF A1-1	22469	32.22	41 A1-2 WARM LOAD 1	25022
34 RF SHELF A1-2	23316	33.01	42 A1-2 WARM LOAD 2	25165
35 DETECTOR/PREAMP	20848	28.46	43 A1-2 WARM LOAD 3	25334
36 A1-1 WARM LOAD 1	24275	24.85	44 A1-2 WARM LOAD 4	25101
37 A1-1 WARM LOAD 2	24056	24.84	45 A1-2 WARM LOAD C	25015
38 A1-1 WARM LOAD 3	24264	24.88	THERMAL REFERENCE	25331
[21] UP		[22] DOWN		

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

MSB

/

LSB

0

TEST DATA SHEET 31
Digital-A Data Output Nadir Mode Synch Sequence,
Unit I.D./Serial Number and Digital-B Serial Data Verification
Sections [I], [II], and [III] (Paragraph 3.2.4.3.4.4)

Step	Element (For Ref)	Description	Recorded Value	Required Value	Pass/Fail
[I]	0001	Sync Sequence Byte 1	255	255	PASS
	0002	Sync Sequence Byte 2	255	255	
	0003	Sync Sequence Byte 3	255	255	
[II]	0004	Unit I.D. and Serial N	33	*	
[III]	0005	Digital-B Data Byte 1	16	16	
	0006	Digital-B Data Byte 2	14	14	
	0007	Digital-B Data Byte 3	0	0	
	0008	Digital-B Data Byte 4	0	0	PASS

* AMSU A1 Identification Words
(data entered in decimal system)

Binary

Decimal

AMSU-A1 S/N 101

00000001

1

AMSU-A1 S/N 102

00000101

5

AMSU-A1 S/N 103

00001001

9

AMSU-A1 S/N 104

00001101

13

AMSU-A1 S/N 105

00010001

17

AMSU-A1 S/N 106

00010101

21

AMSU-A1 S/N 107

00011001

25

AMSU-A1 S/N 108

00011101

29

AMSU-A1 S/N 109

00100001

33

Circle Test: **CPT** LPT

FINAL

Op. 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-CP
Date

Ken Shaw
Test Systems Engineer

7/15/00
Date

Judith Horney
Quality Control

7/15/00
Date

6 Apr 99

TEST DATA SHEET 32

Digital-A Data Output Nadir Mode Radiometer Data Section [V] (Paragraph 3.2.4.3.4.4)

BP	A1-2 Channel-3 (50.3 GHz)				A1-1 Channel-9 (57.290344 GHz)			
	Element (For Ref)	Position*	Required**	Pass/Fail	Element (For Ref)	Position*	Required**	Pass/Fail
01	0018	16802	16,500 ± 4000	PASS	0030	16298	16,500 ± 4000	PASS
02	0052	16801			0064	16301		
03	0086	16806			0098	16302		
04	0120	16803			0132	16303		
05	0154	16803			0166	16297		
06	0188	16810			0200	16301		
07	0222	16810			0234	16301		
08	0256	16800			0268	16297		
09	0290	16801			0302	16298		
10	0324	16806			0336	16296		
11	0356	16809			0370	16298		
12	0392	16803			0404	16295		
13	0426	16806			0438	16298		
14	0460	16804			0472	16295		
15	0494	16802			0506	16298		
16	0528	16798			0540	16295		
17	0562	16809			0574	16302		
18	0596	16808			0608	16300		
19	0630	16803			0642	16299		
20	0664	16802			0676	16297		
21	0698	16805			0710	16301		
22	0732	16805			0744	16298		
23	0766	16812			0778	16299		
24	0800	16809			0812	16300		
25	0834	16803			0846	16301		
26	0868	16804			0880	16295		
27	0902	16800			0914	16297		
28	0936	16803			0948	16299		
29	0970	16805			0982	16302		
30	1004	16806	16,500 ± 4000		1016	16300	16,500 ± 4000	
CC	1038	0	0		1050	0	0	
WC	1190	0	0	PASS	1202	0	0	PASS

* Actual counts from computer printout. Rewriting counts on this data sheet is optional.

** Required = 16,500 ± 4000 counts (Unless otherwise indicated).

Circle Test: CPT LPT

METSAT/AMSU-AT System P/N IS-1331720

Shop Order: 787920

S/N: 109

Customer Representative
(Flight Hardware Only)

7-17-00

Date

Test Systems Engineer

Quality Control



7/15/00

Date

7/15/00

Date

TEST DATA SHEET 33 (Sheet 1 of 2)
Nadir Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.4)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1090	A1-1 Warm Load 1	24.67	25 ± 15	PASS
1092	A1-1 Warm Load 2	26.02	25 ± 15	↓
1094	A1-1 Warm Load 3	27.99	25 ± 15	
1096	A1-1 Warm Load 4	30.03	25 ± 15	
1098	A1-1 Warm Load Center	31.32	25 ± 15	
1100	A1-2 Warm Load 1	33.59	25 ± 15	↓
1102	A1-2 Warm Load 2	35.77	25 ± 15	
1104	A1-2 Warm Load 3	35.59	25 ± 15	
1106	A1-2 Warm Load 4	34.78	25 ± 15	
1108	A1-2 Warm Load Center	30.56	25 ± 15	↓
1110	Local Oscillator Channel 7	32.61	25 ± 15	
1112	Local Oscillator Channel 8	35.00	25 ± 15	
1114	Local Oscillator Channel 15	34.39	25 ± 15	
1116	PLL LO #2 Channels 9-14	31.05	25 ± 15	↓
1118	PLL LO #1 Channels 9-14	37.28	25 ± 15	PASS
1120	PLLO (Reference Oscillator)**/ Not used ***	N/A 	25 ± 15	N/A 
1122	Mixer I.F. Amp. Channel 3	34.10	25 ± 15	PASS
1124	Mixer I.F. Amp. Channel 4	34.32	25 ± 15	↓
1126	Mixer I.F. Amp. Channel 5	33.82	25 ± 15	
1128	Mixer I.F. Amp. Channel 6	31.80	25 ± 15	
1130	Mixer I.F. Amp. Channel 7	32.44	25 ± 15	
1132	Mixer I.F. Amp. Channel 8	34.35	25 ± 15	↓
1134	Mixer I.F. Amp. Channels 9-14	31.23	25 ± 15	
1136	Mixer I.F. Amp. Channel 15	34.21	25 ± 15	PASS

- * Value is from the STE printout sheets. Copying data to this sheet is optional.
 ** For S/N 101 through 104.
 *** For S/N 105 and up.

(Continued on Sheet 2)

TEST DATA SHEET 33 (Sheet 2 of 2)
Nadir Mode Temperature Sensors Section [VI] (Paragraph 3.2.4.3.4.4)

Thermistor Sensors		Recorded Value* (deg. C)	Required Value (deg. C)	Pass/ Fail
Element	Description			
1138	I.F. Amp. Channel 11-14	34.00	25 ± 15	PASS
1140	I.F. Amp. Channel 9	34.13	25 ± 15	
1142	I.F. Amp. Channel 10	34.13	25 ± 15	
1144	I.F. Amp. Channel 11	31.55	25 ± 15	
1146	DC/DC Converter	34.02	25 ± 15	
1148	I.F. Amp. Channel 13	31.52	25 ± 15	
1150	I.F. Amp. Channel 14	31.48	25 ± 15	
1152	I.F. Amp. Channel 12	31.46	25 ± 15	
1154	RF Shelf A1-1	32.57	25 ± 15	
1156	RF Shelf A1-2	33.41	25 ± 15	
1158	Detector Preamp Assy.	28.76	25 ± 15	
1160	Scan Motor A1-1	25.04	25 ± 15	
1162	Scan Motor A1-2	25.01	25 ± 15	
1164	Feed Horn A1-1	25.05	25 ± 15	
1166	Feed Horn A1-2	25.05	25 ± 15	
1168	R.F. Mux A1-1	25.16	25 ± 15	
1170	R.F. Mux A1-2	27.02	25 ± 15	
1172	Local Oscillator Channel 3	26.96	25 ± 15	
1174	Local Oscillator Channel 4	27.10	25 ± 15	
1176	Local Oscillator Channel 5	27.10	25 ± 15	
1178	Local Oscillator Channel 6	27.04	25 ± 15	↓
1180	Temp Sensor Ref Voltage Count	25331	**	PASS

* Value is from the STE printout sheets. Copying data to this sheet is optional.

** = Count of 24,552 +1765,-1308.

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Ken Shaw
Test Systems Engineer

7/15/00
Date

Judith Norvey
Quality Control

(24) 7/15/00
Date

AMSU A1-33 A1.EXE:62 NADIR MODE P1 15-JUL-00 17:10:39 SCAN NUMBER 759
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000
[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

COMMANDS
[9] MODULE POWER = CONNECT ANTENNA IN COLD CAL POSIT = NO [15]
[10] SURVIVAL HEATER POWER = OFF ANTENNA IN NADIR POSITION = YES [16]
[11] MODULE TOTALLY OFF = ON ANTENNA IN FULL SCAN MODE = NO [17]
[12] SCANNER A1 - 1 POWER = ON PLL POWER = PLL0 # 1 [18]
[13] SCANNER A1 - 2 POWER = ON COLD CAL POSITION MSB = ZERO [19]
[14] ANTENNA IN WARM CAL POSIT = NO COLD CAL POSITION LSB = ZERO [20]

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 3

324.344

TDS 31 to 33

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

FINAL CPT

TEST ENG: Ken Shan DATE: 7/15/00
QUALITY: ~~Indip. Donnell~~ 7/15/00

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
1	SYNC SEQUENCE	1	572	NADIR SAMPLE	17
2	SYNC SEQUENCE	2	574		
3	SYNC SEQUENCE	3	576		
4	UNIT ID AND SERIAL NO	1	578		
5	DIGITAL B DATA	1	580		
6	DIGITAL B DATA	2	582		
7	DIGITAL B DATA	3	584		
8	DIGITAL B DATA	4	586		
10	REFLECTOR 1 POSITION	1	588	REFLECTOR 1 POSITION	18
12	REFLECTOR 2 POSITION	1	590	REFLECTOR 2 POSITION	18
14	REFL 1 POS	1	592	REFL 1 POS	18
16	REFL 2 POS	1	594	REFL 2 POS	18
18	NADIR SAMPLE	1	596	NADIR SAMPLE	18
20		3	598		
22		4	600		
24		5	602		
26		6	604		
28		7	606		
30		8	608		
32		9	610		
34		10	612		
36		11	614		
38		12	616		
40		13	618		
42		14	620		
44	REFLECTOR 1 POSITION	2	622	REFLECTOR 1 POSITION	19
46	REFLECTOR 2 POSITION	2	624	REFLECTOR 2 POSITION	19
48	REFL 1 POS	2	626	REFL 1 POS	19
50	REFL 2 POS	2	628	REFL 2 POS	19
52	NADIR SAMPLE	2	630	NADIR SAMPLE	19
54		3	632		
56		4	634		
58		5	636		
60		6	638		
62		7	640		
64		8	642		
66		9	644		
68		10	646		
70		11	648		
72		12	650		
74		13	652		
76		14	654		
78	REFLECTOR 1 POSITION	3	656	REFLECTOR 1 POSITION	20
80	REFLECTOR 2 POSITION	3	658	REFLECTOR 2 POSITION	20
82	REFL 1 POS	3	660	REFL 1 POS	20
84	REFL 2 POS	3	662	REFL 2 POS	20
86	NADIR SAMPLE	3	664	NADIR SAMPLE	20
88		4	666		
90		5	668		
92		6	670		

NADIR MODE

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
94	CH	16445	672	CH	16450
96	CH	14986	674	CH	14987
98	CH	16302	676	CH	16297
100	CH	15998	678	CH	16002
102	CH	16166	680	CH	16167
104	CH	16595	682	CH	16598
106	CH	16358	684	CH	16360
108	CH	16733	686	CH	16713
110	CH	14853	688	CH	14852
112	REFLECTOR 1 POSITION	2155	690	REFLECTOR 1 POSITION	2154
114	REFLECTOR 2 POSITION	1969	692	REFLECTOR 2 POSITION	1969
116	REFL 1 POS 4	2155	694	REFL 1 POS 21	2154
118	REFL 2 POS 4	1969	696	REFL 2 POS 21	1969
120	NADIR SAMPLE	16803	698	NADIR SAMPLE	16805
122	CH	16303	700	CH	16299
124	CH	16871	702	CH	16870
126	CH	16738	704	CH	16739
128	CH	16450	706	CH	16444
130	CH	14988	708	CH	14986
132	CH	16303	710	CH	16301
134	CH	15999	712	CH	15998
136	CH	16165	714	CH	16170
138	CH	16588	716	CH	16587
140	CH	16362	718	CH	16367
142	CH	16747	720	CH	16722
144	CH	14853	722	CH	14852
146	REFLECTOR 1 POSITION	2155	724	REFLECTOR 1 POSITION	2154
148	REFLECTOR 2 POSITION	1969	726	REFLECTOR 2 POSITION	1969
150	REFL 1 POS 5	2154	728	REFL 1 POS 22	2154
152	REFL 2 POS 5	1969	730	REFL 2 POS 22	1969
154	NADIR SAMPLE	16803	732	NADIR SAMPLE	16805
156	CH	16300	734	CH	16301
158	CH	16868	736	CH	16870
160	CH	16740	738	CH	16741
162	CH	16445	740	CH	16447
164	CH	14986	742	CH	14985
166	CH	16297	744	CH	16298
168	CH	15996	746	CH	16003
170	CH	16165	748	CH	16165
172	CH	16588	750	CH	16592
174	CH	16354	752	CH	16355
176	CH	16722	754	CH	16738
178	CH	14853	756	CH	14852
180	REFLECTOR 1 POSITION	2154	758	REFLECTOR 1 POSITION	2154
182	REFLECTOR 2 POSITION	1969	760	REFLECTOR 2 POSITION	1969
184	REFL 1 POS 6	2154	762	REFL 1 POS 23	2154
186	REFL 2 POS 6	1969	764	REFL 2 POS 23	1969
188	NADIR SAMPLE	16810	766	NADIR SAMPLE	16812
190	CH	16301	768	CH	16301
192	CH	16870	770	CH	16867

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
194	CH 6	16739	772	CH 6	16742
196	CH 7	16447	774	CH 7	16448
198	CH 8	14985	776	CH 8	14990
200	CH 9	16301	778	CH 9	16299
202	CH 10	15998	780	CH 10	15994
204	CH 11	16171	782	CH 11	16171
206	CH 12	16586	784	CH 12	16592
208	CH 13	16354	786	CH 13	16351
210	CH 14	16723	788	CH 14	16744
212	CH 15	14852	790	CH 15	14852
214	REFLECTOR 1 POSITION	2154	792	REFLECTOR 1 POSITION	2154
216	REFLECTOR 2 POSITION	1969	794	REFLECTOR 2 POSITION	1969
218	REFL 1 POS	2154	796	REFL 1 POS	2154
220	REFL 2 POS	1969	798	REFL 2 POS	1969
222	NADIR SAMPLE	16810	800	NADIR SAMPLE	16809
224	CH 3	16302	802	CH 3	16303
226	CH 4	16868	804	CH 4	16868
228	CH 5	16741	806	CH 5	16738
230	CH 6	16446	808	CH 6	16445
232	CH 7	14982	810	CH 7	14986
234	CH 8	16301	812	CH 8	16300
236	CH 9	15998	814	CH 9	16000
238	CH 10	16167	816	CH 10	16165
240	CH 11	16593	818	CH 11	16592
242	CH 12	16376	820	CH 12	16356
244	CH 13	16747	822	CH 13	16729
246	CH 14	14853	824	CH 14	14853
248	REFLECTOR 1 POSITION	2154	826	REFLECTOR 1 POSITION	2154
250	REFLECTOR 2 POSITION	1969	828	REFLECTOR 2 POSITION	1969
252	REFL 1 POS	2154	830	REFL 1 POS	2154
254	REFL 2 POS	1969	832	REFL 2 POS	1969
256	NADIR SAMPLE	16800	834	NADIR SAMPLE	16803
258	CH 3	16301	836	CH 3	16303
260	CH 4	16865	838	CH 4	16867
262	CH 5	16738	840	CH 5	16740
264	CH 6	16446	842	CH 6	16445
266	CH 7	14985	844	CH 7	14988
268	CH 8	16297	846	CH 8	16301
270	CH 9	15999	848	CH 9	16002
272	CH 10	16169	850	CH 10	16171
274	CH 11	16596	852	CH 11	16600
276	CH 12	16346	854	CH 12	16366
278	CH 13	16735	856	CH 13	16737
280	CH 14	14852	858	CH 14	14852
282	REFLECTOR 1 POSITION	2154	860	REFLECTOR 1 POSITION	2154
284	REFLECTOR 2 POSITION	1969	862	REFLECTOR 2 POSITION	1969
286	REFL 1 POS	2154	864	REFL 1 POS	2154
288	REFL 2 POS	1969	866	REFL 2 POS	1969
290	NADIR SAMPLE	16801	868	NADIR SAMPLE	16804
292	CH 3	16301	870	CH 3	16301

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
294	CH	16868	872	CH	16866
296	CH	16741	874	CH	16741
298	CH	16448	876	CH	16447
300	CH	14985	878	CH	14991
302	CH	16298	880	CH	16295
304	CH	15997	882	CH	15997
306	CH	16166	884	CH	16165
308	CH	16592	886	CH	16594
310	CH	16364	888	CH	16358
312	CH	16742	890	CH	16741
314	CH	14851	892	CH	14852
316	REFLECTOR 1 POSITION 10	2154	894	REFLECTOR 1 POSITION 27	2154
318	REFLECTOR 2 POSITION 10	1969	896	REFLECTOR 2 POSITION 27	1969
320	REFL 1 POS 10	2154	898	REFL 1 POS 27	2155
322	REFL 2 POS 10	1969	900	REFL 2 POS 27	1969
324	NADIR SAMPLE	16806	902	NADIR SAMPLE	16800
326	CH	16298	904	CH	16297
328	CH	16866	906	CH	16870
330	CH	16739	908	CH	16739
332	CH	16446	910	CH	16449
334	CH	14983	912	CH	14991
336	CH	16296	914	CH	16297
338	CH	16002	916	CH	15993
340	CH	16167	918	CH	16168
342	CH	16592	920	CH	16587
344	CH	16374	922	CH	16349
346	CH	16735	924	CH	16729
348	CH	14852	926	CH	14852
350	REFLECTOR 1 POSITION 11	2154	928	REFLECTOR 1 POSITION 28	2155
352	REFLECTOR 2 POSITION 11	1969	930	REFLECTOR 2 POSITION 28	1969
354	REFL 1 POS 11	2154	932	REFL 1 POS 28	2155
356	REFL 2 POS 11	1969	934	REFL 2 POS 28	1969
358	NADIR SAMPLE	16809	936	NADIR SAMPLE	16803
360	CH	16303	938	CH	16301
362	CH	16871	940	CH	16869
364	CH	16739	942	CH	16739
366	CH	16450	944	CH	16449
368	CH	14986	946	CH	14988
370	CH	16298	948	CH	16299
372	CH	15995	950	CH	15996
374	CH	16162	952	CH	16163
376	CH	16595	954	CH	16590
378	CH	16360	956	CH	16361
380	CH	16732	958	CH	16734
382	CH	14852	960	CH	14851
384	REFLECTOR 1 POSITION 12	2154	962	REFLECTOR 1 POSITION 29	2155
386	REFLECTOR 2 POSITION 12	1969	964	REFLECTOR 2 POSITION 29	1969
388	REFL 1 POS 12	2154	966	REFL 1 POS 29	2155
390	REFL 2 POS 12	1969	968	REFL 2 POS 29	1969
392	NADIR SAMPLE	16803	970	NADIR SAMPLE	16805

ELEMENT	DESCRIPTION	VALUE	ELEMENT	DESCRIPTION	VALUE
394	CH 4	16302	972	CH 4	16300
396	CH 5	16865	974	CH 5	16868
398	CH 6	16739	976	CH 6	16739
400	CH 7	16445	978	CH 7	16446
402	CH 8	14980	980	CH 8	14984
404	CH 9	16295	982	CH 9	16302
406	CH 10	16001	984	CH 10	15999
408	CH 11	16162	986	CH 11	16167
410	CH 12	16595	988	CH 12	16591
412	CH 13	16361	990	CH 13	16367
414	CH 14	16736	992	CH 14	16752
416	CH 15	14853	994	CH 15	14853
418	REFLECTOR 1 POSITION 13	2154	996	REFLECTOR 1 POSITION 30	2155
420	REFLECTOR 2 POSITION 13	1969	998	REFLECTOR 2 POSITION 30	1969
422	REFL 1 POS 13 2ND LOOK	2154	1000	REFL 1 POS 30 2ND LOOK	2155
424	REFL 2 POS 13 2ND LOOK	1969	1002	REFL 2 POS 30 2ND LOOK	1969
426	NADIR SAMPLE 13	16806	1004	NADIR SAMPLE 30	16806
428	CH 3	16300	1006	CH 3	16301
430	CH 4	16865	1008	CH 4	16870
432	CH 5	16742	1010	CH 5	16739
434	CH 6	16442	1012	CH 6	16444
436	CH 7	14986	1014	CH 7	14986
438	CH 8	16298	1016	CH 8	16300
440	CH 9	15997	1018	CH 9	15999
442	CH 10	16166	1020	CH 10	16166
444	CH 11	16593	1022	CH 11	16596
446	CH 12	16368	1024	CH 12	16360
448	CH 13	16737	1026	CH 13	16744
450	CH 14	14852	1028	CH 14	14853
452	CH 15	2154	1030	CH 15	0E
454	REFLECTOR 1 POSITION 14	1969	1032	REFLECTOR 1 COLD CAL POS	0E
456	REFLECTOR 2 POSITION 14	2154	1034	REFLECTOR 2 COLD CAL POS	0E
458	REFL 1 POS 14 2ND LOOK	1969	1036	REFL 1 COLD CAL 2ND LOOK	0E
460	REFL 2 POS 14 2ND LOOK	16804	1038	REFL 2 COLD CAL 2ND LOOK	0
462	NADIR SAMPLE 14	16229	1040	COLD CAL DATA 1	0
464	CH 3	16865	1042	CH 3	0
466	CH 4	16741	1044	CH 4	0
468	CH 5	16445	1046	CH 5	0
470	CH 6	14987	1048	CH 6	0
472	CH 7	16295	1050	CH 7	0
474	CH 8	16002	1052	CH 8	0
476	CH 9	16167	1054	CH 9	0
478	CH 10	16593	1056	CH 10	0
480	CH 11	16359	1058	CH 11	0
482	CH 12	16711	1060	CH 12	0
484	CH 13	14852	1062	CH 13	0
486	CH 14	2154	1064	CH 14	0
488	REFLECTOR 1 POSITION 15	1969	1066	REFLECTOR 1 COLD CAL DATA 2	0
490	REFLECTOR 2 POSITION 15	2154	1068	REFLECTOR 2 COLD CAL DATA 2	0
492	REFL 1 POS 15 2ND LOOK	1969	1070	REFL 1 POS 30 2ND LOOK	0
	REFL 2 POS 15 2ND LOOK	16806		REFL 2 POS 30 2ND LOOK	0

ELEMENT	DESCRIPTION	VALUE	TEMPERATURE	DEG C
1090	SCAN MOTOR A1-1	18585	24.67	
1092	SCAN MOTOR A1-2	19912	26.02	
1094	FEEDHORN A1-1	20620	27.99	
1096	FEEDHORN A1-2	21685	30.03	
1098	RF MUX A1-1	22599	31.32	
1100	RF MUX A1-2	23780	33.59	
1102	LOCAL OSCILLATOR CHANNEL 3	24930	35.77	
1104	LOCAL OSCILLATOR CHANNEL 4	25003	35.59	
1106	LOCAL OSCILLATOR CHANNEL 5	24317	34.78	
1108	LOCAL OSCILLATOR CHANNEL 6	22010	30.56	
1110	LOCAL OSCILLATOR CHANNEL 7	23200	32.61	
1112	LOCAL OSCILLATOR CHANNEL 8	24571	35.00	
1114	LOCAL OSCILLATOR CHANNEL 15	24404	34.39	
1116	PLL LO #2 CHANNELS 9 THROUGH 14	15429	31.05	
1118	PLL LO #1 CHANNELS 9 THROUGH 14	25671	37.28	
1120	SPARE (NOT USED)	32767	51.27	
1122	MIXER/IF AMPLIFIER CHANNEL 3	24493	34.10	
1124	MIXER/IF AMPLIFIER CHANNEL 4	24038	34.32	
1126	MIXER/IF AMPLIFIER CHANNEL 5	22807	33.82	
1128	MIXER/IF AMPLIFIER CHANNEL 6	23016	31.80	
1130	MIXER/IF AMPLIFIER CHANNEL 7	24343	32.44	
1132	MIXER/IF AMPLIFIER CHANNEL 8	22767	34.35	
1134	MIXER/IF AMPLIFIER CH 9 THRU 14	23820	31.23	
1136	MIXER/IF AMPLIFIER CHANNEL 15	23913	34.21	
1138	IF AMPLIFIER CHANNEL 11 THRU 14	23945	34.00	
1140	IF AMPLIFIER CHANNEL 9	24115	34.13	
1142	IF AMPLIFIER CHANNEL 10	22625	31.55	
1144	IF AMPLIFIER CHANNEL 11	23996	34.02	
1146	DC/DC CONVERTER	22612	31.52	
1148	IF AMPLIFIER CHANNEL 13	22706	31.48	
1150	IF AMPLIFIER CHANNEL 14	22524	31.46	
1152	IF AMPLIFIER CHANNEL 12	22650	32.57	
1154	RF SHELF A1-1	23523	33.41	
1156	RF SHELF A1-2	21007	33.76	
1158	DETECTOR/PREAMPLIFIER ASSEMBLY	24369	28.04	
1160	A1-1 WARM LOAD 1	24143	25.01	
1162	A1-1 WARM LOAD 2	24354	25.05	
1164	A1-1 WARM LOAD 3	24315	25.05	
1166	A1-1 WARM LOAD 4	24385	25.16	
1168	A1-1 WARM LOAD CENTER	25173	25.02	
1170	A1-2 WARM LOAD 1	25320	26.96	
1172	A1-2 WARM LOAD 2	25252	27.10	
1174	A1-2 WARM LOAD 3	25164	27.10	
1176	A1-2 WARM LOAD 4	25331	27.04	
1178	A1-2 WARM LOAD CENTER			
1180	TEMP SENSOR REFERENCE VOLTAGE			

DESCRIPTION

STATUS

STATUS

STATUS

SCANNER A1-1 POWER	ON		ON
SCANNER A1-2 POWER	ON		ON
PLL POWER	PLLO # 1	PLLO # 1	PLLO # 1
ANTENNA IN WARM CAL POSITION MODE	NO	NO	NO
ANTENNA IN COLD CAL POSITION MODE	NO	NO	NO
ANTENNA IN NADIR POSITION MODE	YES	YES	YES
ANTENNA IN FULL SCAN MODE	NO	NO	NO
SURVIVAL HEATER POWER	OFF	OFF	OFF
MODULE POWER	CONNECT	CONNECT	CONNECT
COLD CAL POSITION MSB	ZERO	ZERO	ZERO
COLD CAL POSITION LSB	ZERO	ZERO	ZERO

ANALOG DATA

DESCRIPTION

VALUE

VALUE

DEG C

DEG C

A1-1 SCANNER MOTOR TEMPERATURE	216	216	25.1	216	25.1
A1-2 SCANNER MOTOR TEMPERATURE	217	217	26.2	217	26.2
A1-1 RF SHELF TEMPERATURE	219	219	28.8	219	28.8
A1-2 RF SHELF TEMPERATURE	221	222	31.5	222	32.8
A1-1 WARM LOAD TEMPERATURE	216	216	26.1	216	26.1
A1-2 WARM LOAD TEMPERATURE	218	218	29.0	218	29.0

DESCRIPTION

VALUE

VALUE

AMPS/
VOLTS

AMPS/
VOLTS

A1-1 ANTENNA DRIVE MOTOR CURRENT (AVRG)	4	4	1.86	4	1.86
A1-2 ANTENNA DRIVE MOTOR CURRENT (AVRG)	4	4	1.86	4	1.86
SIGNAL PROCESSING +15 VDC	170	169	14.90	169	14.90
ANTENNA DRIVE +15 VDC	169	169	14.90	169	14.90
SIGNAL PROCESSING -15 VDC	147	147	-15.05	147	-15.05
ANTENNA DRIVE -15 VDC	146	146	-15.05	146	-15.05
RECEIVER AMPLIFIER +8 VDC	155	155	7.95	155	7.95
SIGNAL PROCESSOR +5 VDC	144	144	4.99	144	4.99
ANTENNA DRIVE +5 VDC	144	144	4.96	144	4.96
RECEIVER MIXER/IF +10 VDC	168	168	9.89	168	9.89
PHASE LOCK LOOP (CHANNEL 9/14)	168	168	14.99	168	14.99
PHASE LOCK LOOP (CHANNEL 9/14)	143	143	-15.00	143	-15.00
L.O. VOLTAGE (CHANNEL 7)	171	171	9.99	171	9.99
L.O. VOLTAGE (CHANNEL 8)	170	170	9.94	170	9.94
L.O. VOLTAGE (CHANNEL 6)	171	171	9.97	171	9.97
L.O. VOLTAGE (CHANNEL 3)	171	171	9.98	171	9.98
L.O. VOLTAGE (CHANNEL 4)	171	171	9.99	171	9.99
L.O. VOLTAGE (CHANNEL 5)	170	170	9.95	170	9.95
PLLO # 2 LOCK DETECT	4	4	0.08	4	0.08
PLLO # 1 LOCK DETECT	219	219	4.38	219	4.38
L.O. VOLTAGE (CHANNEL 15)	169	169	14.90	169	14.90

PRT TEMPERATURES

VARIABLE TARGET

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
615	42.00	601	14.00
616	43.00	602	15.00
617	44.00	603	16.00
618	45.00	604	17.00
619	46.00	605	18.00
620	47.00	606	19.00
621	48.00	607	20.00
622	49.00	608	21.00
623	50.00	609	22.00
624	51.00	610	23.00
625	52.00	611	24.00
626	53.00	612	25.00
627	67.00	613	69.00
628	68.00	614	70.00
629	71.00	630	72.00
631	26.00	632	27.00

FIXED TARGET

BASEPLATE

THERMOCOUPLE TEMPERATURES

FIXED TARGET SHROUD

VARIABLE TARGET SHROUD

FIXED TARGET N2

VARIABLE TARGET N2

HEATER N2

FIXED TARGET FLOW METER

VARIABLE TARGET FLOW METER

BASEPLATE HEATER N2

BASEPLATE N2

BASEPLATE FLOW METER

ADJUNCT RADIATORS

A1-1		A1-2	
NO.	DEG K	NO.	DEG K
558	5.00	537	34.00
559	6.00	538	35.00
550	7.00	524	36.00
551	8.00	525	37.00
506	57.00	502	30.00
507	58.00	503	31.00
516	59.00	511	32.00
517	60.00	512	33.00
514	1.00	509	38.00
515	2.00	510	39.00
508	63.00	504	61.00
518	64.00	513	62.00
519	3.00	520	4.00
521	9.00	522	10.00
523	65.00		
575	73.00	577	74.00
579	75.00	581	76.00

AMSU A1-33 A1.EXE;62 NADIR MODE 5 P1 15-JUL-00 17:11:15 SCAN NUMBER 763
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 1 TO 18

1 A1-1	SCANR MOTOR	216	26.06	DEG C	10	ANTENNA DRIVE	15VDC	14.96
2 A1-2	SCANR MOTOR	217	27.40	DEG C	11	SIGNAL PROCESSING	-15VDC	-15.01
3 A1-1	RF SHELF	219	29.58	DEG C	12	ANTENNA DRIVE	-15VDC	-15.01
4 A1-2	RF SHELF	222	32.90	DEG C	13	RECEIVER AMPLIFIER	8VDC	7.96
5 A1-1	WARM LOAD	216	26.39	DEG C	14	SIGNAL PROCESSOR	5 VDC	5.00
6 A1-2	WARM LOAD	218	29.28	DEG C	15	ANTENNA DRIVE	5 VDC	4.99
7 ANT A1-1	DRIVE MOTOR CURRENT		1.88		16	RECEIVER MIXER/IF	10VDC	9.90
8 ANT A1-2	DRIVE MOTOR CURRENT		1.88		17	PHASE LOCK LOOP	15VDC	15.04
9 SIGNAL PROCESSING			15.00		18	PHASE LOCK LOOP	-15VDC	-14.95
[21]	UP		[22]	DOWN				

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 NADIR MODE 5 P1 15-JUL-00 17:11:22 SCAN NUMBER 764
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 10 TO 27

10 ANTENNA DRIVE	15VDC	14.96	19 L.O.	VOLTAGE	CH 8	10.00
11 SIGNAL PROCESSING	-15VDC	-15.02	20 L.O.	VOLTAGE	CH 7	9.98
12 ANTENNA DRIVE	-15VDC	-15.00	21 L.O.	VOLTAGE	CH 6	10.02
13 RECEIVER AMPLIFIER	8VDC	7.95	22 L.O.	VOLTAGE	CH 3	9.99
14 SIGNAL PROCESSOR	5 VDC	4.99	23 L.O.	VOLTAGE	CH 4	10.01
15 ANTENNA DRIVE	5 VDC	4.99	24 L.O.	VOLTAGE	CH 5	10.00
16 RECEIVER MIXER/IF	10VDC	9.90	25 PLL0	# 2 LOCK DETECT		0.08
17 PHASE LOCK LOOP CH9/14	15VDC	15.03	26 PLL0	# 1 LOCK DETECT		4.39
18 PHASE LOCK LOOP CH9/14	-15VDC	-14.95	27 L.O.	VOLTAGE	CH15	14.98
[21] UP	[22] DOWN					

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62 NADIR MODE 5 P1 15-JUL-00 17:11:56 SCAN NUMBER 768
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

CHANNEL 3		BP DATA		BP DATA		BP DATA		BP DATA	
BP DATA		BP DATA		BP DATA		BP DATA		BP DATA	
1	16810	9	16808	17	16799	25	16802		
2	16810	10	16803	18	16801	26	16801		
3	16806	11	16806	19	16799	27	16802		
4	16806	12	16803	20	16805	28	16800		
5	16802	13	16805	21	16802	29	16803		
6	16808	14	16804	22	16803	30	16805		
7	16803	15	16805	23	16807	CC	0		
8	16803	16	16800	24	16808	WC	0		
		[22]		DOWN					

[21] UP

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62 NADIR MODE 5 P1 15-JUL-00 17:12:16 SCAN NUMBER 771
[5] DIGITAL A DATA ELEMENT 00 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

RADIOMETRIC DATA

CHANNEL		9	
BP	DATA	BP	DATA
1	16294	9	16297
2	16296	10	16297
3	16300	11	16297
4	16297	12	16300
5	16300	13	16295
6	16299	14	16298
7	16297	15	16298
8	16297	16	16298
[22] DOWN		17	16297
		18	16293
		19	16296
		20	16298
		21	16299
		22	16293
		23	16298
		24	16297
		25	16300
		26	16297
		27	16302
		28	16298
		29	16299
		30	16300
		CC	0
		WC	0

[21] UP

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 NADIR MODE 5 P1 15-JUL-00 17:12:41 SCAN NUMBER 774
 [5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

		REFLECTOR POSITIONS									
		1		2		3		4		5	
BP	LOOK	1	LOOK	2	BP	LOOK	1	LOOK	2	BP	LOOK
1	2155	2154	2154	2154	17	2154	2154	2154	2154	25	2154
2	2155	2154	2154	2154	18	2154	2154	2154	2154	26	2155
3	2154	2154	2154	2154	19	2154	2154	2154	2154	27	2155
4	2154	2154	2154	2154	20	2154	2154	2154	2154	28	2154
5	2154	2154	2154	2154	21	2154	2154	2154	2154	29	2154
6	2154	2154	2154	2154	22	2154	2154	2154	2154	30	2154
7	2154	2154	2154	2154	23	2154	2154	2154	2154	CC	0
8	2154	2154	2154	2154	24	2154	2154	2154	2154	WC	0
[21] UP		[22] DOWN									

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
 SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1 EXE:62 NADIR MODE P1 15-JUL-00 17:12:49 SCAN NUMBER 775
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

		REFLECTOR POSITIONS									
BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2	BP	LOOK 1	LOOK 2
1	1969	1969	9	1969	1969	17	1969	1969	25	1969	1969
2	1969	1969	10	1969	1969	18	1969	1969	26	1969	1969
3	1969	1969	11	1969	1969	19	1969	1969	27	1969	1969
4	1969	1969	12	1969	1969	20	1969	1969	28	1969	1969
5	1969	1969	13	1969	1969	21	1969	1969	29	1969	1969
6	1969	1969	14	1969	1969	22	1969	1969	30	1969	1969
7	1969	1969	15	1969	1969	23	1969	1969	CC	0	0
8	1969	1969	16	1969	1969	24	1969	1969	WC	0	0
[21]	UP			[22]	DOWN						

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62 NADIR MODE P1 15-JUL-00 17:13:29 SCAN NUMBER 780
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

DIGITAL A TEMPERATURES 1 TO 16			
NO	DATA	TEMP C	NO
1	SCAN MOTOR A1-1	18569	24.64
2	SCAN MOTOR A1-2	19894	25.99
3	FEEDHORN A1-1	20632	28.01
4	FEEDHORN A1-2	21692	30.04
5	RF MUX A1-1	22613	31.35
6	RF MUX A1-2	23798	33.63
7	LO CHANNEL 3	24956	35.83
8	LO CHANNEL 4	25021	35.62
[21] UP		[22] DOWN	
		9 LO CHANNEL 5	24330
		10 LO CHANNEL 6	22021
		11 LO CHANNEL 7	23214
		12 LO CHANNEL 8	24586
		13 LO CHANNEL 15	24419
		14 PLLO #2 CH 9/14	15444
		15 PLLO #1 CH 9/14	25686
		16 PLLO REFERENCE	32767
			34.81
			30.58
			32.64
			35.02
			34.42
			31.08
			37.31
			51.27

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE:62 NADIR MODE 5 P1 15-JUL-00 17:13:35 SCAN NUMBER 781
[5] DIGITAL A DATA ELEMENT 5 DIGITAL B DATA BYTE 1 00010000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

NO	DIGITAL A TEMPERATURES 17 TO 32		DATA	TEMP C	
	DATA	TEMP C			
17	MIXER IF CH 3	24512	25 IF AMP CH 11/14	23930	34.03
18	MIXER IF CH 4	24515	26 IF AMP CH 9	23961	34.16
19	MIXER IF CH 5	24056	27 IF AMP CH 10	24131	34.17
20	MIXER IF CH 6	22820	28 IF AMP CH 11	22639	31.57
21	MIXER IF CH 7	23031	29 DC/DC CONVERTER	24010	34.05
22	MIXER IF CH 8	24360	30 IF AMP CH 13	22627	31.55
23	MIXER IF CH 9/14	22781	31 IF AMP CH 14	22721	31.51
24	MIXER IF CH 15	23835	32 IF AMP CH 12	22539	31.48
[21]	UP				

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

[6] DIGITAL B DATA ELEMENT 00
[7] ANALOG DATA ELEMENT 00

DIGITAL A TEMPERATURES 31 TO 46										
NO	DATA			TEMP C	NO	DATA			TEMP C	
31	IF AMP	CH 14	22723	31.51	39	A1-1	WARM LOAD	4	24322	
32	IF AMP	CH 12	22540	31.49	40	A1-1	WARM LOAD	C	24392	
33	RF SHELF	A1-1	22667	32.61	41	A1-2	WARM LOAD	1	25184	
34	RF SHELF	A1-2	23538	33.44	42	A1-2	WARM LOAD	2	25328	
35	DETECTOR/PREAMP		21018	28.78	43	A1-2	WARM LOAD	3	25495	
36	A1-1	WARM LOAD 1	24373	25.05	44	A1-2	WARM LOAD	4	25265	
37	A1-1	WARM LOAD 2	24151	25.03	45	A1-2	WARM LOAD	C	25174	
38	A1-1	WARM LOAD 3	24361	25.07	THERMAL REFERENCE					25331
				[22] DOWN						
[21] UP										

POWER [4] ON SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

TEST DATA SHEET 34
Analog Telemetry Verification by Way of Connector J6 (Paragraph 3.2.4.3.5.1)

	From	Description	To	Measured (volts)	Required (volts)	Pass/Fail
03	J6-02	RF Shelf A1-1 Temp.	J1-10	<u>4.48</u> V	3.5 ± 2 V	P
01	J6-03	A1-1 Scan Motor Temp.	J1-10	<u>4.4</u> V	3.5 ± 2 V	P
05	J6-04	Warm Load A1-1 Temp.	J1-10	<u>4.4</u> V	3.5 ± 2 V	P
04	J6-21	RF Shelf A1-2 Temp.	J1-10	<u>4.48</u> V	3.5 ± 2 V	P
02	J6-22	A1-2 Scan Motor Temp.	J1-10	<u>4.4</u> V	3.5 ± 2 V	P
06	J6-23	Warm Load A1-2 Temp.	J1-10	<u>4.4</u> V	3.5 ± 2 V	P
25	J6-06	PLLO No. 2 Lock detect	J2-03	<u>0.0</u> V	***	P
07	J6-08	A1-1 Drive Motor Curr.	J2-03	<u>1.88</u> V	3.5 ± 2 V	P
10	J6-09	+15 V Antenna Drive	J2-03	<u>3.48</u> V	3.5 ± 2 V	P
15	J6-10	+5 V Antenna Drive	J2-03	<u>3.12</u> V	3.5 ± 2 V	P
09	J6-11	+15 V Signal Processing	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
14	J6-12	+5 V Signal Processing	J2-03	<u>3.04</u> V	3.5 ± 2 V	P
22	J6-13	L.O. Voltage Channel 3	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
24	J6-14	L.O. Voltage Channel 5	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
20	J6-15	L.O. Voltage Channel 7	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
16	J6-16	+15 V PLL LO Ch 9-14	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
17	J6-17	*	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
27	J6-18	L.O. Voltage Channel 15	J2-03	<u>3.52</u> V	3.5 ± 2 V	P
26	J6-25	PLLO No. 1 Lock detect	J2-03	<u>4.48</u> V	***	P
08	J6-27	A1-2 Drive Motor Curr.	J2-03	<u>1.92</u> V	3.5 ± 2 V	P
12	J6-28	-15 V Antenna Drive	J2-03	<u>3.16</u> V	3.5 ± 2 V	P
11	J6-29	-15 V Signal Processing	J2-03	<u>3.04</u> V	3.5 ± 2 V	P
23	J6-30	L.O. Voltage Channel 4	J2-03	<u>3.48</u> V	3.5 ± 2 V	P
21	J6-31	L.O. Voltage Channel 6	J2-03	<u>3.56</u> V	3.5 ± 2 V	P
19	J6-32	L.O. Voltage Channel 8	J2-03	<u>3.56</u> V	3.5 ± 2 V	P
18	J6-33	-15 V PLL LO Ch 9-14	J2-03	<u>2.96</u> V	3.5 ± 2 V	P
13	J6-34	**	J2-03	<u>3.24</u> V	3.5 ± 2 V	P

* +8.5 V PLL LO Ch 9-14 for S/N 101-104, +10V Mixer Amp for S/N 105 and above.

** +8 V Receiver for S/N 101-104, +8 V IF Amp for S/N 105 and above.

*** 4.5 ± 0.5 when locked, 0.5 ± 0.5 when unlocked or OFF. One must be locked.

Circle Test: CPT LPT

0A:0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

Ray Huppert
Test Systems Engineer

7-15-00
Date

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Judith Hervey
Quality Control

7-15-00
Date

TEST DATA SHEET 35 (Sheet 1 of 2)
Analog Telemetry Signals by Way of the STE (Paragraph 3.2.4.3.5.2)

	Description	(*)	Measured (Deg. C)	Required (Deg. C)	Pass/Fail
01	A1-1 Scanner Motor	Temp	<u>26.19</u>	25 ± 15	<u>P</u>
02	A1-2 Scanner Motor	Temp	<u>27.64</u>	25 ± 15	<u>P</u>
03	A1-1 RF Shelf	Temp	<u>29.86</u>	25 ± 15	<u>F</u>
04	A1-2 RF Shelf	Temp	<u>32.41</u>	25 ± 15	<u>F</u>
05	A1-1 Warm Load	Temp	<u>26.47</u>	25 ± 15	<u>P</u>
06	A1-2 Warm Load	Temp	<u>29.65</u>	25 ± 15	<u>F</u>
			(mAmps)	(mAmps)	
07	Ant A1-1 Drv Motor Current		<u>40.93</u>	125 mA (Max)	<u>F</u>
08	Ant A1-2 Drv Motor Current		<u>40.14</u>	125 mA (Max)	<u>F</u>

(*) Data from the printout sheet. Rewriting data on this space is optional.

(Continued on sheet 2)

Circle Test: Final CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Ray Berkeley
Test Systems Engineer
Julia Hervey
Quality Control

7-15-00
Date

7-15-00
Date

TEST DATA SHEET 35 (Sheet 2 of 2)
Analog Telemetry Signals by Way of the STE (Paragraph 3.2.4.3.5.2)

	Description	(*)	Measured (volts)	Required (volts)	Pass/ Fail
09	Signal Processing	+15 V	+ <u>14.99</u>	15.0 ± 0.5 V	<u>P</u>
10	Antenna Drive	+15 V	+ <u>15.13</u>	15.0 ± 0.5 V	<u>P</u>
11	Signal Processing	-15 V	+ <u>15.01</u>	-15.0 ± 0.5 V	<u>P</u>
12	Antenna Drive	-15 V	- <u>14.91</u>	-15.0 ± 0.5 V	<u>P</u>
13	Receiver	+8 V	+ <u>7.96</u>	8.0 ± 0.5 V	<u>P</u>
14	Sig Processing	+5 V	+ <u>5.00</u>	5.0 ± 0.5 V	<u>P</u>
15	Antenna Drive	+5 V	+ <u>5.05</u>	5.0 ± 0.5 V	<u>P</u>
16	Phase Lock Loop Ch 9-14 (a)/	+8.5 V	<u>N/A</u>	8.5 ± 0.5 V	<u>N/A</u>
	Receiver/Mixer IF (b)	+10 V	+ <u>9.89</u>	10.0 ± 0.5 V	<u>P</u>
17	Phase Lock Loop Ch 9-14	+15 V	+ <u>15.02</u>	15.0 ± 0.5 V	<u>P</u>
18	Phase Lock Loop Ch 9-14	-15 V	- <u>14.97</u>	-15.0 ± 0.5 V	<u>P</u>
19	L.O. #8	Ch-8	+ <u>10.00</u>	(**)__ ± 0.5 V	<u>P</u>
20	L.O. #7	Ch-7	+ <u>10.00</u>	(**)__ ± 0.5 V	<u>P</u>
21	L.O. #6	Ch-6	+ <u>9.99</u>	(**)__ ± 0.5 V	<u>P</u>
22	L.O. #3	Ch-3	+ <u>9.98</u>	(**)__ ± 0.5 V	<u>P</u>
23	L.O. #4	Ch-4	+ <u>10.00</u>	(**)__ ± 0.5 V	<u>P</u>
24	L.O. #5	Ch-5	+ <u>9.98</u>	(**)__ ± 0.5 V	<u>P</u>
25	PLLO No. 2 Lock Detect		+ <u>0.08</u>	(***)	<u>P</u>
26	PLLO No. 1 Lock Detect		+ <u>4.39</u>	(***)	<u>P</u>
27	L.O. #15	Ch-15	+ <u>14.97</u>	(**)__ ± 0.5 V	<u>P</u>

(*) Data from the printout sheet. Rewriting data on this space is optional.

(**) GDO voltages from the manufacturer data sheet for S/N 101-104; DRO CH3-8 10V, GDO CH15 15V for S/N 105 and above.

(***) Locked PLO voltage 0 to +15 V, other PLO voltage ±15.0 V; one must be locked for S/N 101-104. Locked PLO voltage 4.0 ± 1.0 V, other PLO voltage 0.0 ± 0.2 V, one must be locked for S/N 105 and above.

(a) For S/N 101 through 104. (b) For S/N 105 and up.

Circle Test: Final CPT LPT

09:0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920

S/N: 109

Ray Herberg
Test Systems Engineer

7-15-00

Date

Joseph Sanford
Customer Representative
(Flight Hardware Only)

7-17-00
Date

Judith Herberg
Quality Control

7-15-00
Date

AMSU A1-33 A1.EXE;62 FULL SCAN MODE P1 15-JUL-00 17:03:50 SCAN NUMBER 709
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

[7] ANALOG DATA ELEMENT 00

ANALOG DATA 1 TO 18

1	A1-1	SCANR MOTOR	216	26.19	DEG C	10	ANTENNA DRIVE	15VDC	15.13
2	A1-2	SCANR MOTOR	218	27.64	DEG C	11	SIGNAL PROCESSING	-15VDC	-15.01
3	A1-1	RF SHELF	219	29.86	DEG C	12	ANTENNA DRIVE	-15VDC	-14.91
4	A1-2	RF SHELF	221	32.41	DEG C	13	RECEIVER AMPLIFIER	8VDC	7.96
5	A1-1	WARM LOAD	216	26.47	DEG C	14	SIGNAL PROCESSOR	5 VDC	5.00
6	A1-2	WARM LOAD	218	29.65	DEG C	15	ANTENNA DRIVE	5 VDC	5.05
7	ANT A1-1	DRIVE MOTOR CURRENT		40.93		16	RECEIVER MIXER/IF	10VDC	9.89
8	ANT A1-2	DRIVE MOTOR CURRENT		40.14		17	PHASE LOCK LOOP CH9/14	15VDC	15.02
9	SIGNAL PROCESSING	+15VDC		14.99		18	PHASE LOCK LOOP CH9/14	-15VDC	-14.97
[21]	UP	[22]	DOWN						

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

AMSU A1-33 A1.EXE;62 FULL SCAN MODE P1 15-JUL-00 17:03:58 SCAN NUMBER 709
[5] DIGITAL A DATA ELEMENT 0000

[6] DIGITAL B DATA ELEMENT 00

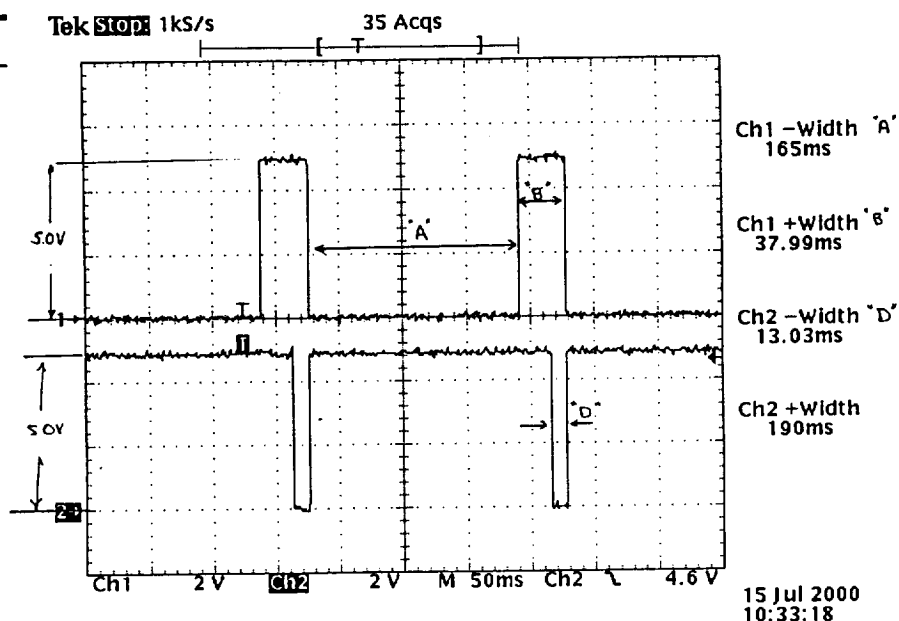
[7] ANALOG DATA ELEMENT 00

ANALOG DATA 10 TO 27

10 ANTENNA DRIVE	15VDC	15.06	19 L.O.	VOLTAGE	CH 8	10.00
11 SIGNAL PROCESSING	-15VDC	-15.01	20 L.O.	VOLTAGE	CH 7	10.00
12 ANTENNA DRIVE	-15VDC	-14.94	21 L.O.	VOLTAGE	CH 6	9.99
13 RECEIVER AMPLIFIER	8VDC	7.97	22 L.O.	VOLTAGE	CH 3	9.98
14 SIGNAL PROCESSOR	5 VDC	5.00	23 L.O.	VOLTAGE	CH 4	10.00
15 ANTENNA DRIVE	5 VDC	5.03	24 L.O.	VOLTAGE	CH 5	9.98
16 RECEIVER MIXER/IF	10VDC	9.91	25 PLL0 # 2	LOCK DETECT		0.08
17 PHASE LOCK LOOP CH9/14	15VDC	15.01	26 PLL0 # 1	LOCK DETECT		4.39
18 PHASE LOCK LOOP CH9/14	-15VDC	-14.96	27 L.O.	VOLTAGE	CH15	14.97
[21] UP	[22] DOWN					

POWER [4] ON
SCREEN ONLY [2] PRINT [3] FULL [1] RETURN
SELECT TOUCHSCREEN BUTTON 2

TEST DATA SHEET 36 Integrate/Hold and Dump Signal Verification (Paragraph 3.2.4.3.6.1)



S/O: 787920 OP: 0830 TDS 36 TEST ENG: Ray Hershberg DATE: 7-15-00
P/N: 1331720-3-TST SN: 109 FINAL CPT QUALITY: [Signature] DATE: 7-15-00

Parameter	Measured	Required	Pass/ Fail
Scope Channel-1: Integration/Hold			
Time Measured (A)*	165.0 ms	165 ms \pm 10%	PASS
Time Measured (B)*	37.99 ms	35 ms \pm 10%	PASS
Amplitude Measured	5.0 V	5.0 \pm 0.2 V	PASS
Scope Channel-2: Dump Signal			
Time Measured (D)*	13.03 ms	9 ms to 15 ms	PASS
Amplitude Measured	5.0 V	5.0 \pm 0.2 V	PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: **FINAL CPT** LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

OP: 0830

Ray Hershberg
Test Systems Engineer

7-15-00
Date

[Signature]
Customer Representative
(Flight Hardware Only)

7-15-00
Date

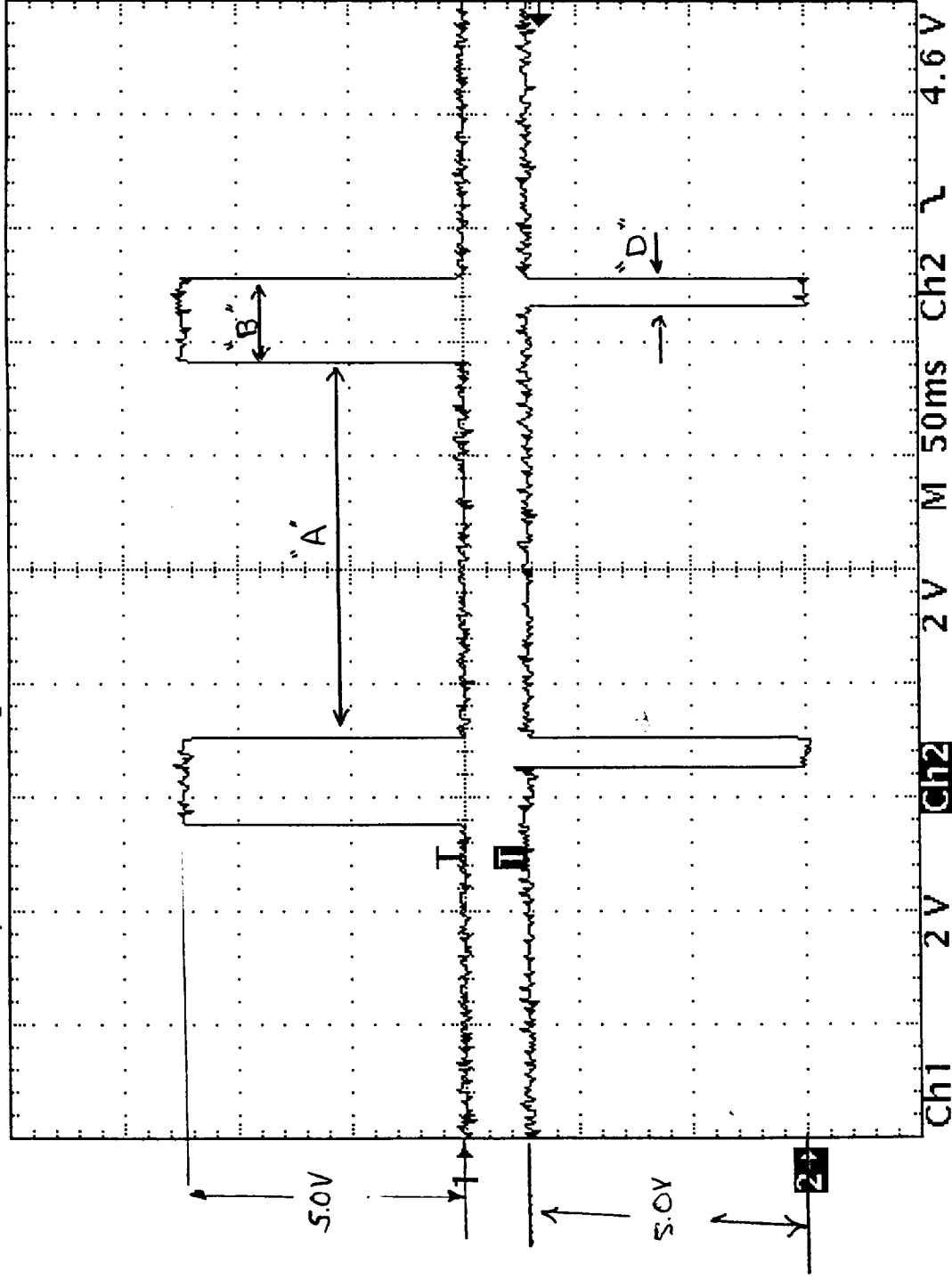
[Signature]
Quality Control

7-15-00
Date

Tek Stop: 1KS/s

35 Acqs

[T]



Ch1 - width "A"
165ms

Ch1 + width "B"
37.99ms

Ch2 - width "D"
13.03ms

Ch2 + width
190ms

15 Jul 2000
10:33:18

3.2436.1

S/O: 787920 CP: 0830

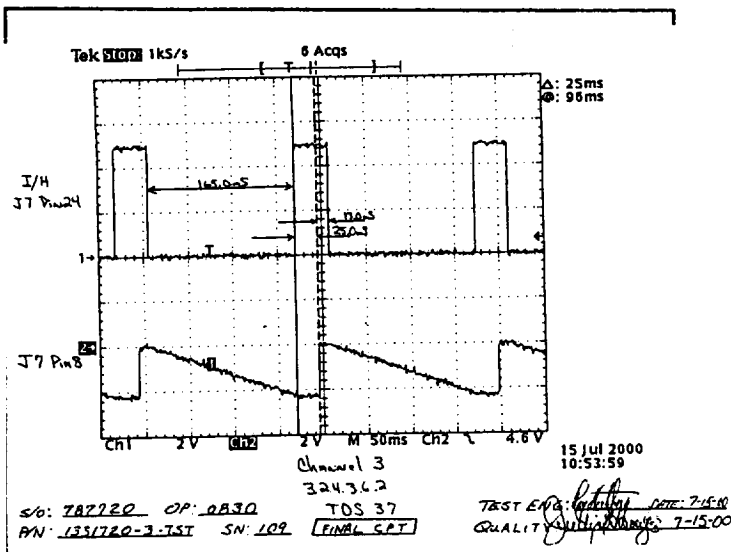
PN: 1331720-3-TST SN: 109

TDS 36

FINAL CPT

TEST ENG: *[Signature]* DATE: 7-15-00
QUALITY: *[Signature]* 7-15-00

TEST DATA SHEET 37 Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

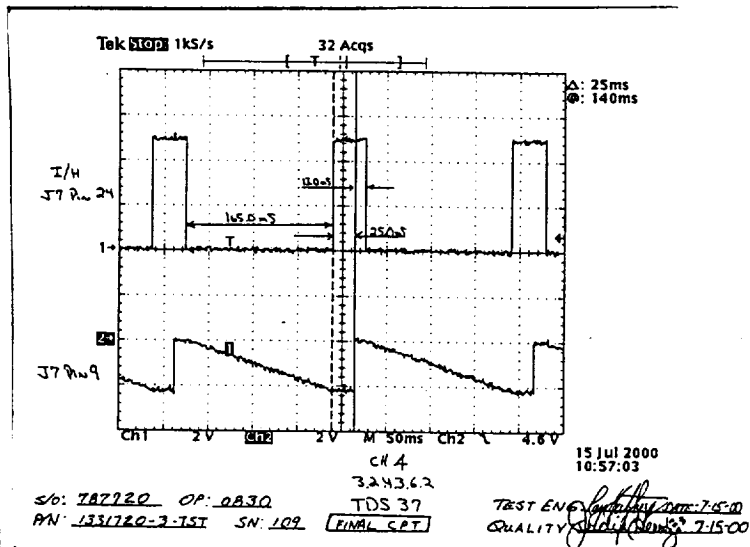


Channel 03
Frequency: 50.3 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS



Channel 04
Frequency: 52.8 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

OP: 0830

Shop Order: 78720 SN: 109

Ja Sanford
Customer Representative
(Flight Hardware Only)

7-15-00
Date

Ray H. Berg
Test Systems Engineer
7-15-00
Date

Gregory J. Berg
Quality Control
07-15-00
Date

Tek Stop 1ks/s

32 Acqs

[T]

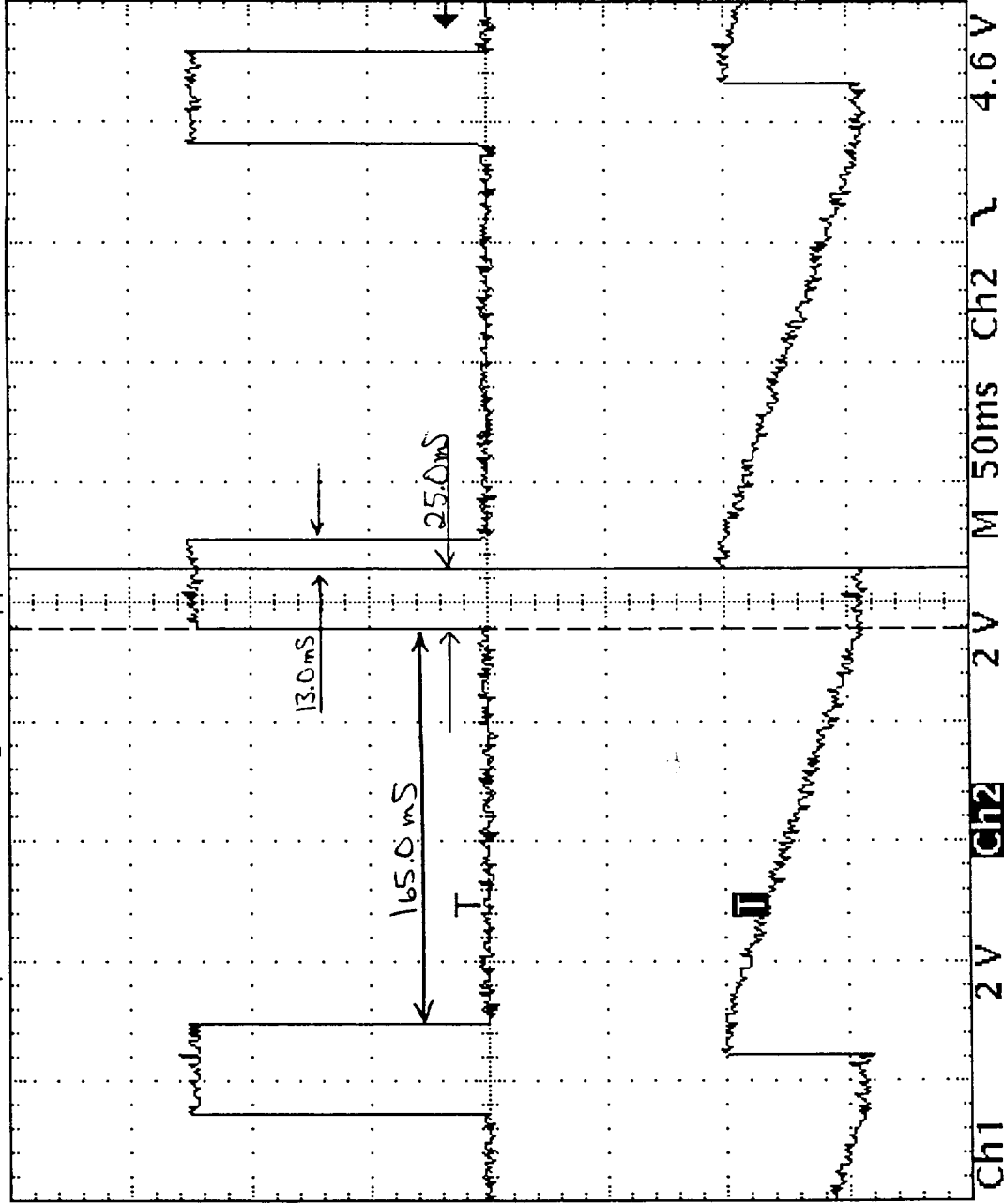
Δ : 25ms
@: 140ms

I/H
J7 Pin 24

1 →

2 →

J7 Pin 9



15 Jul 2000
10:57:03

CH 4

3.243.62

TDS 37

FINAL CPT 1

CP: 0830

SN: 109

DATE: 7-15-00

7-15-00

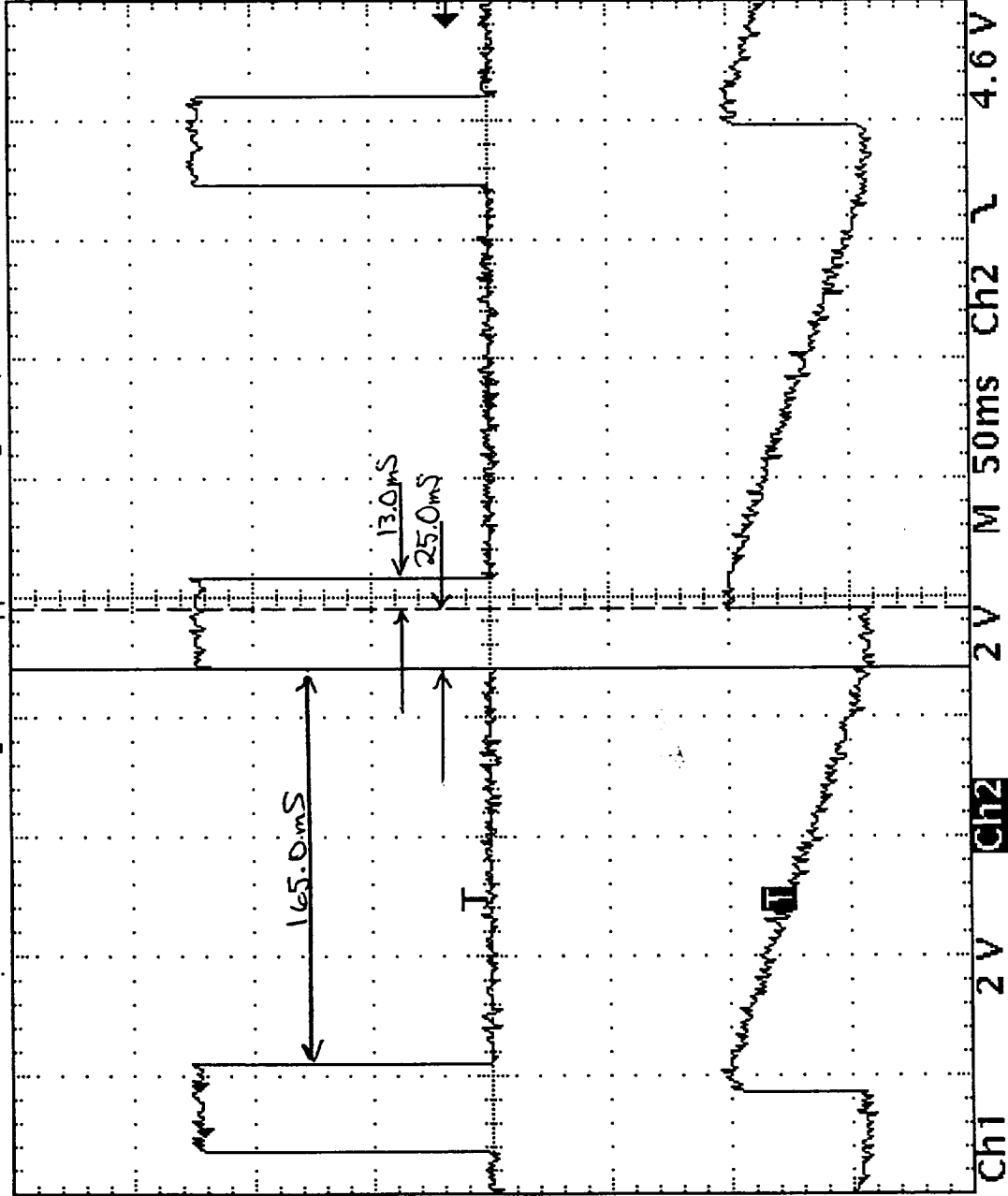
TEST ENG. *[Signature]*

QUALITY CONTROL *[Signature]*

Tek Stop: 1ks/s

6 Acqs

Δ : 25ms
@: 96ms



I/H
J7 Pin24

J7 Pin8

15 Jul 2000
10:53:59

Channel 3

3.24.3.6.2

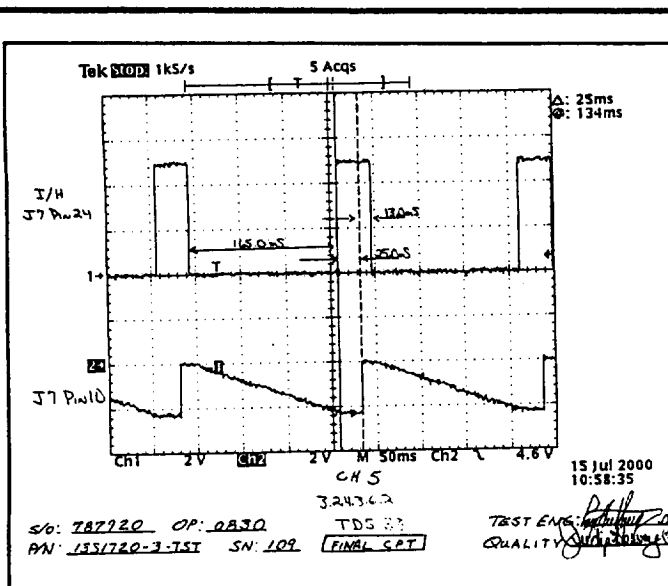
S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

FINAL CPT

TEST END: *[Signature]* DATE: 7-15-00
QUALITY: *[Signature]* 7-15-00

TEST DATA SHEET 38
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

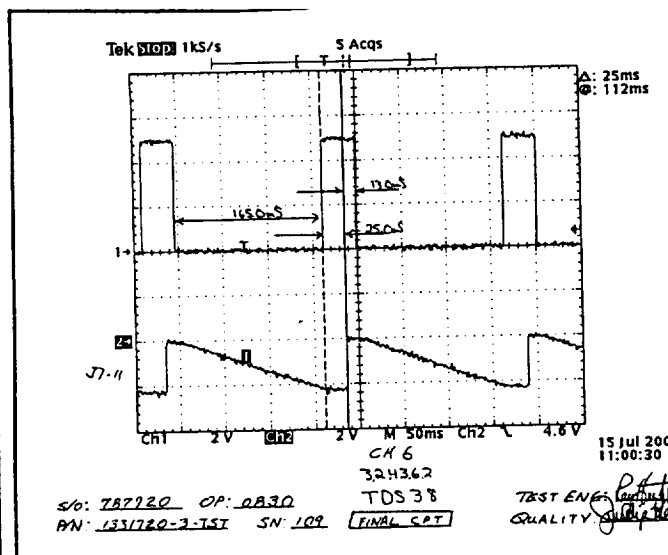


Channel 05
Frequency: 53.596 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS



Channel 06
Frequency: 54.4 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: FINAL CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op 0830
Shop Order: 787920 SN 109

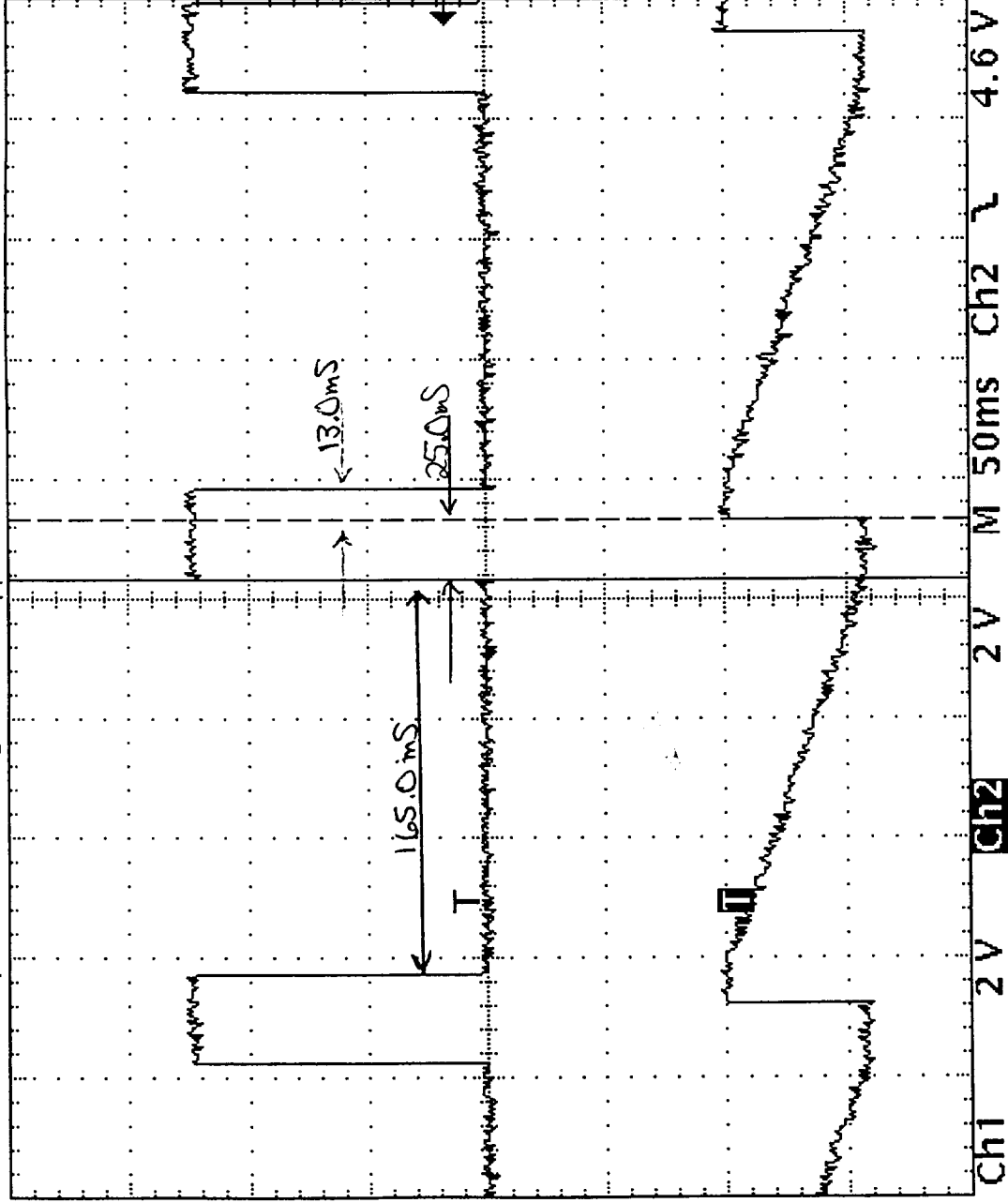
[Signature] 7-15-00
Customer Representative (Flight Hardware Only) Date

[Signature] 7-15-00
Test Systems Engineer Date
[Signature] 7-15-00
Quality Control Date

Tek Stop: 1ks/s

5 Acqs

Δ : 25ms
@: 134ms



I/H
J7 Pin 24

J7 Pin 10

15 Jul 2000
10:58:35

CH 5

3.24300

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

TDS 33

FINAL CPT

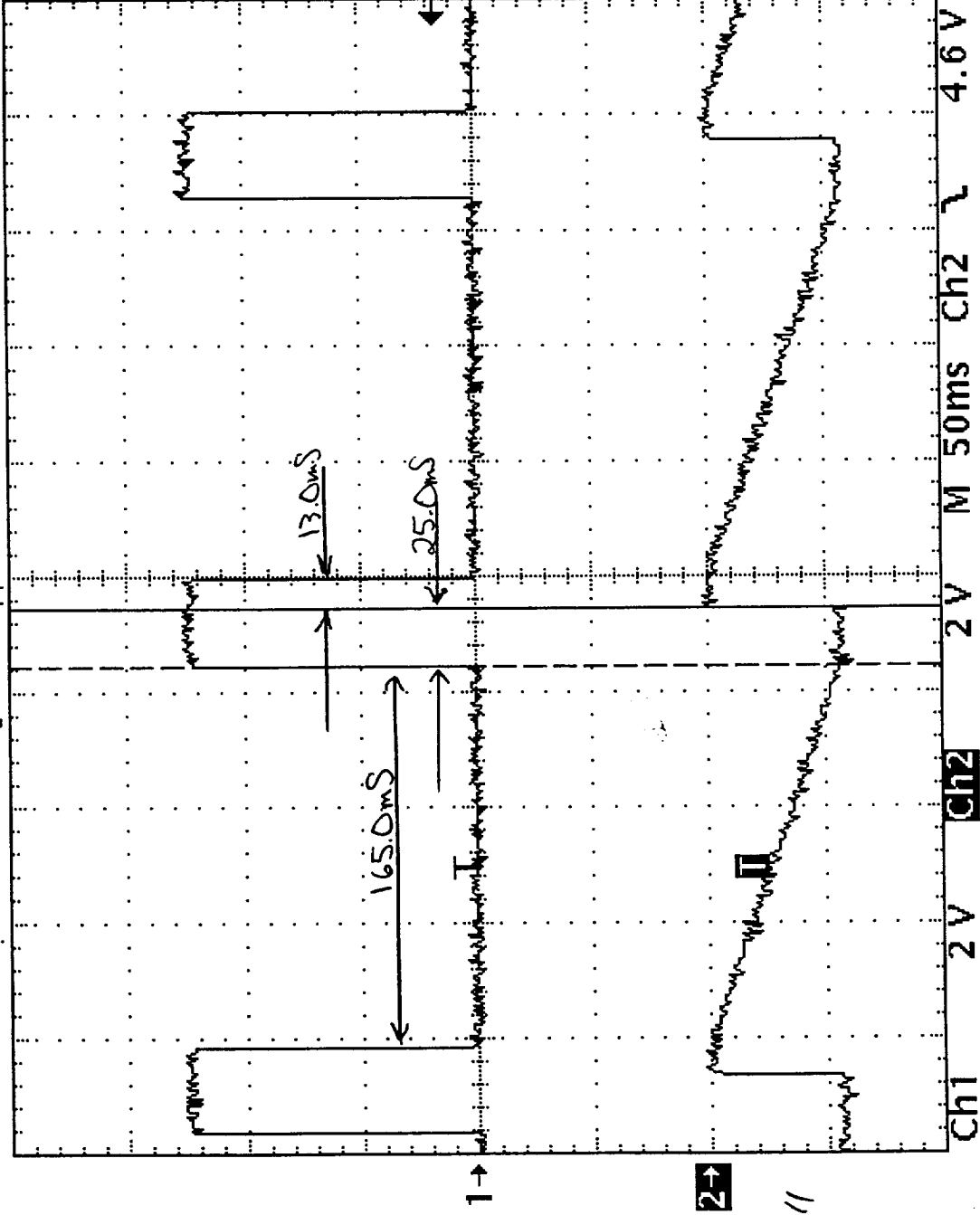
TEST ENG: *[Signature]* DATE: 7-15-00
QUALITY: *[Signature]* 7-15-00

Tek Stop: 1ks/s

5 Acqs

[T]

Δ : 25ms
@: 112ms



J7-11

15 Jul 2000
11:00:30

CH 6

32.436.2

TDS 38

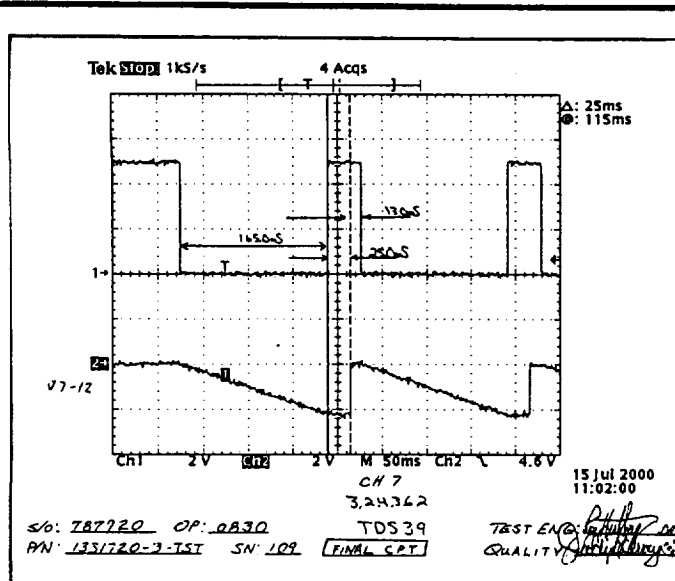
S/O: 787920 CP: 0830

PN: 1321720-3-TST SN: 109

FINAL CPT

TEST ENG: [Signature]
QUALITY: [Signature]
DATE: 7-15-00
7/15/00

TEST DATA SHEET 39
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

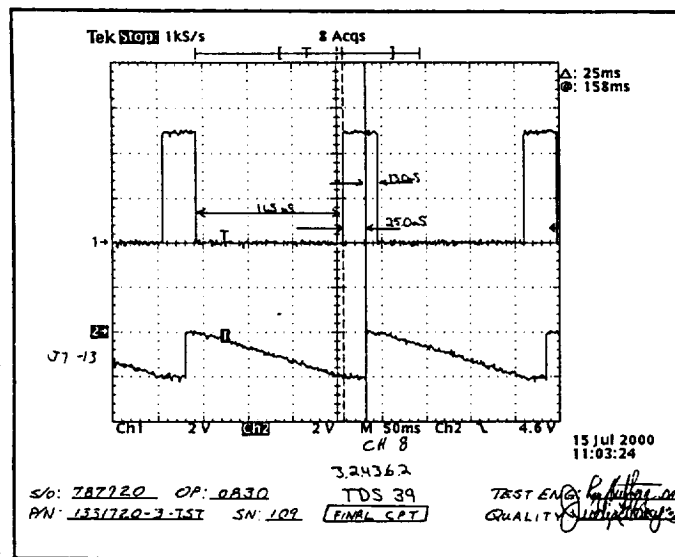


Channel 07
Frequency: 54.94 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS



Channel 08
Frequency: 55.5 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787920 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

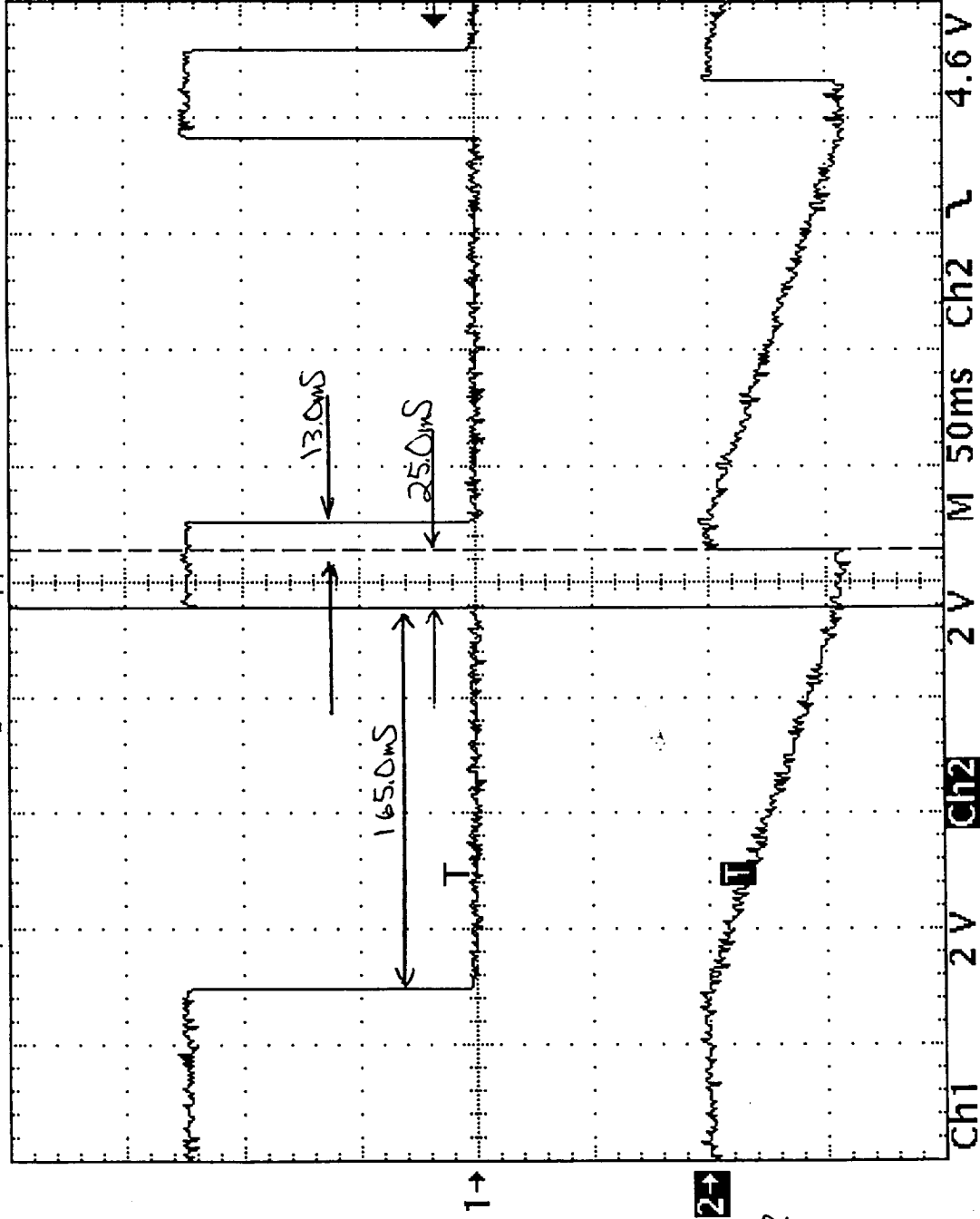
7-15-00
Date

Op 0830
Raymond L. Huber 7-15-00
Test Systems Engineer Date
Judith Vervey 7-15-00
Quality Control Date

Tek Stop: 1kS/s

4 Acqs

[T]



Δ : 25ms
@: 115ms

15 Jul 2000
11:02:00

CH 7

3,24,362

S/O: 787920 CP: 0830

PAN: 1331720-3-TST SN: 109

TDS 39

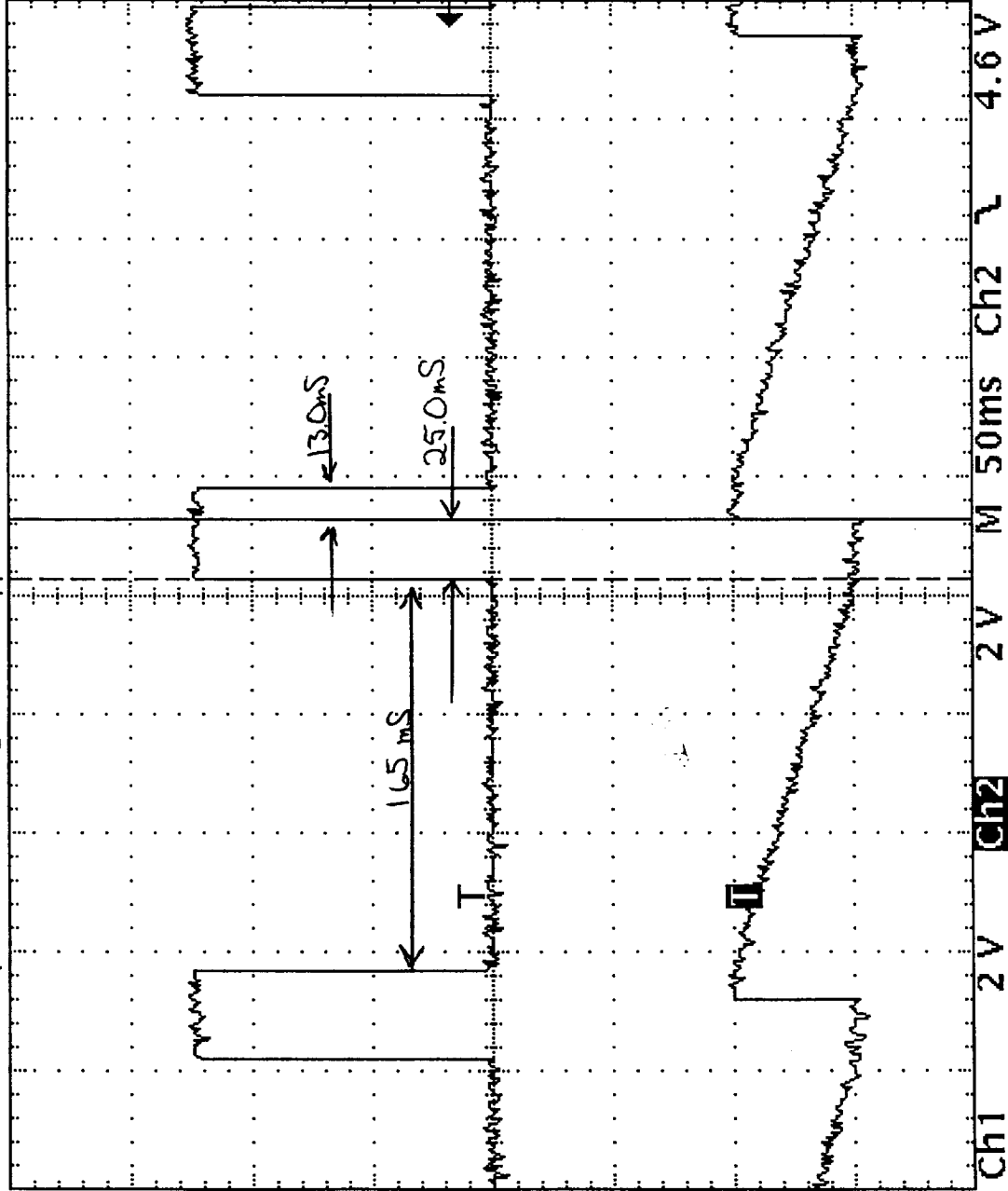
FINAL CPT

TEST ENG: [Signature] DATE: 7-15-00
QUALITY: [Signature] 7/15/00

Tek Stop 1ks/s

8 Acqs

Δ : 25ms
@: 158ms



15 Jul 2000
11:03:24

J7-13

3.24362

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

TDS 39

FINAL CPT

TEST ENG: Ray G. G. DATE: 7-15-00
DIA IT: [Signature] 7-15-00

TEST DATA SHEET 40 Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

Tek **STOP** 1kS/s 6 Acqs
 Δ: 25ms
 Φ: 117ms
 CH 9
 3.24362
 S/O: 787720 OP: 0830 TDS 40
 P/N: 1331720-3-TST SN: 109 **FINAL CPT**
 TEST END: 7-15-00
 QUALITY: 2/15/00

Channel 09
Frequency: 57.2903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

Tek **STOP** 1kS/s 8 Acqs
 Δ: 25ms
 Φ: 140ms
 CH 10
 3.24362
 S/O: 787720 OP: 0830 TDS 40
 P/N: 1331720-3-TST SN: 109 **FINAL CPT**
 TEST END: 7-15-00
 QUALITY: 7-15-00

Channel 10
Frequency: 57.2903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: **Final CPT** LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op 0830
Shop Order: 78792C SN: 109

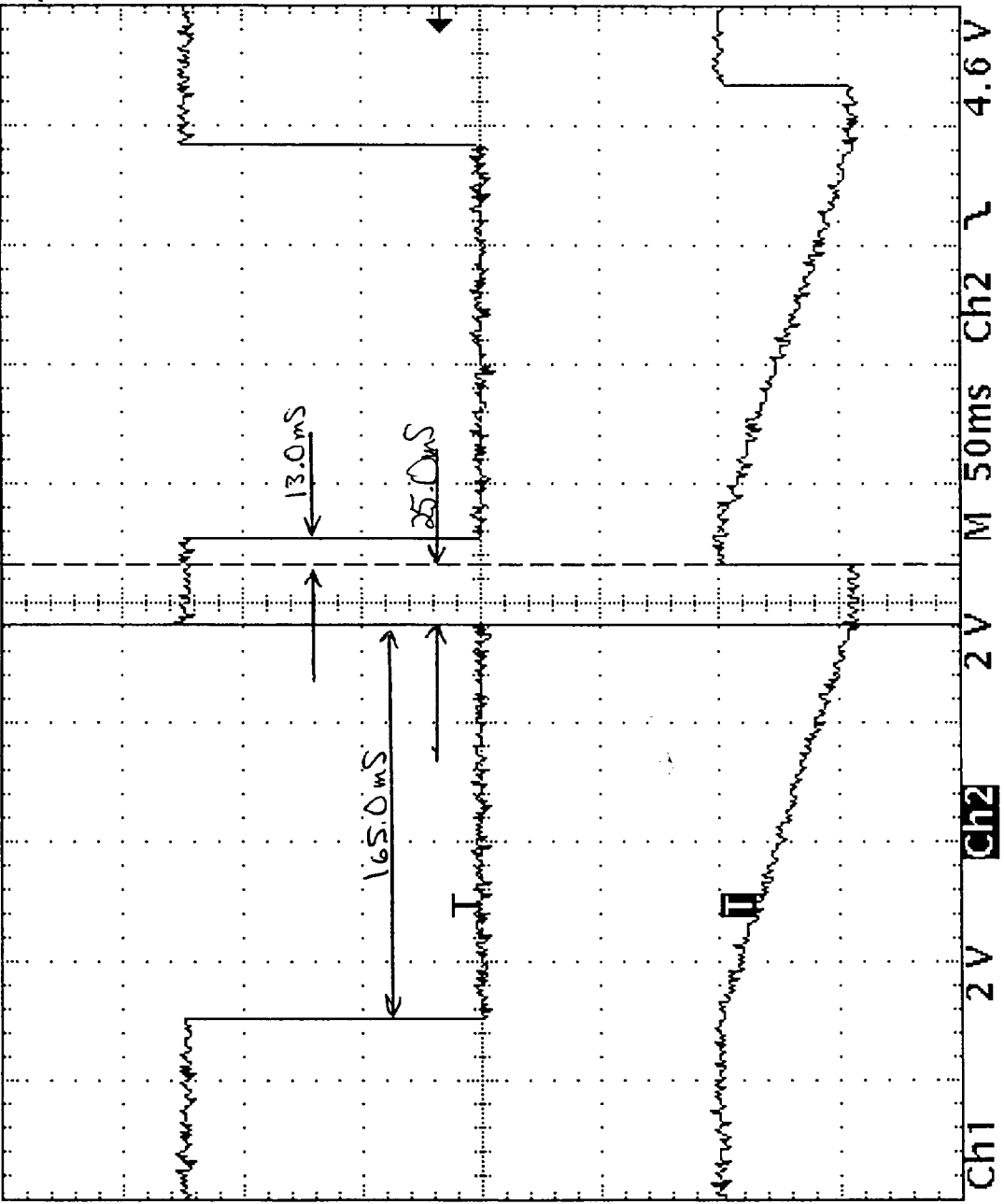
J. Sanford
Customer Representative
(Flight Hardware Only)
7-15-00
Date

Ray Herberg
Test Systems Engineer
7-15-00
Date

Judith Herberg
Quality Control
259
7-15-00
Date

Tek Stop: 1ks/s

6 Acqs



Δ : 25ms
@: 117ms

15 Jul 2000
11:04:48

CH 9

3.24.3.62

TDS40

FINAL CPT

TEST ENG: Anthony DATE: 7-15-00
QUALITY: Good 7/15/00

S/O: 787920 CP: 0830

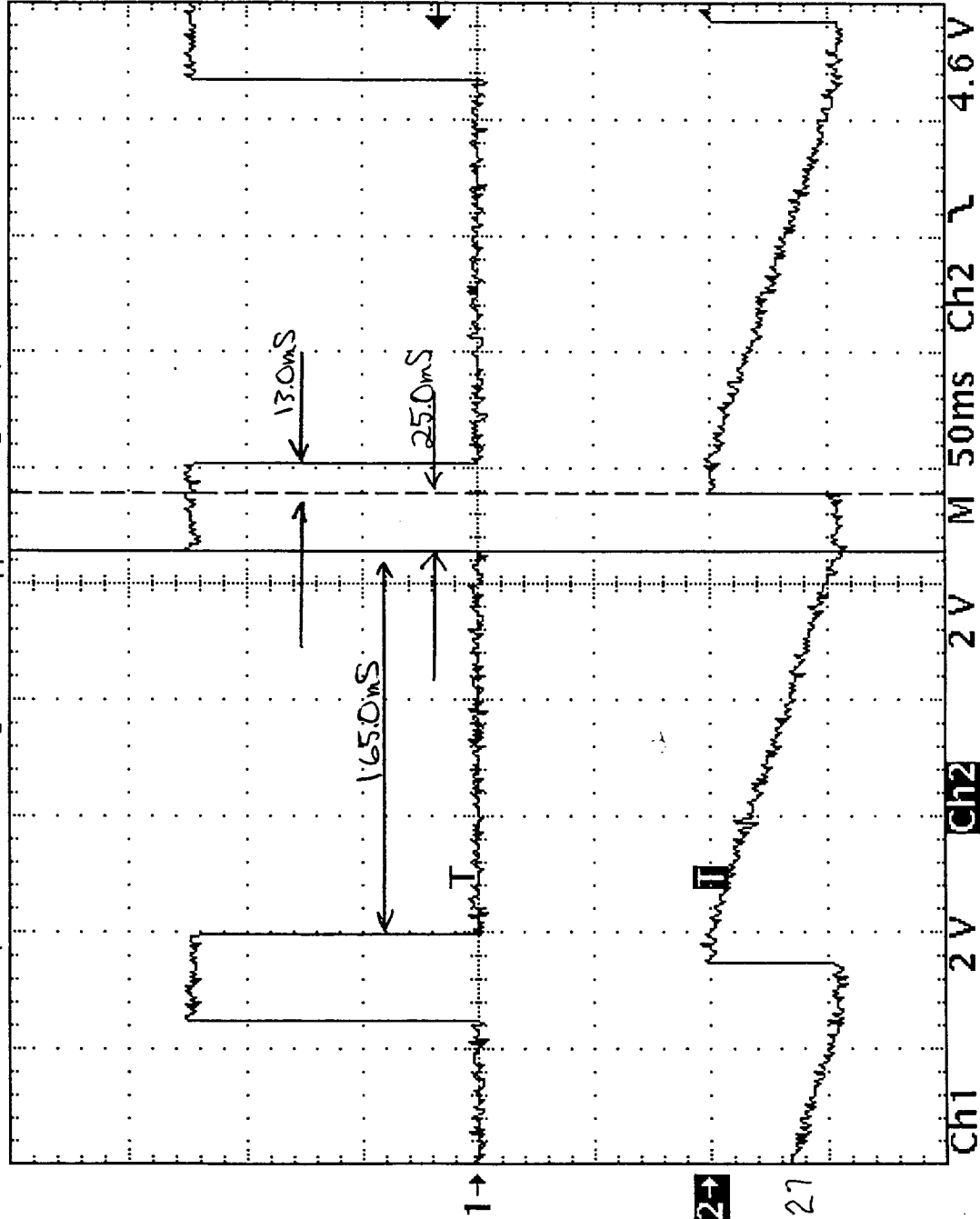
P/N: 1331720-3-TST SN: 109

J7-14

Tek Stop: 1ks/s

8 Acqs

Δ : 25ms
@: 140ms



J7-27

15 Jul 2000
11:06:28

CH 10

3.24362

TDS 40

CP: 0830

S/O: 787920

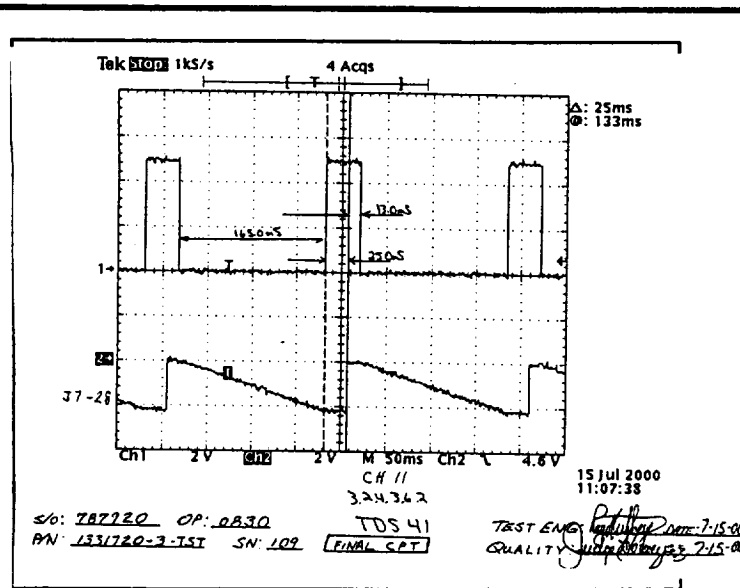
SN: 109

PN: 1331720-3-TST

FINAL CPT

TEST ENG: Kayden
DATE: 7-15-00
DUAL IT: 7-15-00

TEST DATA SHEET 41
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)

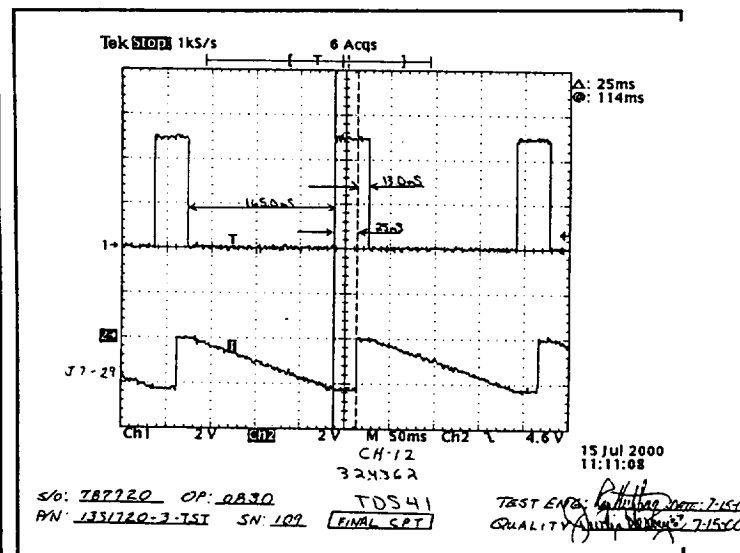


Channel 11
Frequency: 57.3903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS



Channel 12
Frequency: 57.3903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: Op. 0830 787220 SN: 109

J. Sanford 7-15-00
Customer Representative (Flight Hardware Only) Date

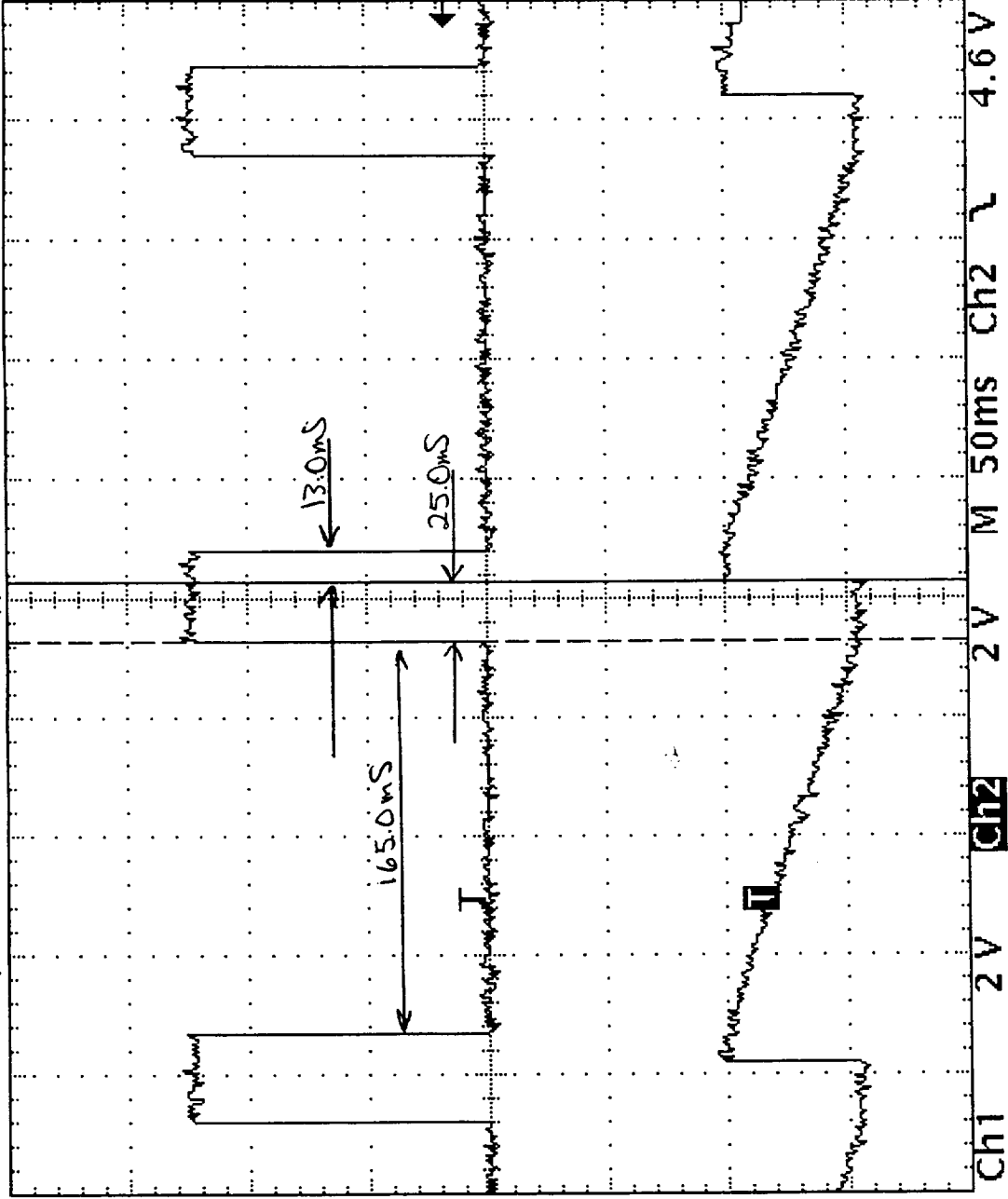
Ray H. [Signature] 7-15-00
Test Systems Engineer Date
[Signature] 7-15-00
Quality Control Date

Tek Stop 1kS/s

4 Acqs

[T]

Δ : 25ms
@: 133ms



15 Jul 2000
11:07:38

CH 11

3.24, 3.62

S/O: 787920 CP: 0830

P/N: 1331720-3-TST SN: 109

TDS 41

FINAL CPT

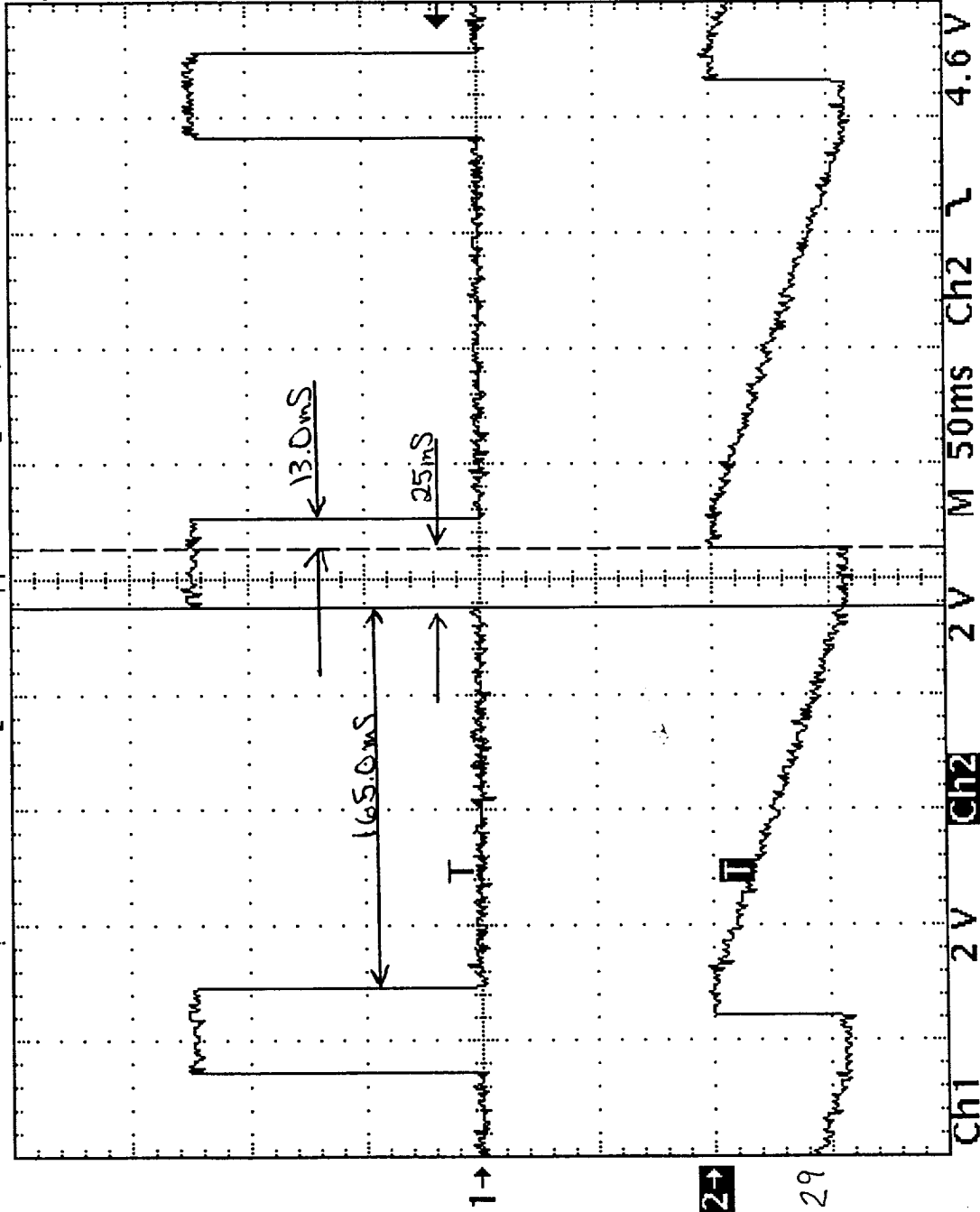
TEST ENG: [Signature] DATE: 7-15-00
QUALITY: [Signature] DATE: 7-15-00

J7-28

Tek Stop! 1ks/s

6 Acqs

Δ : 25ms
@: 114ms



15 Jul 2000
11:11:08

CH-12

3.24362

S/O: 787920 OP: 0830

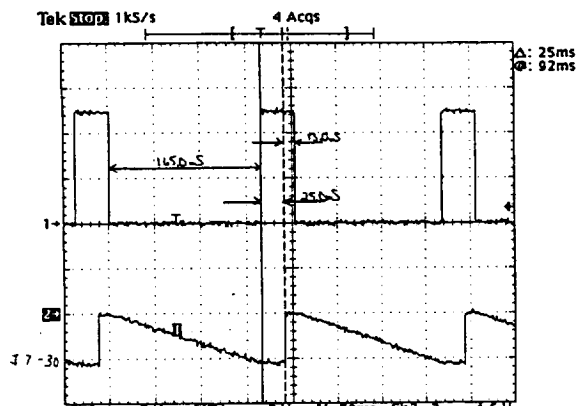
PRI: 1331770-3-TST SN: 109

TDS41

FINAL CPT

TEST ENG: *[Signature]* DATE: 7-15-00
DIA IT & *[Signature]* 7-15-00

TEST DATA SHEET 42 Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)



S/O: 787720 OP: 0830
P/N: 1331720-3-TST SN: 109

CH 13
3.24.362
TDS 42

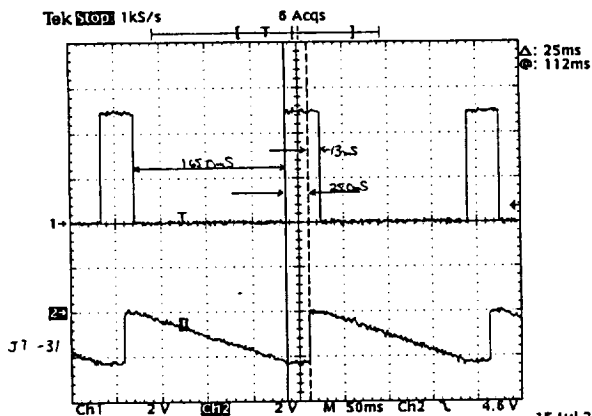
TEST ENG: [Signature] DATE: 7-15-00
QUALITY CONTROL: [Signature] DATE: 7-15-00

Channel 13
Frequency: 57.3903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS



S/O: 787720 OP: 0830
P/N: 1331720-3-TST SN: 109

CH 14
3.24.362
TDS 42

TEST ENG: [Signature] DATE: 7-15-00
QUALITY CONTROL: [Signature] DATE: 7-15-00

Channel 14
Frequency: 57.3903 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms ± 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 25.0 ms
Required 25 ms ± 10%
Pass/Fail PASS

DUMP (D) *
Measured 13.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787720

S/N: 109

Customer Representative
(Flight Hardware Only)

Date

Test Systems Engineer

Quality Control

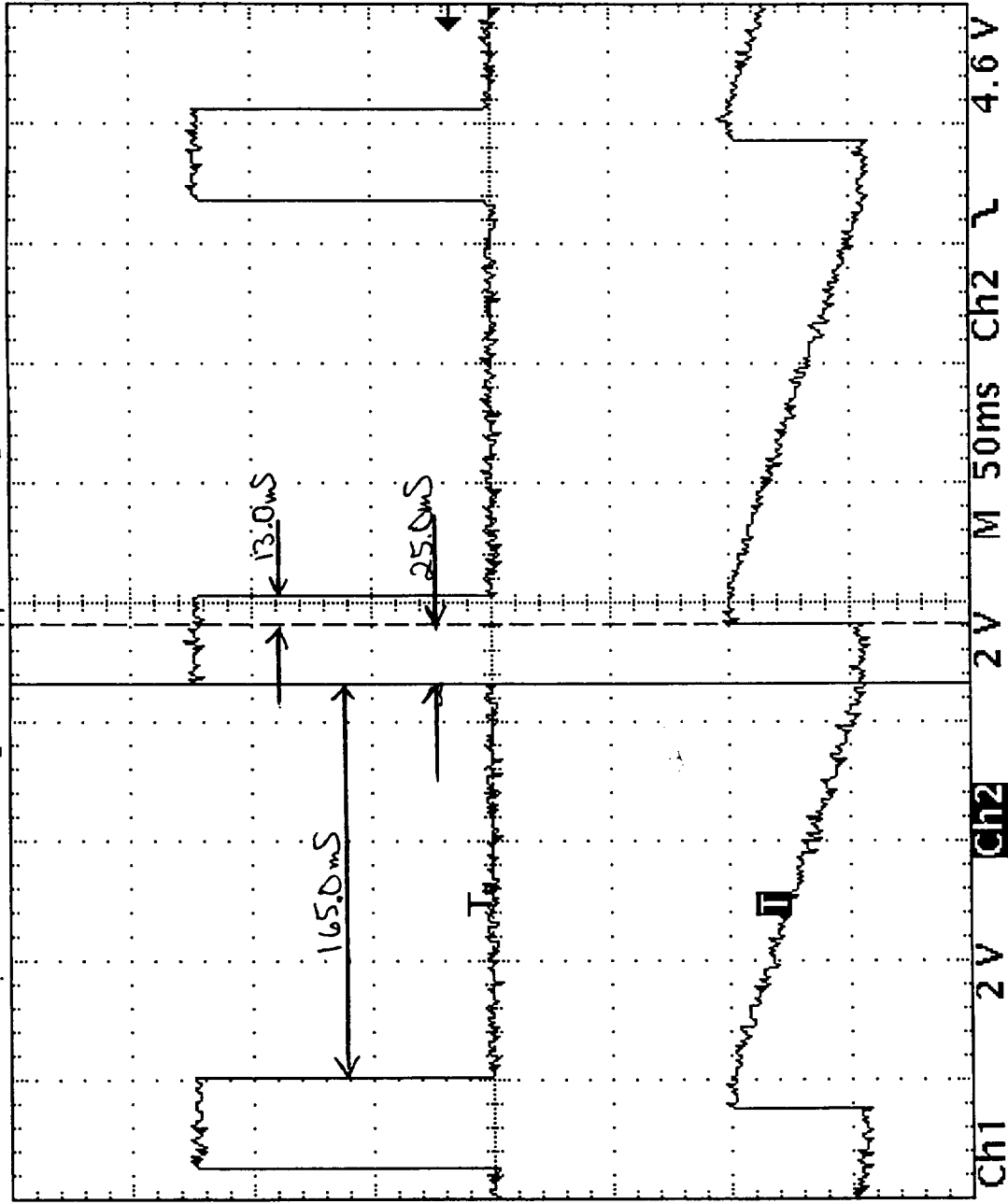
Date

Date

Tek Stop: 1KS/s

4 Acqs

[T]



15 Jul 2000
11:26:14

CH 13

3.24.3.62

TDS42

S/O: 787920 OP: 0830

P/N: 1331720-3-TST SN: 109

FINAL CPT

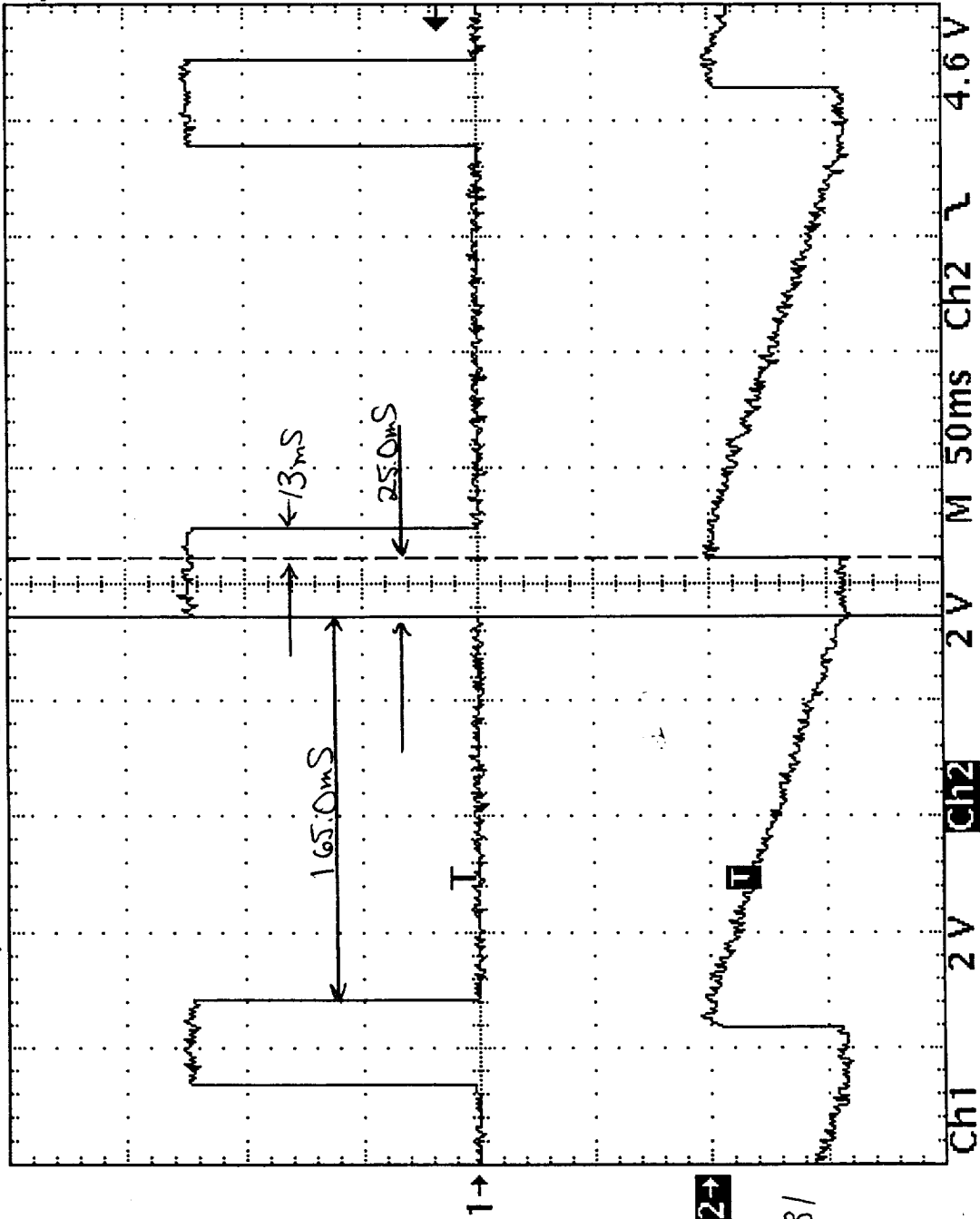
TEST ENG: *[Signature]* DATE: 7-15-00
QUALITY: *[Signature]* 7-15-00

Tek Stop! 1ks/s

6 Acqs

[T]

Δ : 25ms
@: 112ms



15 Jul 2000
11:17:20

CH 14

3.24362

S/O: 787920 OP: 0830

DAI: 1331720-3-TST S/N: 109

TOS 42

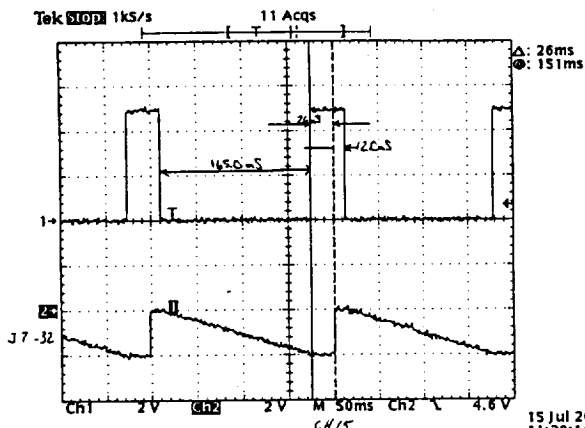
FINA1 CAT1

TEST ENG: *[Signature]* DATE: 7-15-00

DAI: 1117 *[Signature]* 7-15-00

J7 -31

TEST DATA SHEET 43
Integration Time (Analog Output) Verification (Paragraph 3.2.4.3.6.2)



S/O: 787720 OP: 0830
PN: 1331720-3-TST SN: 109

CH1 2V 50ms CH2 4.6V
3.24.362 TDS 43

TEST ENG: [Signature] QUALITY: [Signature]
DATE: 7-15-00 TIME: 2-15:00

Channel 15
Frequency: 89 GHz

INTEGRATION (X) *
Measured 165.0 ms
Required 165 ms \pm 10%
Pass/Fail PASS

HOLD (B-D) *
Measured 26.0 ms
Required 25 ms \pm 10%
Pass/Fail PASS

DUMP (D) *
Measured 12.0 ms
Required 9 ms to 15 ms
Pass/Fail PASS

* Refer to Figure 2 for waveform configuration.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787980

SN: 109

Customer Representative
(Flight Hardware Only)

Date

Test Systems Engineer

Quality Control

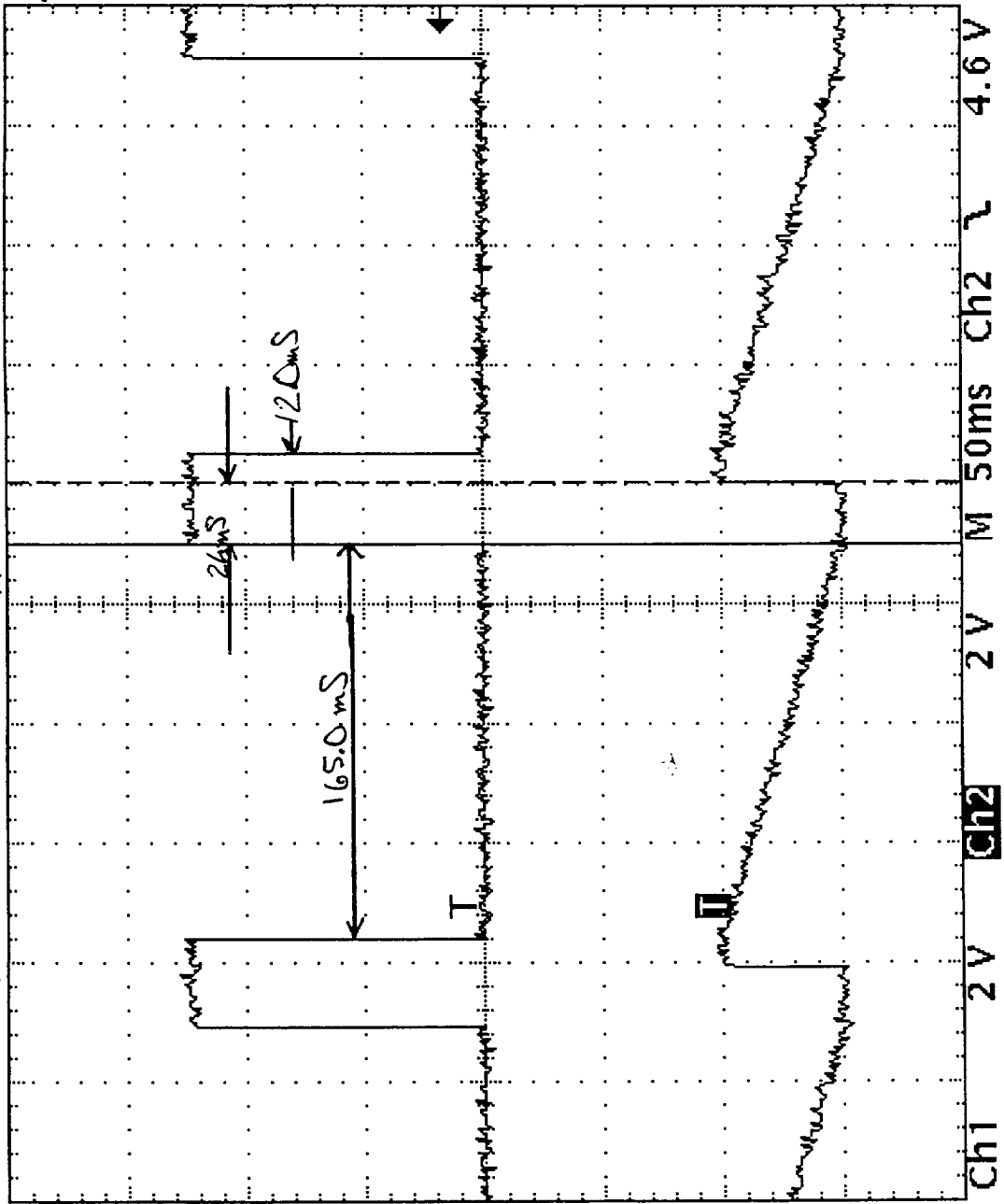
Date

Date

Tek Stop: 1KS/s

11 Acqs

[T]



Δ : 26ms
@: 151ms

15 Jul 2000
11:20:14

CH15
3.24.362

S/O: 787920 OP: 0830

P/N: 1331720-3-IST SN: 109

TDS 43

FINAL CPT

TEST ENG: [Signature] DATE: 7-15-00
QUALITY: [Signature] 7-15-00

TEST DATA SHEET 44
PLLO No. 1 Verification (Paragraph 3.2.4.3.6.3)
PLLO No. 2 Verification (Paragraph 3.2.4.3.6.4)

PLLO NO. 1
PLLO No. 1 dc Level 4.48 V Required: * Pass/Fail PASS

PLLO NO. 2
PLLO No. 2 dc Level 4.48 V Required: * Pass/Fail PASS

* -15 to +15 V dc level for S/N 101 - S/N 104, 4.0 ± 1.0 V for S/N 105 and above.

Circle Test: Final
CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Op. 0830

Shop Order: 787920

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)



7-15-00
Date

Ray Burkberg
Test Systems Engineer

Judith Denny
Quality Control



7-15-00
Date

7-15-00
Date

TEST DATA SHEET 49
Receiver Input Signals (Paragraph 3.2.4.4.1)

CH 9 through 14 PLLO	PRT Temp (°C)		Measured * Frequency	Requirements **	Pass/ Fail
PLLO No. 1	PLO No. 1	Xtal *** Osc.	57.2903406 Hz	57290.334 MHz ± 50 kHz	Pass
	35.76 °C				
PLLO No. 2	PLO No. 2	Xtal *** Osc.	57.2903406 Hz	57290.334 MHz ± 50 kHz	Pass
	32.84 °C				

- * Attach spectrum analyzer plots.
** = At 18°C
*** PRT not connected on S/N 105 and above.

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 0A: 0830 787929 S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

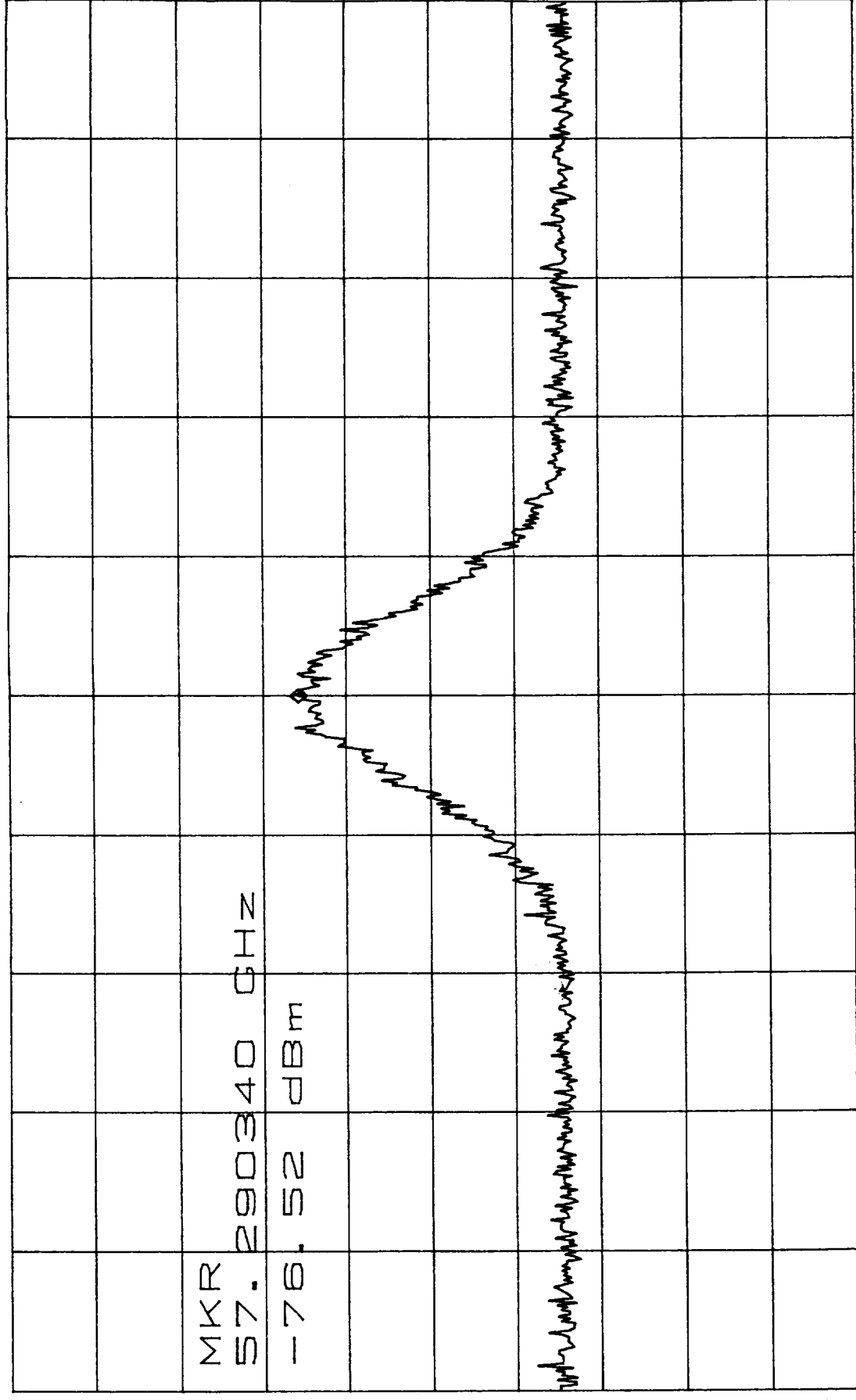
7-15-00
Date

Ray G. Gubug 7-17-00
Test Systems Engineer Date
Judith Hervey 7-18-00
Quality Control Date

332507-1 RKN

100

57.290340GHN



324.41

PLLO#1

CENTER 57.290340CHN

TDS 49

Z
A
L
S

2000-2001

Oscar Wilde : 0/5

TESTING

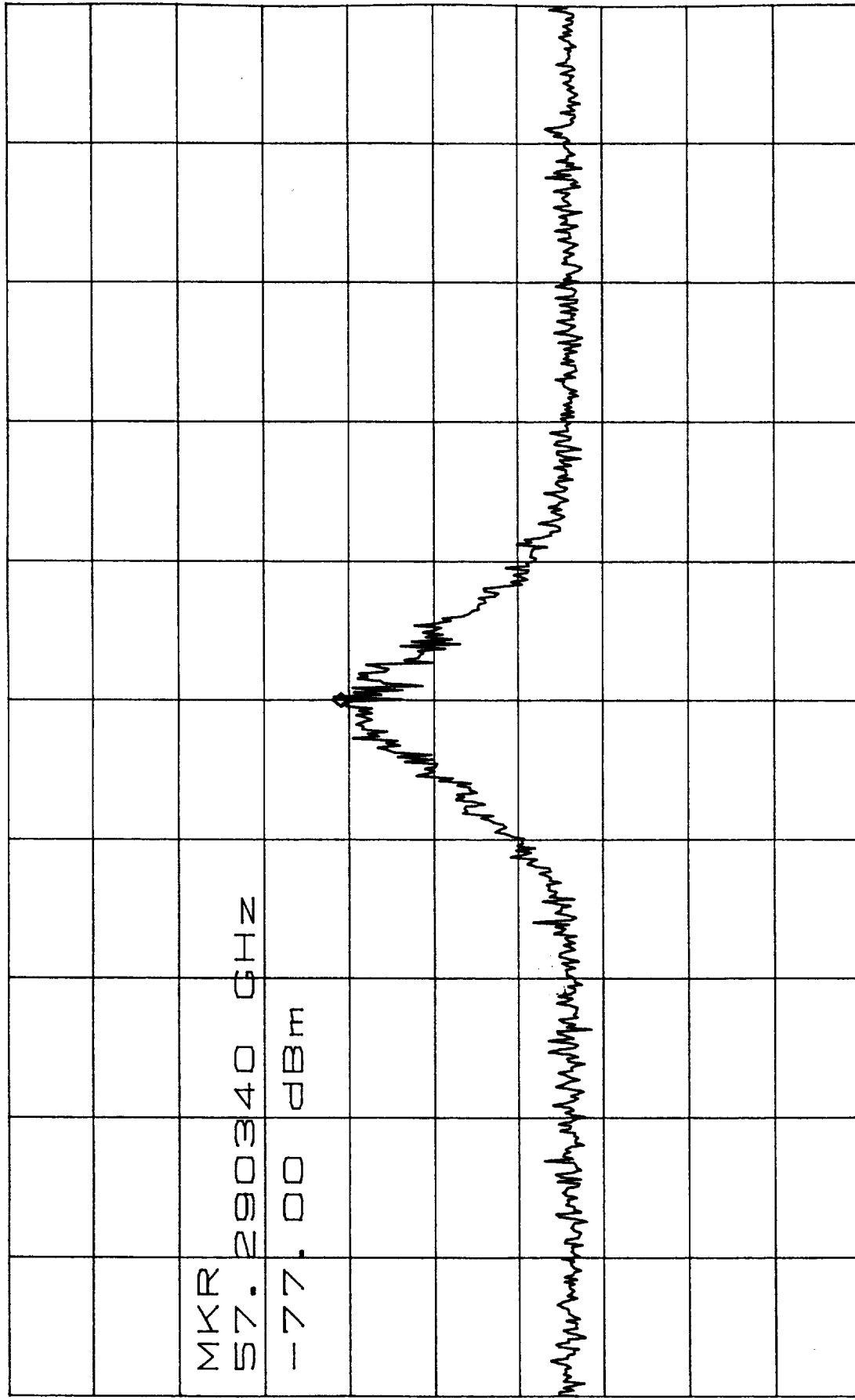
7-19-00

P/N: 1331720-3-75T SN: 109

FINAL CPT

QUALITY Jockey ³/₄ 7-18

CL 30.0dB MKR -77.00dBm
RL -73.0dBm 1dB/ 57.290340GHz



3.2.4.7.1 PL40#2

CENTER 57.290340GHz

SPAN 20

7-17-00

S/O: ~~788720~~ 30000 * VBW 3000Hz

TEST: ~~EMV~~

7-17-00

P/N: 1331720-3-TST SN: 109 [FINAL CPT]

QUALITY ~~7-18-00~~

TEST DATA SHEET 50 (Sheet 1 of 2)
Radiometer "Relative" NEΔT Verification* (Paragraph 3.2.4.4.2.2)

Channels 3, 4, 5, 6, 7, 8, and 15. PLLO No. 1 (Channels 9 through 14)

Channel Number>	3	4	5	6
NEΔT (Average of 5 data)	<u>.264</u>	<u>.164</u>	<u>.167</u>	<u>.133</u>
Pass/Fail	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>
NEΔT (Specified) K **	0.40	0.25	0.25	0.25
Channel Number>	7	8	9	10
NEΔT (Average of 5 data)	<u>.151</u>	<u>.225</u>	<u>.173</u>	<u>.208</u>
Pass/Fail	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>
NEΔT (Specified) K **	0.25	0.25	0.25	0.40
Channel Number>	11	12	13	14
NEΔT (Average of 5 data)	<u>.259</u>	<u>.306</u>	<u>.467</u>	<u>.810</u>
Pass/Fail	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>
NEΔT (Specified) K **	0.40	0.60	0.80	1.20
Channel Number>	15			
NEΔT (Average of 5 data)	<u>.138</u>			
Pass/Fail	<u>Pass</u>			
NEΔT (Specified) K **	0.50			

* Baseline data for acceptance tests. Use first CPT or first LPT data along with specification value for pass/fail criteria

** For reference only

Circle Test: Final CPT LPT

OP: 0830

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 787929

S/N: 109

J. Sanford
Customer Representative
(Flight Hardware Only)

7-18-00
Date

Ray Buehler
Test Systems Engineer

7-18-00
Date

Judith Hurrey
Quality Control

7-18-00
Date

A1 FUNCTIONAL TEST RESULTS
18-JUL-00

08:02:44

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	295.96	16748.0	14127.0	0.082	0.218
4	295.96	16485.0	13755.0	0.079	0.144
5	295.96	16973.0	14042.0	0.074	0.173
6	295.61	16905.0	13807.0	0.070	0.136
7	295.61	16616.0	13922.0	0.080	0.137
8	295.96	15222.0	13071.0	0.100	0.261
9	295.61	16486.0	13638.0	0.076	0.193
10	295.61	16236.0	13448.0	0.077	0.199
11	295.61	16730.0	13893.0	0.076	0.269
12	295.61	17157.0	14205.0	0.073	0.361
13	295.61	16932.0	13592.0	0.065	0.513
14	295.61	17488.0	14413.0	0.070	0.891
15	295.61	15088.0	13559.0	0.141	0.138

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

	CH3	CH4	CH5	CH6	CH7	CH8	CH9	CH10	CH11	CH12	CH13	CH14	CH15
Run # 1	.218	.144	.173	.136	.157	.261	.193	.199	.269	.361	.513	.891	.138
Run # 2	.294	.178	.152	.131	.158	.204	.164	.213	.255	.316	.460	.729	.155
Run # 3	.301	.167	.182	.130	.148	.224	.172	.205	.245	.343	.482	.818	.140
Run # 4	.265	.168	.166	.135	.139	.204	.161	.211	.258	.315	.476	.818	.136
Run # 5	.242	.162	.164	.131	.171	.230	.174	.210	.271	.350	.405	.796	.119
SAM	$\frac{137}{5}$	$\frac{818}{5}$	$\frac{837}{5}$	$\frac{663}{5}$	$\frac{753}{5}$	$\frac{1123}{5}$	$\frac{864}{5}$	$\frac{1038}{5}$	$\frac{1298}{5}$	$\frac{1529}{5}$	$\frac{2336}{5}$	$\frac{4052}{5}$	$\frac{688}{5}$
AVE	.264	.164	.167	.133	.151	.225	.173	.208	.259	.306	.467	.810	.138

PLLO # 1

Run # 1

A1 FUNCTIONAL TEST RESULTS
 18-JUL-00

08:04:36

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	295.98	16751.0	14159.0	0.083	0.294
4	295.98	16475.0	13776.0	0.080	0.178
5	295.98	16967.0	14067.0	0.074	0.152
6	295.62	16900.0	13766.0	0.069	0.131
7	295.62	16610.0	13884.0	0.079	0.158
8	295.98	15212.0	13087.0	0.102	0.204
9	295.62	16482.0	13607.0	0.075	0.164
10	295.62	16231.0	13414.0	0.077	0.213
11	295.62	16718.0	13856.0	0.075	0.255
12	295.62	17145.0	14164.0	0.072	0.316
13	295.62	16920.0	13548.0	0.064	0.460
14	295.62	17472.0	14369.0	0.069	0.729
15	295.62	15081.0	13530.0	0.139	0.155

PLLO #1
 Run #2

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
 SELECT TOUCHSCREEN BUTTON 2

A1 FUNCTIONAL TEST RESULTS
18-JUL-00

08:05:48

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.00	16749.0	14170.0	0.084	0.301
4	296.00	16469.0	13784.0	0.080	0.167
5	296.00	16964.0	14076.0	0.075	0.182
6	295.62	16896.0	13754.0	0.069	0.130
7	295.62	16608.0	13873.0	0.079	0.148
8	296.00	15206.0	13091.0	0.102	0.224
9	295.62	16480.0	13598.0	0.075	0.172
10	295.62	16227.0	13404.0	0.076	0.205
11	295.62	16711.0	13842.0	0.075	0.245
12	295.62	17137.0	14151.0	0.072	0.343
13	295.62	16911.0	13533.0	0.064	0.482
14	295.62	17459.0	14353.0	0.069	0.818
15	295.62	15077.0	13519.0	0.138	0.140

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

PLLO # 1
Run # 3

A1 FUNCTIONAL TEST RESULTS
A1.EXE;62 18-JUL-00

08:06:44

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.00	16751.0	14187.0	0.084	0.265
4	296.00	16465.0	13797.0	0.081	0.168
5	296.00	16962.0	14089.0	0.075	0.166
6	295.62	16894.0	13746.0	0.068	0.135
7	295.62	16606.0	13866.0	0.079	0.139
8	296.00	15203.0	13100.0	0.103	0.204
9	295.62	16478.0	13592.0	0.075	0.161
10	295.62	16225.0	13398.0	0.076	0.211
11	295.62	16704.0	13832.0	0.075	0.258
12	295.62	17131.0	14142.0	0.072	0.315
13	295.62	16905.0	13523.0	0.064	0.476
14	295.62	17450.0	14340.0	0.069	0.818
15	295.62	15074.0	13511.0	0.138	0.136

PLLO #1

Run #4

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

A1 FUNCTIONAL TEST RESULTS
 18-JUL-00

08:07:56

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.02	16749.0	14204.0	0.085	0.242
4	296.02	16459.0	13811.0	0.082	0.162
5	296.02	16958.0	14103.0	0.076	0.164
6	295.63	16891.0	13729.0	0.068	0.131
7	295.63	16602.0	13851.0	0.078	0.171
8	295.63	15197.0	13109.0	0.103	0.230
9	296.02	16477.0	13581.0	0.074	0.174
10	295.63	16221.0	13385.0	0.076	0.210
11	295.63	16696.0	13816.0	0.075	0.271
12	295.63	17124.0	14124.0	0.072	0.350
13	295.63	16898.0	13505.0	0.064	0.405
14	295.63	17442.0	14319.0	0.069	0.796
15	295.63	15070.0	13500.0	0.137	0.119

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
 SELECT TOUCHSCREEN BUTTON 2

PLLO #1

Run #5

6 Apr 99

TEST DATA SHEET 50 (Sheet 2 of 2)
Radiometer "Relative" NEAT Verification* (Paragraph 3.2.4.4.2.2)

PLLO No. 2 (Channels 9 through 14)

Channel Number>	9	10	11	12
NEAT (Average of 5 data)	<u>.168</u>	<u>.219</u>	<u>.245</u>	<u>.325</u>
Pass/Fail	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>	<u>Pass</u>
NEAT (Specified) K **	0.25	0.40	0.40	0.60
Channel Number>	13	14		
NEAT (Average of 5 data)	<u>.455</u>	<u>.734</u>		
Pass/Fail	<u>Pass</u>	<u>Pass</u>		
NEAT (Specified) K **	0.80	1.20		

* Baseline data for acceptance tests. Use first CPT or first LPT data along with specification value for pass/fail criteria

** For reference only

Circle Test: Final CPT LPT

METSAT/AMSU-A1 System P/N IS-1331720

Shop Order: 78792BS/N: 109

Ray H. Hines
Test Systems Engineer

Date

John Sanford
Customer Representative
(Flight Hardware Only)

Date

Judith Hervey
Quality Control

Date

A1 FUNCTIONAL TEST RESULTS
18-JUL-00

08:38:20

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.80	16773.0	16762.0	1.000	3.383
4	296.80	16355.0	16342.0	1.000	2.084
5	296.80	16895.0	16881.0	1.000	1.977
6	295.56	16837.0	13804.0	0.071	0.145
7	295.56	16548.0	13910.0	0.082	0.140
8	296.80	15074.0	15063.0	1.000	2.217
9	295.56	16405.0	13615.0	0.077	0.166
10	295.56	16150.0	13423.0	0.079	0.197
11	295.56	16547.0	13799.0	0.078	0.216
12	295.56	16973.0	14111.0	0.075	0.297
13	295.56	16746.0	13506.0	0.067	0.480
14	295.56	17237.0	14274.0	0.073	0.701
15	295.56	15006.0	13519.0	0.145	0.186

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

PLLO#2

R

Run # 1

	CH 9	CH 10	CH 11	CH 12	CH 13	CH 14
Run # 1	.166	.197	.216	.297	.480	.701
Run # 2	.154	.233	.263	.315	.417	.742
Run # 3	.175	.219	.234	.304	.497	.786
Run # 4	.166	.207	.252	.352	.421	.679
Run # 5	.180	.239	.260	.358	.462	.765
sum						
5	.841	1.095	1.225	1.626	2.277	3.673
AVE	.1682	.219	.245	.3252	.4554	.7346

A1 FUNCTIONAL TEST RESULTS
A1.EXE;62 18-JUL-00

08:39:24

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.83	16773.0	16764.0	1.000	3.006
4	296.83	16351.0	16338.0	1.000	2.112
5	296.83	16893.0	16879.0	1.000	2.275
6	295.56	16834.0	13791.0	0.071	0.159
7	295.56	16546.0	13899.0	0.081	0.166
8	296.83	15068.0	15057.0	1.000	2.392
9	295.56	16401.0	13603.0	0.077	0.154
10	295.56	16146.0	13410.0	0.079	0.233
11	295.56	16542.0	13786.0	0.078	0.263
12	295.56	16968.0	14096.0	0.075	0.315
13	295.56	16743.0	13488.0	0.066	0.417
14	295.56	17232.0	14255.0	0.072	0.742
15	295.56	15004.0	13506.0	0.144	0.153

PLLO#2

Run # 2

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

A1 FUNCTIONAL TEST RESULTS
A1.EXE;62 18-JUL-00

08:40:28

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.87	16774.0	16761.0	1.000	3.356
4	296.87	16348.0	16334.0	1.000	1.815
5	296.87	16892.0	16878.0	1.000	1.987
6	295.56	16833.0	13797.0	0.071	0.155
7	295.56	16545.0	13903.0	0.082	0.162
8	296.87	15065.0	15053.0	1.000	2.253
9	295.56	16398.0	13606.0	0.077	0.175
10	295.56	16144.0	13412.0	0.079	0.219
11	295.56	16537.0	13787.0	0.078	0.234
12	295.56	16964.0	14098.0	0.075	0.304
13	295.56	16736.0	13490.0	0.066	0.497
14	295.56	17225.0	14257.0	0.073	0.786
15	295.56	15002.0	13507.0	0.144	0.160

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 1

PLLO #2
Run #3

A1 FUNCTIONAL TEST RESULTS
A1.EXE;62 18-JUL-00

08:41:32

CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.91	16774.0	16761.0	1.000	3.157
4	296.91	16344.0	16331.0	1.000	2.270
5	296.91	16892.0	16877.0	1.000	1.908
6	295.56	16832.0	13799.0	0.071	0.136
7	295.56	16543.0	13904.0	0.082	0.163
8	296.91	15062.0	15051.0	1.000	2.227
9	295.56	16398.0	13607.0	0.077	0.166
10	295.56	16141.0	13412.0	0.079	0.207
11	295.56	16533.0	13786.0	0.078	0.252
12	295.56	16959.0	14097.0	0.075	0.352
13	295.56	16731.0	13491.0	0.067	0.421
14	295.56	17218.0	14257.0	0.073	0.679
15	295.56	15000.0	13505.0	0.144	0.148

PLLO #2
Run #4

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
SELECT TOUCHSCREEN BUTTON 2

A1 FUNCTIONAL TEST RESULTS
 18-JUL-00

08:43:00


CH	WARM TEMP	WARM COUNTS	COLD COUNTS	GAIN	DELTA T
3	296.95	16773.0	16760.0	1.000	3.690
4	296.95	16341.0	16325.0	1.000	1.710
5	296.95	16888.0	16872.0	1.000	2.413
6	295.56	16829.0	13798.0	0.071	0.136
7	295.56	16541.0	13903.0	0.082	0.142
8	296.95	15058.0	15045.0	1.000	2.236
9	295.56	16393.0	13604.0	0.077	0.180
10	295.56	16138.0	13409.0	0.079	0.239
11	295.56	16526.0	13782.0	0.079	0.260
12	295.56	16952.0	14093.0	0.075	0.358
13	295.56	16724.0	13487.0	0.067	0.462
14	295.56	17210.0	14252.0	0.073	0.765
15	295.56	14997.0	13501.0	0.144	0.181

[2] PRINT SCREEN [3] PRINT RAW DATA [4] PRINT HISTOGRAM

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH
 SELECT TOUCHSCREEN BUTTON 1

PLLO #2
 Run # 5

 NASA National Aeronautics and Space Administration				Report Documentation Page			
1. Report No. ---		2. Government Accession No. ---		3. Recipient's Catalog No. ---			
4. Title and Subtitle Integrated Advanced Microwave Sounding Unit-A (AMSU-A), Performance Verification Report				5. Report Date July 2000			
				6. Performing Organization Code ---			
7. Author(s) R. Haigh				8. Performing Organization Report No. 11669			
9. Performing Organization Name and Address Aerojet 1100 W. Hollyvale Azusa, CA 91702				10. Work Unit No. ---			
				11. Contract or Grant No. NAS 5-32314			
12. Sponsoring Agency Name and Address NASA Goddard Space Flight Center Greenbelt, Maryland 20771				13. Type of Report and Period Covered Final			
				14. Sponsoring Agency Code ---			
15. Supplementary Notes ---							
16. ABSTRACT (Maximum 200 words) This is the Performance Verification Report, Final Comprehensive Performance Test Report, P?N 1331720-3-TST, S/N 109/A1, for the Integrated Advanced Microwave Sounding Unit-A (AMSU-A).							
17. Key Words (Suggested by Author(s)) EOS Microwave System				18. Distribution Statement Unclassified --- Unlimited			
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				22. Price ---			

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4. TITLE AND SUBTITLE Integrated Advanced Microwave Sounding Unit-A (AMSU-A), Performance Verification Report			5. FUNDING NUMBERS NAS 5-32314	
6. AUTHOR(S) B. Delgado-Barger				
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Aerojet 1100 W. Hollyvale Azusa, CA 91702			8. PERFORMING ORGANIZATION REPORT NUMBER 11669 July 2000	
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) NASA Goddard Space Flight Center Greenbelt, Maryland 20771			10. SPONSORING/MONITORING AGENCY REPORT NUMBER ---	
11. SUPPLEMENTARY NOTES ---				
12a. DISTRIBUTION/AVAILABILITY STATEMENT ---			12b. DISTRIBUTION CODE ---	
13. ABSTRACT (Maximum 200 words) This is the Performance Verification Report, Final Comprehensive Performance Test Report, P?N 1331720-3-TST, S/N 109/A1, for the Integrated Advanced Microwave Sounding Unit-A (AMSU-A).				
14. SUBJECT TERMS EOS Microwave System			15. NUMBER OF PAGES	
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